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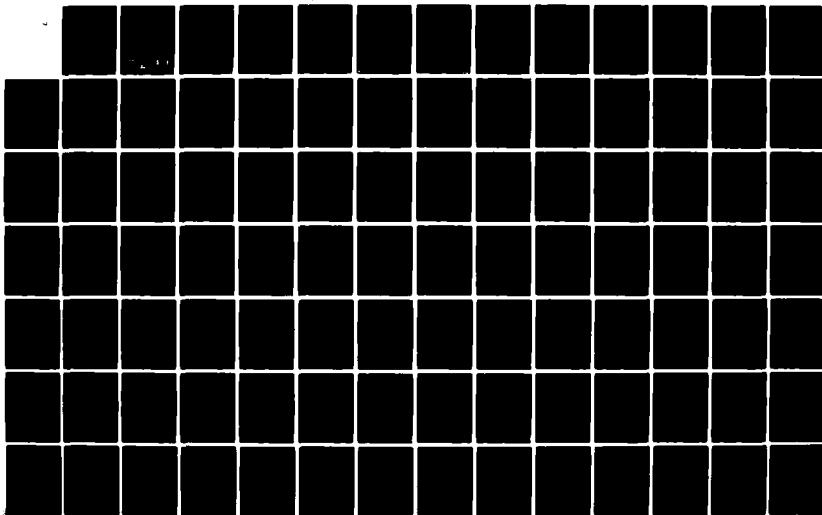
ORBITAL RADIATION STUDY FOR INCLINED CIRCULAR
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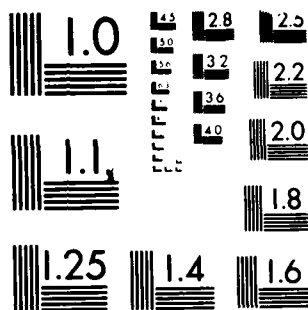
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ORBITAL RADIATION STUDY FOR INCLINED CIRCULAR TRAJECTORIES

Contract N00173-81-MP-06630

E.G. STASSINOPOULOS

NOVEMBER 1981

NASA

National Aeronautics and
Space Administration

Goddard Space Flight Center,
Greenbelt, Maryland 20771

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Orbital Radiation Study for Inclined
Circular Trajectories

E. G. Stassinopoulos

NASA-Goddard Space Flight Center
Sciences Directorate
National Space Science Data Center



November 1981

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1. Introduction

At the request of and with support from the Naval Research Laboratory*, a comprehensive study was conducted to determine the space radiation environment of a series of high inclination geocentric orbits for a mission duration of five years.)

Following the precedent established with ~~previous studies~~, the external (surface incident) charged particle radiation, predicted for the satellite was determined by orbital flux integration for six independent trajectories, (see section 3). The latest standard models of the environment were used in the calculations, (see section 5).

Magnetic field definitions for the six nominal circular trajectories were obtained from a current field model, (see section 3).

Spatial and temporal variations or conditions affecting the static environment models were considered and accounted for, wherever possible.

Finally, limited shielding and dose evaluations were performed for simple infinite slab and spherical geometries.

Results, given in graphical and tabular form, are analyzed, explained, and discussed. Conclusions are presented and commented on.

2. SPECIFICATION OF ORBITS

The analysis was based on nominal circular orbits with inclination of 60 degrees and altitudes of 1667, 2593, 3889, 5186, 6389, and 10371 kilometers.

3. GENERATION OF TRAJECTORIES

Six separate flight path emphemerides were generated for the specified orbits with the GEODYN-BLCONV System¹ for trajectories of 24-hour duration defined at 1-minute intervals. The length of simulated orbit time and the integration stepsize were especially selected so as to provide sufficient point density to insure an adequate sampling of the ambient radiation environment when flying the trajectories through the models. The trajectories were subsequently converted from geodetic polar to magnetic B-L coordinates with McIlwain's INVAR program of 1965² and the field routine ALLMAG,³ which now utilizes the BARRACLOUGH 1975 field model.⁴ The field computations were extrapolated to the tentative mission epoch of 1989.5 with linear time terms representing secular variations of the field.

4. FLIGHT PATH EXPOSURE TO TRAPPING DOMAINS

The specified nominal flight-path configurations display a significant characteristic of high inclination orbits in magnetic L-space: they traverse almost the entire terrestrial radiation belt twice during each revolution, moving back and forth through regions of low L values (the inner zone: $1.0 < L \leq 2.8$), regions of high L values (the outer zone: $2.8 < L < 12$), and

*This work funded by the Naval Electronics Systems Command under the NRL Nuclear Survivability/Vulnerability Program.

regions outside the trapping domain (external). Occasionally, some revolutions will also enter regions of space where no particle trapping can occur because of atmospheric cut-off conditions; that is, trajectory segments may have a combination of magnetic B and L values that place them outside the atmospheric cut-off limits of the models.

These excursions and the "external" visitations afford the satellite an amount of flux-free time, which may be of substantial duration (see section 11, C).

5. TRAPPED PARTICLE ENVIRONMENT MODELS

The fluxes in this study were obtained from current NSSDC models: the solar maximum AE6 for the inner zone electrons⁵, the new interim model AEI7 for the outer zone electrons⁶, and the solar maximum version of the new AP8 model⁷ for energetic trapped protons. It should be noted that the interim AEI7 does not reflect solar cycle variations in its present state. However, this model was issued in two versions, the AEI7-HI and the AEI7-LO, in order to account for differences in the data sets used in their construction. The LO version was used in this effort. All models describe an average static environment at a given epoch.

6. ORBITAL FLUX INTEGRATIONS

Orbital flux integrations were performed with the UNIFLUX⁸ and the SOFIP⁹ systems. UNIFLUX provides L-band distributions and exposure times with B-L bin breakdown, while SOFIP provides the dose and shield data.

7. GEOMAGNETIC SHIELDING AND SOLAR FLARE PROTONS

Low altitude high inclination orbits experience a significant amount of geomagnetic shielding from cosmic rays of solar or galactic origin in the energy range $E > 10$ MeV. Therefore, it may be assumed that the spacecraft will only intermittently be exposed to the unattenuated interplanetary solar flare proton intensities of all energies above 10 MeV. To a first approximation, the fluxes may also be considered omnidirectional and isotropic, probably to within 10-15%.

Usually, geomagnetic shielding effects on geocentric missions are being evaluated with simple rigidity considerations because of substantial diurnal variations in the cutoff latitude associated with geomagnetic tail effects (2-4 degrees) and storm-induced changes (> 4 degrees). The simple analysis used here assumed that energetic solar protons of all energies above 10 MeV have free access to all magnetospheric regions external to a dipole shell of $L=5$ earth radii, which is equivalent to a cut-off latitude of about 63 degrees.

Predictions of solar flare proton fluxes at 1 AU are obtained as a function of mission duration τ and confidence level Q^* on the basis of a probabilistic analysis¹⁰ using a modified type of Poisson statistics by a computerized model SOLPRO¹¹ that includes the distinction between "ordinary" (OR) and

* Q denotes the degree of confidence one wishes to assign to the results, namely that for the specified mission duration the calculated fluences are the smallest values which will not be exceeded by actually encountered intensities.

"anomalously large" (AL) events and the probability of occurrence of the latter. Both AL- and OR- event fluences are non-linear functions of Q and τ . For these predictions, only high quality comprehensive satellite measurements (not ground observations) are being used, covering almost the entire 20th solar cycle. There have been indications that descriptions of the solar flare environment in interplanetary space (at 1 AU), derived from interpretations and extrapolations of ground based measurements, have not been very accurate.

It should be noted that the statistics cannot predict when an AL event will occur; only the probability that one will occur in a given length of time. And it must be remembered that a single AL event will impart its total fluence within two to four days.

This implies that for unmanned satellites with mission durations of $\tau \geq 1$ year, OR-event fluences are not significant because probabilistic theory predicts the possible occurrence of at least one AL event, even for the lowest allowable confidence level ($Q=80\%$).

8. FLUX DATA: TYPE, QUALITY, AND VARIATIONS

The trapped particle flux data available from the models represent omnidirectional, integral intensities that one would expect to obtain as average values over periods in excess of six months. But over most regions of magnetospheric space ($L \geq 2$ earth radii), short term excursions can vary from these values by factors of 10^2 to 10^3 , depending on the particle energies and on the type and intensity of the causative event. These variations do affect the investigated missions because their trajectories enter regions of space where L is greater than 2 earth radii. Also, trapped particle populations experience changes due to: (a) local time (LT) dependence, and (b) solar cycle dependence. Both are of some consequence to these missions. The former is significant for spacecraft that sample regions of $L > 5$ earth radii, which are visited by the specified missions. To compensate for these variations, the model provides LT-averaged values, which should yield an adequate approximation for missions of long duration ($\tau \geq 1$ year). The solar cycle variations have been taken into account by selecting the appropriate models for each period, where available.

Generally, solar cycle variations have opposite effects on each particle species:

	<u>Solar Min</u>	<u>Solar Max</u>
Electrons	lower	higher
Protons	higher	lower

The solar cycle changes, as derived from a comparison of the corresponding models, are functions of energy E and magnetic parameter L . For the inner zone electrons, they may range from a factor of 1 to a factor of 5.

Protons are only affected in the vicinity of the atmospheric cutoff regions. No changes of consequence have been observed in the heart of the proton trapping domain. Proton changes have about the same range as those of the electrons.

It is necessary to emphasize that the calculations, although based on the best data available for the past epochs, can only serve as approximations for the future.

It also should be noted that a basic uncertainty factor of 2 is defined for the flux values of the AP8 and the AE5 models, while the AE6 is characterized by an average uncertainty factor of 5. No uncertainty factor has yet been defined for the interim AEI7.

9. DOSE AND SHIELDING EVALUATION

Doses were calculated from the total orbit integrated, surface incident, omnidirectional, integral fluences by existing shielding codes¹², as functions of various aluminum shield thicknesses and geometries.

A simple procedure was followed, not involving solid angle sectoring or three-dimensional ray tracing considerations. Instead, a simple two-dimensional geometry with a cosine law for the incident spectra, and a three-dimensional spherical geometry were considered. (See comment in section 11D-III)

Bremsstrahlung calculations were performed with the same codes.

10. RESULTS: PRESENTATION DESCRIPTION

This section describes the form and format in which the results, derived from the Orbital Flux Integration (OFI) process, are presented for practical use. Except where otherwise specified, all particle data in this report relate to integral, omnidirectional fluxes or fluences.

A. Tabular Presentations

The outcome of all calculations is summarized in Tables 1 to 60. The tables are arranged in six sets, where every set pertains to one specific type of data. The first two sets have two similar members for every trajectory considered in the study: one for trapped protons and one for electrons, in that order. The next three sets contain only one member for each trajectory. The sixth set contains three similar members for each trajectory. A more detailed description of the tables is provided in the following paragraphs.

I. L-band Tabulations: Tables 1-12

Tabulation of total orbit-integrated fluence distributions by L-bands for selected energy thresholds, in units of particles per square centimeter, normalized to 5 MeV and .5 MeV for protons and electrons, respectively.

The tables contain 48 L-bands of equal size covering the range from $L=1.0$ to $L > 10.4$ earth radii in constant increments of .2 earth radii.

II. Spectral Profiles: Tables 13-24

Tabulation of average orbit-integrated spectral distributions. Composite spectra are given in units of: fluxes per square centimeter per second, fluxes per square centimeter per day, and total fluences per specified mission duration (5 years). For the electrons, the latter are also given in terms of

inner and outer zone contributions. Functionally derived differential fluxes are listed in the last columns for both species of particles.

Total orbit-integrated spectra in percent, for energy intervals ΔE corresponding to the energy levels of the L-band tables, are also given in terms of average instantaneous and daily intensities.

An exposure index (for the normalization energies used in the L-band tables) is listed for nine successive intensity ranges varying by one order of magnitude, in terms of processed exposure duration (in hours) and total number of particles accumulated while in that intensity range for the indicated number of hours.

III. Peaks and Totals Per Orbit: Tables 25-30

These tables contain the absolute instantaneous peak fluxes and the total fluences accumulated during each successive revolution, as obtained from the nominal trajectories for the investigated flight duration (24 hours of mission time).

Specifically, there are nine columns on these tables. Column 1 is an orbit counting device, based on:

- a) the orbit period when the trajectory is circular and lies in the equatorial plane;
- b) the physical perigee in all elliptical flight-path cases; and,
- c) the equatorial crossing for circular inclined trajectories.

Column 2 gives the peak flux. Columns 3, 4 and 5 indicate the spacecraft position in geocentric coordinates at which the predicted peak flux was encountered. Columns 6, 7 and 8 determine respectively the relative orbit time and the magnetic B-L coordinates for this event. For the purpose of orbital radiation studies, all simulated trajectories start at $t_0 = 0$ hours. Finally, the last column indicates the total predicted flux to be encountered during that particular orbit. It is advisable to disregard the last line on this table because many times that orbit is incomplete and the fluxes or positions shown do not correspond to true peaks.

IV. Time-Accounting and Exposure-Analysis: Tables 31-36

The "EXPOSURE-ANALYSIS" summary indicates what percent of its total lifetime T the satellite spends in "flux-free" regions of space, what percent of its total lifetime it spends in high intensity proton and electron domains, and while so exposed, what percent of its total flux it accumulates.

In the context of this study, the term "flux-free" applies to all regions of space where trapped particle fluxes are less than one proton or electron per square centimeter per second, having energies $E > 5$ MeV, and $E > .5$ MeV, respectively. By definition, this includes all regions external to the Van Allen radiation belts.

The concept of "trapped particle fluxes" is meant to include stably trapped, pseudo trapped, and transient fluxes, as long as they are part of or contained in the environment models used and, in the case of transients or pseudos, their sources are considered powerful enough to supply them continuously in substantial numbers.

Similarly, as "high intensity" are defined those regions of space where the instantaneous, integral, omnidirectional, trapped-particle flux is greater than 10^3 protons with energies $E > 5$ MeV, and greater than 10^5 electrons with energies $E > 5$ MeV.

The values given in these tables are statistical averages, obtained over extended intervals of mission time. However, they may vary significantly from one orbit to the next, when individual revolutions are considered.

The "TIME-ACCOUNT" breakdown shows what percent of its total time the satellite spends in the "inner zone" ($1.0 \leq L < 2.8$) and in the "outer zone" ($2.8 \leq L \leq 11.0$) electron trapping domains, and also the percent of time spent in regions external to the latter ($L > 11.0$).

It should be noted that the confinement of the outer zone within the boundary of the $L=11.0$ earth radii volume is arbitrary and has no physical meaning. It is intended only as a simplification to facilitate the calculations. The region considered "external" in this study ($L > 11.0$) is still partially a domain of the outer zone, at least as far out as $L=12.0$ earth radii, according to the current environment models.

A last item on this table: the inner zone time is further subdivided into two parts: the percentage of time spent outside ($L < 1.1$) and inside ($1.1 \leq L < 2.8$) the trapping domain.

V. Solar Proton Fluences and Exposure Factor: Tables 37-42

For the specified mission duration τ (printed in the sub-title), and dipole cut-off shell ($L=5$ earth radii, shown in the header), this table lists the solar proton fluence-spectra (in units of particles per square centimeter) at five discrete confidence levels Q (given at the top of each column).

The exposure factors (in percent of total mission duration) obtained from the geomagnetic shielding analysis are also listed for four dipole cut-off shell values (in earth radii).

VI. Total Dose and Components: Tables 43-60

These tables list doses in units of rads_{AL} as a function of aluminum shield thickness, given in three ways: range s in grams per square centimeter, depth t in millimeters, and depth z in mils.

Electron, bremsstrahlung, and proton contributions to the overall sumtotal dose are given separately. Electron and proton doses are further broken down into their respective constituents; namely, inner zone and outer zone for the former, trapped and solar flare for the latter.

The specific mission duration for which the doses have been calculated is indicated in the table headline.

Caution: the AL-event solar flare protons are not contributed gradually over the investigated mission duration ($\tau = 5$ years) but are imparted in toto in a relatively short burst, that is, within approximately 2-4 days per AL event.

B. Graphical Presentation

Some of the tabulated data are also plotted in Figures 1 to 36, and 49 to 102 with additional Figures 37 to 48 containing plots of flight path data. Positional flux and dose data are plotted in Figures 103-162. As with the tables, the computer plots are arranged in eight sets, where again each set pertains to one specific type of data. The first three sets have two similar members for each trajectory investigated: one for each particle species. The next two sets (fourth and fifth) contain one member for each trajectory considered. The sixth set contains nine similar members for each trajectory, providing three graphs (for respective depth ranges) for each of three geometries. The seventh set contains two similar members for each trajectory: one for each particle species. Finally, the last set has eight members for each trajectory.

I. Time and Flux Histograms: Figures 1-12

These plots show two curves superimposed on the same graph; namely, one each for the variables "time" and "flux". Both are given on a semi-log scale as functions of the parameter L (earth radii), within the range $1 < L \leq 10$, and for constant L -bands of .1 earth radius width. The plots are plotted:

- a) by a plain curve, the characteristic trajectory intensities as obtained from the orbital integration process in terms of averaged integral particle fluxes above a given energy.
- b) by a contour marked with symbols, the percent total lifetime (%T) spent in each L interval.

The logarithmic ordinate relates to the time-flux variables. The printed numbers are powers of 10 and pertain to the fluxes; the scale values for the time curves are given in the upper part of the ordinate label: from 10^{-3} to 10^2 percent of T , the type of particles, their integral energy, and the units, are all given in the lower part of the label. The label on the top of the graph identifies the trajectory.

II. Spectral Profiles: Figures 13-24

A graphical presentation of the final composite spectral distribution, obtained from the orbital integration process. The plots are semi-log graphs, where the abscissa is a linear energy scale for integral particle energies E , in MeV, and the ordinate is a logarithmic scale for the fluxes, given in daily averages for energies greater than E ; the printed scale values are powers of 10.

III. Peaks Per Orbit: Figures 25-36

Here the absolute peak intensities, encountered per period (1 period = 1 revolution = 1 orbit), are plotted for the duration of the flight-time processed in the analysis. The logarithmic ordinate, with scale values in powers of 10, relates to instantaneous particle fluxes of the environment at the indicated energy thresholds, while the abscissa is a linear orbit enumeration.

IV. Trajectory World Map Projections: Figures 37-42

These graphs depict the surface trace of the geocentrically projected subsatellite positions. The trajectories are plotted for several revolutions on a global map produced by a Miller Cylindrical Projection method. The contours of the continents have been omitted for clarity. The positions of equatorial crossing, of physical perigee, or of period commencement are indicated by numbers identifying the orbits shown in the graphs. For all trajectories, the distance between successive sequential numbers is a measure of the orbit precession.

V. Flight Path Tracing in B-L Space: Figures 43-48

Plots showing trajectory traces in B-L space on a semi-log scale. Several orbits are depicted, each identified by its sequential number. The magnetic equator is entered on all plots. The logarithmic ordinate relates to the field strength B in gauss; the printed values are exponents of 10. L is given in earth radii on the linear abscissa.

VI. Dose-Depth Curves by Geometry: Figures 49-102

Plots of final depth-dose values for the indicated mission duration. Normally, these plots show composite curves for bremsstrahlung, combined electrons (inner and outer zone), combined protons (trapped and solar flare), and sumtotal of all contributions. In the present case, the respective contours consist of inner and outer zone electrons separately and of trapped and solar flare protons, separately, of composite bremsstrahlung, and the sumtotal.

For ease of use and in order to provide a greater resolution at the more sensitive range of depths (namely the thinner shields) three plots have been generated per processed trajectory, for shield-ranges and subdivisions increasing by one order of magnitude.

The logarithmic ordinate, with scale values in powers of 10, relates to aluminum dose in units of rads. The linear abscissa is the shield thickness, given in three different units: range s in grams per square centimeter, depth t in millimeters, depth t in mils.

VII. Positional Flux Plots: Figures 103-114

Plots of instantaneous omnidirectional trapped particle fluxes (electrons and protons) at (up to 10) specified threshold energy levels ($>MeV$), for a selected orbit (usually worst case revolution through heart of SAA).

The logarithmic ordinate, with scale values in powers of 10, relates to the number of particles per square centimeter per second. The linear abscissa is the relative time, in minutes or hours, from the beginning of the selected orbital pass.

VIII. Positional Dose Plots: Figures 115-162

Plots of instantaneous omnidirectional trapped particle dose values at (up to 10) specified shield thicknesses (omnidirectional isotropic incidence, cosine-theta distribution) for a selected orbit (usually worst case revolution through heart of SAA). Separate plots are generated (if present) for: electron dose (including bremsstrahlung), proton dose, and total dose (no solar proton contributions are included) for dose at transmission surface of aluminum slab shields, dose in semi-infinite aluminum medium, and dose at center of aluminum spheres.

The logarithmic ordinate, with scale values in powers of 10, relates to the respective dose in units of rads-aluminum. The linear abscissa is the relative time, in minutes or hours, from the beginning of the selected orbital pass.

11. RESULTS: ANALYSIS AND DISCUSSION

In this section, some of the presented tabular or graphical study-results are discussed, with occasional comments as to their use, limits, and applications.

A. Spectral Profiles

Characteristic features of the near earth radiation environment are strong altitude and inclination dependencies. However, at high inclination values ($30^\circ < i < 90^\circ$) small changes in inclination will not produce changes in flux levels and spectral distributions as significant as those produced by small changes in altitude. The greatest inclination dependent variations occur in the range $0^\circ \leq i \leq 30^\circ$.

I. Protons: The protons exhibit relatively hard, almost uniform spectra in the investigated inclination-altitude regime.

It should be noted that a characteristic softening of the high energy tail of the spectrum appears at the 6389 km altitude orbit, near $E > 400$ MeV, as a consequence of the limit of the volume of space occupied by these particles. The extent of the proton trapping domain along the magnetic equator is inversely related to their energy. This is strikingly demonstrated by the spectrum of the 10371 km altitude orbit, which falls off sharply at about $E > 100$ MeV and which indicates that according to the standard model there are no protons with energies $E > 150$ MeV contained in the region of space sampled by this trajectory.

II. Electrons: The electrons show complex variable spectra. Inner zone and outer zone average, orbit-integrated, composite intensities rise non-uniformly with altitude, particularly at energies above 3.75 MeV with differences reaching up to several orders of magnitude at $E > 6$ MeV. Spectra extend to higher energies as height increases.

The inner zone spectra fall rapidly off to zero flux in the energy range from 4 to 5 MeV and they are therefore more benign than their harder outer-zone counterparts, which extend to energies of about 7 MeV.

At low altitudes (1667 and 2593 km), the inner zone contributions prevail up to about $E > 2.75$ MeV, but for all energies $E > 2$ MeV the inner zone spectra are always softer than the outer zone spectra (in agreement with the models).

However, at high altitudes (e.g. the 10371 km orbit), the inner zone contributions become insignificant even at low energies ($.1 < E(\text{MeV}) < .7$).

B. Peaks Per Orbit

The absolute peaks per revolution have been obtained for standard processing energies: $E > 5$ MeV for protons and $E > .5$ MeV for electrons. Other energy selections produce different peak curves in an inverse relationship: lower energies yield higher and more expanded contours, and vice versa.

Peak contours of inclined circular trajectories display amplitude variations and sometimes discontinuities (flux-free time) that follow periodic patterns based on the daily cycle of revolutions. For fixed energies, amplitudes and discontinuities are function of: (a) inclination i , and (b) altitude h .

Variations in either i or h may produce significant changes in the amplitude of the peak curves and in the duration of the discontinuities: up to several orders of magnitude for the former, and completely eliminating the latter.

For the investigated trajectories at the given, fixed, inclination ($i = 60^\circ$), the following observations can be made:

a) protons: at the processed energy of $E > 5$ MeV only small variations are obtained for the low altitude orbits at 1667 and 2593 km; the difference between maximum and minimum peak intensity predicted is about a factor of 6 and 13, respectively. These differences disappear at higher altitudes, while at the same time the peak flux levels rise by about two orders of magnitude, from $\sim 7 \times 10^4$ particles per square centimeter at 1667 km to $\sim 7 \times 10^6$ particles per square centimeter at 5186 km.

b) electrons: at the processed energy of $E > .5$ MeV the electron peaks appear very similar to those of the protons, as described in the previous paragraph. But the electron peak fluxes start at higher levels and reach an upper limit at the same altitude as the protons only with smaller differences: from $\sim 3 \times 10^6$ particles per square centimeter at 1667 km to $\sim 1 \times 10^7$ particles per square centimeter at 5186 km.

C. Flux-Free Time

Some comments on this topic have been provided in the previous section and in section 10/IV. Here a more detailed discussion will be given.

Flux-free time (FFT) intervals are an important feature of certain orbital configurations. They may occur over short orbit segments (partial FFT per period) or over the entire length of a revolution (total FFT per period). In terms of geomagnetic geometry, the FFTs establish the duration for which the trajectory lies outside the trapping domain of the corresponding particle species, evaluated at the given energies. Or conversely, they are a measure of the degree to which the trajectory is exposed to the charged particle trapping domains.

The number of consecutive flux-free orbits of circular trajectories is primarily a function of altitude and inclination and to a lesser degree a function of particle energy. Of the investigated flight paths, and for the selected energies (electrons: $E > .5$ MeV, protons: $E > 5$ MeV), none shows any completely flux-free revolutions per day. The total FFT, which derives completely from partially exposed revolutions (see "Exposure Analysis," Tables #31 to 36), in percent of total mission duration, can be summarized as follows:

<u>Altitude</u>	<u>Protons</u>	<u>Electrons</u>
(km)	($E > 5$ MeV)	($E > .5$ MeV)
1667	22%	2%
2593	24%	3%
3889	30%	4%
5186	35%	6%
6389	39%	7%
10371	51%	11%

Higher energies will yield longer FFTs because the more energetic particles occupy a smaller volume of space.

D. Dose and Shielding

The calculated doses display features characteristic of the terrestrial radiation environment: at medium-to-high shield thicknesses, small contributions from relatively benign and low intensity electron spectra combined with major contributions from comparatively hard and intense proton spectra; at thin shield thicknesses the electrons predominate.

Depending on altitude, the proton doses prevail for shield thicknesses from greater than 16 to greater than 200 mils of aluminum. At high altitudes (e.g. 10371 km orbit) the opposite is true: protons are the major dose contributors only for shield thicknesses of less than 5 mils of aluminum.

Significant, however, is the fact that for aluminum the proton dose is only a weak function of shield thickness, as it shows very little attenuation over the evaluated depth range. Thus, in order to get an appreciable reduction in the dose, say by one order of magnitude, a 20-fold increase in shield thickness is necessary. The same is true for the bremsstrahlung dose. However, at low altitudes, in comparison to the proton contributions, the bremsstrahlung dose is so small (about 2 orders of magnitude lower) that it may be disregarded.

I. Decay and Degradation

The total doses obtained for each of the six investigated trajectories for the five year mission duration are substantial. In terms of electronics decay or materials degradation, the doses to be experienced on this mission inside the satellite, that is, behind a skin of about 80 mils of equivalent aluminum, are severe even for insensitive components or equipment:

<u>Altitudes (km)</u>	<u>Spherical Dose</u>
1667	~294K rads
2593	~1093K rads
3889	~1558K rads
5186	~1266K rads
6389	~1099K rads
10371	~1682K rads

II. Contamination and Interference

It should be remembered, that the direct or indirect effects of the radiation environment may also be a nuisance in terms of instrument interference or measurement contamination. If such is the case, some remedies may be available.

III. Possible Improvements

In the event that the magnitude of total dose or degree of radiation penetration behind the skin of the satellite is of importance to the mission, four possibilities exist to reduce the radiation effects on instrument and components:

- a) build or design an instrument less sensitive to radiation and construct the on-board and/or on-ground data processing software to remove or suppress radiation-induced noise
 - b) change the orbit by any combination of the elements eccentricity, altitude, and inclination so as to achieve a more benign environment
 - c) change the mission epoch: solar max for reduced proton intensities, solar min for reduced electron intensities
 - d) provide increased shielding either by geometry or by weight or by a combination of both;
- by geometry: perform a 3-D analysis (solid angle sectoring) and rearrange other equipment on board the satellite in order to provide maximum protection to sensitive part over greatest possible fraction of solid angle.*
- by weight: place additional shields around sensitive part as needed. Clearly options (a), (b), and (d) are most readily accessible.

*A powerful computer package for complex radiation shielding and transport calculations is now operational at GSFC. It is capable of addressing such topics as: (a) material mixtures, cross sections for protons, electrons, heavy charge particles, and neutrons, including source spectra and response functions; (b) source geometry, detector geometry, surfaces, rays, bodies, regions, body intersections, body unions, simple meshes, design bodies, spacecraft rays, with diverse features such as combinatorial options, translate-rotate-replicate capabilities, etc.; (c) heavy charged particle applications-1D transport by numerical integration, small volume pulse height (soft errors), 3D ray trace sectoring, 3D adjoint Monte Carlo; (d) electron bremsstrahlung-1D transport by numerical integration and by adjoint Monte

Carlo, small volume pulse height, 3D ray trace sectoring, 3D forward and adjoint Monte Carlo, energy deposition, charging distributions.

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ORBITAL FLUX STUDY IN COSMOPOLIC PARTICLE ENVIRONMENTS: VETTER, A.E. 1972 FOR SOLAR MAXIMUM
 UNCERTAINTY FACTORS APPLIED FOR THIS RUN ARE: FOR PROTONS (AP) - JF= 1.0; FOR INNER ZONE ELECTRONS (AF) - JF= 1.0
 MAGNETIC COORDINATES H AND L COMPUTED BY INVARA OF 1972 WITH ALL MAG. MODEL 4: DAPKAC (UGH ET AL. 1968-TRM 1975) TIME= 1989.5
 ** VEHICLE: NWLXLS-MAX **
 ** FOR INFORMATION OR EXPLANATION CONTACT: C. STASSINGPULS AT NASA-JSC, CUTE FUL. GREENGLT, MAPYLAND 20771-1FL (301)-344-8067 **

 ** SPECTRAL DISTRIBUTION: NORMALIZED BY FLUX OF ENERGY GREATER THAN 5.000MEV **

ENERGY LEVELS >(MEV)	1.0-1.2	1.2-1.4	1.4-1.6	1.6-1.8	1.8-2.0	2.0-2.2	2.2-2.4	2.4-2.6	2.6-2.8	2.8-3.0	3.0-3.2	3.2-3.4
1000	1.05E 00	1.10E 00	1.15E 00	1.20E 00	1.25E 00	1.30E 00	1.35E 00	1.40E 00	1.45E 00	1.50E 00	1.55E 00	1.60E 00
2000	1.02E 00	1.05E 00	1.08E 00	1.12E 00	1.15E 00	1.18E 00	1.22E 00	1.25E 00	1.28E 00	1.32E 00	1.35E 00	1.38E 00
5000	1.00E 00	1.00E 00	1.00E 00	1.00E 00	1.00E 00	1.00E 00	1.00E 00	1.00E 00	1.00E 00	1.00E 00	1.00E 00	1.00E 00
8000	5.83E-01	9.03E-01	1.27E-01	1.54E-01	1.82E-01	2.10E-01	2.38E-01	2.66E-01	2.94E-01	3.22E-01	3.50E-01	3.78E-01
10000	6.45E-01	8.38E-01	1.27E-01	1.54E-01	1.82E-01	2.10E-01	2.38E-01	2.66E-01	2.94E-01	3.22E-01	3.50E-01	3.78E-01
25000	8.72E-01	5.69E-01	4.31E-01	1.82E-01	8.00E-02	3.24E-02	1.29E-02	5.68E-03	2.49E-03	1.10E-03	5.00E-04	2.51E-03
50000	7.46E-01	4.20E-01	3.02E-01	1.04E-01	3.01E-02	7.20E-03	2.14E-03	5.81E-04	1.67E-04	4.90E-05	1.40E-05	4.00E-06
100000	5.53E-01	2.59E-01	1.73E-01	4.38E-02	7.63E-03	1.39E-03	3.00E-04	6.90E-05	1.50E-05	3.30E-06	7.40E-07	1.60E-07
500000	2.32E-02	9.25E-03	2.23E-03	8.98E-04	3.50E-04	1.39E-04	5.00E-05	1.80E-05	6.00E-06	2.00E-06	7.00E-07	2.50E-07
NORMFLUX=	1.06E 06	2.64E 08	1.57E 08	6.44E 07	3.40E 07	2.17E 07	1.13E 07	5.46E 06	1.51E 06	4.14E 05	1.25E 05	5.65E 04
ENERGY LEVELS >(MEV)	3.4-3.6	3.6-3.8	3.8-4.0	4.0-4.2	4.2-4.4	4.4-4.6	4.6-4.8	4.8-5.0	5.0-5.2	5.2-5.4	5.4-5.6	5.6-5.8
1000	3.85E 05	1.98E 07	2.10E 07	1.13E 07	6.54E 06	2.23E 06	2.41E 06	4.41E 06	2.61E 06	2.61E 06	3.69E 06	1.06E 06
2000	1.38E 02	5.22E 03	2.62E 03	4.99E 02	2.39E 02	1.00E 01	3.00E 00	1.00E 00	1.00E 00	1.00E 00	1.00E 00	1.00E 00
5000	1.00E 00	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
8000	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
10000	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
25000	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
50000	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
100000	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
500000	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
NORMFLUX=	6.45E 03	3.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
ENERGY LEVELS >(MEV)	5.8-6.0	6.0-6.2	6.2-6.4	6.4-6.6	6.6-6.8	6.8-7.0	7.0-7.2	7.2-7.4	7.4-7.6	7.6-7.8	7.8-8.0	8.0-8.2
1000	3.25E 05	3.20E 05	2.25E 04	1.49E 04	3.09E 04	3.42E 03	1.25E 02	3.69E 00	0.0	0.0	0.0	0.0
2000	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
5000	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
8000	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
10000	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
25000	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
50000	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
100000	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
500000	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
NORMFLUX=	0.0	3.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
ENERGY LEVELS >(MEV)	8.2-8.4	8.4-8.6	8.6-8.8	8.8-9.0	9.0-9.2	9.2-9.4	9.4-9.6	9.6-9.8	9.8-10.0	10.0-10.2	10.2-10.4	10.4-10.6
1000	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
2000	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
5000	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
8000	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
10000	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
25000	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
50000	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
100000	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
500000	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
NORMFLUX=	0.0	3.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0

TABLE 1

[illegible]

TABLE 2

ORBITAL FLUX STUDY WITH COMPOSITE PARTICLE ENVIRONMENTS: VETTES APB: AEB: AEL7 FOR SLAR MAXIMUM UNIFLUX OF 1979
 ** UNCERTAINTY FACTORS (UF) APPLIED FOR THIS RUN ARE: FOR PROTONS (APB) - JF= 1.0; FOR INVER ZONE FLCTPCNS (AEO) - UF= 1.0
 ** MAGNETIC COORDINATES B AND L COMPUTED BY INVARA OF 1972 WITH ALLMAG, MODEL 4: BARRACUGH ET AL, 161-TPM 1975 * TIME= 1989.5
 ** VEHICLE: NW 2215-MAY ** INCLINATION: 250.00 ** PERIGEE: 250.00 ** APOGEE: 250.00 ** BUL GELLY TAPE: J05A06 ** PERIOD: 2.348 **
 ** FOR INFORMATION OR EXPLANATION CONTACT E.G. STASSINGHOULS AT NASA-JSC, CODE 501, GREENLEAF, MARLAND 20771, TEL (301)-344-8067 **
 ***** PRINTING *****
 ** SPECTRAL DISTRIBUTION : NORMALIZED BY FLUX OF ENERGY GREATER THAN 5. COCHEV **

ENERGY LEVELS > (MEV) 1.0-1.2* 1.2-1.4* 1.4-1.6* 1.6-1.8* 1.8-2.0* 2.0-2.2* 2.2-2.4* 2.4-2.6* 2.6-2.8* 2.8-3.0* 3.0-3.2* 3.2-3.4*

0.0 1.23E 00 1.41E 00 1.91E 03 3.64E 03 1.05E 01 3.98E 01 1.51E 02 6.54E 02 2.07E 03 1.48E 04 8.16E 04
 2.000 0.0 1.23E 00 1.41E 00 1.91E 03 3.64E 03 1.05E 01 3.98E 01 1.51E 02 6.54E 02 2.07E 03 1.48E 04 8.16E 04
 5.000 0.0 1.00E 00 1.00E 00 1.00E 00 1.00E 00 1.00E 00 1.00E 00 1.00E 00 1.00E 00 1.00E 00 1.00E 00 1.00E 00
 8.000 0.0 8.38E-01 7.36E-01 5.98E-01 4.64E-01 3.69E-01 3.07E-01 2.48E-01 1.86E-01 1.37E-01 9.42E-02 6.42E-02 4.56E-02
 10.000 0.0 7.34E-01 5.84E-01 4.14E-01 2.70E-01 2.07E-01 1.61E-01 1.22E-01 8.11E-02 5.16E-02 3.37E-02 2.16E-02 1.41E-02
 25.000 0.0 4.02E-01 2.12E-01 1.02E-01 5.16E-02 2.74E-02 1.42E-02 6.08E-03 3.42E-03 5.23E-04 0.0 0.0 0.0
 50.000 0.0 2.73E-01 1.25E-01 5.21E-02 1.78E-02 6.16E-03 1.99E-03 8.51E-04 9.13E-05 0.0 0.0 0.0
 100.00 0.0 1.59E-01 6.93E-02 2.21E-02 5.97E-03 1.23E-03 0.0 0.0 0.0 0.0 0.0 0.0
 500.00 0.0 3.27E-03 1.06E-03 9.86E-05 3.06E-07 0.0 0.0 0.0 0.0 0.0 0.0 0.0

NORMFLUX= 0.0 3.12E 08 2.22E 09 6.24E 08 2.25E 08 1.14E 08 5.65E 07 2.18E 07 7.16E 06 2.26E 06 6.38E 05 1.67E 05

ENERGY LEVELS > (MEV) 3.4-3.6* 3.6-3.8* 3.8-4.0* 4.0-4.2* 4.2-4.4* 4.4-4.6* 4.6-4.8* 4.8-5.0* 5.0-5.2* 5.2-5.4* 5.4-5.6* 5.6-5.8*

0.0 3.34E 05 4.40E 06 6.35E 07 3.79E 07 2.99E 07 2.46E 07 1.87E 07 2.34E 07 1.45E 07 0.0 0.0 0.0
 2.000 1.09E 02 9.98E 02 6.94E 02 1.56E 03 4.38E 02 1.14E 02 3.00E 02 1.82E 01 0.0 0.0 0.0 0.0
 5.000 1.00E 00 1.00E 00 1.00E 00 1.00E 00 1.00E 00 1.00E 00 1.00E 00 1.00E 00 1.00E 00 1.00E 00 1.00E 00
 8.000 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0
 10.000 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0
 25.000 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0
 50.000 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0
 100.00 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0
 500.00 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0

NORMFLUX= 3.95E 04 1.88E 03 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0

ENERGY LEVELS > (MEV) 5.8-6.0* 6.0-6.2* 6.2-6.4* 6.4-6.6* 6.6-6.8* 6.8-7.0* 7.0-7.2* 7.2-7.4* 7.4-7.6* 7.6-7.8* 7.8-8.0* 8.0-8.2*

0.0 3.44E 06 1.11E 06 1.11E 06 2.61E 05 2.58E 05 2.53E 04 1.58E 04 1.11E 03 2.53E 02 2.53E 02 2.53E 02 2.53E 02
 2.000 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0
 5.000 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0
 8.000 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0
 10.000 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0
 25.000 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0
 50.000 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0
 100.00 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0
 500.00 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0

NORMFLUX= 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0

ENERGY LEVELS > (MEV) 8.2-8.4* 8.4-8.6* 8.6-8.8* 8.8-9.0* 9.0-9.2* 9.2-9.4* 9.4-9.6* 9.6-9.8* 9.8-10.0* 10.0-10.2* 10.2-10.4* 10.4-10.6*

0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0
 2.000 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0
 5.000 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0
 8.000 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0
 10.000 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0
 25.000 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0
 50.000 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0
 100.00 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0
 500.00 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0

NORMFLUX= 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0

TABLE 3

[illegible]

TABLE 4

ENERGY LEVEL < (HRT)	1-0-1.0	1-0-1.2	1-0-1.4	1-0-1.6	1-0-1.8	1-0-2.0	1-0-2.2	1-0-2.4	1-0-2.6	1-0-2.8	1-0-3.0	1-0-3.2	1-0-3.4	1-0-3.6	1-0-3.8	1-0-4.0	1-0-4.2	1-0-4.4	1-0-4.6	1-0-4.8	1-0-5.0	1-0-5.2	1-0-5.4	1-0-5.6	1-0-5.8	1-0-6.0	1-0-6.2	1-0-6.4	1-0-6.6	1-0-6.8	1-0-7.0	1-0-7.2	1-0-7.4	1-0-7.6	1-0-7.8	1-0-8.0	1-0-8.2	1-0-8.4	1-0-8.6	1-0-8.8	1-0-9.0	1-0-9.2	1-0-9.4	1-0-9.6	1-0-9.8	1-0-10.0	1-0-10.2	1-0-10.4	1-0-10.6	1-0-10.8	1-0-11.0	1-0-11.2	1-0-11.4	1-0-11.6	1-0-11.8	1-0-12.0	1-0-12.2	1-0-12.4	1-0-12.6	1-0-12.8	1-0-13.0	1-0-13.2	1-0-13.4	1-0-13.6	1-0-13.8	1-0-14.0	1-0-14.2	1-0-14.4	1-0-14.6	1-0-14.8	1-0-15.0	1-0-15.2	1-0-15.4	1-0-15.6	1-0-15.8	1-0-16.0	1-0-16.2	1-0-16.4	1-0-16.6	1-0-16.8	1-0-17.0	1-0-17.2	1-0-17.4	1-0-17.6	1-0-17.8	1-0-18.0	1-0-18.2	1-0-18.4	1-0-18.6	1-0-18.8	1-0-19.0	1-0-19.2	1-0-19.4	1-0-19.6	1-0-19.8	1-0-20.0	1-0-20.2	1-0-20.4	1-0-20.6	1-0-20.8	1-0-21.0	1-0-21.2	1-0-21.4	1-0-21.6	1-0-21.8	1-0-22.0	1-0-22.2	1-0-22.4	1-0-22.6	1-0-22.8	1-0-23.0	1-0-23.2	1-0-23.4	1-0-23.6	1-0-23.8	1-0-24.0	1-0-24.2	1-0-24.4	1-0-24.6	1-0-24.8	1-0-25.0	1-0-25.2	1-0-25.4	1-0-25.6	1-0-25.8	1-0-26.0	1-0-26.2	1-0-26.4	1-0-26.6	1-0-26.8	1-0-27.0	1-0-27.2	1-0-27.4	1-0-27.6	1-0-27.8	1-0-28.0	1-0-28.2	1-0-28.4	1-0-28.6	1-0-28.8	1-0-29.0	1-0-29.2	1-0-29.4	1-0-29.6	1-0-29.8	1-0-30.0	1-0-30.2	1-0-30.4	1-0-30.6	1-0-30.8	1-0-31.0	1-0-31.2	1-0-31.4	1-0-31.6	1-0-31.8	1-0-32.0	1-0-32.2	1-0-32.4	1-0-32.6	1-0-32.8	1-0-33.0	1-0-33.2	1-0-33.4	1-0-33.6	1-0-33.8	1-0-34.0	1-0-34.2	1-0-34.4	1-0-34.6	1-0-34.8	1-0-35.0	1-0-35.2	1-0-35.4	1-0-35.6	1-0-35.8	1-0-36.0	1-0-36.2	1-0-36.4	1-0-36.6	1-0-36.8	1-0-37.0	1-0-37.2	1-0-37.4	1-0-37.6	1-0-37.8	1-0-38.0	1-0-38.2	1-0-38.4	1-0-38.6	1-0-38.8	1-0-39.0	1-0-39.2	1-0-39.4	1-0-39.6	1-0-39.8	1-0-40.0	1-0-40.2	1-0-40.4	1-0-40.6	1-0-40.8	1-0-41.0	1-0-41.2	1-0-41.4	1-0-41.6	1-0-41.8	1-0-42.0	1-0-42.2	1-0-42.4	1-0-42.6	1-0-42.8	1-0-43.0	1-0-43.2	1-0-43.4	1-0-43.6	1-0-43.8	1-0-44.0	1-0-44.2	1-0-44.4	1-0-44.6	1-0-44.8	1-0-45.0	1-0-45.2	1-0-45.4	1-0-45.6	1-0-45.8	1-0-46.0	1-0-46.2	1-0-46.4	1-0-46.6	1-0-46.8	1-0-47.0	1-0-47.2	1-0-47.4	1-0-47.6	1-0-47.8	1-0-48.0	1-0-48.2	1-0-48.4	1-0-48.6	1-0-48.8	1-0-49.0	1-0-49.2	1-0-49.4	1-0-49.6	1-0-49.8	1-0-50.0	1-0-50.2	1-0-50.4	1-0-50.6	1-0-50.8	1-0-51.0	1-0-51.2	1-0-51.4	1-0-51.6	1-0-51.8	1-0-52.0	1-0-52.2	1-0-52.4	1-0-52.6	1-0-52.8	1-0-53.0	1-0-53.2	1-0-53.4	1-0-53.6	1-0-53.8	1-0-54.0	1-0-54.2	1-0-54.4	1-0-54.6	1-0-54.8	1-0-55.0	1-0-55.2	1-0-55.4	1-0-55.6	1-0-55.8	1-0-56.0	1-0-56.2	1-0-56.4	1-0-56.6	1-0-56.8	1-0-57.0	1-0-57.2	1-0-57.4	1-0-57.6	1-0-57.8	1-0-58.0	1-0-58.2	1-0-58.4	1-0-58.6	1-0-58.8	1-0-59.0	1-0-59.2	1-0-59.4	1-0-59
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TABLE 6

***** ORBITAL FLUX STUDY WITH COMPOSITE PARTICLE ENRICHMENTS: VARIOUS APPROX. 4-17 FOR SOLAR MAXIMUM *****
 ** UNCERTAINTY FACTORS (UF) APPLIED FOR THIS RUN ARE: FOR PROTONS (APB) - 1.0, FOR ALPHA - 2.0, FOR ELECTRONS - 1.0 *****
 ** MAGNETIC COORDINATES H AND L CALCULATED BY INVADE OF 1972 WITH ALLMASS APPROPRIATE ELECTRON FLUX TABLES: 1972-1973, 1973-1974, 1974-1975, 1975-1976, 1976-1977, 1977-1978, 1978-1979 *****
 ** THESE ARE MAXIMUMS *****
 ** FOR INFORMATION OF EXPLANATION: CONTACT E.O. STASSELCOULDS AT NASA-JSC, COMPLEX 210, CHASE BLVD., HOUSTON, TEXAS 77058 *****

 ** SPECTRAL DISTRIBUTION: NORMALIZED BY FLUX OF ENERGY PER ATOM PER SECOND *****

ENERGY LEVELS >(MEV)	1-5-6-7-8-9-10-11-12-13-14-15-16-17-18-19-20-21-22-23-24-25-26-27-28-29-30-31-32-33-34-35-36-37-38-39-40-41-42-43-44-45-46-47-48-49-50-51-52-53-54-55-56-57-58-59-60-61-62-63-64-65-66-67-68-69-70-71-72-73-74-75-76-77-78-79-80-81-82-83-84-85-86-87-88-89-90-91-92-93-94-95-96-97-98-99-100-101-102-103-104-105-106-107-108-109-110-111-112-113-114-115-116-117-118-119-120-121-122-123-124-125-126-127-128-129-130-131-132-133-134-135-136-137-138-139-140-141-142-143-144-145-146-147-148-149-150-151-152-153-154-155-156-157-158-159-160-161-162-163-164-165-166-167-168-169-170-171-172-173-174-175-176-177-178-179-180-181-182-183-184-185-186-187-188-189-190-191-192-193-194-195-196-197-198-199-200-201-202-203-204-205-206-207-208-209-210-211-212-213-214-215-216-217-218-219-220-221-222-223-224-225-226-227-228-229-230-231-232-233-234-235-236-237-238-239-240-241-242-243-244-245-246-247-248-249-250-251-252-253-254-255-256-257-258-259-260-261-262-263-264-265-266-267-268-269-270-271-272-273-274-275-276-277-278-279-280-281-282-283-284-285-286-287-288-289-290-291-292-293-294-295-296-297-298-299-300-301-302-303-304-305-306-307-308-309-310-311-312-313-314-315-316-317-318-319-320-321-322-323-324-325-326-327-328-329-330-331-332-333-334-335-336-337-338-339-340-341-342-343-344-345-346-347-348-349-350-351-352-353-354-355-356-357-358-359-360-361-362-363-364-365-366-367-368-369-370-371-372-373-374-375-376-377-378-379-380-381-382-383-384-385-386-387-388-389-390-391-392-393-394-395-396-397-398-399-400-401-402-403-404-405-406-407-408-409-410-411-412-413-414-415-416-417-418-419-420-421-422-423-424-425-426-427-428-429-430-431-432-433-434-435-436-437-438-439-440-441-442-443-444-445-446-447-448-449-450-451-452-453-454-455-456-457-458-459-460-461-462-463-464-465-466-467-468-469-470-471-472-473-474-475-476-477-478-479-480-481-482-483-484-485-486-487-488-489-490-491-492-493-494-495-496-497-498-499-500-501-502-503-504-505-506-507-508-509-510-511-512-513-514-515-516-517-518-519-520-521-522-523-524-525-526-527-528-529-530-531-532-533-534-535-536-537-538-539-540-541-542-543-544-545-546-547-548-549-550-551-552-553-554-555-556-557-558-559-560-561-562-563-564-565-566-567-568-569-570-571-572-573-574-575-576-577-578-579-580-581-582-583-584-585-586-587-588-589-590-591-592-593-594-595-596-597-598-599-600-601-602-603-604-605-606-607-608-609-610-611-612-613-614-615-616-617-618-619-620-621-622-623-624-625-626-627-628-629-630-631-632-633-634-635-636-637-638-639-640-641-642-643-644-645-646-647-648-649-650-651-652-653-654-655-656-657-658-659-660-661-662-663-664-665-666-667-668-669-670-671-672-673-674-675-676-677-678-679-680-681-682-683-684-685-686-687-688-689-690-691-692-693-694-695-696-697-698-699-700-701-702-703-704-705-706-707-708-709-710-711-712-713-714-715-716-717-718-719-720-721-722-723-724-725-726-727-728-729-730-731-732-733-734-735-736-737-738-739-740-741-742-743-744-745-746-747-748-749-750-751-752-753-754-755-756-757-758-759-760-761-762-763-764-765-766-767-768-769-770-771-772-773-774-775-776-777-778-779-780-781-782-783-784-785-786-787-788-789-790-791-792-793-794-795-796-797-798-799-800-801-802-803-804-805-806-807-808-809-810-811-812-813-814-815-816-817-818-819-820-821-822-823-824-825-826-827-828-829-830-831-832-833-834-835-836-837-838-839-840-841-842-843-844-845-846-847-848-849-850-851-852-853-854-855-856-857-858-859-860-861-862-863-864-865-866-867-868-869-870-871-872-873-874-875-876-877-878-879-880-881-882-883-884-885-886-887-888-889-890-891-892-893-894-895-896-897-898-899-900-901-902-903-904-905-906-907-908-909-910-911-912-913-914-915-916-917-918-919-920-921-922-923-924-925-926-927-928-929-930-931-932-933-934-935-936-937-938-939-940-941-942-943-944-945-946-947-948-949-950-951-952-953-954-955-956-957-958-959-960-961-962-963-964-965-966-967-968-969-970-971-972-973-974-975-976-977-978-979-980-981-982-983-984-985-986-987-988-989-990-991-992-993-994-995-996-997-998-999-1000-1001-1002-1003-1004-1005-1006-1007-1008-1009-1010-1011-1012-1013-1014-1015-1016-1017-1018-1019-1020-1021-1022-1023-1024-1025-1026-1027-1028-1029-1030-1031-1032-1033-1034-1035-1036-1037-1038-1039-1040-1041-1042-1043-1044-1045-1046-1047-1048-1049-1050-1051-1052-1053-1054-1055-1056-1057-1058-1059-1060-1061-1062-1063-1064-1065-1066-1067-1068-1069-1070-1071-1072-1073-1074-1075-1076-1077-1078-1079-1080-1081-1082-1083-1084-1085-1086-1087-1088-1089-1090-1091-1092-1093-1094-1095-1096-1097-1098-1099-1100-1101-1102-1103-1104-1105-1106-1107-1108-1109-1110-1111-1112-1113-1114-1115-1116-1117-1118-1119-1120-1121-1122-1123-1124-1125-1126-1127-1128-1129-1130-1131-1132-1133-1134-1135-1136-1137-1138-1139-1140-1141-1142-1143-1144-1145-1146-1147-1148-1149-1150-1151-1152-1153-1154-1155-1156-1157-1158-1159-1160-1161-1162-1163-1164-1165-1166-1167-1168-1169-1170-1171-1172-1173-1174-1175-1176-1177-1178-1179-1180-1181-1182-1183-1184-1185-1186-1187-1188-1189-1190-1191-1192-1193-1194-1195-1196-1197-1198-1199-1200-1201-1202-1203-1204-1205-1206-1207-1208-1209-1210-1211-1212-1213-1214-1215-1216-1217-1218-1219-1220-1221-1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222-2223-2224-2225-2226-2227-2228-2229-2230-2231-2232-2233-2234-2235-2236-2237-2238-2239-2240-2241-2242-2243-2244-2245-2246-2247-2248-2249-2250-2251-2252-2253-2254-2255-2256-2257-2258-2259-2260-2261-2262-2263-2264-2265-2266-2267-2268-2269-2270-2271-2272-2273-2274-2275-2276-2277-2278-2279-2280-2281-2282-2283-2284-2285-2286-2287-2288-2289-2290-2291-2292-2293-2294-2295-2296-2297-2298-2299-2300-2301-2302-2303-2304-2305-2306-2307-2308-2309-2310-2311-2312-2313-2314-2315-2316-2317-2318-2319-2320-2321-2322-2323-2324-2325-2326-2327-2328-2329-2330-2331-2332-2333-2334-2335-2336-2337-2338-2339-2340-2341-2342-2343-2344-2345-2346-2347-2348-2349-2350-2351-2352-2353-2354-2355-2356-2357-2358-2359-2360-2361-2362-2363-2364-2365-2366-2367-2368-2369-2370-2371-2372-2373-2374-2375-2376-2377-2378-2379-2380-2381-2382-2383-2384-2385-2386-2387-2388-2389-2390-2391-2392-2393-2394-2395-2396-2397-2398-2399-2400-2401-2402-2403-2404-2405-2406-2407-2408-2409-2410-2411-2412-2413-2414-2415-2416-2417-2418-2419-2420-2421-2422-2423-2424-2425-2426-2427-2428-2429-2430-2431-2432-2433-2434-2435-2436-2437-2438-2439-2440-2441-2442-2443-2444-2445-2446-2447-2448-2449-2450-2451-2452-2453-2454-2455-2456-2457-2458-2459-2460-2461-2462-2463-2464-2465-2466-2467-2468-2469-2470-2471-2472-2473-2474-2475-2476-2477-2478-2479-2480-2481-2482-2483-2484-2485-2486-2487-2488-2489-2490-2491-2492-2493-2494-2495-2496-2497-2498-2499-2500-2501-2502-2503-2504-2505-2506-2507-2508-2509-2510-2511-2512-2513-2514-2515-2516-2517-2518-2519-2520-2521-2522-2523-2524-2525-2526-2527-2528-2529-2530-2531-2532-2533-2534-2535-2536-2537-2538-2539-2540-2541-2542-2543-2544-2545-2546-2547-2548-2549-2550-2551-2552-2553-2554-2555-2556-2557-2558-2559-2560-2561-2562-2563-2564-2565-2566-2567-2568-2569-2570-2571-2572-2573-2574-2575-2576-2577-2578-2579-2580-2581-2582-2583-2584-2585-2586-2587-2588-2589-2590-2591-2592-2593-2594-2595-2596-2597-2598-2599-2600-2601-2602-2603-2604-2605-2606-2607-2608-2609-2610-2611-2612-2613-2614-2615-2616-2617-2618-26
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ENERGY LEVELS > (MEV)	1.0-1.2	1.2-1.4	1.4-1.6	1.6-1.8	1.8-2.0	2.0-2.2	2.2-2.4	2.4-2.6	2.6-2.8	2.8-3.0	3.0-3.2	3.2-3.4	3.4-3.6	3.6-3.8	3.8-4.0	4.0-4.2	4.2-4.4	4.4-4.6	4.6-4.8	4.8-5.0	5.0-5.2	5.2-5.4	5.4-5.6	5.6-5.8	5.8-6.0	6.0-6.2	6.2-6.4	6.4-6.6	6.6-6.8	6.8-7.0	7.0-7.2	7.2-7.4	7.4-7.6	7.6-7.8	7.8-8.0	8.0-8.2	8.2-8.4	8.4-8.6	8.6-8.8	8.8-9.0	9.0-9.2	9.2-9.4	9.4-9.6	9.6-9.8	9.8-10.0	10.0-10.2	10.2-10.4	10.4-10.6	10.6-10.8	10.8-11.0	11.0-11.2	11.2-11.4	11.4-11.6	11.6-11.8	11.8-12.0	12.0-12.2	12.2-12.4	12.4-12.6	12.6-12.8	12.8-13.0	13.0-13.2	13.2-13.4	13.4-13.6	13.6-13.8	13.8-14.0	14.0-14.2	14.2-14.4	14.4-14.6	14.6-14.8	14.8-15.0	15.0-15.2	15.2-15.4	15.4-15.6	15.6-15.8	15.8-16.0	16.0-16.2	16.2-16.4	16.4-16.6	16.6-16.8	16.8-17.0	17.0-17.2	17.2-17.4	17.4-17.6	17.6-17.8	17.8-18.0	18.0-18.2	18.2-18.4	18.4-18.6	18.6-18.8	18.8-19.0	19.0-19.2	19.2-19.4	19.4-19.6	19.6-19.8	19.8-20.0	20.0-20.2	20.2-20.4	20.4-20.6	20.6-20.8	20.8-21.0	21.0-21.2	21.2-21.4	21.4-21.6	21.6-21.8	21.8-22.0	22.0-22.2	22.2-22.4	22.4-22.6	22.6-22.8	22.8-23.0	23.0-23.2	23.2-23.4	23.4-23.6	23.6-23.8	23.8-24.0	24.0-24.2	24.2-24.4	24.4-24.6	24.6-24.8	24.8-25.0	25.0-25.2	25.2-25.4	25.4-25.6	25.6-25.8	25.8-26.0	26.0-26.2	26.2-26.4	26.4-26.6	26.6-26.8	26.8-27.0	27.0-27.2	27.2-27.4	27.4-27.6	27.6-27.8	27.8-28.0	28.0-28.2	28.2-28.4	28.4-28.6	28.6-28.8	28.8-29.0	29.0-29.2	29.2-29.4	29.4-29.6	29.6-29.8	29.8-30.0	30.0-30.2	30.2-30.4	30.4-30.6	30.6-30.8	30.8-31.0	31.0-31.2	31.2-31.4	31.4-31.6	31.6-31.8	31.8-32.0	32.0-32.2	32.2-32.4	32.4-32.6	32.6-32.8	32.8-33.0	33.0-33.2	33.2-33.4	33.4-33.6	33.6-33.8	33.8-34.0	34.0-34.2	34.2-34.4	34.4-34.6	34.6-34.8	34.8-35.0	35.0-35.2	35.2-35.4	35.4-35.6	35.6-35.8	35.8-36.0	36.0-36.2	36.2-36.4	36.4-36.6	36.6-36.8	36.8-37.0	37.0-37.2	37.2-37.4	37.4-37.6	37.6-37.8	37.8-38.0	38.0-38.2	38.2-38.4	38.4-38.6	38.6-38.8	38.8-39.0	39.0-39.2	39.2-39.4	39.4-39.6	39.6-39.8	39.8-40.0	40.0-40.2	40.2-40.4	40.4-40.6	40.6-40.8	40.8-41.0	41.0-41.2	41.2-41.4	41.4-41.6	41.6-41.8	41.8-42.0	42.0-42.2	42.2-42.4	42.4-42.6	42.6-42.8	42.8-43.0	43.0-43.2	43.2-43.4	43.4-43.6	43.6-43.8	43.8-44.0	44.0-44.2	44.2-44.4	44.4-44.6	44.6-44.8	44.8-45.0	45.0-45.2	45.2-45.4	45.4-45.6	45.6-45.8	45.8-46.0	46.0-46.2	46.2-46.4	46.4-46.6	46.6-46.8	46.8-47.0	47.0-47.2	47.2-47.4	47.4-47.6	47.6-47.8	47.8-48.0	48.0-48.2	48.2-48.4	48.4-48.6	48.6-48.8	48.8-49.0	49.0-49.2	49.2-49.4	49.4-49.6	49.6-49.8	49.8-50.0	50.0-50.2	50.2-50.4	50.4-50.6	50.6-50.8	50.8-51.0	51.0-51.2	51.2-51.4	51.4-51.6	51.6-51.8	51.8-52.0	52.0-52.2	52.2-52.4	52.4-52.6	52.6-52.8	52.8-53.0	53.0-53.2	53.2-53.4	53.4-53.6	53.6-53.8	53.8-54.0	54.0-54.2	54.2-54.4	54.4-54.6	54.6-54.8	54.8-55.0	55.0-55.2	55.2-55.4	55.4-55.6	55.6-55.8	55.8-56.0	56.0-56.2	56.2-56.4
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TABLE 9

[illegible]

2 A.11E 01

ENVELOPE	1	2	3	4	5	6	7	8	9	10	11	12	13	14	15	16	17	18	19	20	21	22	23	24	25	26	27	28	29	30	31	32	33	34	35	36	37	38	39	40	41	42	43	44	45	46	47	48	49	50	51	52	53	54	55	56	57	58	59	60	61	62	63	64	65	66	67	68	69	70	71	72	73	74	75	76	77	78	79	80	81	82	83	84	85	86	87	88	89	90	91	92	93	94	95	96	97	98	99	100
ENVELOPE	1	2	3	4	5	6	7	8	9	10	11	12	13	14	15	16	17	18	19	20	21	22	23	24	25	26	27	28	29	30	31	32	33	34	35	36	37	38	39	40	41	42	43	44	45	46	47	48	49	50	51	52	53	54	55	56	57	58	59	60	61	62	63	64	65	66	67	68	69	70	71	72	73	74	75	76	77	78	79	80	81	82	83	84	85	86	87	88	89	90	91	92	93	94	95	96	97	98	99	100
LEVEL	1	2	3	4	5	6	7	8	9	10	11	12	13	14	15	16	17	18	19	20	21	22	23	24	25	26	27	28	29	30	31	32	33	34	35	36	37	38	39	40	41	42	43	44	45	46	47	48	49	50	51	52	53	54	55	56	57	58	59	60	61	62	63	64	65	66	67	68	69	70	71	72	73	74	75	76	77	78	79	80	81	82	83	84	85	86	87	88	89	90	91	92	93	94	95	96	97	98	99	100
LEVEL	1	2	3	4	5	6	7	8	9	10	11	12	13	14	15	16	17	18	19	20	21	22	23	24	25	26	27	28	29	30	31	32	33	34	35	36	37	38	39	40	41	42	43	44	45	46	47	48	49	50	51	52	53	54	55	56	57	58	59	60	61	62	63	64	65	66	67	68	69	70	71	72	73	74	75	76	77	78	79	80	81	82	83	84	85	86	87	88	89	90	91	92	93	94	95	96	97	98	99	100
LEVEL	1	2	3	4	5	6	7	8	9	10	11	12	13	14	15	16	17	18	19	20	21	22	23	24	25	26	27	28	29	30	31	32	33	34	35	36	37	38	39	40	41	42	43	44	45	46	47	48	49	50	51	52	53	54	55	56	57	58	59	60	61	62	63	64	65	66	67	68	69	70	71	72	73	74	75	76	77	78	79	80	81	82	83	84	85	86	87	88	89	90	91	92	93	94	95	96	97	98	99	100
LEVEL	1	2	3	4	5	6	7	8	9	10	11	12	13	14	15	16	17	18	19	20	21	22	23	24	25	26	27	28	29	30	31	32	33	34	35	36	37	38	39	40	41	42	43	44	45	46	47	48	49	50	51	52	53	54	55	56	57	58	59	60	61	62	63	64	65	66	67	68	69	70	71	72	73	74	75	76	77	78	79	8																				

***** SPECTRUM IN PERCENT DELTA ENERGY *****									
ENERGY RANGES	AVERAGED #/CM ² /2/SEC	AVERAGED TOTAL FLUX #/CM ² /2/DAY	SPECTRUM PERCENT	ENERGY LEVELS >(MEV)	AVERAGED INITIAL FLUX #/CM ² /2/SEC	AVERAGED INITIAL FLUX #/CM ² /2/DAY	AVERAGED INITIAL FLUX #/CM ² /2/TAU	AVRAGED DIFFER. FLUX #/CM ² /2/SEC	AVRAGED DIFFER. FLUX #/CM ² /2/TAU
1000-2.000	2.094E 05	1.809E 10	61.185	9.000E	0.134E 22L 05	2.956E 10	5.375E 12	2.556E 03	
2.000-3.000	1.381E 03	2.809E 08	3.837	1.000E	0.021E 12E 05	1.594E 10	4.224E 13	1.484E 03	
3.000-4.000	1.381E 03	2.809E 08	3.837	1.000E	0.021E 12E 05	1.594E 10	4.224E 13	1.484E 03	
4.000-5.000	1.381E 03	2.809E 08	3.837	1.000E	0.021E 12E 05	1.594E 10	4.224E 13	1.484E 03	
5.000-6.000	1.381E 03	2.809E 08	3.837	1.000E	0.021E 12E 05	1.594E 10	4.224E 13	1.484E 03	
6.000-7.000	1.381E 03	2.809E 08	3.837	1.000E	0.021E 12E 05	1.594E 10	4.224E 13	1.484E 03	
7.000-8.000	1.381E 03	2.809E 08	3.837	1.000E	0.021E 12E 05	1.594E 10	4.224E 13	1.484E 03	
8.000-9.000	1.381E 03	2.809E 08	3.837	1.000E	0.021E 12E 05	1.594E 10	4.224E 13	1.484E 03	
9.000-10.000	1.381E 03	2.809E 08	3.837	1.000E	0.021E 12E 05	1.594E 10	4.224E 13	1.484E 03	
10.000-11.000	1.381E 03	2.809E 08	3.837	1.000E	0.021E 12E 05	1.594E 10	4.224E 13	1.484E 03	
11.000-12.000	1.381E 03	2.809E 08	3.837	1.000E	0.021E 12E 05	1.594E 10	4.224E 13	1.484E 03	
12.000-13.000	1.381E 03	2.809E 08	3.837	1.000E	0.021E 12E 05	1.594E 10	4.224E 13	1.484E 03	
13.000-14.000	1.381E 03	2.809E 08	3.837	1.000E	0.021E 12E 05	1.594E 10	4.224E 13	1.484E 03	
14.000-15.000	1.381E 03	2.809E 08	3.837	1.000E	0.021E 12E 05	1.594E 10	4.224E 13	1.484E 03	
15.000-16.000	1.381E 03	2.809E 08	3.837	1.000E	0.021E 12E 05	1.594E 10	4.224E 13	1.484E 03	
16.000-17.000	1.381E 03	2.809E 08	3.837	1.000E	0.021E 12E 05	1.594E 10	4.224E 13	1.484E 03	
17.000-18.000	1.381E 03	2.809E 08	3.837	1.000E	0.021E 12E 05	1.594E 10	4.224E 13	1.484E 03	
18.000-19.000	1.381E 03	2.809E 08	3.837	1.000E	0.021E 12E 05	1.594E 10	4.224E 13	1.484E 03	
19.000-20.000	1.381E 03	2.809E 08	3.837	1.000E	0.021E 12E 05	1.594E 10	4.224E 13	1.484E 03	
20.000-21.000	1.381E 03	2.809E 08	3.837	1.000E	0.021E 12E 05	1.594E 10	4.224E 13	1.484E 03	
21.000-22.000	1.381E 03	2.809E 08	3.837	1.000E	0.021E 12E 05	1.594E 10	4.224E 13	1.484E 03	
22.000-23.000	1.381E 03	2.809E 08	3.837	1.000E	0.021E 12E 05	1.594E 10	4.224E 13	1.484E 03	
23.000-24.000	1.381E 03	2.809E 08	3.837	1.000E	0.021E 12E 05	1.594E 10	4.224E 13	1.484E 03	
24.000-25.000	1.381E 03	2.809E 08	3.837	1.000E	0.021E 12E 05	1.594E 10	4.224E 13	1.484E 03	
25.000-26.000	1.381E 03	2.809E 08	3.837	1.000E	0.021E 12E 05	1.594E 10	4.224E 13	1.484E 03	
26.000-27.000	1.381E 03	2.809E 08	3.837	1.000E	0.021E 12E 05	1.594E 10	4.224E 13</		

TABLE 13

***** SPECTRUM IN PERCENT DELTA ENERGY *****									
ENERGY RANGES		AVERAGED TOTAL FLUX		SPECTRUM		***** COMPOSITE UNIT SPECTRUM *****			
#/CM**2/SEC		#/CM**2/SEC		PERCENT		TAU = 5.0000 YHLS			
ENERGY	AVG	TOTAL	AVG	TOTAL	PERCENT	INTEGRAL	INTEGRAL	INTEGRAL	INTEGRAL
>(MEV)						LEVELS	FLUX	FLUX	FLUX
#/CM**2/SEC		#/CM**2/SEC		#/CM**2/SEC		#/CM**2/SEC		#/CM**2/SEC	
1.000-5.000	1.038E 02	8.994E 11	4.742E 10	1.783	1.783	1.038E 02	8.994E 11	4.742E 10	1.783
5.000-1.000	2.698E 04	3.231E 10	1.783	1.783	1.783	2.698E 04	3.231E 10	1.783	1.783
1.000-1.500	3.758E 04	3.272E 09	0.247	0.247	0.247	3.758E 04	3.272E 09	0.247	0.247
1.500-2.000	1.347E 04	1.154E 09	0.089	0.089	0.089	1.347E 04	1.154E 09	0.089	0.089
2.000-3.000	8.336E 03	7.202E 08	0.355	0.355	0.355	8.336E 03	7.202E 08	0.355	0.355
3.000-4.000	1.910E 03	1.642E 08	0.013	0.013	0.013	1.910E 03	1.642E 08	0.013	0.013
4.000-5.000	1.259E 02	2.765E 07	0.002	0.002	0.002	1.259E 02	2.765E 07	0.002	0.002
5.000-6.000	1.873E 01	1.618E 06	0.300	0.300	0.300	1.873E 01	1.618E 06	0.300	0.300
6.000-OVER	1.441E 02	1.243E 01	0.000	0.000	0.000	1.441E 02	1.243E 01	0.000	0.000
TOTAL	1.062E 07	9.172E 11	70.104			1.062E 07	9.172E 11	70.104	
***** EXPOSURE INDEX: ENERGY>5.000 MEV *****									
INTENSITY		EXPOSURE		TOTAL # OF					
RANGES		DURATION		A-CUMULATED					
#/CM**2/SEC		(HOURS)		PARTICLES					
2.000-FLUX	0.917	0.0	0.0	0.0	0.0	0.917	0.0	0.0	0.0
1.00-1.0E	0.350	7.418E	02	2.284E	02	0.350	7.418E	02	2.284E
1.0E-1.0E2	0.133	2.284E	02	1.155E	02	0.133	2.284E	02	1.155E
1.0E-1.0E3	0.717	1.155E	02	6.154E	07	0.717	1.155E	02	6.154E
1.0E-1.0E4	2.467	6.154E	07	9.148E	08	2.467	6.154E	07	9.148E
1.0E-1.0E5	2.083	9.148E	08	1.343E	10	2.083	9.148E	08	1.343E
1.0E-1.0E6	10.800	1.343E	10	1.421F	10	10.800	1.343E	10	1.421F
1.0E-1.0E7	2.233	1.421F	10	0.0	0.0	2.233	1.421F	10	0.0
1.0E-OVER	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
TOTAL	24.000	2.462E	10			24.000	2.462E	10	

TABLE 16

***** ORBITAL FLUX STUDY WITH COMPOSITE PARTICLE ENVIRONMENTS: VETTES AP4; AEO, AP17 FOR SCALAR MAXIMUM UNIFLUX OF 1979 *****
 ** UNCEP TANTIC COORDINATES (UP) APPLIED FOR THIS RUN ARE: FOR PROTONS (APR) - UP= 1.3; FOR INNER ZONE ELECTRONS (AEO) - UP= 1.0
 ** MAGNETIC COORDINATES (UP) COMPUTED BY INABA OF 1972 WITH ALL TAG MODEL 4; DARRACLOUGH ET AL. 168-TRM 1975 & TIME= 1989.5
 ** VEHICLE : WILKINS-35-MAX ** INCLINATION = 60DEG ** PERIGEE = 3489KM ** APOGEE = 3839KM ** B/L CRUISE TAPE: TD5446 ** PERIOD= 2.876
 ** FOR INFORMATION OR EXPLANATION CONTACT E.C. STASINPOULOS AT NASA-GSPC CODE 631, GREENBELT, MARYLAND 20771 TEL. (301)-344-6367 *****
 ***** ERTONS *****

***** SPECTRUM IN PERCENT DELTA ENERGY *****
 ENERGY AVERAGED SPECTRUM
 RANGES TOTAL FLUX AVERAGED TOTAL FLUX
 > (MEV) #/CM**2/SEC #/CM**2/DAY PERCENT
 1000-2-000 2.547E 06 2-201E 11 03-270
 2-000-5-000 1.482E 05 1-281E 10 3-624
 5-000-10-000 7.726E 04 6-676E 09 1-885
 10-000-25-000 3.075E 04 2-677E 09 0-752
 25-000-50-000 4.745E 04 4-099E 09 1-100
 50-000-100-000 4.418E 03 3-817E 08 0-038
 100-0-500-0 1.559E 03 1-347E 08 0-000
 500-0-OVER 1.139E 01 9-841E 07 0-000
 1-019E 05 8-804E 05 0-000
 TOTAL 2-858E 06 2-470E 11 69.368

***** EXPOSURE INDEX: ENERGY>5.000 MEV *****
 INTENSITY EXPOSURE TOTAL # OF
 RANGES DURATION ACCUMULATED
 #/CM**2/SEC (HOURS) PARTICLES
 ZERC FLUX 7-233 0-0
 1-20-1-21 0-283 7-943E 03
 1-21-1-22 1-117 1-725E 05
 1-22-1-23 1-467 2-107E 06
 1-23-1-24 1-833 2-735E 07
 1-24-1-25 3-817 6-138E 08
 1-25-1-26 7-883 1-200E 10
 1-26-1-27 0-367 1-407E 09
 1-27-OVER 0-0 0-0
 TOTAL 24-000 1-405E 10

*** COMPOSITE ORBIT SPECTRUM *** TAU= 5.0000 YEAR(S)
 ENERGY AVERAGED AVERAGED AVERAGED
 LEVELS INTEGR. FLUX INTEGR. FLUX INTEGR. FLUX
 > (MEV) #/CM**2/SEC #/CM**2/DAY #/CM**2/TAU
 0.000E-014 0.031E 00 3.537E 11 1.4
 0.000E-013 0.858E 06 2.470E 10 1.4
 0.000E-012 7.807E 05 6.676E 09 1.4
 1.000E-011 4.745E 04 4.099E 09 1.4
 2.000E-010 3.075E 04 2.677E 09 1.4
 3.000E-009 4.418E 03 3.817E 08 1.4
 4.000E-008 1.559E 03 1.347E 08 1.4
 5.000E-007 1.139E 01 9.841E 07 1.4
 6.000E-006 1.019E 05 8.804E 05 1.4
 7.000E-005 1.347E 04 5.453E 04 1.4
 8.000E-004 3.339E 04 2.470E 03 1.4
 9.000E-003 1.009E 03 7.155E 03 1.4
 10.000E-002 5.453E 03 3.339E 03 1.4
 15.000E-001 1.009E 03 3.339E 03 1.4
 20.000E-000 5.453E 03 3.339E 03 1.4
 30.000E-000 3.339E 03 3.339E 03 1.4
 40.000E-000 3.339E 03 3.339E 03 1.4
 50.000E-000 3.339E 03 3.339E 03 1.4
 60.000E-000 3.339E 03 3.339E 03 1.4
 70.000E-000 3.339E 03 3.339E 03 1.4
 80.000E-000 3.339E 03 3.339E 03 1.4
 90.000E-000 3.339E 03 3.339E 03 1.4
 100.000E-000 3.339E 03 3.339E 03 1.4
 110.000E-000 3.339E 03 3.339E 03 1.4
 120.000E-000 3.339E 03 3.339E 03 1.4
 130.000E-000 3.339E 03 3.339E 03 1.4
 140.000E-000 3.339E 03 3.339E 03 1.4
 150.000E-000 3.339E 03 3.339E 03 1.4
 160.000E-000 3.339E 03 3.339E 03 1.4
 170.000E-000 3.339E 03 3.339E 03 1.4
 180.000E-000 3.339E 03 3.339E 03 1.4
 190.000E-000 3.339E 03 3.339E 03 1.4
 200.000E-000 3.339E 03 3.339E 03 1.4
 210.000E-000 3.339E 03 3.339E 03 1.4
 220.000E-000 3.339E 03 3.339E 03 1.4
 230.000E-000 3.339E 03 3.339E 03 1.4
 240.000E-000 3.339E 03 3.339E 03 1.4
 250.000E-000 3.339E 03 3.339E 03 1.4
 260.000E-000 3.339E 03 3.339E 03 1.4
 270.000E-000 3.339E 03 3.339E 03 1.4
 280.000E-000 3.339E 03 3.339E 03 1.4
 290.000E-000 3.339E 03 3.339E 03 1.4
 300.000E-000 3.339E 03 3.339E 03 1.4
 310.000E-000 3.339E 03 3.339E 03 1.4
 320.000E-000 3.339E 03 3.339E 03 1.4
 330.000E-000 3.339E 03 3.339E 03 1.4
 340.000E-000 3.339E 03 3.339E 03 1.4
 350.000E-000 3.339E 03 3.339E 03 1.4
 360.000E-000 3.339E 03 3.339E 03 1.4
 370.000E-000 3.339E 03 3.339E 03 1.4
 380.000E-000 3.339E 03 3.339E 03 1.4
 390.000E-000 3.339E 03 3.339E 03 1.4
 400.000E-000 3.339E 03 3.339E 03 1.4
 410.000E-000 3.339E 03 3.339E 03 1.4
 420.000E-000 3.339E 03 3.339E 03 1.4
 430.000E-000 3.339E 03 3.339E 03 1.4
 440.000E-000 3.339E 03 3.339E 03 1.4
 450.000E-000 3.339E 03 3.339E 03 1.4
 460.000E-000 3.339E 03 3.339E 03 1.4
 470.000E-000 3.339E 03 3.339E 03 1.4
 480.000E-000 3.339E 03 3.339E 03 1.4
 490.000E-000 3.339E 03 3.339E 03 1.4
 500.000E-000 3.339E 03 3.339E 03 1.4

TABLE 17

TABLE 18

***** SPECTRUM IN PERCENT (ALTA ENERGY) *****				***** SPECTRUM ***** TEL. ***** (COC Y) A (C)			
ENERGY RANGES (X MEV)	AVERAGE TOTAL FLUX /CM ² SEC	AVERAGE TOTAL FLUX PERCENT	SPECTRUM	ENERGY RANGES (X MEV)	AVERAGE TOTAL FLUX /CM ² SEC	AVERAGE TOTAL FLUX PERCENT	SPECTRUM
1.000 - 2.000	5.504E 06	4.750 11	4.750 11	4.000 - 5.000	4.000E 06	4.000E 06	4.000E 06
2.000 - 3.000	4.867E 06	4.095 10	4.095 10	5.000 - 6.000	3.500E 06	3.500E 06	3.500E 06
3.000 - 4.000	4.867E 06	4.095 10	4.095 10	6.000 - 7.000	3.500E 06	3.500E 06	3.500E 06
4.000 - 5.000	4.867E 06	4.095 10	4.095 10	7.000 - 8.000	3.500E 06	3.500E 06	3.500E 06
5.000 - 6.000	4.867E 06	4.095 10	4.095 10	8.000 - 9.000	3.500E 06	3.500E 06	3.500E 06
6.000 - 7.000	4.867E 06	4.095 10	4.095 10	9.000 - 10.000	3.500E 06	3.500E 06	3.500E 06
7.000 - 8.000	4.867E 06	4.095 10	4.095 10	10.000 - 11.000	3.500E 06	3.500E 06	3.500E 06
8.000 - 9.000	4.867E 06	4.095 10	4.095 10	11.000 - 12.000	3.500E 06	3.500E 06	3.500E 06
9.000 - 10.000	4.867E 06	4.095 10	4.095 10	12.000 - 13.000	3.500E 06	3.500E 06	3.500E 06
10.000 - 11.000	4.867E 06	4.095 10	4.095 10	13.000 - 14.000	3.500E 06	3.500E 06	3.500E 06
11.000 - 12.000	4.867E 06	4.095 10	4.095 10	14.000 - 15.000	3.500E 06	3.500E 06	3.500E 06
12.000 - 13.000	4.867E 06	4.095 10	4.095 10	15.000 - 16.000	3.500E 06	3.500E 06	3.500E 06
13.000 - 14.000	4.867E 06	4.095 10	4.095 10	16.000 - 17.000	3.500E 06	3.500E 06	3.500E 06
14.000 - 15.000	4.867E 06	4.095 10	4.095 10	17.000 - 18.000	3.500E 06	3.500E 06	3.500E 06
15.000 - 16.000	4.867E 06	4.095 10	4.095 10	18.000 - 19.000	3.500E 06	3.500E 06	3.500E 06
16.000 - 17.000	4.867E 06	4.095 10	4.095 10	19.000 - 20.000	3.500E 06	3.500E 06	3.500E 06
17.000 - 18.000	4.867E 06	4.095 10	4.095 10	20.000 - 21.000	3.500E 06	3.500E 06	3.500E 06
18.000 - 19.000	4.867E 06	4.095 10	4.095 10	21.000 - 22.000	3.500E 06	3.500E 06	3.500E 06
19.000 - 20.000	4.867E 06	4.095 10	4.095 10	22.000 - 23.000	3.500E 06	3.500E 06	3.500E 06
20.000 - 21.000	4.867E 06	4.095 10	4.095 10	23.000 - 24.000	3.500E 06	3.500E 06	3.500E 06
21.000 - 22.000	4.867E 06	4.095 10	4.095 10	24.000 - 25.000	3.500E 06	3.500E 06	3.500E 06
22.000 - 23.000	4.867E 06	4.095 10	4.095 10	25.000 - 26.000	3.500E 06	3.500E 06	3.500E 06
23.000 - 24.000	4.867E 06	4.095 10	4.095 10	26.000 - 27.000	3.500E 06	3.500E 06	3.500E 06
24.000 - 25.000	4.867E 06	4.095 10	4.095 10	27.000 - 28.000	3.500E 06	3.500E 06	3.500E 06
25.000 - 26.000	4.867E 06	4.095 10	4.095 10	28.000 - 29.000	3.500E 06	3.500E 06	3.500E 06
26.000 - 27.000	4.867E 06	4.095 10	4.095 10	29.000 - 30.000	3.500E 06	3.500E 06	3.500E 06
27.000 - 28.000	4.867E 06	4.095 10	4.095 10	30.000 - 31.000	3.500E 06	3.500E 06	3.500E 06
28.000 - 29.000	4.867E 06	4.095 10	4.095 10	31.000 - 32.000	3.500E 06	3.500E 06	3.500E 06
29.000 - 30.000	4.867E 06	4.095 10	4.095 10	32.000 - 33.000	3.500E 06	3.500E 06	3.500E 06

TOTAL	24,000	20,250	10
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TABLE 20

[illegible]

198LE 202

[illegible]

TABLE 23

[illegible]

TABLE 24

***** ORBITAL FLUX STUDY WITH COMPOSITE PARTICLE ENVIRONMENTS: VECTIS APRIL 1972, ALL 17 FOR 5 LAPS MAXIMUM *****
 ** UNCERTAINTY FACTORS (U) APPLIED FOR THIS RUN ARE: FOR PROTONS (APR) - U = 1.0; FOR INDIAN ZONE ELECTRIC (APR) - U = 1.0 *****
 ** MAGNETIC COORDINATES B AND L COMPUTED BY INVARA OF 1972 WITH ALL MAG. MODEL 4: MAGNETIC COORDINATES: 1972-1975, TIME = 1989.5 *****
 ** VEHICLE: MEXI-SWAX AS INCLINATION: LOWER AS PERIGEE: 16672.4 KM AS AROUND: 16672.4 KM AS AROUND: 16672.4 KM AS AROUND: 16672.4 KM *****
 ** FOR INFORMATION OR EXPLANATION CONTACT E.S. STASSINGHOLZ AT NASA-JSC (CODE 631), DR. JAMES T. WARDLAW 20771 JPL (3311)-344-8067 *****

 ** TABLE OF PEAK AND TOTAL FLUXES PER PERIOD: ENERGY 25.000 MEV *****

PERIOD NUMBER	PEAK FLUX #/CM ² /SEC	POSITION AT WHICH ENCOUNTERED (DEG)	LATITUDE (DEG)	ALTITUDE (KM)	ORBIT TIME (HOURS)	FLUXES (CAUS)	TIME (L)	PER ORBIT #/CM ² /ORBIT
1	2.242E 04	-105.648	-16.14	1667.73	1.10000	3.18331	1.31	4.247E 07
2	2.505E 04	-11.437	-18.42	1669.46	3.46066	3.18084	1.31	4.647E 07
3	5.055E 04	-22.515	-24.40	1669.46	4.91667	3.12519	1.30	6.826E 07
4	6.826E 04	-49.059	-18.70	1669.46	7.25000	3.12346	1.29	7.525E 07
5	4.500E 04	-77.238	-18.99	1662.52	9.56666	3.12383	1.33	4.273E 07
6	2.156E 04	-106.763	-13.85	1667.54	11.89667	3.18324	1.29	3.831E 07
7	1.158E 04	-97.422	4.33	1567.14	11.94333	3.18422	1.27	1.630E 07
8	1.772E 04	-56.526	-3.97	1559.03	14.95000	3.15845	1.42	2.631E 07
9	2.803E 04	-13.543	-17.58	1669.46	17.33666	3.15997	1.55	3.863E 07
10	4.655E 04	-23.076	-19.98	1559.43	18.69999	3.13241	1.40	5.559E 07
11	2.722E 04	-46.922	-17.33	1663.53	21.33333	3.12475	1.38	6.666E 07
12	4.764E 04	-74.954	-20.69	1561.64	23.56666	3.12714	1.35	5.422E 07

***** ORBITAL FLUX STUDY WITH COMPOSITE PARTICLE ENVIRONMENTS: VECTIS APRIL 1972, ALL 17 FOR 5 LAPS MAXIMUM *****
 ** UNCERTAINTY FACTORS (U) APPLIED FOR THIS RUN ARE: FOR PROTONS (APR) - U = 1.0; FOR INDIAN ZONE ELECTRIC (APR) - U = 1.0 *****
 ** MAGNETIC COORDINATES B AND L COMPUTED BY INVARA OF 1972 WITH ALL MAG. MODEL 4: MAGNETIC COORDINATES: 1972-1975, TIME = 1989.5 *****
 ** VEHICLE: MEXI-SWAX AS INCLINATION: LOWER AS PERIGEE: 16672.4 KM AS AROUND: 16672.4 KM AS AROUND: 16672.4 KM AS AROUND: 16672.4 KM *****
 ** FOR INFORMATION OR EXPLANATION CONTACT E.S. STASSINGHOLZ AT NASA-JSC (CODE 631), DR. JAMES T. WARDLAW 20771 JPL (3311)-344-8067 *****

 ** TABLE OF PEAK AND TOTAL FLUXES PER PERIOD: ENERGY 25.000 MEV *****

PERIOD NUMBER	PEAK FLUX #/CM ² /SEC	POSITION AT WHICH ENCOUNTERED (DEG)	LATITUDE (DEG)	ALTITUDE (KM)	ORBIT TIME (HOURS)	FLUXES (CAUS)	TIME (L)	PER ORBIT #/CM ² /ORBIT
1	1.921E 04	-105.923	-20.32	1663.30	1.10067	3.15111	1.30	2.114E 07
2	2.051E 04	-16.575	-19.13	1667.20	3.46184	3.16321	1.41	1.810E 07
3	2.627E 04	-16.768	-19.71	1667.07	4.91833	3.13611	1.41	1.561E 07
4	2.571E 04	-54.034	-25.24	1669.46	7.25000	3.12774	1.41	1.264E 07
5	2.242E 04	-68.028	-25.43	1673.33	9.56666	3.16331	1.41	2.331E 07
6	1.643E 04	-68.028	4.18	1667.09	11.89333	3.15174	1.43	1.874E 07
7	1.920E 04	-49.074	4.33	1669.46	14.95000	3.15333	1.37	1.204E 07
8	1.751E 04	-49.074	-0.83	1669.46	17.33666	3.15709	1.39	1.600E 07
9	2.137E 04	-16.740	-0.83	1669.46	19.69999	3.12711	1.43	2.500E 07
10	2.761E 04	-16.740	-11.57	1669.46	21.33666	3.13774	1.43	2.715E 07
11	2.007E 04	-44.313	-20.43	1662.15	23.56667	3.12643	1.44	3.277E 07
12	2.633E 04	-73.634	-20.53	1564.00	25.59999	3.13644	1.43	1.395E 07

TABLE 25

***** ORBITAL FLUX STUDY WITH COMPOSITE PARTICLE ENVIRONMENTS: VECTORS APPLIED TO ALL 17 FOR SILENT MAXIMUM *****
 ** UNCERTAINTY FACTORS (UF) APPLIED FOR THIS RUN ARE: FOR PARTICLES (APR) - UF = 1.0; FOR THERMAL ZONE ELECTRONS (AUG) - UF = 1.0 *****
 ** MAGNETIC COORDINATES D AND L COMPUTED BY INVARIANCE OF 1972 WITH ALL MAG. MODEL 4: DATA CLUSTER ETAL. 104-104 1975 * TIME = 1989.5 *****
 ** VEHICLE: NALX2-SMALL AN INCLINATION CONTACT LUGS STATIONED AT NAVA-55FC CODE 631. GROUND STATION: WAWYLAND 20771 TEL (331)-344-4007 *****
 ** FOR INFORMATION OR EXPLANATION CONTACT LUGS STATIONED AT NAVA-55FC CODE 631. GROUND STATION: WAWYLAND 20771 TEL (331)-344-4007 *****
 ***** TABLE OF PEAK AND TOTAL FLUXES PER PERIOD: ENERGY > 0.0001 eV *****

PERIOD NUMBER	PEAK FLUX ENCOUNTERED #/CM ² /SEC	POSITION AT WHICH ENCOUNTERED (DEG)	LONGITUDE (DEG)	LATITUDE (DEG)	ALTITUDE (KM)	HEIGHT (M)	FLUXES (AUG)	LINE (AUG)	TOTAL FLUX PER PERIOD #/CM ² /GHEIT
1	1.524E 05	-110.840	-12.36	-12.36	2580.65	1.26567	3.13417	1.43	3.335E 04
2	2.084E 05	-110.840	-12.36	-12.36	2580.65	1.26567	3.13417	1.43	3.335E 04
3	3.353E 05	-28.675	-7.59	-7.59	2593.77	4.66667	3.13417	1.43	7.726E 04
4	3.182E 05	-68.455	-15.95	-15.95	2593.77	4.66667	3.13417	1.43	5.550E 04
5	1.780E 05	-101.894	-11.14	-11.14	2593.77	4.66667	3.13417	1.43	4.510E 04
6	1.244E 05	-133.599	-4.13	-4.13	2593.77	4.66667	3.13417	1.43	2.634E 04
7	1.551E 05	-24.297	-2.82	-2.82	2593.77	4.66667	3.13417	1.43	2.190E 04
8	2.420E 05	-32.803	-4.24	-4.24	2593.77	4.66667	3.13417	1.43	2.673E 04
9	3.496E 05	-36.014	-9.04	-9.04	2580.65	4.66667	3.13417	1.43	4.502E 04
10	3.230E 05	-64.424	-14.26	-14.26	2587.95	4.66667	3.13417	1.43	4.321E 04

***** ORBITAL FLUX STUDY WITH COMPOSITE PARTICLE ENVIRONMENTS: VECTORS APPLIED TO ALL 17 FOR SILENT MAXIMUM *****
 ** UNCERTAINTY FACTORS (UF) APPLIED FOR THIS RUN ARE: FOR PARTICLES (APR) - UF = 1.0; FOR THERMAL ZONE ELECTRONS (AUG) - UF = 1.0 *****
 ** MAGNETIC COORDINATES D AND L COMPUTED BY INVARIANCE OF 1972 WITH ALL MAG. MODEL 4: DATA CLUSTER ETAL. 104-104 1975 * TIME = 1989.5 *****
 ** VEHICLE: NALX2-SMALL AN INCLINATION CONTACT LUGS STATIONED AT NAVA-55FC CODE 631. GROUND STATION: WAWYLAND 20771 TEL (331)-344-4007 *****
 ** FOR INFORMATION OR EXPLANATION CONTACT LUGS STATIONED AT NAVA-55FC CODE 631. GROUND STATION: WAWYLAND 20771 TEL (331)-344-4007 *****
 ***** TABLE OF PEAK AND TOTAL FLUXES PER PERIOD: ENERGY > 0.0001 eV *****

PERIOD NUMBER	PEAK FLUX ENCOUNTERED #/CM ² /SEC	POSITION AT WHICH ENCOUNTERED (DEG)	LONGITUDE (DEG)	LATITUDE (DEG)	ALTITUDE (KM)	HEIGHT (M)	FLUXES (AUG)	LINE (AUG)	TOTAL FLUX PER PERIOD #/CM ² /GHEIT
1	3.978E 04	-106.609	-16.74	-16.74	2587.05	1.30300	3.13417	1.43	4.404E 04
2	4.451E 04	-20.431	-3.59	-3.59	2592.97	4.66667	3.13417	1.43	5.550E 04
3	5.388E 04	-20.431	-3.59	-3.59	2592.97	4.66667	3.13417	1.43	5.550E 04
4	5.212E 04	-63.431	-15.75	-15.75	2594.16	4.66667	3.13417	1.43	5.550E 04
5	4.136E 04	-103.431	-13.46	-13.46	2594.16	4.66667	3.13417	1.43	4.404E 04
6	3.478E 04	-135.431	-4.56	-4.56	2593.43	4.66667	3.13417	1.43	4.404E 04
7	4.036E 04	-31.744	-2.03	-2.03	2593.43	4.66667	3.13417	1.43	4.404E 04
8	4.721E 04	-31.744	-2.03	-2.03	2593.43	4.66667	3.13417	1.43	4.404E 04
9	5.423E 04	-36.018	-9.04	-9.04	2587.95	4.66667	3.13417	1.43	4.404E 04
10	5.267E 04	-60.914	-14.26	-14.26	2587.95	4.66667	3.13417	1.43	4.404E 04

TABLE 26

PERIOD NUMBER	PEAK FLUX ENCOUNTERED 8/CM ² /SEC	POSITION AT WHICH LONGITUDE LATITUDE (DEG)	LOCATION AT WHICH ENCOUNTERED (DEG)	CABIT TIME (HOURS)	FIELD(H) (GAUSS)	LINE(L) (E.R.)	TOTAL FLUX PER AREA 8/CM ² /CM ²
1	7.315E-05	-116.921	-7.64	5.51697	0.07470	1.43	1.74E-09
2	1.013E-06	-8.535	-1.38	5.71933	0.06435	1.70	2.09E-09
3	1.313E-05	-57.904	-1.17	4.51637	0.06435	1.70	1.55E-09
4	9.878E-05	-7.904	-2.15	11.33333	0.06435	1.95	1.29E-09
5	9.772E-05	-136.930	-4.35	15.3300	0.07442	1.65	1.43E-09
6	1.081E-06	-107.926	-3.77	15.76687	0.06435	1.70	1.53E-09
7	1.081E-06	-18.984	-3.35	18.71609	0.06435	1.70	1.53E-09
8	1.081E-06	-17.750	-3.91	21.53333	0.06435	1.70	1.53E-09

PERIOD NUMBER	PEAK FLUX ENCOUNTERED #/CM ² /SEC	POSITION AT WHICH ENCOUNTERED LONGITUDE (DEG)	ALTITUDE (DEG)	CADIT TIME (HOURS)	FIELD (U (GRUSS))	LINE (I) (I. P.)	TOTAL FLUX PER CADIT #/CM ² /CADIT
1	7.323	-115.9	-8.64	1.3167	0.6723	1.64	3.945
2	7.323	-115.9	-8.64	1.3167	0.6723	1.64	3.945
3	9.318	-57.37	0.37	1.5000	0.6630	1.71	2.643
4	9.318	-57.37	0.37	1.5000	0.6630	1.71	2.643
5	9.695	-116.30	-2.15	1.3500	0.6743	1.61	1.733
6	9.280	-107.26	-4.36	1.5767	0.6633	1.65	1.733
7	9.175	-107.26	-4.36	1.5767	0.6633	1.65	1.733
8	9.175	-107.26	-4.36	1.5767	0.6633	1.65	1.733
9	9.175	-57.750	-1.91	2.1633	0.6630	1.73	2.143

TABLE 27

***** ORBITAL FLUX STUDY WITH COMPOSITE PARTICLE ENVIRONMENTS: VETTES A601 A617 FOR SCLAR MAXIMUM ***** UNIFLUX CF 1979
 ** UNCERTAINTY FACTORS OF 1 APPLIED TO ALL DATA: 1972 WITH ALL MAG. MODELS ** JFE 1.01 EPOCH ZONE ELECTRONICS (A61) - UFE 1.0
 ** MAGNETIC COORDINATES 3 AND 4 CLAMPED TO INVADE DE 1972 WITH ALL MAG. MODELS ** JFE 1.01 EPOCH ZONE ELECTRONICS (A61) - UFE 1.0
 ** FOR INFORMATION ON EXPLANATION CONTACT: STASSINGERBULLS AT NASA-CSCG, 3101 C. C. TOWNSHIP, MARYLAND 20771, TEL. (301)-344-8067 **
 ** TABLE OF PEAK AND TOTAL FLUXES PER PERIOD: ENERGY > 100 MeV **

NUMBER	PERIOD	POSITION AT WHICH ENCOUNTERED	CREAT TIME	FILED(S)	LINE(L)	TOTAL FLUX PER ORBIT
NUMBER	PERIOD	CONJUNCTION (DEG)	(HOURS)	(COUNTS)	(E.P.)	#/CM**2/CRHIT
1	0.432E CC	54.533	0.11567	5.54200	1.57	5.215E 10
2	0.432E CC	168.055	5.54333	5.54275	1.57	5.174E 10
3	0.432E CC	107.624	5.54500	5.54275	1.56	5.194E 10
4	0.432E CC	-154.535	15.49500	5.54275	1.55	5.194E 10
5	0.432E CC	-154.537	15.49667	5.54275	1.55	5.194E 10
6	0.432E CC	104.507	20.00000	5.54275	1.56	5.194E 10

***** ORBITAL FLUX STUDY WITH COMPOSITE PARTICLE ENVIRONMENTS: VETTES A601 A617 FOR SCLAR MAXIMUM ***** UNIFLUX CF 1979
 ** UNCERTAINTY FACTORS OF 1 APPLIED TO ALL DATA: 1972 WITH ALL MAG. MODELS ** JFE 1.01 EPOCH ZONE ELECTRONICS (A61) - UFE 1.0
 ** MAGNETIC COORDINATES 3 AND 4 CLAMPED TO INVADE DE 1972 WITH ALL MAG. MODELS ** JFE 1.01 EPOCH ZONE ELECTRONICS (A61) - UFE 1.0
 ** FOR INFORMATION ON EXPLANATION CONTACT: STASSINGERBULLS AT NASA-CSCG, 3101 C. C. TOWNSHIP, MARYLAND 20771, TEL. (301)-344-8067 **
 ** TABLE OF PEAK AND TOTAL FLUXES PER PERIOD: ENERGY > 100 MeV **

NUMBER	PERIOD	POSITION AT WHICH ENCOUNTERED	CREAT TIME	FILED(S)	LINE(L)	TOTAL FLUX PER ORBIT
NUMBER	PERIOD	CONJUNCTION (DEG)	(HOURS)	(COUNTS)	(E.P.)	#/CM**2/CRHIT
1	0.432E CC	54.533	0.11567	5.54200	1.57	5.215E 10
2	0.432E CC	168.055	5.54333	5.54275	1.57	5.174E 10
3	0.432E CC	107.624	5.54500	5.54275	1.56	5.194E 10
4	0.432E CC	-154.535	15.49500	5.54275	1.55	5.194E 10
5	0.432E CC	-154.537	15.49667	5.54275	1.55	5.194E 10
6	0.432E CC	104.507	20.00000	5.54275	1.56	5.194E 10

TABLE 29

***** ORBITA FLUX STUDY WITH COMPOSITE PARTIAL ENVIRONMENTAL EFFECTS: VETES AFBI AGO 1979 FOR JLV JLV *****
 ** UNCERTAINTY FACTORS (U) APPLIED FOR THIS RUN ARE: FOR FACTORS (AFBI) U=1.00 FOR FACTORS (AFBI) U=1.00
 ** MAGNETIC COORDINATES (U) AND L COMPUTED BY INVA OF 1972 WITH ALL MAGNETIC COORDINATES (AFBI) U=1.00
 ** VECTORS (U) AND L COMPUTED BY INVA OF 1972 WITH ALL MAGNETIC COORDINATES (AFBI) U=1.00
 ** FOR INFORMATION ON EXPLANATION CONTACT 5.2. STASSINCEPULS AT NASA-CSCG 20771 TEL (301) 340-8067
 ** TABLE OF PEAK AND TOTAL FLUXES PER PERIOD: ENERGY 50-100 MeV *****

ORBITA NUMBER	FLUX ENCOUNTERED #/CM**2/SEC	POSITION AT WHICH ENCOUNTERED LONGITUDE (DEG)	LATITUDE (DEG)	HEIGHT TIME (HOURS)	FIELD (G)	LIVE (S)	TOTAL FLUX PER ORBIT #/CM**2/CR311
1	4.802 05	81.250	10.66	0.11667	0.01761	5.81	1.510E 05
2	4.810E 05	122.076	10.54	0.11667	0.01761	5.81	1.510E 05
3	4.810E 05	122.076	10.54	0.11667	0.01761	5.81	1.510E 05
4	4.810E 05	122.076	10.54	0.11667	0.01761	5.81	1.510E 05

***** ORBITA FLUX STUDY WITH COMPOSITE PARTIAL ENVIRONMENTAL EFFECTS: VETES AFBI AGO 1979 FOR JLV JLV *****
 ** UNCERTAINTY FACTORS (U) APPLIED FOR THIS RUN ARE: FOR FACTORS (AFBI) U=1.00 FOR FACTORS (AFBI) U=1.00
 ** MAGNETIC COORDINATES (U) AND L COMPUTED BY INVA OF 1972 WITH ALL MAGNETIC COORDINATES (AFBI) U=1.00
 ** VECTORS (U) AND L COMPUTED BY INVA OF 1972 WITH ALL MAGNETIC COORDINATES (AFBI) U=1.00
 ** FOR INFORMATION ON EXPLANATION CONTACT 5.2. STASSINCEPULS AT NASA-CSCG 20771 TEL (301) 340-8067
 ** TABLE OF PEAK AND TOTAL FLUXES PER PERIOD: ENERGY 50-100 MeV *****

ORBITA NUMBER	FLUX ENCOUNTERED #/CM**2/SEC	POSITION AT WHICH ENCOUNTERED LONGITUDE (DEG)	LATITUDE (DEG)	HEIGHT TIME (HOURS)	FIELD (G)	LIVE (S)	TOTAL FLUX PER ORBIT #/CM**2/CR311
1	4.802 05	81.250	10.66	0.11667	0.01761	5.81	1.510E 05
2	4.810E 05	122.076	10.54	0.11667	0.01761	5.81	1.510E 05
3	4.810E 05	122.076	10.54	0.11667	0.01761	5.81	1.510E 05
4	4.810E 05	122.076	10.54	0.11667	0.01761	5.81	1.510E 05

TABLE 30

TABLE -

TABLE -

NVLX1:S-MAX

NVLX1:S-MAX

CIRCULAR

CIRCULAR

INCLINATION: 60 DEG

INCLINATION: 60 DEG

PERIGEE: 1667 KM

PERIGEE: 1667 KM

APUGEE: 1667 KM

APUGEE: 1667 KM

**** EXPOSURE ANALYSIS ****

* PERCENT OF TOTAL LIFETIME SPENT INSIDE AND *

* OUTSIDE THE TRAPPED-PARTICLE RADIATION BELT *

ELECTRONS LE

(E > 5.000 MEV)

INLET ZONE -IT- : 67.99 %

(1.0 < L < 2.8)

PERCENT OF TOTAL LIFE-

TIME SPENT IN HIGH-

INTENSITY REGIONS* OF

VAN ALLEN BELTS :

PERCENT OF TOTAL LIFE-

TIME SPENT IN HIGH-

INTENSITY REGIONS* OF

VAN ALLEN BELTS :

PERCENT OF TOTAL DAILY

FLUX ACCUMULATED IN

HIGH-INTENSITY REGIONS*

PERCENT OF TOTAL DAILY

FLUX ACCUMULATED IN

HIGH-INTENSITY REGIONS*

PERCENT OF TOTAL DAILY

FLUX ACCUMULATED IN

HIGH-INTENSITY REGIONS*

PERCENT OF TOTAL DAILY

FLUX ACCUMULATED IN

HIGH-INTENSITY REGIONS*

PERCENT OF TOTAL DAILY

FLUX ACCUMULATED IN

HIGH-INTENSITY REGIONS*

PERCENT OF TOTAL DAILY

FLUX ACCUMULATED IN

HIGH-INTENSITY REGIONS*

PERCENT OF TOTAL DAILY

FLUX ACCUMULATED IN

HIGH-INTENSITY REGIONS*

PERCENT OF TOTAL DAILY

FLUX ACCUMULATED IN

HIGH-INTENSITY REGIONS*

TABLE 31

TABLE -

TABLE -

F.M.XP25C-MAX

F.M.XP25C-MAX

CIRCULAR

CIRCULAR

INCLINATION: 60 DEG

INCLINATION: 60 DEG

PERFECT: 2553 KM

PERFECT: 2553 KM

APERT: 2553 KM

APERT: 2553 KM

**** EXPERT ANALYSIS ****

* PERCENT OF TOTAL LIFETIME SPENT INSIDE AND *

* OUTSIDE THE TRAPPED-PARTICLE RADIATION BELT *

ELECTRONIC LC

ELECTRONIC LC

(1.25-0.000000)

(1.25-0.000000)

PERCENT OF TOTAL LIFE-

PERCENT OF TOTAL LIFE-

TIME SPENT IN FLUX-FAST

TIME SPENT IN FLUX-FAST

REGION-OF-SPACE *

REGION-OF-SPACE *

PERCENT OF TOTAL LIFE-

PERCENT OF TOTAL LIFE-

TIME SPENT IN HIGH-

TIME SPENT IN HIGH-

INTENSITY REGIONS* OF

INTENSITY REGIONS* OF

VAN ALLEN BELTS :

VAN ALLEN BELTS :

PERCENT OF TOTAL DAILY

PERCENT OF TOTAL DAILY

FLUX ACCUMULATED IN

FLUX ACCUMULATED IN

HIGH-INTENSITY REGIONS:

HIGH-INTENSITY REGIONS:

62.30 %

62.30 %

62.30 %

62.30 %

62.30 %

62.30 %

62.30 %

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62.30 %

TABLE 32

TABLE -

NVLX3:S-MAX
CIRCULAR
INCLINATION: 60 DEG
PERIGEE: 3889 KM
APOGEE: 3889 KM

* PERCENT OF TOTAL LIFETIME SPENT INSIDE AND *
* OUTSIDE THE TRAPPED-PARTICLE RADIATION BELT *

INNER ZONE -71- : 54.51 %
(1.0 < L < 2.8)
OUTER ZONE -70- : 42.08 %
(2.8 < L < 11.0)
EXTERNAL -7E- : 3.40 %
(L > 11.0)
TOTAL : 100.00 %

*TIME IN INNER ZONE MAY BE SUBDIVIDED AS FOLLOWS:

OUTSIDE TRAPPING REGION : 0.0 %
(1.0 < L < 1.1)
INSIDE TRAPPING REGION : 54.51 %
(1.1 < L < 2.8)

TABLE -

NVLX3:S-MAX
CIRCULAR
INCLINATION: 60 DEG
PERIGEE: 3889 KM
APOGEE: 3889 KM

**** EXPOSURE ANALYSIS ****

PROTONS
(E>5.000MEV)
ELECTRONS LO
(E>5000MEV)

PERCENT OF TOTAL LIFE-
TIME SPENT IN FLUX-FREE
REGIONS* OF SPACE :

4.17 %

PERCENT OF TOTAL LIFE-
TIME SPENT IN HIGH-
INTENSITY REGIONS* OF
VAN ALLEN BELTS :

81.94 %

PERCENT OF TOTAL DAILY
FLUX ACCUMULATED IN
HIGH-INTENSITY REGIONS:

99.71 %

* <1 PARTICLE/CM**2/SEC
* >1.E5 EL/CM**2/SEC OR 1.E3 PA/CM**2/SEC

TABLE 33

TABLE

TABLE

NALDA(S-NA)
 INCLINATION: 60 DEG
 PERIGEE: 5100 KM
 APOGEE: 5100 KM

* PERCENT OF TOTAL LIFETIME SPENT INSIDE AN
 * OUTSIDE THE TRAPPED-PARTICLE RADIATION BELT *

INNER ZONE -11- : 40.33 %
 11.0 < L < 1.8
 OUTER ZONE -10- : 40.67 %
 1.8 < L < 11.0
 INTERNAL -7E- : 5.00 %
 (L > 11.0)
 TOTAL : 100.00 %

* TIME IN INNER ZONE MAY BE SUBDIVIDED AS FOLLOWS:

OUTSIDE TRAPPING REGION : 0.0 %
 (1.0 < L < 1.1)
 INSIDE TRAPPING REGION : 40.33 %
 (1.1 < L < 2.0)

**** EXPOSURE ANALYSIS ****

FRACTIONS ELCTEONS LU
 (E>5000MEV) (E>5000MEV)
 PERCENT OF TOTAL LIFE-
 TIME SPENT IN FLUX-FREE
 REGIONS OF SPACE : 34.00 %
 PERCENT OF TOTAL LIFE-
 TIME SPENT IN HIGH-
 INTENSITY REGIONS OF
 VAN ALLEN BELTS : 65.97 %
 PERCENT OF TOTAL DAILY
 FLUX ACCUMULATED IN
 HIGH-INTENSITY REGIONS : 50.84 %

* <1 PARTICLE/CM**2/SEC
 * >1.0E5 EL/CM**2/SEC OF 1.0E3 PR/CM**2/SEC

TABLE 34

TABLE -

RADIATION-MAX

CIRCULAR

Irradiation: 40 DEG

PERIOD: 4369 KM

ADJ. CORR: 4369 KM

**** EXPOSURE ANALYSIS ****

PROTONS
(0.25.000000 V)ELECTRONS LC
(0.25.000000 V)

PERCENT OF TOTAL LIFE-

TIME SPENT IN FLUX-FIELD

REGIONS OF SPACE

14.44 %

6.51 %

PERCENT OF TOTAL LIFE-

TIME SPENT IN FICH-

INTENSITY REGIONS OF

VAN ALLEN BELT:

60.14 %

46.67 %

PERCENT OF TOTAL CAILY

FLUX ACCUMULATED IN

HIGH-INTENSITY REGIONS:

66.66 %

66.66 %

* < 1 PART/CM-MINUTE/SEC

* 1.000 FLUX/CM-MINUTE/SEC OR 1.000 10/CM-MINUTE/SEC

TABLE -

RADIATION-MAX

CIRCULAR

Irradiation: 40 DEG

PERIOD: 4369 KM

ADJ. CORR: 4369 KM

* PERCENT OF TOTAL LIFETIME SPENT INSIDE AND *

* OUTSIDE THE TEMPOLO-PARTICLE RADIATION BELT *

INNER REGION - I-1- :

42.36 %

(1.0 < L < 2.5)

OUTER REGION - I-2- :

51.64 %

(2.5 < L < 11.0)

EXTERNAL - I-3- :

5.69 %

(L > 11.0)

TOTAL

100.00 %

* TIME IN WHICH TIME MAY BE SURVIVED AS FOLLOWS:

OUTSIDE TRAPPING REGION :

0.00 %

(1.0 < L < 1.1)

INSIDE TRAPPING REGION :

42.36 %

(1.1 < L < 2.5)

TABLE -

17 APR - MAY
 CIRCULAR
 DIST. DIST. 1972 - 1973
 10.10 10.10 10.10
 10.10 10.10 10.10

* PERCENT OF TOTAL TIME SPENT INSIDE AND
 * OUTSIDE RADIATION BELT

INSIDE BELT - 11.0% : 16.63%
 (1.0% < 1.0%)
 OUTSIDE BELT - 11.0% : 72.71%
 (1.0% < 11.0%)
 TOTAL : 100.00%

TIME IN RADIATION BELT CAN BE SUBDIVIDED AS FOLLOWS:

OUTSIDE RADIATION BELT : 0.0%
 (1.0% < 1.0%)
 INSIDE RADIATION BELT : 16.63%
 (1.0% < 1.0%)

TABLE -

17 APR - MAY
 CIRCULAR
 DIST. DIST. 1972 - 1973
 10.10 10.10 10.10
 10.10 10.10 10.10

*** CIRCULAR ANALYSIS ***

ELECTRICITY
 (10.0000000)

PERCENT OF TOTAL TIME

TIME SPENT IN RADIATION BELT

PERCENT OF TOTAL TIME

TIME SPENT IN RADIATION BELT

INTENSITY RECORDS OF

VAN ALLEN BELTS :

PERCENT OF TOTAL DAILY

FLUX ACCUMULATED IN

HIGH-INTENSITY RECORDS :

PERCENT OF TOTAL DAILY

FLUX ACCUMULATED IN

HIGH-INTENSITY RECORDS :

PERCENT OF TOTAL DAILY

FLUX ACCUMULATED IN

HIGH-INTENSITY RECORDS :

PERCENT OF TOTAL DAILY

FLUX ACCUMULATED IN

HIGH-INTENSITY RECORDS :

PERCENT OF TOTAL DAILY

FLUX ACCUMULATED IN

HIGH-INTENSITY RECORDS :

PERCENT OF TOTAL DAILY

FLUX ACCUMULATED IN

HIGH-INTENSITY RECORDS :

PERCENT OF TOTAL DAILY

FLUX ACCUMULATED IN

HIGH-INTENSITY RECORDS :

PERCENT OF TOTAL DAILY

FLUX ACCUMULATED IN

HIGH-INTENSITY RECORDS :

TABLE 36

***** ORBITAL FLUX STUDY WITH COMPOSITE PARTICLE ENVIRONMENTAL VECTORS APPL. ALG. 1.01 FOR SOLAR MAXIMUM *****
 ** UNCERTAINTY FACTORS (UF) APPLIED FOR THIS RUN ARE: FOR PROTONS (APB) - UF = 1.0; FOR INNER ZONE ELECTRONS (AE6) - UF = 1.0; FOR OUTER ZONE ELECTRONS (AE6) - UF = 1.0 *****
 ** MAGNETIC COORDINATES B AND L COMPUTED BY INVARA OF 1972 WITH ALL MAG. MODEL 4: BARRACLOUGH ET AL. 1975; TIME = 1989.5 *****
 ** MEMO: 1. MAY 11-15 MAY 11-15 MAY 11-15 MAY 11-15 MAY 11-15 MAY 11-15 MAY 11-15 MAY 11-15 MAY 11-15 MAY 11-15 *****
 ** FOR INFORMATION OR EXPLANATION CONTACT E.G. STASSINPOULOS AT NASA-GSFC, CODE 601, GREENBELT, MARYLAND 20771, TEL. (301) 344-8867 *****

ENERGY LEVELS > (MEV)	*****EMISSION DURATION T=60. MONTHS *****				GFOMAGNETIC SHIELDING	
	80	85	90	95	DIPOLAR CLIFF SHELL	PERCENT EXPOSURE TIME
10.0	5.636E 09	5.636E 09	7.615E 09	9.394E 09	1.315E 10	1.24
20.0	3.865E 09	3.865E 09	5.153E 09	6.441E 09	9.017E 09	1.18
30.0	2.850E 09	2.850E 09	3.535E 09	4.418E 09	6.239E 09	1.12
40.0	1.817E 09	1.817E 09	2.423E 09	3.028E 09	4.239E 09	1.07
50.0	1.246E 09	1.246E 09	1.661E 09	2.076E 09	2.903E 09	1.03
60.0	8.542E 08	8.542E 08	1.139E 09	1.424E 09	1.947E 09	1.00
70.0	5.857E 08	5.857E 08	7.839E 08	9.762E 08	1.347E 09	0.97
80.0	4.016E 08	4.016E 08	5.355E 08	6.587E 08	4.371E 08	0.94
90.0	2.754E 08	2.754E 08	3.671E 08	4.587E 08	3.425E 08	0.91
100.0	1.858E 08	1.858E 08	2.517E 08	3.147E 08	2.405E 08	0.88
110.0	1.255E 08	1.255E 08	1.726E 08	2.178E 08	1.671E 08	0.85
120.0	8.877E 07	8.877E 07	1.164E 08	1.517E 08	1.142E 08	0.82
130.0	6.066E 07	6.066E 07	8.113E 07	1.058E 08	7.758E 07	0.79
140.0	4.173E 07	4.173E 07	5.564E 07	7.058E 07	5.178E 07	0.76
150.0	2.862E 07	2.862E 07	3.815E 07	4.708E 07	3.477E 07	0.73
160.0	1.922E 07	1.922E 07	2.596E 07	3.273E 07	2.273E 07	0.70
170.0	1.345E 07	1.345E 07	1.744E 07	2.242E 07	1.539E 07	0.67
180.0	9.244E 06	9.244E 06	1.230E 07	1.517E 07	1.052E 07	0.64
190.0	6.352E 06	6.352E 06	8.455E 06	1.074E 07	7.476E 06	0.61
200.0	4.337E 06	4.337E 06	5.782E 06	7.228E 06	5.012E 06	0.58

TABLE 37

 ** ORBITAL FLUX SLDON WITH COMPOSITE PARTICLE ENRICHMENT: VOTES APR: ADO: A.17 FOR SCOR MAXIMUM *** UNIFLUX OF 1975
 ** UNCERTAINTY FACTORS (UF) APPLIED FOR THIS RUN ARE: FOR POSITIVE (APB) - JEE 1.01 FOR NEGATIVE (APB) - JEE 1.01
 ** MAGNETIC COORDINATES B AND L COMPUTED BY INVATA OF 1972 WITH ALL MAG. MODEL 4: BAZAL LUP 13.4AL. LUT-TIN 1974. TIME 1.00
 ** MAGNETIC COORDINATES B AND L COMPUTED BY INVATA OF 1972 WITH ALL MAG. MODEL 4: BAZAL LUP 13.4AL. LUT-TIN 1974. TIME 1.00
 ** MAGNETIC COORDINATES B AND L COMPUTED BY INVATA OF 1972 WITH ALL MAG. MODEL 4: BAZAL LUP 13.4AL. LUT-TIN 1974. TIME 1.00
 ** FOR INFORMATION OF EXPLANATION CONTACT L.G. STASSINGFULDS AT NASA-OSFC/CINL 011. 500 BULL. MAYLAN 2071. 71.6011-34-2067

 ** ENEGETIC JOLAR PROTON FLUENCE *****
 ** ECE CUTTER - IPOLL SHELL L=5 BA *****
 ** ENEGETIC JOLAR PROTON FLUENCE *****
 ** ECE CUTTER - IPOLL SHELL L=5 BA *****
 ***** (PARTICLES/CM^2) *****

ENERGY LEVELS (MEV)	MISSION DURATION TIME MONTHS *****					GEOMAGNETIC SHELTING		
	R0	E5	40	45	55	SHIELD	RECU	RECUAT
10.0	1.0012	1.0018	1.414	1.708	2.473	129	32.45	
15.0	1.0012	1.0018	1.414	1.708	2.473	129	32.45	
20.0	1.0012	1.0018	1.414	1.708	2.473	129	32.45	
25.0	1.0012	1.0018	1.414	1.708	2.473	129	32.45	
30.0	1.0012	1.0018	1.414	1.708	2.473	129	32.45	
35.0	1.0012	1.0018	1.414	1.708	2.473	129	32.45	
40.0	1.0012	1.0018	1.414	1.708	2.473	129	32.45	
45.0	1.0012	1.0018	1.414	1.708	2.473	129	32.45	
50.0	1.0012	1.0018	1.414	1.708	2.473	129	32.45	
55.0	1.0012	1.0018	1.414	1.708	2.473	129	32.45	
60.0	1.0012	1.0018	1.414	1.708	2.473	129	32.45	
65.0	1.0012	1.0018	1.414	1.708	2.473	129	32.45	
70.0	1.0012	1.0018	1.414	1.708	2.473	129	32.45	
75.0	1.0012	1.0018	1.414	1.708	2.473	129	32.45	
80.0	1.0012	1.0018	1.414	1.708	2.473	129	32.45	
85.0	1.0012	1.0018	1.414	1.708	2.473	129	32.45	
90.0	1.0012	1.0018	1.414	1.708	2.473	129	32.45	
95.0	1.0012	1.0018	1.414	1.708	2.473	129	32.45	
100.0	1.0012	1.0018	1.414	1.708	2.473	129	32.45	
105.0	1.0012	1.0018	1.414	1.708	2.473	129	32.45	
110.0	1.0012	1.0018	1.414	1.708	2.473	129	32.45	
115.0	1.0012	1.0018	1.414	1.708	2.473	129	32.45	
120.0	1.0012	1.0018	1.414	1.708	2.473	129	32.45	
125.0	1.0012	1.0018	1.414	1.708	2.473	129	32.45	
130.0	1.0012	1.0018	1.414	1.708	2.473	129	32.45	
135.0	1.0012	1.0018	1.414	1.708	2.473	129	32.45	
140.0	1.0012	1.0018	1.414	1.708	2.473	129	32.45	
145.0	1.0012	1.0018	1.414	1.708	2.473	129	32.45	
150.0	1.0012	1.0018	1.414	1.708	2.473	129	32.45	
155.0	1.0012	1.0018	1.414	1.708	2.473	129	32.45	
160.0	1.0012	1.0018	1.414	1.708	2.473	129	32.45	
165.0	1.0012	1.0018	1.414	1.708	2.473	129	32.45	
170.0	1.0012	1.0018	1.414	1.708	2.473	129	32.45	
175.0	1.0012	1.0018	1.414	1.708	2.473	129	32.45	
180.0	1.0012	1.0018	1.414	1.708	2.473	129	32.45	
185.0	1.0012	1.0018	1.414	1.708	2.473	129	32.45	
190.0	1.0012	1.0018	1.414	1.708	2.473	129	32.45	
195.0	1.0012	1.0018	1.414	1.708	2.473	129	32.45	
200.0	1.0012	1.0018	1.414	1.708	2.473	129	32.45	

TABLE 40

[illegible]

TABLE 42

ORBITAL FLUX STUDY WITH COMPOSITE PARTICLE ENVIRONMENTALITIES. ALSO, A17 FOR SOLAR MAXIMUM
 ** UNCERTAINTY FACTORS (U) APPLIED FOR THIS RUN ARE: JPE = 1.01 FOR INNER ZONE PROTONS (APB) - JPE = 1.01 FOR INNER ZONE ELECTRONS (AEG) - U = 1.0
 ** MAGNETIC COORDINATES 9 AND 1 COMPUTED BY INVARI OF 1972 WITH ALL MAG. MODEL 4: GARRACLOUGH ET AL. 1971-IPM 1975 & TIME = 1989.5
 ** MAGNETIC COORDINATES 9 AND 1 COMPUTED BY INVARI OF 1972 WITH ALL MAG. MODEL 4: GARRACLOUGH ET AL. 1971-IPM 1975 & TIME = 1989.5
 ** FOR INFORMATION IN EXPLANATION CONTACT E.C. STASSINPOULOS AT NASA-GFSCUDF 03: GREENBELL, MARYLAND 20771 TEL (301)-344-8067

*** DCS AT TRANSMISSION SURFACE OF FINITE ALUMINUM SLAB SHIELDS
 *** MISSION DURATION: 5,000 YEARS ***

SHIELD THICKNESS S (ALUMINUM) (G/M ² X 10 ⁻²¹) (MM)	ELECTRONS*			URLEMISM- AELING			PARTIONS			TOTAL DOSE	
	I (RAD-S-ALL)	INNER ZN (RAD-S-ALL)	OUTER ZN (RAD-S-ALL)	TOTAL (RAD-S-ALL)	TRAPPED** (RAD-S-ALL)	SOLAR* (RAD-S-ALL)	TOTAL (RAD-S-ALL)	ALL SOURCES (RAD-S-ALL)			
0.01	0.04	2.257E 07	6.399E 05	2.321E 07	6.980E 05	9.988E 02	6.512E 05	2.392E 07			
0.02	0.07	1.297E 07	3.977E 05	1.297E 07	3.977E 05	1.035E 03	2.342E 05	1.332E 07			
0.03	0.11	7.784E 06	2.882E 05	8.072E 06	2.882E 05	1.035E 03	2.612E 05	8.332E 06			
0.04	0.15	5.084E 06	2.287E 05	5.310E 06	2.287E 05	1.035E 03	2.812E 05	5.532E 06			
0.05	0.19	3.434E 06	1.844E 05	3.620E 06	1.844E 05	1.035E 03	3.012E 05	3.812E 06			
0.06	0.22	2.385E 06	1.544E 05	2.539E 06	1.544E 05	1.035E 03	3.212E 05	2.712E 06			
0.07	0.24	1.683E 06	1.340E 05	1.824E 06	1.340E 05	1.035E 03	3.412E 05	1.982E 06			
0.08	0.27	1.228E 06	1.166E 05	1.345E 06	1.166E 05	1.035E 03	3.612E 05	1.482E 06			
0.09	0.33	9.095E 05	1.037E 05	1.013E 06	1.037E 05	1.035E 03	3.812E 05	1.182E 06			
0.10	0.37	6.875E 05	9.331E 04	7.808E 05	9.331E 04	1.035E 03	4.012E 05	9.302E 05			
0.20	0.74	3.063E 05	4.532E 04	3.517E 05	4.532E 04	8.766E 02	5.552E 04	2.222E 05			
0.30	1.11	1.981E 04	2.888E 04	6.669E 04	8.000E 02	5.035E 02	5.122E 04	1.188E 05			
0.40	1.48	1.095E 04	1.955E 04	3.050E 04	5.620E 02	3.120E 02	3.120E 04	7.902E 04			
0.50	1.85	7.33E 03	1.379E 04	2.112E 04	3.495E 02	2.495E 02	2.495E 04	5.845E 04			
0.60	2.22	4.515E 03	7.192E 03	1.170E 04	2.401E 02	1.984E 02	1.155E 04	4.610E 04			
0.80	2.56	3.124E 03	5.531E 03	8.655E 03	1.620E 02	1.460E 02	2.560E 04	3.270E 04			
1.00	3.70	1.241E 03	1.907E 03	3.148E 03	2.235E 02	2.235E 02	2.235E 04	2.635E 04			
1.20	5.03	5.022E 02	8.947E 02	1.397E 03	1.995E 02	2.995E 02	2.612E 04	2.142E 04			
1.50	5.56	2.19E 03	3.851E 02	4.358E 02	1.755E 02	1.755E 02	1.812E 04	1.872E 04			
1.75	6.48	2.55E 03	1.473E 02	1.537E 02	1.650E 02	1.556E 02	1.670E 04	1.702E 04			
2.00	7.41	2.92E 03	4.904E 01	1.507E 02	1.535E 02	1.535E 02	1.540E 04	1.568E 04			
2.50	9.26	3.65E 03	3.086E 00	3.087E 01	1.300E 02	9.677E 01	1.340E 04	1.372E 04			
3.00	11.11	4.32E 03	5.038E 00	5.038E 02	1.200E 02	7.680E 01	1.216E 04	1.272E 04			
3.50	12.96	5.10E 03	1.291E 00	1.291E 05	1.098E 02	5.797E 01	1.106E 04	1.113E 04			
4.00	14.81	5.83E 03	0.0	0.0	1.008E 02	4.665E 01	1.013E 04	1.021E 04			
4.50	16.67	6.56E 03	0.0	0.0	9.398E 01	3.745E 01	9.347E 03	9.422E 03			
5.00	18.52	7.29E 03	0.0	0.0	8.636E 01	3.096E 01	8.647E 03	8.735E 03			
5.50	20.37	8.02E 03	0.0	0.0	7.523E 01	2.153E 01	7.545E 03	7.603E 03			
6.00	22.22	8.75E 03	0.0	0.0	5.911E 01	1.118E 01	5.922E 03	5.965E 03			
6.50	24.07	9.48E 03	0.0	0.0	4.804E 01	6.266E 00	4.810E 03	4.843E 03			
7.00	25.92	1.02E 04	0.0	0.0	3.287E 01						
7.50	27.77	1.10E 04	0.0	0.0							
8.00	29.62	1.18E 04	0.0	0.0							
8.50	31.47	1.26E 04	0.0	0.0							
9.00	33.32	1.34E 04	0.0	0.0							
9.50	35.17	1.42E 04	0.0	0.0							
10.00	37.04	1.50E 04	0.0	0.0							

* ELECTRON MODELS:

AE6: INNER ZONE-SCLAP MAX
 NO UNCERTAINTY FACTOR WAS APPLIED TO THE MODEL DATA.

* SOLAR PROTON MODEL:

SOLPRO: SOLAR FLARE PROTONS AT 1 AU
 UNATTENUATED, INTERPLANETARY
 FOR CUTOFF DIPOLE SHELL OF 5 E.R.

AF17: OUTER ZONE-INTERIM MODEL WITHOUT SOLAR CYCLE DEPENDENCE.
 FOR ENERGIES ABOVE 1.5 MEV, THIS MODEL CONTAINS UPPER &
 LOWER LIMIT VALUES TO ACCOUNT FOR DISCREPANCY BETWEEN
 EXISTING DATA SETS. THE A17-HI FAVORS VAMPPLA'S FIT
 TO OX-19 DATA WHILE A17-LO IS MORE REPRESENTATIVE OF
 ALL THE DATA SETS PRESENTLY AVAILABLE TO NSSDC.

FOR INFO: 10-9012-7 OF AT EVENTS=4
 84.82X GEOMAGNETIC SHIELDING APPLIED

NOTE: U DENOTES THE DEGREE OF CONFIDENCE ONE WISHES
 TO ASSIGN TO RESULTS, NAMELY THAT FOR THE
 SPECIFIC MISSION DURATION THE CALCULATED
 FLUENCES ARE THE SMALLEST VALUES WHICH WILL
 NOT BE EXCEEDED BY ACTUALLY ENCOUNTERED
 INTENSITIES.

** THE A17-LO VERSION WAS USED FOR THESE CALCULATIONS **

** PROTON MODEL:

APB-MAG: TRAPPED PROTONS-SOLAR MAX
 NO UNCERTAINTY FACTOR WAS APPLIED TO THE MODEL DATA.

IT IS NOT ADVISABLE TO EXTRAPOLATE THE SOLAR
 PROTON SPECTRA EITHER TOWARDS LOWER NOR
 TOWARDS HIGHER ENERGIES BECAUSE THE DATA SETS
 USED IN THE CONSTRUCTION OF THE MODEL IS A
 LIMITED MEASUREMENTS MADE DURING THE 20TH SOLAR
 CYCLE: 1964-1975) DO NOT CONTAIN INFORMATION
 FOR 5-10 AND 10-20 MEV

TABLE 43

TABLE 44

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***** ORBITAL FLOW STUDY WITH COMPOSITE PARTICLE ENHANCEMENTS *****
***** VETTES APPROX. 4817 FOR POLAR MAXIMUM ***** HNPLOX UP 1979
***** UNCERTAINTY FACTORS (UP) APPLIED FOR THIS RUN ARE: *****
***** FOR INNER ZONE ELECTRONS (AOL) - UP= 1.0 *****
***** MAGNETIC COORDINATES B AND L COMPUTED BY INVAPO *****
***** FROM 1972 WITH ALMAG. MODEL 3849A *****
***** VEHICLE: MWL33-5-MAX ** PERIGEE= 3849KM ** ** ** **
***** ** INCLINATION= 60DEG ** ** ** **
***** ** ** ** **
***** FOR INFORMATION OR EXPLANATION CONTACT E.G. STASSINOPOLUS AT NASA-SPC CODE 631 *****
***** GREENBELT, MARYLAND 20771 ***** TEL. (301) 344-8067 *****
*****

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*****
*** COSE AT TRANSMISSION SURFACE (P FINITE ALUMINUM GLASS SHIELDS ***
*****
*** MISSION DURATION: 5.0000 YEAR(S) ***
*****

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[illegible]

• ELECTRON MODELS: STADON NOV13272

AB6: INNER ZONE-SOLAR MAX
NO UNCERTAINTY FACTOR WAS APPLIED TO THE MODEL DATA.

RE17: OUTER ZONE-INTERIM MODEL WITHOUT SOLAR CYCLE DEPENDENCE. FOR ENERGIES ABOVE 1.5 MEV, THIS MODEL CONTAINS UPPER & LOWER LIMIT VALUES TO ACCOUNT FOR DISCREPANCY BETWEEN EXISTING DATA SETS. THE RE17-EL FAVORS VAMPOLA'S FIT TO OVI-19 DATA WHILE RE17-LO IS MORE REPRESENTATIVE OF ALL THE DATA SETS PRESENTLY AVAILABLE TO MSSDC.

>>> THE AEI7-10 VERSION WAS USED FOR THESE CALCULATIONS <<

*** PROTON MODEL:*

AP8-MAC: TRAPPED PROTONS-SOLAR MAX NO UNCERTAINTY FACTOR WAS APPLIED TO THE MODEL DATA.

* SOLAR PROTON MODEL:

SOLPRO: SCALAR FLARE PROTONS AT 1 AU
(UNATTENUATED, INTERPLANETARY)
FOR CUTOFF DIPOLE SHELL CP5 1.E6.

FOR TAC=60.HO., $\eta=90\%$: # OF AL EVENTS=4

NOTE: C DENOTES THE DEGREE OF CONFIDENCE ONE WISHES TO ASSIGN TO RESULTS, NAMELY THAT FOR THE SPECIFIC MISSION DURATION THE CALCULATED FLUENCES WERE THE SMALLEST VALUES WHICH WILL NOT BE EXCEEDED BY ACTUALLY ENCOUNTERED INTENSITIES.

IT IS NOT ADVISABLE TO EXTRAPOLATE THE SOLAR
ELECTRIC ENERGY TOWARDS LOWER NON SETS-
TOWARDS HIGHER ENERGIES OF THE DATA SETS-
USED IN THE COMMISSION OF THE MODEL (SPEL-
LIT) MEASUREMENTS DURING THE 20TH SOLAR
CYCLE (1964-1975). DO NOT CONTAIN INFORMATION
FOR $E < 10$ AND $E > 30$ MEV.

 ORBITAL FLUX STUDY WITH COMPLETE PARTICLE ENVIRONMENTS: VETTER APRI AGO, A17 FOR SOLAR MAXIMUM
 UNCERTAINTY FACTORS (UF) APPLIED FOR THIS RUN ARE: FOR PROTONS (APRI) - UF 1.0; FOR INNER ZONE - ELECTRON (A) - UF 1.0
 MAGNETIC COORDINATES B AND L COMPUTED BY INVARIANT OF 1972 WITH ALLMAG, MODEL 4; BACKGROUND ST. 168-TM 1975; TIME 1500.5
 MAGNETIC COORDINATES B AND L COMPUTED BY INVARIANT OF 1972 WITH ALLMAG, MODEL 4; BACKGROUND ST. 168-TM 1975; TIME 1500.5
 MAGNETIC COORDINATES B AND L COMPUTED BY INVARIANT OF 1972 WITH ALLMAG, MODEL 4; BACKGROUND ST. 168-TM 1975; TIME 1500.5
 FOR INFORMATION OR EXPLANATION CONTACT E.G. STASSINGFELT AT NASA-GSFC, CODE 611, GREENBELT, MARYLAND 20741-1344-4467

 CGSE AT TRANSMISSION SURFACE OF FINITE ALUMINUM SLAB SP100: 400
 MISSION DURATION: 5.0000 YEAR(S) ***

SHIELD THICKNESS		ELECTRONS		GRAMS/CM ²		TRAPED		PROTONS		TOTAL	
S (GCM/CM ²)	T (ALUMINUM)	INNER 20° (DEGS-ALL)	OUTER 20° (DEGS-ALL)	TOTAL (DEGS-ALL)	ALUMINUM TOTAL (GRAMS-ALL)	TRAPED (GRAMS-ALL)	SOLAR (GRAMS-ALL)	TOTAL (GRAMS-ALL)	ALL SOURCES (GRAMS-ALL)	TOTAL (GRAMS-ALL)	
0.01	0.04	1.441E 06	2.251E 06	1.477E 09	0.231E 04	0.328E 04	1.600E 07	0.328E 04	2.170E 08	07	
0.02	0.07	4.245E 07	1.635E 06	5.009E 07	0.939E 04	0.939E 04	1.600E 07	0.939E 04	1.104E 07	07	
0.03	0.11	2.190E 07	1.284E 06	3.309E 07	3.952E 04	1.742E 04	1.600E 07	3.952E 04	7.320E 07	07	
0.04	0.15	1.171E 07	1.022E 06	2.275E 07	2.735E 04	1.742E 04	1.600E 07	2.735E 04	4.631E 07	07	
0.05	0.19	1.124E 07	8.678E 05	1.613E 07	2.534E 04	8.678E 05	1.600E 07	2.534E 04	2.500E 07	07	
0.06	0.22	1.124E 07	7.665E 05	1.168E 07	2.042E 04	8.678E 05	1.600E 07	2.042E 04	1.807E 07	07	
0.07	0.26	7.545E 06	6.733E 05	8.619E 06	1.771E 04	8.678E 05	1.600E 07	1.771E 04	1.46E 07	07	
0.08	0.30	4.245E 06	5.906E 05	4.906E 06	1.571E 04	8.678E 05	1.600E 07	1.571E 04	1.171E 07	07	
0.10	0.37	3.425E 05	2.520E 05	3.045E 05	1.409E 03	8.678E 05	1.600E 07	1.409E 03	1.624E 06	06	
0.20	0.74	3.425E 05	1.444E 05	2.108E 05	0.712E 03	8.678E 05	1.600E 07	0.712E 03	6.393E 05	05	
0.30	1.11	1.267E 04	5.263E 04	1.139E 05	2.930E 03	8.678E 05	1.600E 07	2.930E 03	3.364E 05	05	
0.40	1.46	0.00E 00	0.00E 00	0.00E 00	0.00E 00	8.678E 05	1.600E 07	8.678E 05	2.051E 05	05	
0.50	1.82	0.00E 00	0.00E 00	0.00E 00	0.00E 00	8.678E 05	1.600E 07	8.678E 05	1.235E 05	05	
0.60	2.17	0.00E 00	0.00E 00	0.00E 00	0.00E 00	8.678E 05	1.600E 07	8.678E 05	7.235E 04	04	
0.80	3.52	0.00E 00	0.00E 00	0.00E 00	0.00E 00	8.678E 05	1.600E 07	8.678E 05	4.512E 04	04	
1.00	5.70	0.00E 00	0.00E 00	0.00E 00	0.00E 00	8.678E 05	1.600E 07	8.678E 05	2.614E 04	04	
1.25	8.03	0.00E 00	0.00E 00	0.00E 00	0.00E 00	8.678E 05	1.600E 07	8.678E 05	1.555E 04	04	
1.50	1.03	0.00E 00	0.00E 00	0.00E 00	0.00E 00	8.678E 05	1.600E 07	8.678E 05	1.555E 04	04	
1.75	1.44	0.00E 00	0.00E 00	0.00E 00	0.00E 00	8.678E 05	1.600E 07	8.678E 05	1.555E 04	04	
2.00	1.81	0.00E 00	0.00E 00	0.00E 00	0.00E 00	8.678E 05	1.600E 07	8.678E 05	1.555E 04	04	
2.50	2.72	0.00E 00	0.00E 00	0.00E 00	0.00E 00	8.678E 05	1.600E 07	8.678E 05	1.555E 04	04	
3.00	3.52	0.00E 00	0.00E 00	0.00E 00	0.00E 00	8.678E 05	1.600E 07	8.678E 05	1.555E 04	04	
3.50	4.32	0.00E 00	0.00E 00	0.00E 00	0.00E 00	8.678E 05	1.600E 07	8.678E 05	1.555E 04	04	
4.00	5.12	0.00E 00	0.00E 00	0.00E 00	0.00E 00	8.678E 05	1.600E 07	8.678E 05	1.555E 04	04	
4.50	5.92	0.00E 00	0.00E 00	0.00E 00	0.00E 00	8.678E 05	1.600E 07	8.678E 05	1.555E 04	04	
5.00	6.72	0.00E 00	0.00E 00	0.00E 00	0.00E 00	8.678E 05	1.600E 07	8.678E 05	1.555E 04	04	
5.50	7.52	0.00E 00	0.00E 00	0.00E 00	0.00E 00	8.678E 05	1.600E 07	8.678E 05	1.555E 04	04	
6.00	8.32	0.00E 00	0.00E 00	0.00E 00	0.00E 00	8.678E 05	1.600E 07	8.678E 05	1.555E 04	04	
10.00	14.58	0.00E 00	0.00E 00	0.00E 00	0.00E 00	8.678E 05	1.600E 07	8.678E 05	1.555E 04	04	

* ELECTRON MODELS:
 AEG: INNER ZONE - SOLAR MAX
 AEG: NO UNCERTAINTY FACTOR WAS APPLIED TO THE MODEL DATA.

AET: OUTER ZONE - INTERIM MODEL WITHOUT SOLAR CYCLE DEPENDENCE.
 FOR ENERGIES ABOVE 1.5 MEV, THIS MODEL CONTAINS UPPER &
 LOWER LIMIT VALUES TO ACCOUNT FOR DISCREPANCY BETWEEN
 EXISTING DATA SETS. THE AET-17-HI FAVORS VAMPCLAS FIT
 TO EXISTING DATA WHILE AET-17-LO IS MORE REPRESENTATIVE OF
 ALL THE DATA SETS PRESENTLY AVAILABLE TO NSSCC.
 >> THE AET-17-LO VERSION WAS USED FOR THESE CALCULATIONS <<

** PROTON MODEL:
 APR-MAG: TRAPPED PROTONS - SOLAR MAX
 NO UNCERTAINTY FACTOR WAS APPLIED TO THE MODEL DATA.

* SOLAR PROTON MODEL:
 SUBJECT: SOLAR FLARE PROTONS AT 1 AU
 (UNATTENUATED, INTERPLANETARY)
 FOR CUTOFF RIGIDITY SHELL OF 5 E.E.
 FOR TAP-60-005050: * OF AL EVENTS=4
 76.503 COLUMBIAN SHIP LIVING APPEL

NOTE: Q DETERMINES THE DEGREE OF CONFIDENCE ONE WISHES
 TO ASSIGN TO RESULTS, NAMELY THAT FOR THE
 SPECIFIC MISSION DURATION THE CALCULATED
 FLUXES ARE THE SMALLEST VALUES WHICH WILL
 NOT BE EXCEEDED BY ACTUALLY ENCOUNTERED
 INTERMITTENCIES.

IT IS NOT ADVISABLE TO INTERPOLATE TO SOLAR
 PROTON SPECTRA EITHER TOWARDS LOWER ENERGY SETS
 TOWARDS HIGHER ENERGIES BECAUSE OF DATA SETS
 USED IN THE DETERMINATION OF THE MODEL (SABLE
 LIFE MEASUREMENTS MADE DURING THE 2014 SOLAR
 CYCLE 1968-1975) DO NOT CONTAIN INFORMATION
 FOR 4-510 AND 7-5200 M.V.

[illegible]

TABLE 47

***** DOSE IN SEMI-INFANTILE ALUMINUM MEDIA *****

***** MISSION DURATION: 50000 YEARS(S) *****

• ELECTRON MODELS:

AE6: INNER ZONE-SOLAR MAX
NO UNCERTAINTY FACTOR WAS APPLIED TO THE MODEL DATA.

SOLAR FLARE PHOTONS AT 1 AU
(UNATTENUATED, INTERPLANETARY)
FOR CUTOFF DIPOLE SHELL OF 5 E.R.

AE17: OUTER ZONE-INTERIM MODEL WITHOUT SOLAR CYCLE DEPENDENCE.

NOTE: A DENOTES THE DEGREE OF CONFIDENCE ONE WISHES TO ASSIGN TO RESULTS, NAMELY THAT FOR THE SPECIFIC MISSION DURATION BE CALCULATED FIGURES ARE THE SMALLEST VALUES WHICH WILL NOT BE EXCEEDED BY ACTUALLY ENCOUNTERED INTENSITIES.

IT IS NOT ADVISABLE TO EXTRAPOLATE THE SOLAR CONSTANT EITHER TOWARDS LOWER OR TOWARDS HIGHER ENERGIES BECAUSE THE DATA SETS USED IN THE CONSTRUCTION OF THE MODEL (SATELITE MEASUREMENTS MADE DURING THE 20TH SOLAR CYCLE: 1964-1975) DO NOT CONTAIN INFORMATION FOR $\lambda > 10$ AND $\lambda < 200$ NM.

66 PROTON MODEL:

APR-MAC: TRAPPED PROTONS-SOLAR MAX
NO UNCERTAINTY FACTOR WAS APPLIED TO THE MODEL DATA.

```
***** DOSE IN SEMI-INFINITE ALUMINUM MEDIUM *****
*****
***** MISSION DURATION: 5,000 YEARS *****
```

• ELECTRON MODELS:

AF6: INNER ZONE-SOLAR MAX

INNER ZONE-SOLAR MAX
NO UNCERTAINTY FACTOR WAS APPLIED TO THE MODEL DATA.

+ SOLAR PROTON MODEL :

SDI PRD: SOLAR FLARE PROTONS AT 1 AU

IONAL ATTENUATION, IN THE RELAXATIONARY REGION CUTOFF DIPOLE SMELL OF S.F.P.

FOR TAUF0.MO.O=90%: 7 OF AL EVENTS=4
60.81% OF MAGNETIC SHIELDING APPLIED

AE17: OUTER ZONE-INTERIM MODEL WITHOUT SOLAR CYCLE DEPENDENCE.

OUTER ZONE-INTERIM MODEL WITHOUT SOLAR CYCLE DEPENDENCE.
FOR ENERGIES ABOVE 1.5 MEV, THIS MODEL CONTAINS UPPER &

FOR ENERGIES ABOVE 1.5 MEV, THIS MODEL CONTAINS UPPER & LOWER LIMIT VALUES TO ACCOUNT FOR DISCREPANCY BETWEEN

COVER LIMIT VALUES TO ACCOUNT FOR DISCREPANCY BETWEEN EXISTING DATA SETS. THE AE17-HI FAVORS VAMPULA'S FIT

EXISTING DATA SUGGESTS THE RESULTS FAVOR AMERICA'S POSITION. THE NEW DATA WHILE A FZ-TO-US MORE REPRESENTATIVE OF

ALL THE DATA SETS PRESENTLY AVAILABLE TO NSSDC.

APPENDIX A

THE ACT 17-0 VERSION WAS USED FOR THESE CALCULATIONS

*** PROTON MODEL: *

AP A=MAC: TRAPPED PROTONS=SOLAR MAX

NO UNCERTAINTY FACTOR WAS APPLIED TO THE MODEL DATA.

05 JUL 1961

[illegible]

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* ELECTRON MODELS:
  AEC: INNER ZONE-SOLAR MAX
  NO UNCERTAINTY FACTOR WAS APPLIED TO THE MODEL DATA.

  AE17: OUTER ZONE-INTERIM MODEL WITHOUT SOLAR CYCLE CORRECTIONS.
  FOR ENERGIES ABOVE 150 KEV, THIS MODEL CONTAINS A LOWER
  LOWER LIMIT VALUE TO ACCOUNT FOR DISCREPANCY BETWEEN
  EXISTING DATA SETS. THE AE17-1 FAVORABLE VARIATION FOR
  TO EXISTING DATA WHILE AE17-2C IS MORE REPRESENTATIVE OF
  ALL THE DATA SETS BEING AVAILABLE TO SSC.

  >> THE AE17-2C VERSION WAS USED FOR THESE CALCULATIONS. <<

** PROTON MODEL:
  ADU-MAG: THINLED PHOTOIN-SOLAR MAX
  NO UNCERTAINTY FACTOR WAS APPLIED TO THE MODEL DATA.

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TABLE 52

 ** JOB: BLU STUDY WITH COMPOSITE PARTICLE INFLUENCE: VEEIES ARE: ALU, AE17 FOR ELN MAXIMUM
 ** UNCERTAINTY FACTORS (U) APPLIED FOR THIS RUN ARE: FCF PROTONS (AE17) - JFE 1.0; FCF INFLUENCE ZONE
 ** MAGNETIC COORDINATES 3 AND 4 COMPUTED BY INAPCA OF 1972 WITH ALL MAG. MODEL 4; FCF INFLUENCE ZONE
 ** MAGNETIC COORDINATES 3 AND 4 COMPUTED BY INAPCA OF 1972 WITH ALL MAG. MODEL 4; FCF INFLUENCE ZONE
 ** FOR INFORMATION OR EXPLANATION CONTACT S.G. STASINCOULOS AT NASA-CSCF, CODE 601, GREENBELT, MARYLAND 20771 TEL: 301-344-8007

 ** DISE IN SEUL-INFINITE ALUMINUM MEDIUM *****
 ** MISSION DURATION: 5.0000 YEARS *****

SHIELD THICKNESS		ELECTRICITY		DENSEST		TRAPPED		PROTONS		TOTAL		TOTAL		TOTAL	
S (ALUMINUM)	T (G/CM ²)	INNER ZN	OUTER ZN	TRAPPED	TOTAL	TRAPPED	TOTAL	INNER ZN	OUTER ZN	TRAPPED	TOTAL	INNER ZN	OUTER ZN	TRAPPED	TOTAL
0.01	0.04	1.0	1.782E	0.0	1.359E	0.0	1.359E	0.0	1.359E	0.0	1.359E	0.0	1.359E	0.0	1.359E
0.02	0.07	3.0	5.251E	0.0	4.763E	0.0	4.763E	0.0	4.763E	0.0	4.763E	0.0	4.763E	0.0	4.763E
0.03	0.11	4.0	5.251E	0.0	4.763E	0.0	4.763E	0.0	4.763E	0.0	4.763E	0.0	4.763E	0.0	4.763E
0.04	0.15	4.0	5.251E	0.0	4.763E	0.0	4.763E	0.0	4.763E	0.0	4.763E	0.0	4.763E	0.0	4.763E
0.05	0.19	7.0	2.457E	0.0	2.457E	0.0	2.457E	0.0	2.457E	0.0	2.457E	0.0	2.457E	0.0	2.457E
0.06	0.22	9.0	1.452E	0.0	1.452E	0.0	1.452E	0.0	1.452E	0.0	1.452E	0.0	1.452E	0.0	1.452E
0.07	0.26	10.0	1.452E	0.0	1.452E	0.0	1.452E	0.0	1.452E	0.0	1.452E	0.0	1.452E	0.0	1.452E
0.08	0.30	12.0	8.438E	0.0	8.438E	0.0	8.438E	0.0	8.438E	0.0	8.438E	0.0	8.438E	0.0	8.438E
0.09	0.33	13.0	6.145E	0.0	6.145E	0.0	6.145E	0.0	6.145E	0.0	6.145E	0.0	6.145E	0.0	6.145E
0.10	0.37	15.0	4.518E	0.0	4.518E	0.0	4.518E	0.0	4.518E	0.0	4.518E	0.0	4.518E	0.0	4.518E
0.20	0.74	29.0	3.132E	0.0	3.132E	0.0	3.132E	0.0	3.132E	0.0	3.132E	0.0	3.132E	0.0	3.132E
0.30	1.11	44.0	4.047E	0.0	4.047E	0.0	4.047E	0.0	4.047E	0.0	4.047E	0.0	4.047E	0.0	4.047E
0.40	1.48	58.0	1.648E	0.0	1.648E	0.0	1.648E	0.0	1.648E	0.0	1.648E	0.0	1.648E	0.0	1.648E
0.50	1.85	73.0	4.105E	0.0	4.105E	0.0	4.105E	0.0	4.105E	0.0	4.105E	0.0	4.105E	0.0	4.105E
0.60	2.22	87.0	1.650E	0.0	1.650E	0.0	1.650E	0.0	1.650E	0.0	1.650E	0.0	1.650E	0.0	1.650E
0.80	2.96	117.0	4.765E	0.0	4.765E	0.0	4.765E	0.0	4.765E	0.0	4.765E	0.0	4.765E	0.0	4.765E
1.00	3.70	140.0	1.253E	0.0	1.253E	0.0	1.253E	0.0	1.253E	0.0	1.253E	0.0	1.253E	0.0	1.253E
1.25	4.62	182.0	2.720E	0.0	2.720E	0.0	2.720E	0.0	2.720E	0.0	2.720E	0.0	2.720E	0.0	2.720E
1.50	5.56	219.0	3.747E	0.0	3.747E	0.0	3.747E	0.0	3.747E	0.0	3.747E	0.0	3.747E	0.0	3.747E
1.75	6.48	255.0	5.457E	0.0	5.457E	0.0	5.457E	0.0	5.457E	0.0	5.457E	0.0	5.457E	0.0	5.457E
2.00	7.41	292.0	5.522E	0.0	5.522E	0.0	5.522E	0.0	5.522E	0.0	5.522E	0.0	5.522E	0.0	5.522E
2.50	9.26	365.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
3.00	11.11	437.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
3.50	12.96	510.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
4.00	14.81	583.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
4.50	16.67	656.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
5.00	18.52	729.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
6.00	22.22	873.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
8.00	29.63	1167.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
10.00	37.04	1459.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0

* ELECTRON MODE 51
 AE6: INNER ZONE SOLAR MAX
 NO UNCERTAINTY FACTOR WAS ADDED TO THE MODEL DATA.

AE17: OUTER ZONE-INTERIOR MODEL WITHOUT ELAR CYCLE DEPENDENCE.
 FOR ENERGIES ABOVE 1.0 MEV, THE MODEL CONTAINS UPPER
 LOWER LIMIT VALUES TO ACCOUNT FOR DISCREPANCY BETWEEN
 EXISTING DATA SETS. THE BEST FIT VALUES VARYING FROM
 1.0 TO 10.0 MEV. ALL THE DATA SETS ORIGINALLY AVAILABLE TO NASA.
 ** THE AE17-3 VERSION WAS USED FOR THESE CALCULATIONS **

** PROTON MODEL:
 NO UNCERTAINTY FACTOR WAS ADDED TO THE MODEL DATA.

[illegible][illegible]

* ELECTRON MODELS:

AE6: INNER ZONE-SOLAR MAX
NO UNCERTAINTY FACTOR WAS APPLIED TO THE MODEL DATA

◆ SOLAR PROTON, WFFL:

SOLUBLE: THERM FLAME PROTONS AT 1 AU
 NOT ATTENUATED. INTERPLANETARY
 FOR CUFF DIDDLE SMOEL OF SEE R.

ECU TA.1 = 40.00, 0 = 50% * GF AL E VENT 5 = 4

NOTE: TO REPORT THE DEGREE OF CONFIDENCE THE ISSUES
CONCERNING THE RESULTS, NAMELY THAT FOR THE
SPECIFIC MISSION DURATION THE CALCULATED
VALUES ARE THE SELECT VALUES WHICH WILL
NOT BE EXCEEDED BY ACTUALLY ENCOUNTERED
INTERFERENCES.

IT IS NOT ADVISABLE TO EXTRAPOLATE THE SCALAR PROPERTIES EITHER TOWARDS LOWER OR HIGHER ENERGY TO PICO-ELECTRONS BECAUSE THE DATA SETS USED IN THE CONSTRUCTION OF THE MODEL (SAFETY) LIVES AND EXPERIMENTS MADE DURING THE 20TH SOLAR CYCLE (1944-1957) DO NOT CONTAIN INFORMATION FOR 600 EV AND BELOW.

Practcn Model:

NO UNCERTAINTY FACTOR WAS APPLIED TO THE MODEL DATA.

TABLE 54

***** DOSE AT CENTER OF ALUMINUM SPHERES *****
***** MISSION DURATION: 5 JULY 1965 *****

• ELECTRON MODELS:

AE6: INNER ZONE-SOLAR MAX
NO UNCERTAINTY FACTOR WAS APPLIED TO THE MODEL DATA.

* SC-AR PROTON MODEL:

SOLPRO: SOLAR FLARE PROTONS AT 1 AU
(UNATTENUATED, INTERPLANETARY)
FOR CUTOFF DIPOLE SHELL OF 5 E.R.

$\rho_{04} - \tau_{04} = \rho_0, \omega_0, \omega_0 = 0.04 \div 0.05$

AE17: OUTER ZONE-INTERIM MODEL WITHOUT SOLAR CYCLE DEPENDENCE. FOR ENERGIES ABOVE 1.5 MEV, THIS MODEL CONTAINS UPPER & LOWER LIMIT VALUES TO ACCOUNT FOR DISCREPANCY BETWEEN EXISTING DATA SETS. THE AE17-HI FAVORS VAMPOLA'S FIT TO QM1-19 DATA WHILE AE17-LO IS MORE REPRESENTATIVE OF ALL THE DATA SETS PRESENTLY AVAILABLE TO NSSDC.

NOTE: 3 DENIES THE DEGREE OF CONFIDENCE ONE WISHES TO ASSIGN TO RESULTS, NAMELY THAT FOR THE SPECIFIC MISSION DURATION THE CALCULATED FLUENCES ARE THE SMALLEST VALUES WHICH WILL NOT BE EXCEEDED BY ACTUALLY ENCOUNTERED INTENSITIES.

IT IS NOT ADVISABLE TO EXTRAPOLATE THE SOLAR
PROTON SPECTRA EITHER TOWARDS LOWER NOR
TOWARDS HIGHER ENERGIES BECAUSE THE DATA SETS
USED IN THE CONSTRUCTION OF THE MODEL (SOLAR-
LITE MEASUREMENTS MADE DURING THE 20TH SOLAR
CYCLE: 1964-1975) DO NOT CONTAIN INFORMATION
FROM 1950 AND 1950-1954.

TABLE 55

***** DOSE AT CENTER OF ALUMINUM SPHERES. *****

***** MISSION DURATION: 5,000 YEARS *****

* ELECTRON MODELS:	
AE6:	INNER ZONE-SOLAR MAX NO UNCERTAINTY FACTOR WAS APPLIED TO THE MODEL DATA.
AE17:	OUTER ZONE-INTERIM MODEL WITHOUT SOLAR CYCLE DEPENDENCE. FOR ENERGIES ABOVE 1.5 MEV, THIS MODEL CONTAINS UPPER & LOWER LIMIT VALUES TO ACCOUNT FOR DISCREPANCY BETWEEN EXISTING DATA SETS. THE AE17-41 FAVORS VAMPALA'S FIT TO JULY-19 DATA WHILE AE17-10 IS MORE REPRESENTATIVE OF ALL THE DATA SETS PRESENTLY AVAILABLE TO NSSDC.
+ SOLAR PROTON MODEL:	
SOLPRO:	SOLAR FLARE PROTONS AT 1 AU UNATTENUATED, INTERPLANETARY FOR CUTOFF DIPOL SHIELD OF 5 E.R.
FOR TAUS=60,MO=0.0,90% OF ALL EVENTS=4 86.81% GEOMAGNETIC SHIELDING APPLIED	
NOTE: 1) DENOTES THE DEGREE OF CONFIDENCE ONE WISHES TO ASSIGN TO RESULTS, NAMELY THAT FOR THE SPECIFIC MISSION DURATION THE CALCULATED FLUENCE'S ARE THE SMALLEST VALUES WHICH WILL NOT BE EXCEEDED BY ACTUALLY ENCOUNTERED INTERSTARS.	
>> THE AE17-0 VERSION WAS USED FOR THESE CALCULATIONS <<	

** PROTON MODEL:
 IT IS NOT ADVISABLE TO EXTRAPOLATE THE SOLAR
 PROTON SPECTRA EITHER TOWARDS LOWER NOB
 TOWARDS HIGHER ENERGIES BECAUSE THE DATA SETS
 USED IN THE CONSTRUCTION OF THE MODEL ISALF-
 LITE MEASUREMENTS MADE DURING THE 20TH SOLAR
 CYCLE (1964-1975) DO NOT CONTAIN INFORMATION
 FOR $E < 10$ AND $E > 200$ MEV.

APR-MAG: TRAPPED PROTONS-SOLAR MAX
 NO UNCERTAINTY FACTOR WAS APPLIED TO THE MODEL DATA.


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***** ORBITAL FLUX STUDY WITH COMPOSITE PARTICLE ENVIRONMENTS ***** IMPLICATION OF 1979
***** UNCERTAINTY FACTORS (UF) APPLIED FOR THIS RUN ARE: POB PROTONS (APR)- UF= 1.0; POB INNER ZONE ELECTROFORM (AUG)- UF= 1.0;
***** MAGNETIC COORDINATES B AND L COMPUTED BY INVARS OF 1972 WITH ALMA-8 MODEL = 3683 KM ** BARACHOU ET AL. 1981-TRM 1976, IAPR 1989, P.
***** VEHICLE : MVLX3-S-HAY ** INCLINATION = 60DEG ** PERIGEE = 3689 KM ** APOGEE = 4843 KM ** RAY CARLISLE TAPER: 7.5% PER 1000 S ** PENULTI
***** FOR INFORMATION ON EXPLANATION CONTACT E.G. STASSINOPOULOS AT NASA-35FC CODE 001 GREEK ISLET, MARVELAND 20771 TEL: (301)-343-8067

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***** DOSE AT CENTER OF ALUMINUM SPHERES *****
*****
***** MISSION DURATION: 5.0000 YEAR(S) *****
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[illegible]

ELECTRON MODELS:

AB6: INNER ZONE-SOLAR MAX NO UNCERTAINTY FACTOR WAS APPLIED TO THE MODEL DATA.

* Study Points to Note:

SUBJECT: SOLAR FLARE EMISSION, AT 1 AU
 UNATTENUATED INTERPLANETARY
 FOR CATALOG OF SOLAR FLARE EMISSIONS

2217: OUTP 208L-INTERIM MODEL REV. 1. THE ABOVE SOLAR CYCLE DEPENDENCE. 8-
FOR ENERGIES ABOVE 1.0 MEV. THIS SOLAR CYCLE DEPENDENCE CONTAINS ORDER 5-
EXCESS LIMIT VALUES. TO ACCOUNT FOR DISCREPANCY BETWEEN
EXISTING DATA SETS. THE AP17-70 FAVORS VAMPOLA. 3 FIT
TO 0.01-19 DATA WHILE AP17-70 IS MORE REPRESENTATIVE OF
ALL THE DATA SETS PRESENTLY AVAILABLE TO NSRC.

FOR TAG (C-100) : 0.25
SUPPLIES

CONDUCTED FOR PURPOSES OF CONFIRMING ONE WISHES
TO ASSIST IN THE RESEARCH, ANALYSIS, AND THE
SPECIFIC NATION'S POLITICAL CALCULATED
EVIDENCES ARE THE SHARPEST VALUES WHICH WILL
NOT BE EXCEPTED BY ACTUALLY ENCOUNTERED
UNPREDICTABLE

PROTON MODEL:

AP8-HAC: TRAPPED PROTONS-SOLAR MAX
NO UNCERTAINTY FACTOR WAS
APPLIED TO THE MODEL DAT.

IT IS NOT ANTICIPATED TO EXAPOLATE THE SOLAR PROTON DATA TO LOWER ENERGY RANGES. THE DATA SETS TO HIGH ENERGIES ARE USED TO CORRECT THE SOLAR FLUXES IN THE CORRECTION MADE DURING THE 20TH SOLAR MINIMUM PERIOD (1964-1975). DO NOT CONTAIN INFORMATION FROM 1964 AND 1975. PAGES 10 AND 11.

[illegible]

***** RISE AT CENTER OF ALUMINUM SPRINGS *****
***** DISCUSSION DURATION: 5,000 YEARS(C) *****

SHIELD THICKNESS (MILLIMETER)	T (MILLIMETER)	INNER 2N (MILLIMETER)	CLIFF 2N (MILLIMETER)	ELECTR (NS)	TOTAL T (MILLIMETER)	UPRST T (MILLIMETER)	THMP (MILLIMETER)	SCLAR (MILLIMETER)	PRTONS	TOTAL (MILLIMETER)	TOTAL ALL (MILLIMETER)	TOTAL SURFCS (MILLIMETER)
0.01	1.0	7.485E	0.0	1.77E	7.46E7	2.97E	0.0	0.0	0.0	7.25E	1.04E	0.0
0.02	2.0	4.55E	0.0	1.54E	5.10E7	2.81E	0.0	0.0	0.0	1.00E	4.51E	0.0
0.03	3.0	3.47E	0.0	1.35E	3.55E7	1.92E	0.0	0.0	0.0	1.00E	3.51E	0.0
0.04	4.0	2.78E	0.0	1.17E	1.97E7	1.39E	0.0	0.0	0.0	1.00E	2.78E	0.0
0.05	5.0	1.42E	0.0	1.01E	1.47E7	1.19E	0.0	0.0	0.0	1.00E	1.51E	0.0
0.06	6.0	1.02E	0.0	0.85E	1.12E7	1.00E	0.0	0.0	0.0	1.00E	1.02E	0.0
0.07	7.0	0.41E	0.0	0.35E	0.81E7	0.73E	0.0	0.0	0.0	1.00E	0.41E	0.0
0.08	8.0	0.26E	0.0	0.21E	0.49E7	0.42E	0.0	0.0	0.0	1.00E	0.26E	0.0
0.09	9.0	0.17E	0.0	0.14E	0.31E7	0.28E	0.0	0.0	0.0	1.00E	0.17E	0.0
0.10	10.0	0.11E	0.0	0.09E	0.18E7	0.15E	0.0	0.0	0.0	1.00E	0.11E	0.0
0.20	20.0	0.05E	0.0	0.04E	0.08E7	0.07E	0.0	0.0	0.0	1.00E	0.05E	0.0
0.30	30.0	0.03E	0.0	0.02E	0.04E7	0.03E	0.0	0.0	0.0	1.00E	0.03E	0.0
0.40	40.0	0.02E	0.0	0.01E	0.02E7	0.02E	0.0	0.0	0.0	1.00E	0.02E	0.0
0.50	50.0	0.01E	0.0	0.00E	0.01E7	0.01E	0.0	0.0	0.0	1.00E	0.01E	0.0
0.60	60.0	0.00E	0.0	0.00E	0.00E7	0.00E	0.0	0.0	0.0	1.00E	0.00E	0.0
0.70	70.0	0.00E	0.0	0.00E	0.00E7	0.00E	0.0	0.0	0.0	1.00E	0.00E	0.0
0.80	80.0	0.00E	0.0	0.00E	0.00E7	0.00E	0.0	0.0	0.0	1.00E	0.00E	0.0
0.90	90.0	0.00E	0.0	0.00E	0.00E7	0.00E	0.0	0.0	0.0	1.00E	0.00E	0.0
1.00	100.0	0.00E	0.0	0.00E	0.00E7	0.00E	0.0	0.0	0.0	1.00E	0.00E	0.0

• ELECTFCN MODEL :

AE6: INNER ZONE-SOLAF MAX

~~REG. INNER ZONE-SULFUR MAP~~

4-11-68

SOURCE: CIA; FLA. SECTIONS AT: 1. U.
(UNATTENLA. 2. INTERFLANITARY)
REF. (CUT OFF) - IPCL. (CUT OFF) - E. R. O.

ALB17: OUTER ZONE-INTERIM MODEL WITH 12 SCAR CYCLE DEPENDENCE.
FOR ENERGIES ABOVE 15 MEV, THIS MODEL CONTAINS SUPPLEMENTARY
LOWERING LIMIT VALUES TO ACCOUNT FOR DISCREPANCY BETWEEN
EXISTING DATA SETS. THE ALB17 FACTORS VARY AS $E^{-1.5}$ FOR
OUTER ZONE DATA WHILE ALB17-LLC IS CLOSE REPRESENTATIVE OF
ALL THE DATA SETS PRESENTLY AVAILABLE TO NSDDC.

[illegible]

*** PROTON MODEL ***

~~ADD-MAC: IRAPREC PRODIGAS-SCLAR MAX~~
NO UNCERTAINTY FACTOR WAS APPLIED TO THE MODEL DATA.

THE UNIVERSITY OF CHICAGO
 LIBRARY
 540 EAST 58TH STREET
 CHICAGO, ILL. 60637
 TEL: 773-936-5000
 FAX: 773-936-5000
 WWW: WWW.CHICAGO.EDU

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** ORBITAL FLUX STUDY WITH IMPULSIVE PARTICLE ENVIRONMENT: VETTES A97, A60, AF17 FOR 75 CM MAXIMUM JNIFLUX CF 1979
** UNCERTAINTY FACTOR OF 2 APPLIED TO THIS RUN AREA: CCF BETTINS (A80) - DIFF = 1.0; CCF PINKUSZON ELECTRONIC (A66) - UF = 1.0
** MAGNETIC COEFFICIENTS HAVE BEEN COMPUTED BY INVAFA OF 1972 AND ALL-SCALE MODEL #1: EPOCHAL DIFF. T.A.: 16-HR 1975 % TIME = 1989.5
** MAGNETIC COEFFICIENTS HAVE BEEN MATCHED TO THE DATA SET: 44 DEGREE = 6.304KM ** MARLINER TAPE: 1D54K4 86 PERIOD = 1.588
** FOR INFORMATION ON EXPANATING CONTACTS AT NASA-CFPC,COL FOLI, CHECKED BY MARFLAND 2C771.TE.C(301)-3AA-8087

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***** DUSEL AT CENTER OF ALUMINUM SPHERE *****

***** MISSION DURATION: 6,000 YEARS! *****

SPEED TICKETS		T		LND 2 PM		ELECTRON		TOTAL		TRANSC		CH 1 +		TOTAL		TOTAL	
5	10	15	20	25	30	35	40	45	50	55	60	65	70	75	80	85	90
0.01	0.04	0.07	0.10	0.13	0.16	0.19	0.22	0.25	0.28	0.31	0.34	0.37	0.40	0.43	0.46	0.49	0.52
0.03	0.11	0.19	0.27	0.35	0.43	0.51	0.59	0.67	0.75	0.83	0.91	0.99	1.07	1.15	1.23	1.31	1.39
0.04	0.12	0.20	0.28	0.36	0.44	0.52	0.60	0.68	0.76	0.84	0.92	1.00	1.08	1.16	1.24	1.32	1.40
0.05	0.13	0.21	0.29	0.37	0.45	0.53	0.61	0.69	0.77	0.85	0.93	1.01	1.09	1.17	1.25	1.33	1.41
0.06	0.14	0.22	0.30	0.38	0.46	0.54	0.62	0.70	0.78	0.86	0.94	1.02	1.10	1.18	1.26	1.34	1.42
0.07	0.15	0.23	0.31	0.39	0.47	0.55	0.63	0.71	0.79	0.87	0.95	1.03	1.11	1.19	1.27	1.35	1.43
0.08	0.16	0.24	0.32	0.40	0.48	0.56	0.64	0.72	0.80	0.88	0.96	1.04	1.12	1.20	1.28	1.36	1.44
0.09	0.17	0.25	0.33	0.41	0.49	0.57	0.65	0.73	0.81	0.89	0.97	1.05	1.13	1.21	1.29	1.37	1.45
0.10	0.18	0.26	0.34	0.42	0.50	0.58	0.66	0.74	0.82	0.90	0.98	1.06	1.14	1.22	1.30	1.38	1.46
0.11	0.19	0.27	0.35	0.43	0.51	0.59	0.67	0.75	0.83	0.91	0.99	1.07	1.15	1.23	1.31	1.39	1.47
0.12	0.20	0.28	0.36	0.44	0.52	0.60	0.68	0.76	0.84	0.92	1.00	1.08	1.16	1.24	1.32	1.40	1.48
0.13	0.21	0.29	0.37	0.45	0.53	0.61	0.69	0.77	0.85	0.93	1.01	1.09	1.17	1.25	1.33	1.41	1.49
0.14	0.22	0.30	0.38	0.46	0.54	0.62	0.70	0.78	0.86	0.94	1.02	1.10	1.18	1.26	1.34	1.42	1.50
0.15	0.23	0.31	0.39	0.47	0.55	0.63	0.71	0.79	0.87	0.95	1.03	1.11	1.19	1.27	1.35	1.43	1.51
0.16	0.24	0.32	0.40	0.48	0.56	0.64	0.72	0.80	0.88	0.96	1.04	1.12	1.20	1.28	1.36	1.44	1.52
0.17	0.25	0.33	0.41	0.49	0.57	0.65	0.73	0.81	0.89	0.97	1.05	1.13	1.21	1.29	1.37	1.45	1.53
0.18	0.26	0.34	0.42	0.50	0.58	0.66	0.74	0.82	0.90	0.98	1.06	1.14	1.22	1.30	1.38	1.46	1.54
0.19	0.27	0.35	0.43	0.51	0.59	0.67	0.75	0.83	0.91	0.99	1.07	1.15	1.23	1.31	1.39	1.47	1.55
0.20	0.28	0.36	0.44	0.52	0.60	0.68	0.76	0.84	0.92	1.00	1.08	1.16	1.24	1.32	1.40	1.48	1.56
0.21	0.29	0.37	0.45	0.53	0.61	0.69	0.77	0.85	0.93	1.01	1.09	1.17	1.25	1.33	1.41	1.49	1.57
0.22	0.30	0.38	0.46	0.54	0.62	0.70	0.78</										

* ELECTRONIC *

AF6: UNRECORDED. AAR
NO CERTAINLY - ACTION WAS APPLIED TO THE NCOLL DATA.

ACT17: TOTAL ZONE-INTER-10 MODEL WITHOUT SOLAR CYCLE DEPENDENCE.
 FOR ENERGY VALUES 1.0 MW, THIS MODEL CONTAINS DEEPER
 LAYER FLUX VALUES IN ACCORDANCE WITH DISCREPANCY BETWEEN
 EXISTING DATA SETS. THE ACT17-10 FAVORS VASCULAR FIT
 INCLUDING DATA SETS. ACT17-10 IS MORE REPRESENTATIVE OF
 ACT-10 DATA SETS AND ACTIVITY VARIATION TO R5000.

• 25 •

ADM-MAC: ISADDO 1111155ULAN MAX
NO OTHER TAINITY PAY ICD WAS ADDED TO THE MONTI DATE.

• 501 AR 7 • 198

SCOPE: POLAR PLATE PROTONS AT 1 AU
UNATTENUATED, INTERPLANETARY
FIELD OFF DIPOLE SHELL OF S.E.R.

FIG. 1. α (C.M.O.) OF AL EVENTS OF 1.0 (C.M.O.) MAGNETIC SHIELDING APPLIED

NOTE: 3. MEASURES THE DEGREE OF CONFIDENCE ONE WISHES TO ASSIGN TO RESULTS, NAMELY THAT FOR THE SPECIFIC MICH DURATION THE CALCULATED FLEW WAS ABOVE THE SMALLEST VALUES WHICH WILL NOT BE EXCEEDED BY ACTUALLY OCCURRED INSTANCES.

IT IS NOT AVAILABLE TO EXTRAPOLATE THE SOLAR
CONSTANT EITHER TOWARDS LOWER OR
TOWARDS HIGHER ENERGIES BECAUSE THE DATA SETS
USED IN THE CONSTRUCTION OF THE MODEL SATELLITE
LIFE TIME AGREEMENTS MADE DURING THE 20TH SOLAR
CYCLE (1964-1976) DO NOT CONTAIN INFORMATION
FOR 100 AND 25000 MEV.

```

*** ORBITAL FLUX, STARRY HILLS *** UNIFLUX CF 1979 ***
*** ENHANCEMENTS: VETTES A95, A-9, AE17 FOR SCLM, MAXIMUM ***
*** FOR THE ENHANCEMENTS: CF=1.00, FOR THE ENHANCEMENTS: ***
*** UNCESTAINITY FACTORS OF 1.00, FOR THE ENHANCEMENTS: ***
*** MAGNETIC COEFFICIENTS: 1.00, FOR THE ENHANCEMENTS: ***
*** FOR INFORMATION OR FOR ANIMATE: 1.00, FOR THE ENHANCEMENTS: ***

```

```
*****  
***** OCE AT CENTER OF ALUMINUM SPHERES *****  
*****  
*****  
***** VLECTION EQUATION: 5.000 YEARS(S) *****  
*****  
*****
```

STATE	1960	1961	1962	1963	1964	1965	1966	1967	1968	1969	1970	1971	1972	1973	1974	1975	1976	1977	1978	1979	1980	1981	1982	1983	1984	1985	1986	1987	1988	1989	1990	1991	1992	1993	1994	1995	1996	1997	1998	1999	2000	2001	2002	2003	2004	2005	2006	2007	2008	2009	2010	2011	2012	2013	2014	2015	2016	2017	2018	2019	2020	2021	2022	2023	2024	2025	2026	2027	2028	2029	2030	2031	2032	2033	2034	2035	2036	2037	2038	2039	2040	2041	2042	2043	2044	2045	2046	2047	2048	2049	2050	2051	2052	2053	2054	2055	2056	2057	2058	2059	2060	2061	2062	2063	2064	2065	2066	2067	2068	2069	2070	2071	2072	2073	2074	2075	2076	2077	2078	2079	2080	2081	2082	2083	2084	2085	2086	2087	2088	2089	2090	2091	2092	2093	2094	2095	2096	2097	2098	2099	2100	2101	2102	2103	2104	2105	2106	2107	2108	2109	2110	2111	2112	2113	2114	2115	2116	2117	2118	2119	2120	2121	2122	2123	2124	2125	2126	2127	2128	2129	2130	2131	2132	2133	2134	2135	2136	2137	2138	2139	2140	2141	2142	2143	2144	2145	2146	2147	2148	2149	2150	2151	2152	2153	2154	2155	2156	2157	2158	2159	2160	2161	2162	2163	2164	2165	2166	2167	2168	2169	2170	2171	2172	2173	2174	2175	2176	2177	2178	2179	2180	2181	2182	2183	2184	2185	2186	2187	2188	2189	2190	2191	2192	2193	2194	2195	2196	2197	2198	2199	2200	2201	2202	2203	2204	2205	2206	2207	2208	2209	2210	2211	2212	2213	2214	2215	2216	2217	2218	2219	2220	2221	2222	2223	2224	2225	2226	2227	2228	2229	2230	2231	2232	2233	2234	2235	2236	2237	2238	2239	2240	2241	2242	2243	2244	2245	2246	2247	2248	2249	2250	2251	2252	2253	2254	2255	2256	2257	2258	2259	2260	2261	2262	2263	2264	2265	2266	2267	2268	2269	2270	2271	2272	2273	2274	2275	2276	2277	2278	2279	2280	2281	2282	2283	2284	2285	2286	2287	2288	2289	2290	2291	2292	2293	2294	2295	2296	2297	2298	2299	2300	2301	2302	2303	2304	2305	2306	2307	2308	2309	2310	2311	2312	2313	2314	2315	2316	2317	2318	2319	2320	2321	2322	2323	2324	2325	2326	2327	2328	2329	2330	2331	2332	2333	2334	2335	2336	2337	2338	2339	2340	2341	2342	2343	2344	2345	2346	2347	2348	2349	2350	2351	2352	2353	2354	2355	2356	2357	2358	2359	2360	2361	2362	2363	2364	2365	2366	2367
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[illegible]

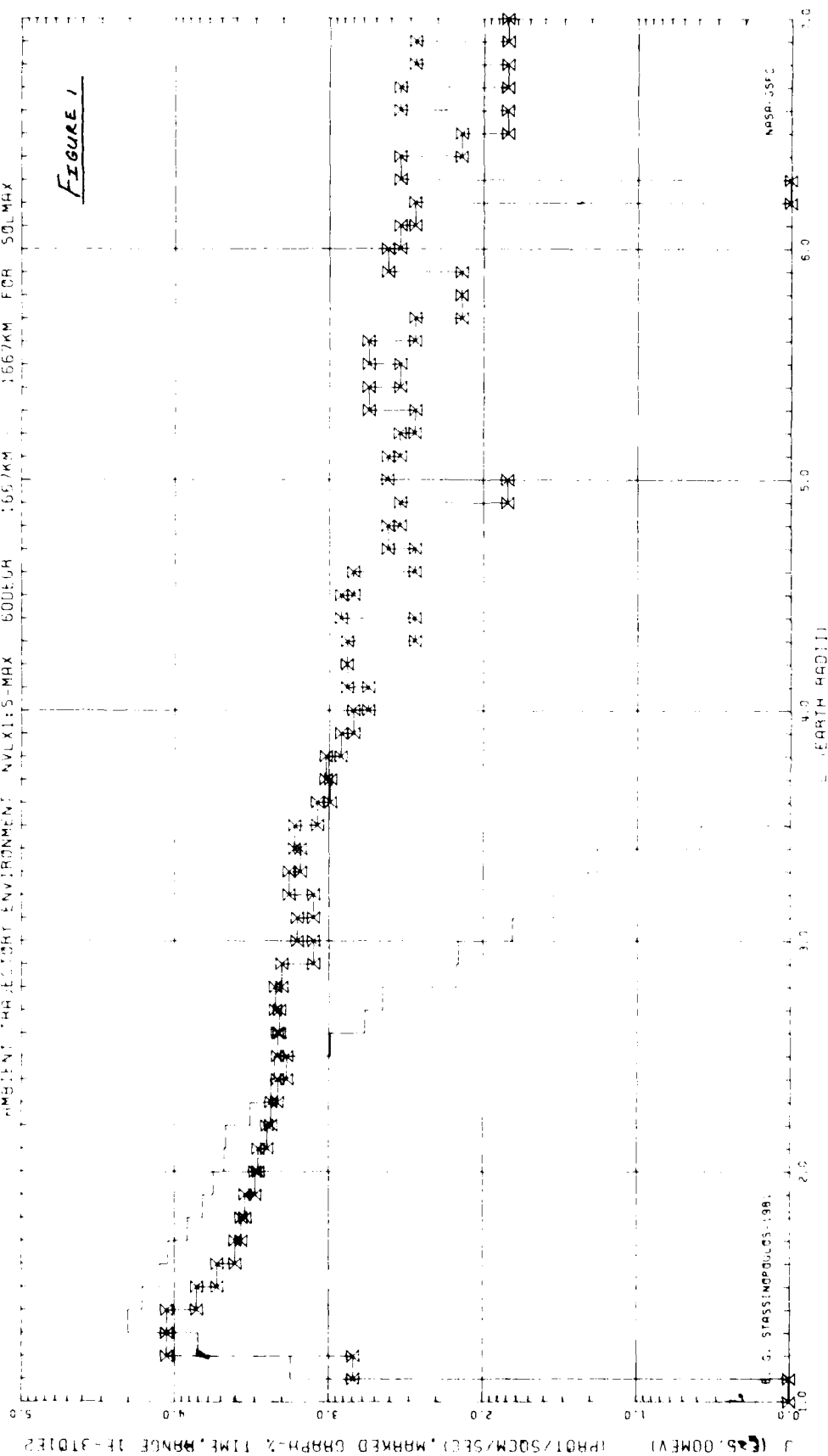
TABLE 60

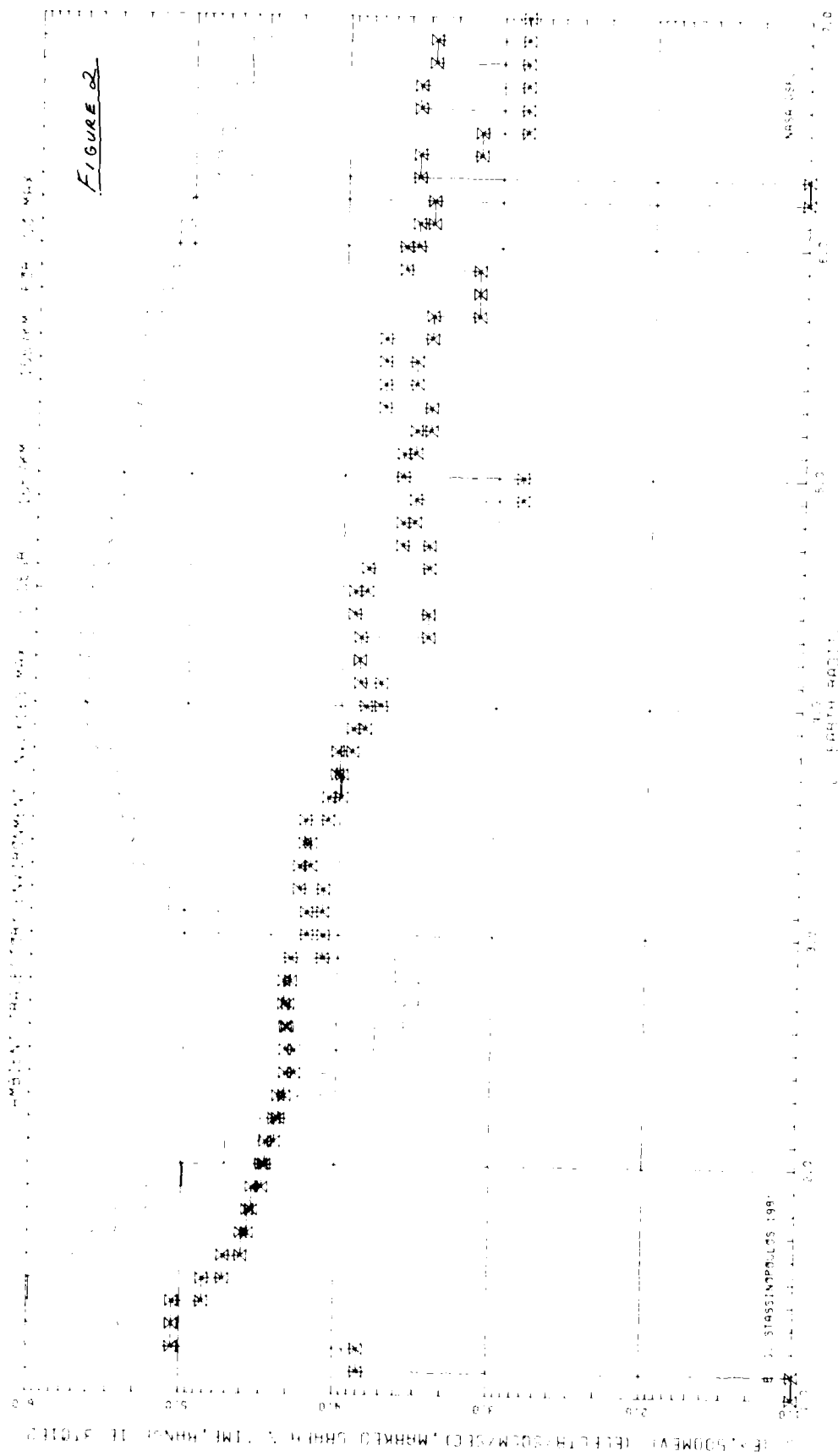
```

ENVIRONMENT:  "HARVESTING ENVIRONMENT"  NVLX1:5-MAX  600000  :667KM  FOR  SQLMAX

```

FIGURE 1





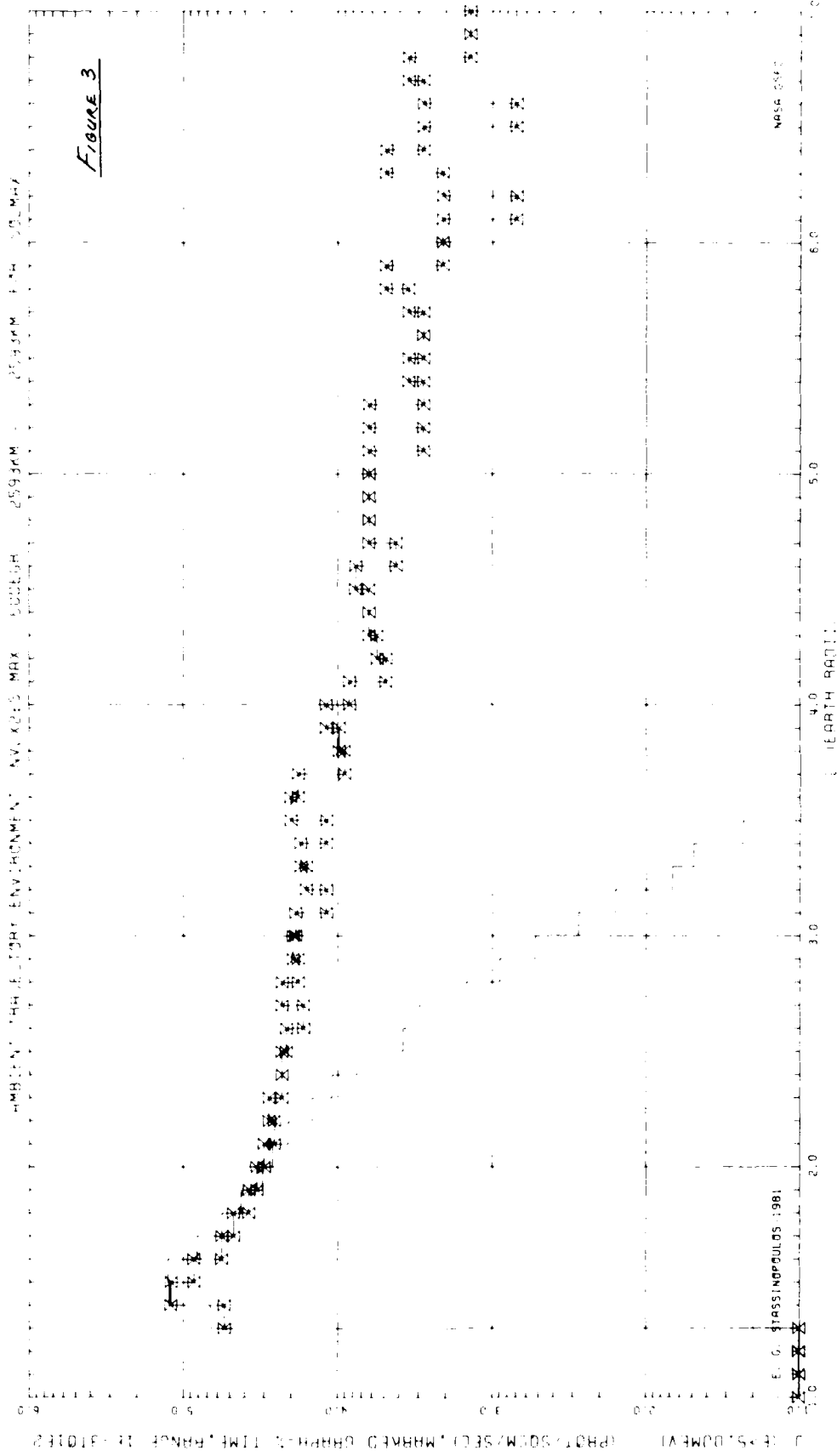


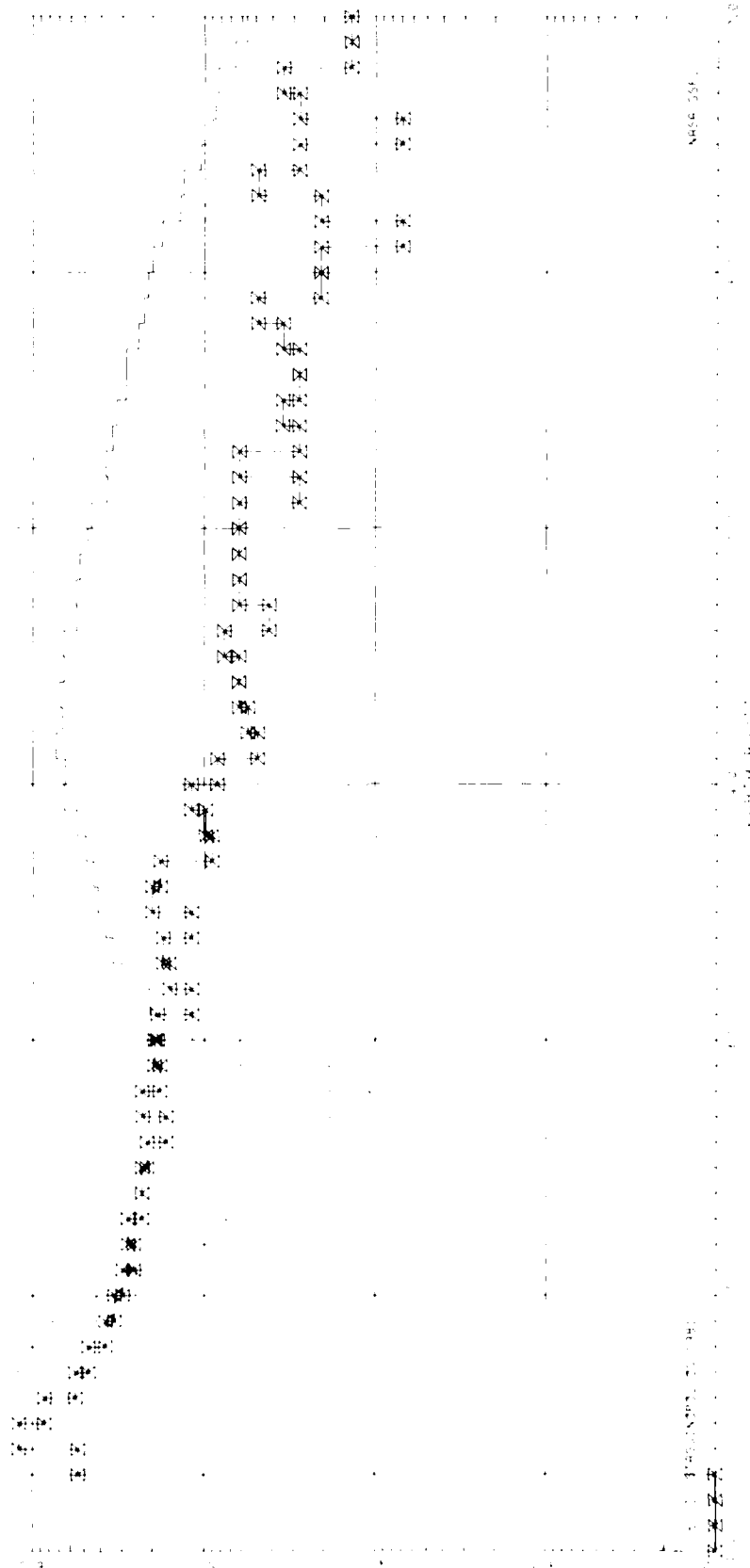
FIGURE 3

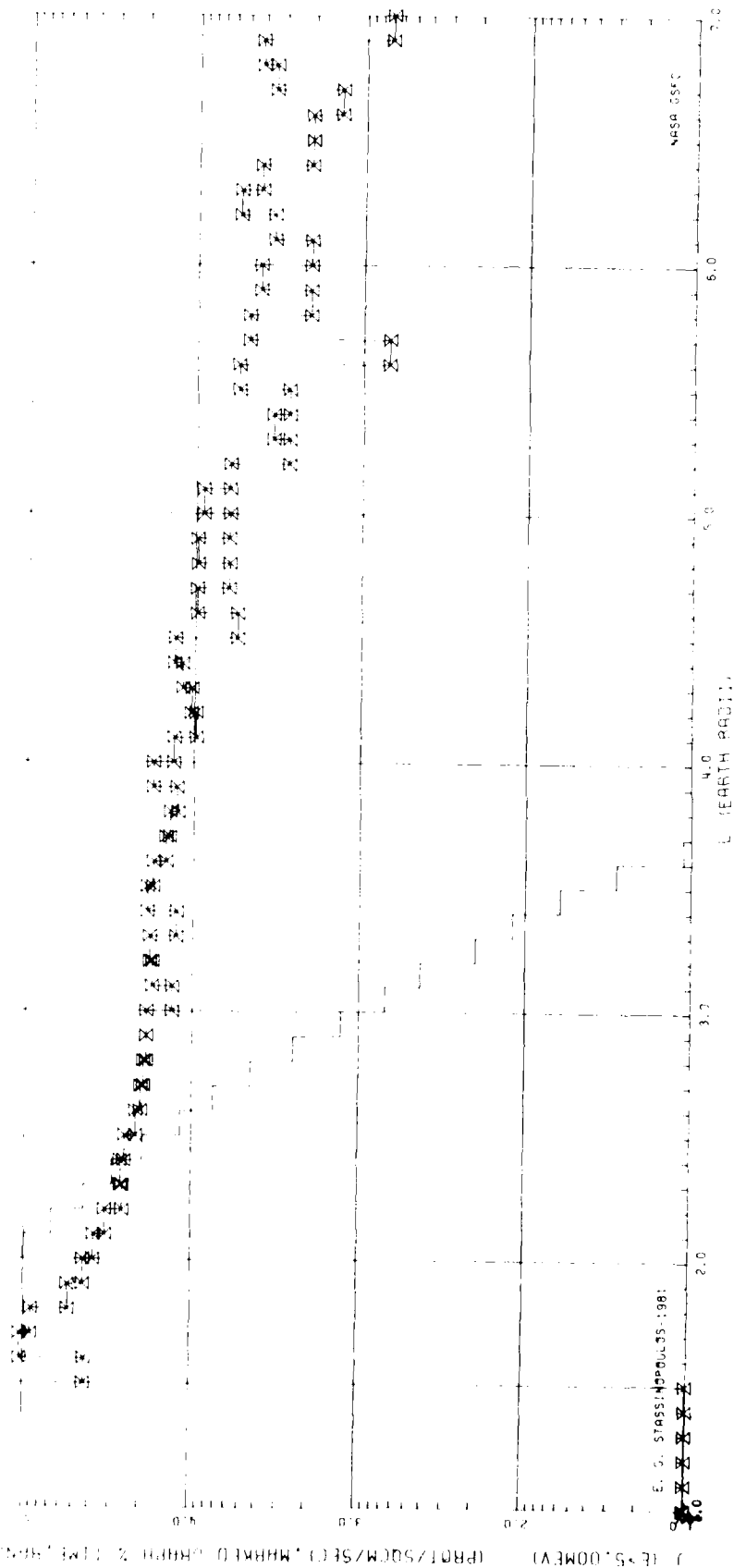
AMBLEN, THE EIGHT ENVIRONMENT, NY, K215 MAX, 50010H, 2593PM, 14H, 30 MAY

E. G. STASSINOPOULOS (1981)

[illegible]

FIGURE 4

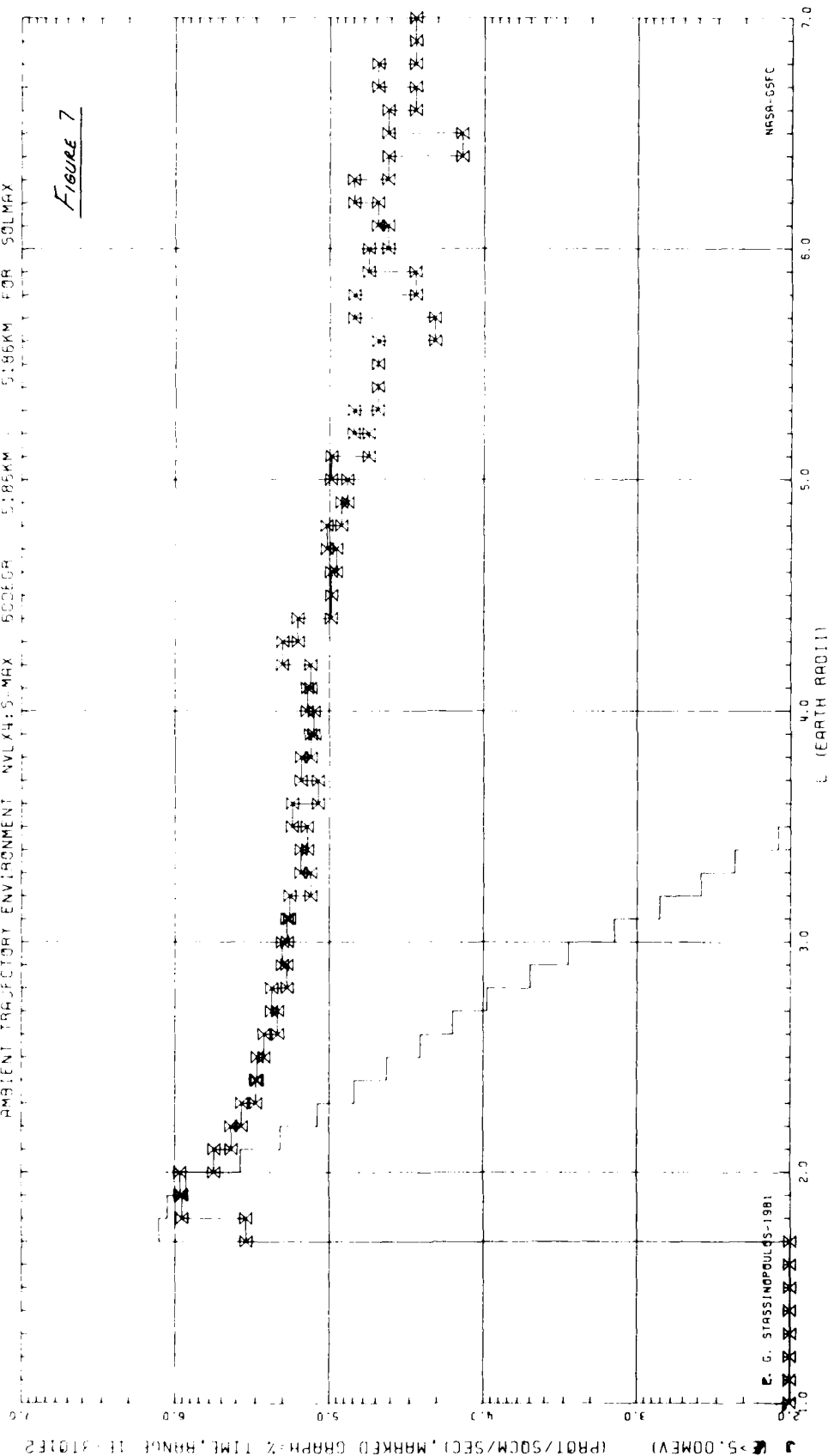


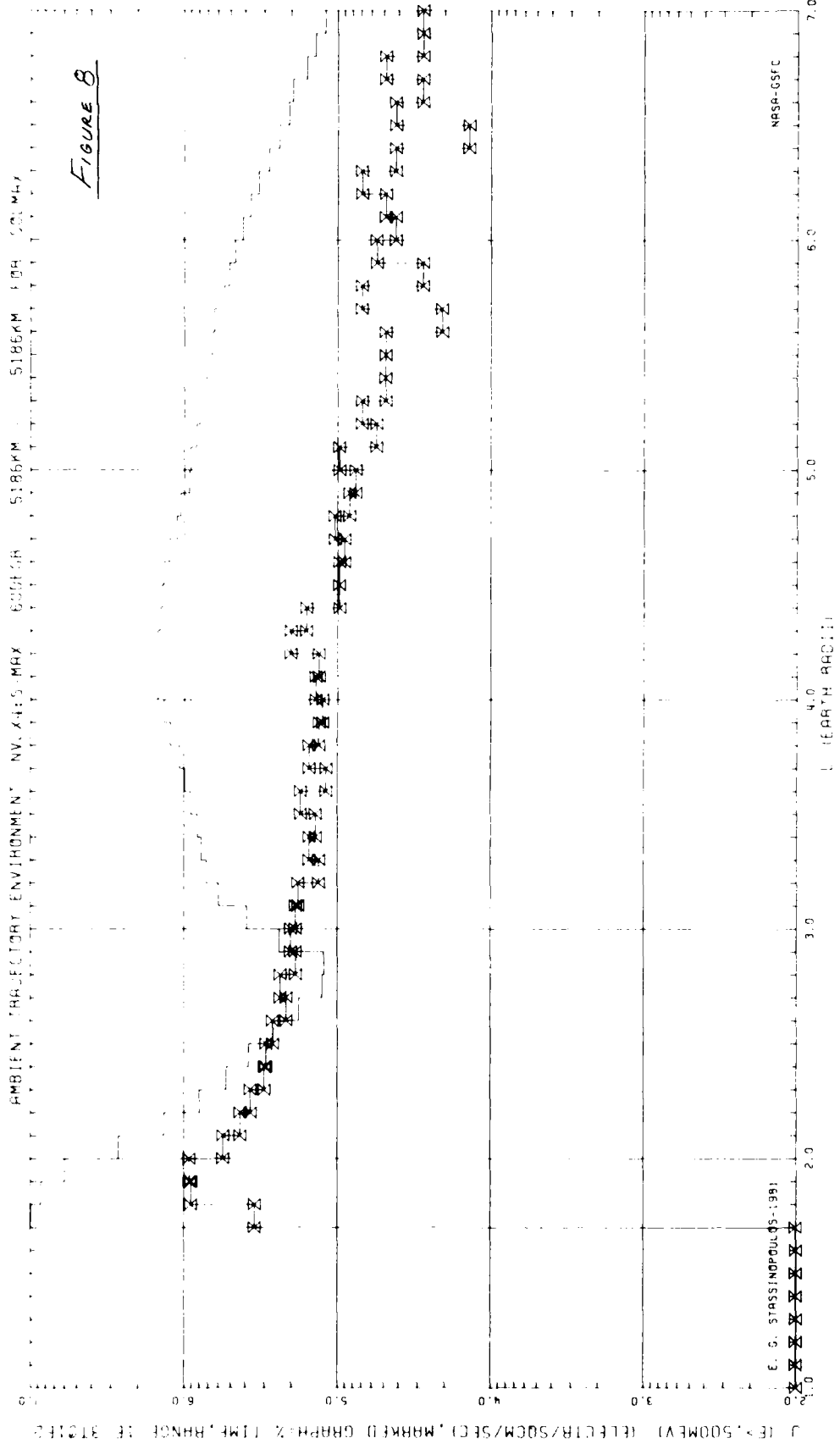
[illegible]

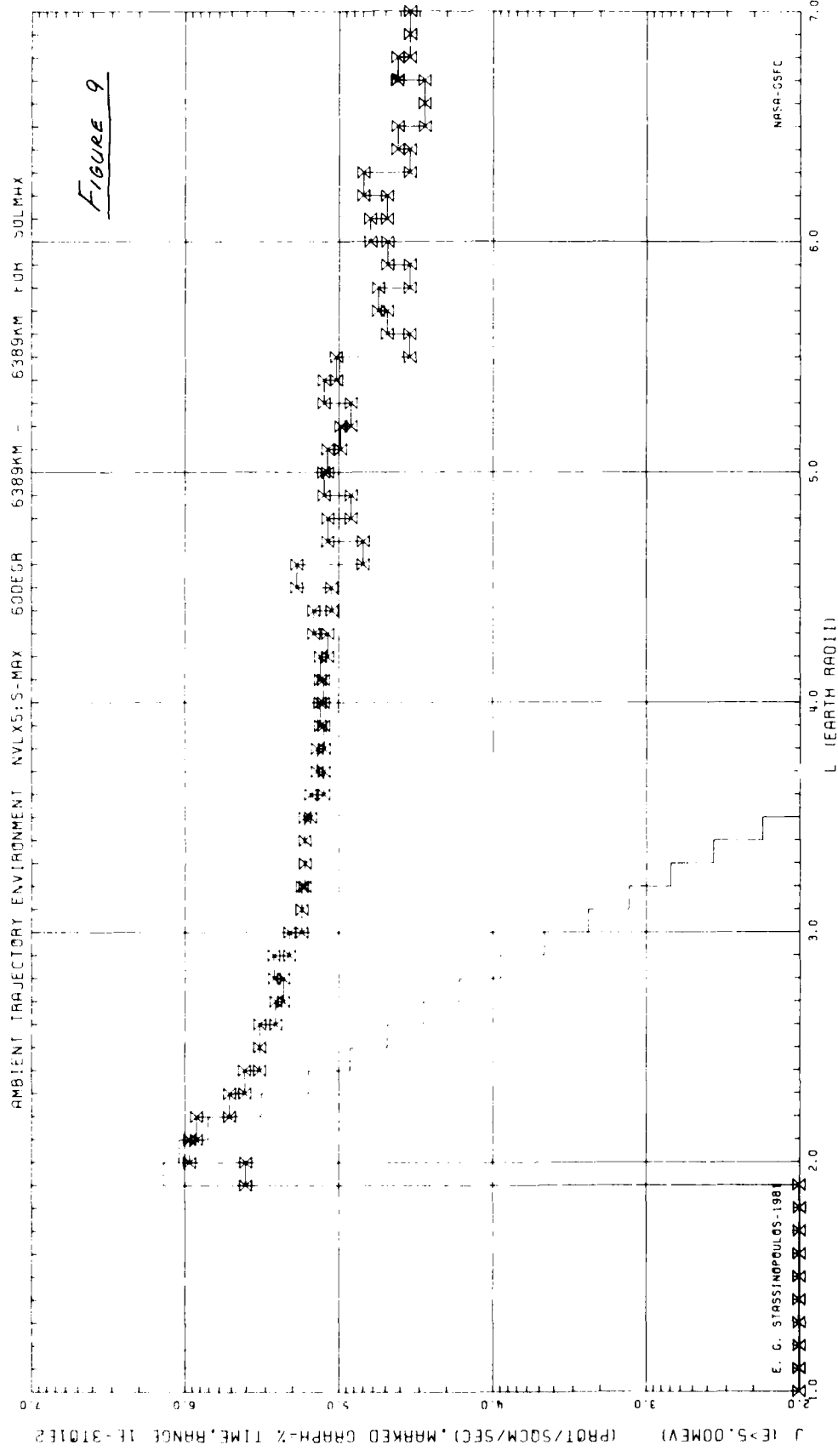


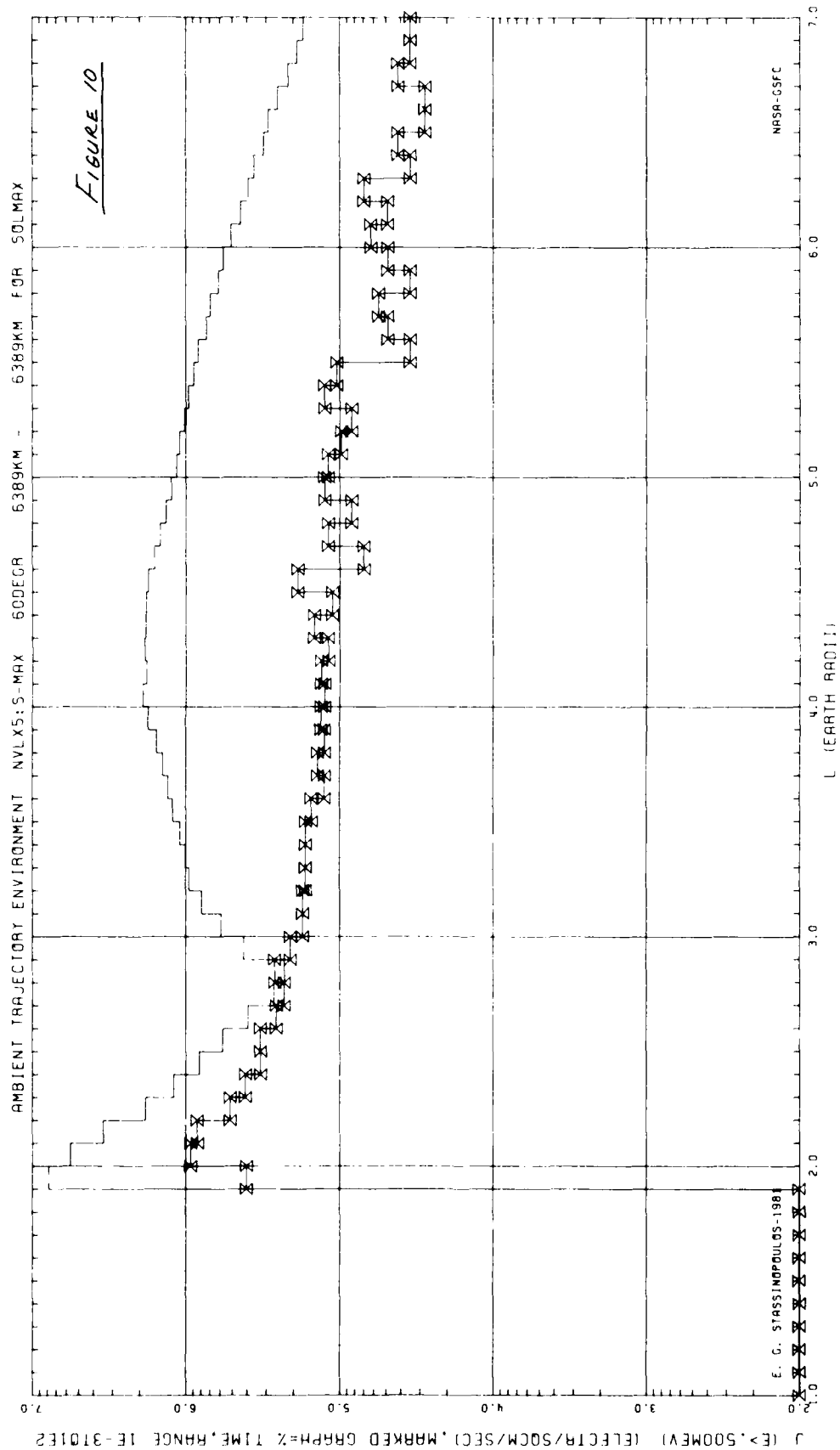
PROBENT TRAJECTORY ENVIRONMENT NVLX4:5-MAX 5006GR 5185KM 5185KM FOR SOLMAX

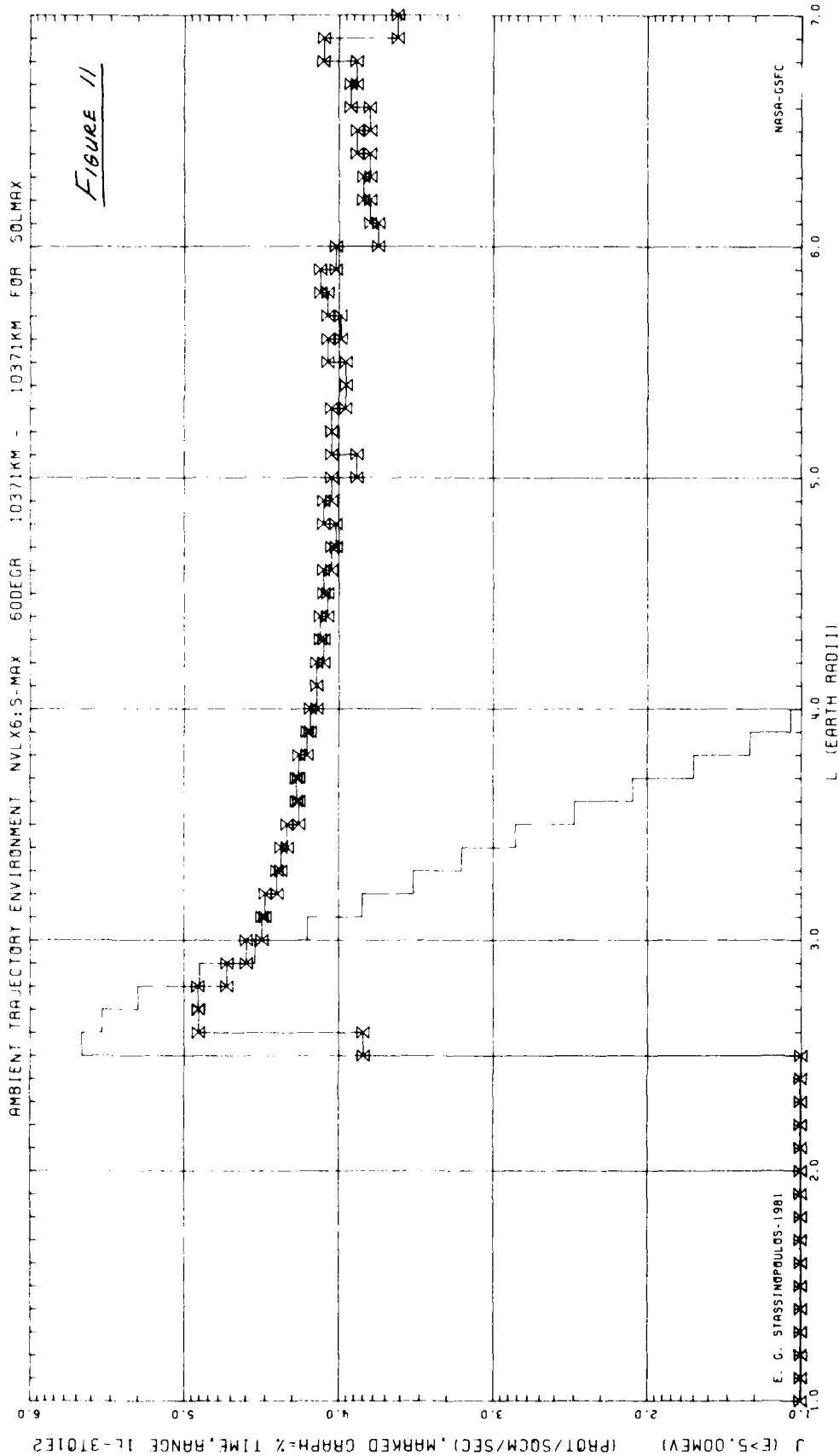
Figure 7





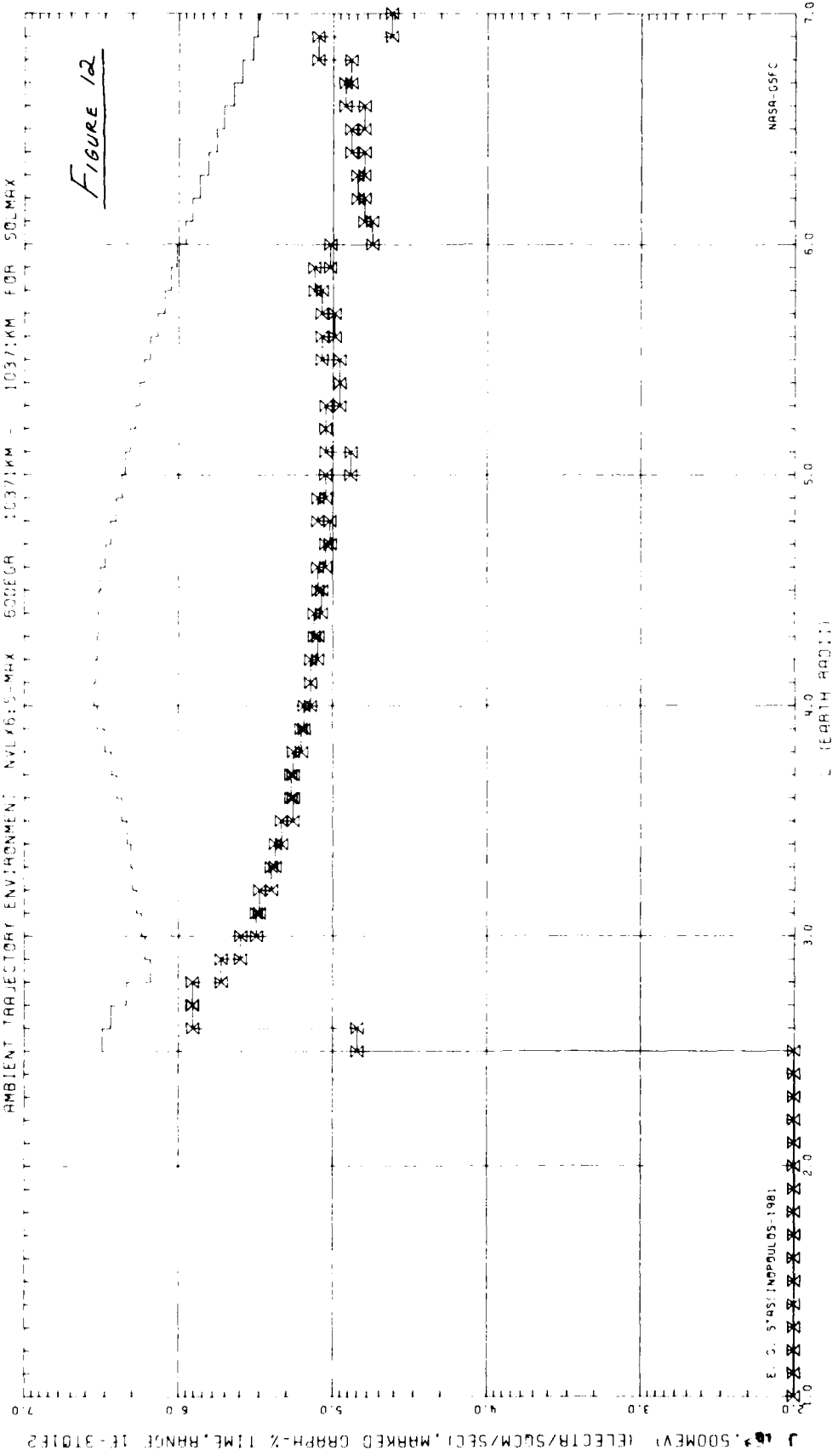






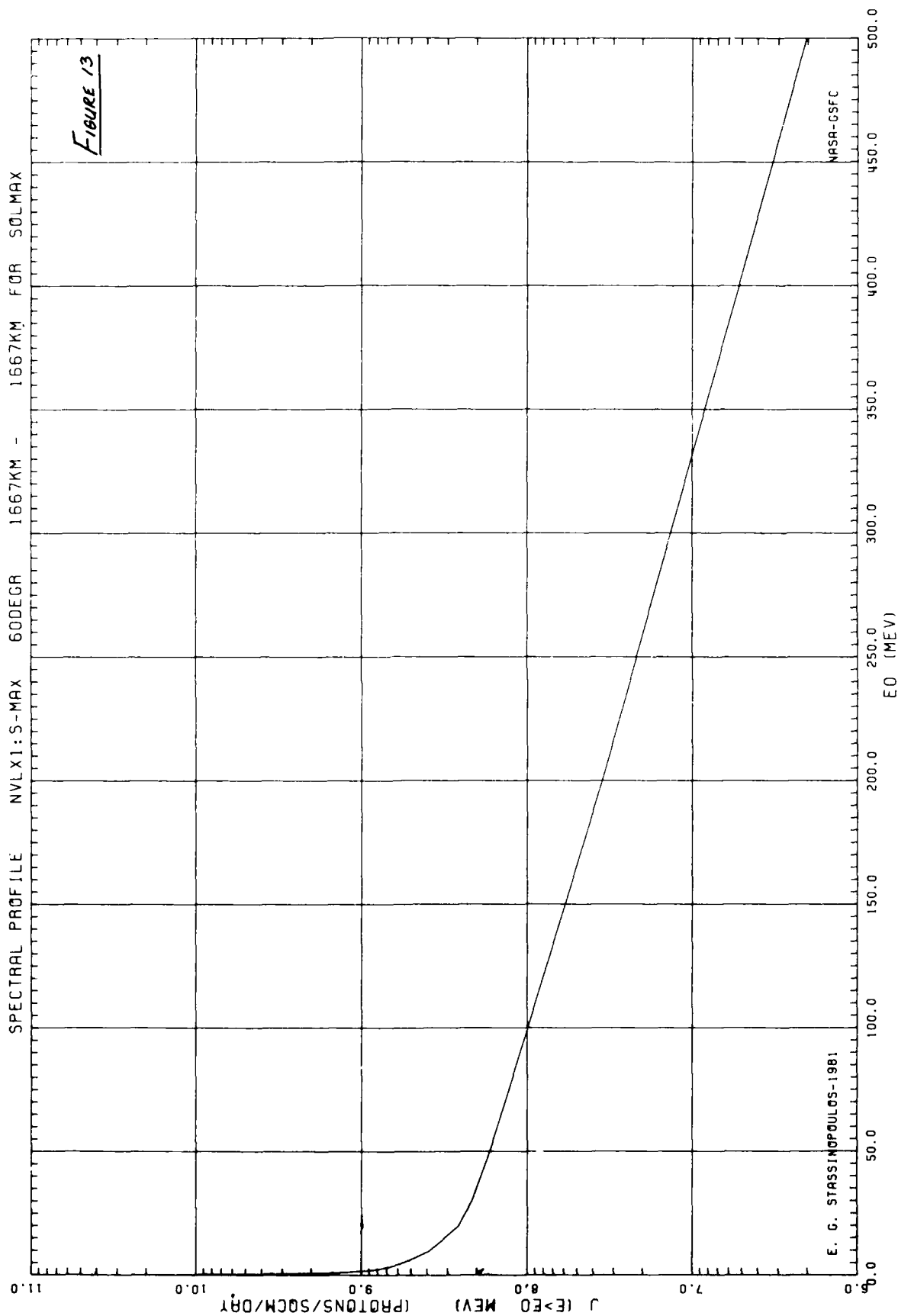
AMBIENT TRAJECTORY ENVIRONMENT: NALX6:5-MAX 600EGR 10371KM 10371KM FOR SOL MAX

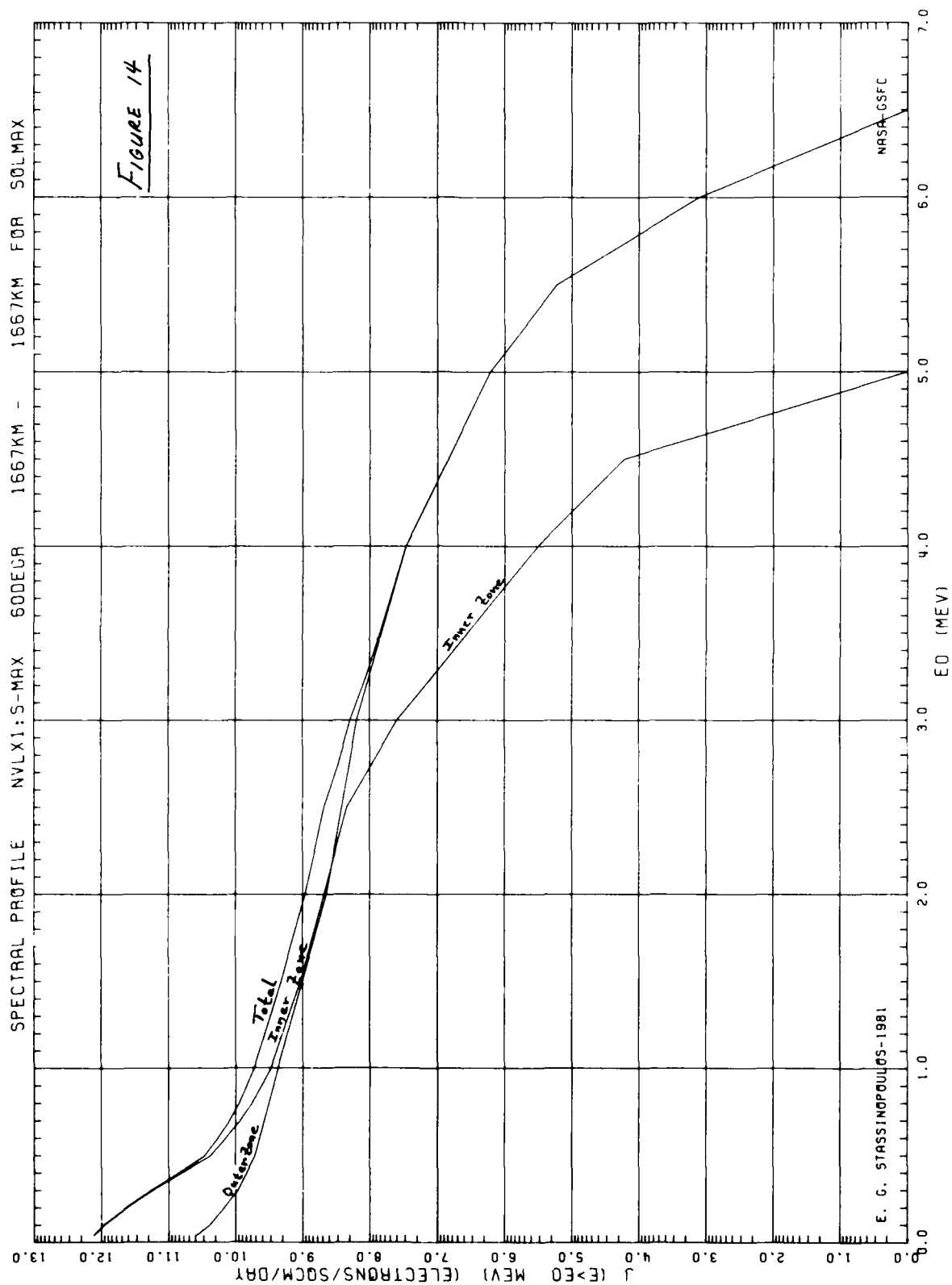
FIGURE 12

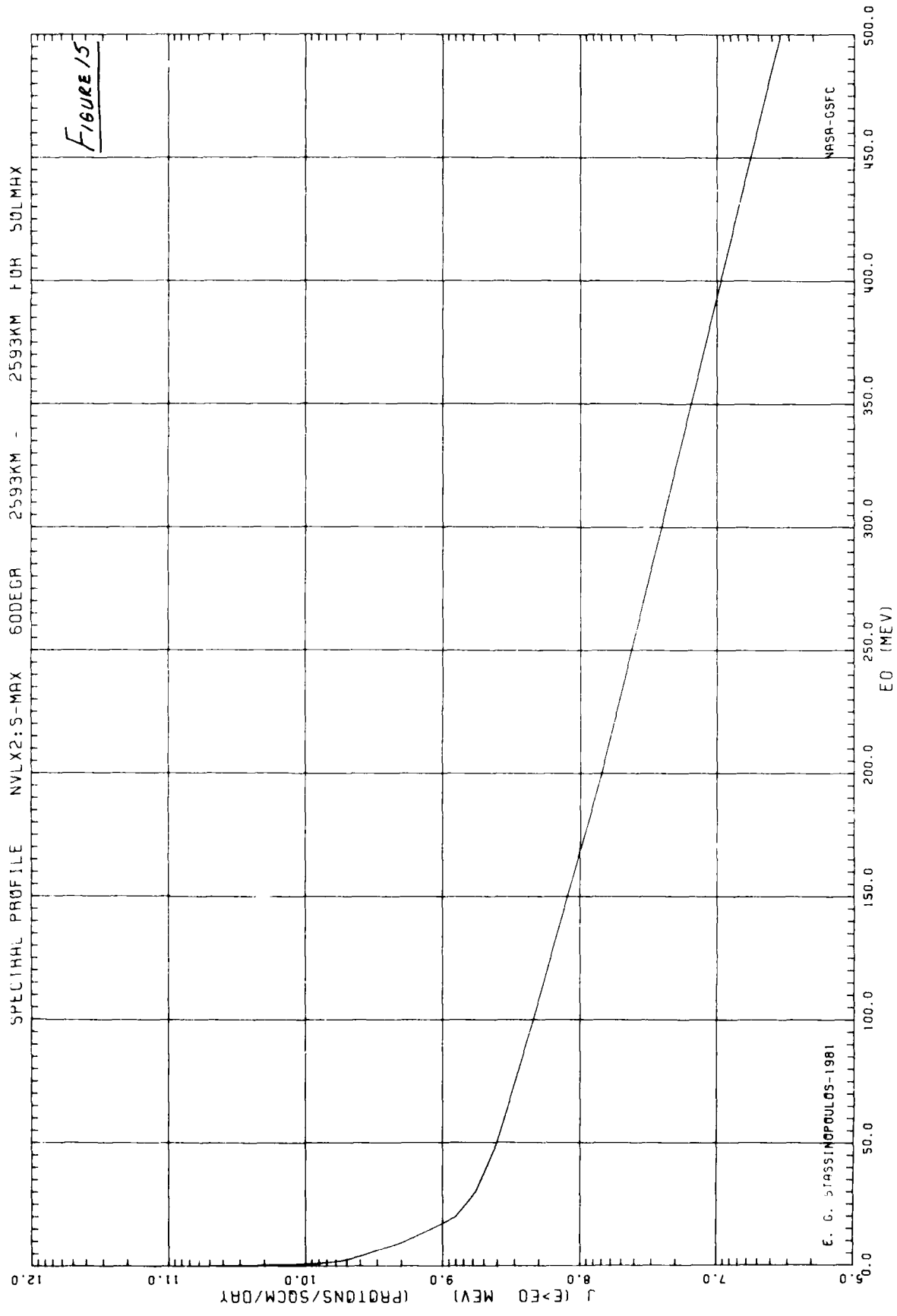


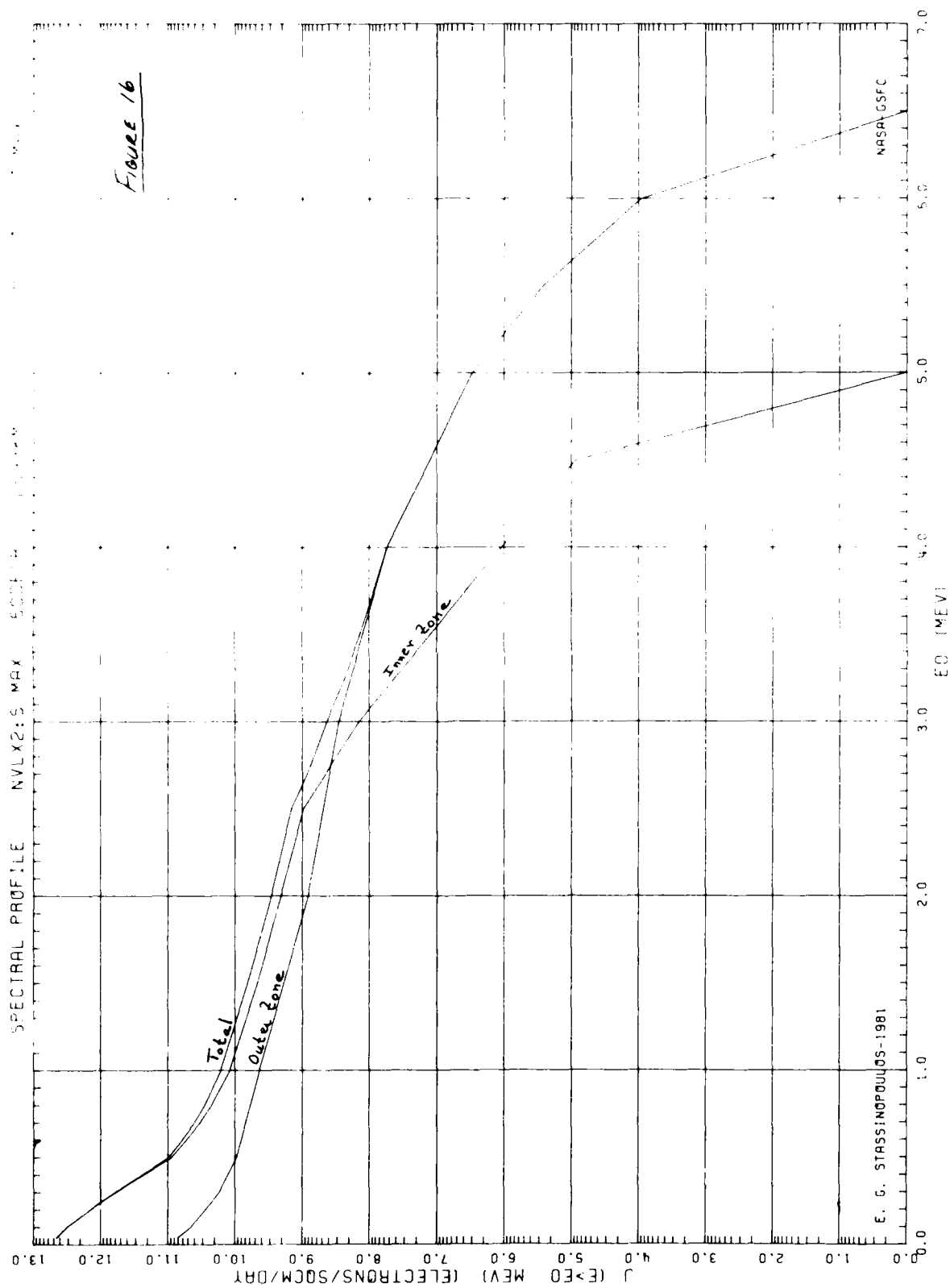
E. S. 5°05' INPAULOS-1981

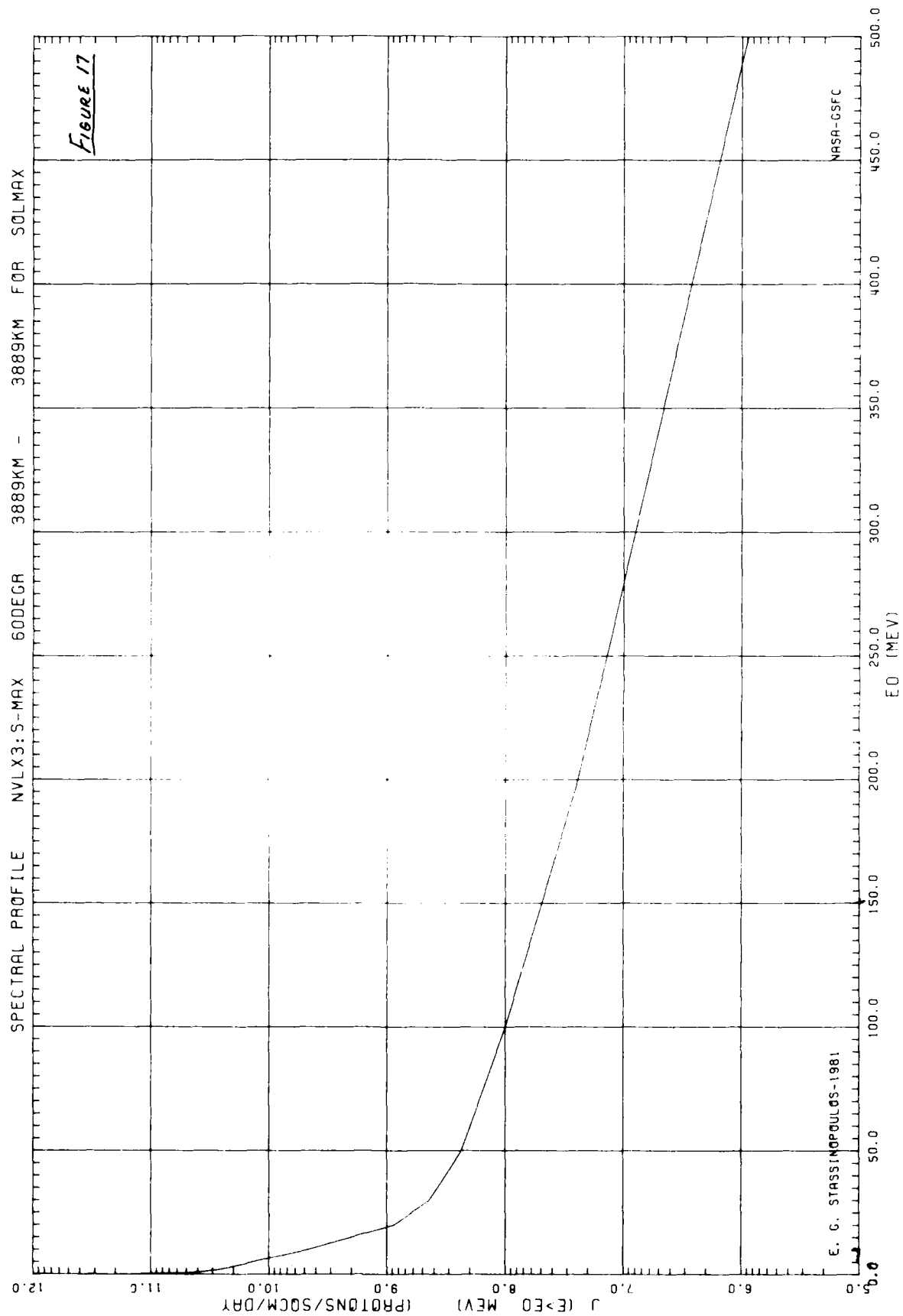
NASA-OSFC

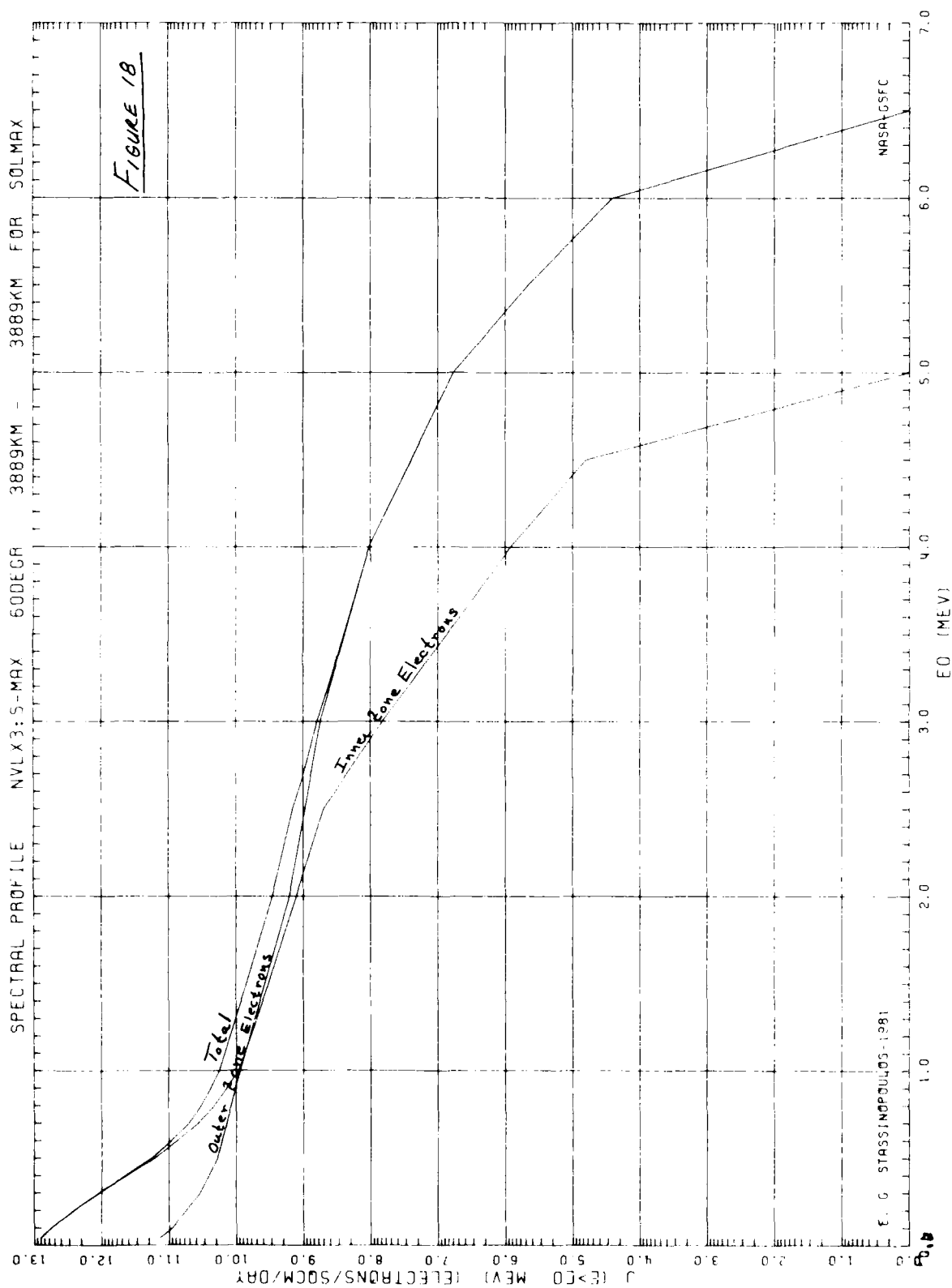












AD-A141 849

ORBITAL RADIATION STUDY FOR INCLINED CIRCULAR
TRAJECTORIES(U) NATIONAL AERONAUTICS AND SPACE
ADMINISTRATION GREENBELT MD GO... E G STASSINOPOULOS

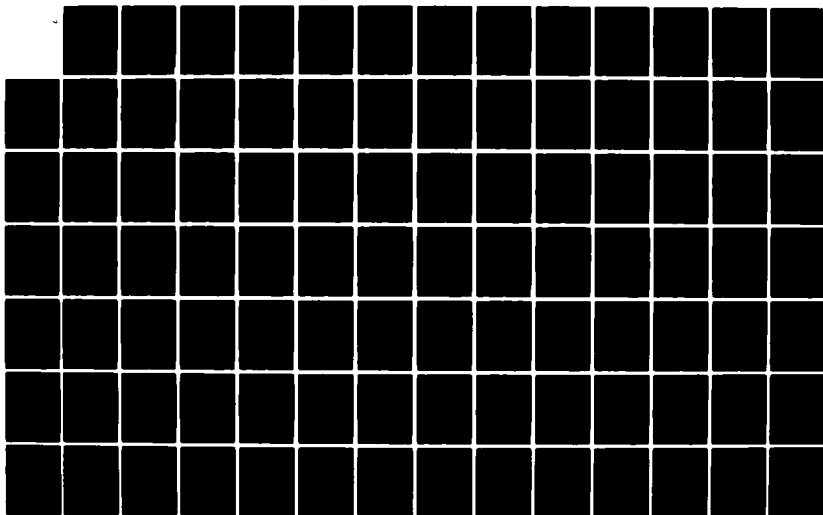
2/5

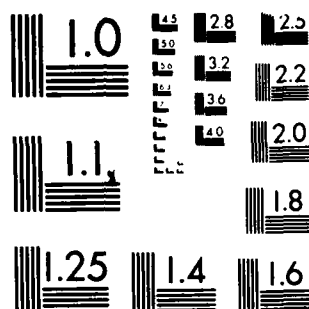
UNCLASSIFIED

NOV 81 NASA-GSFC-X-601-81-28

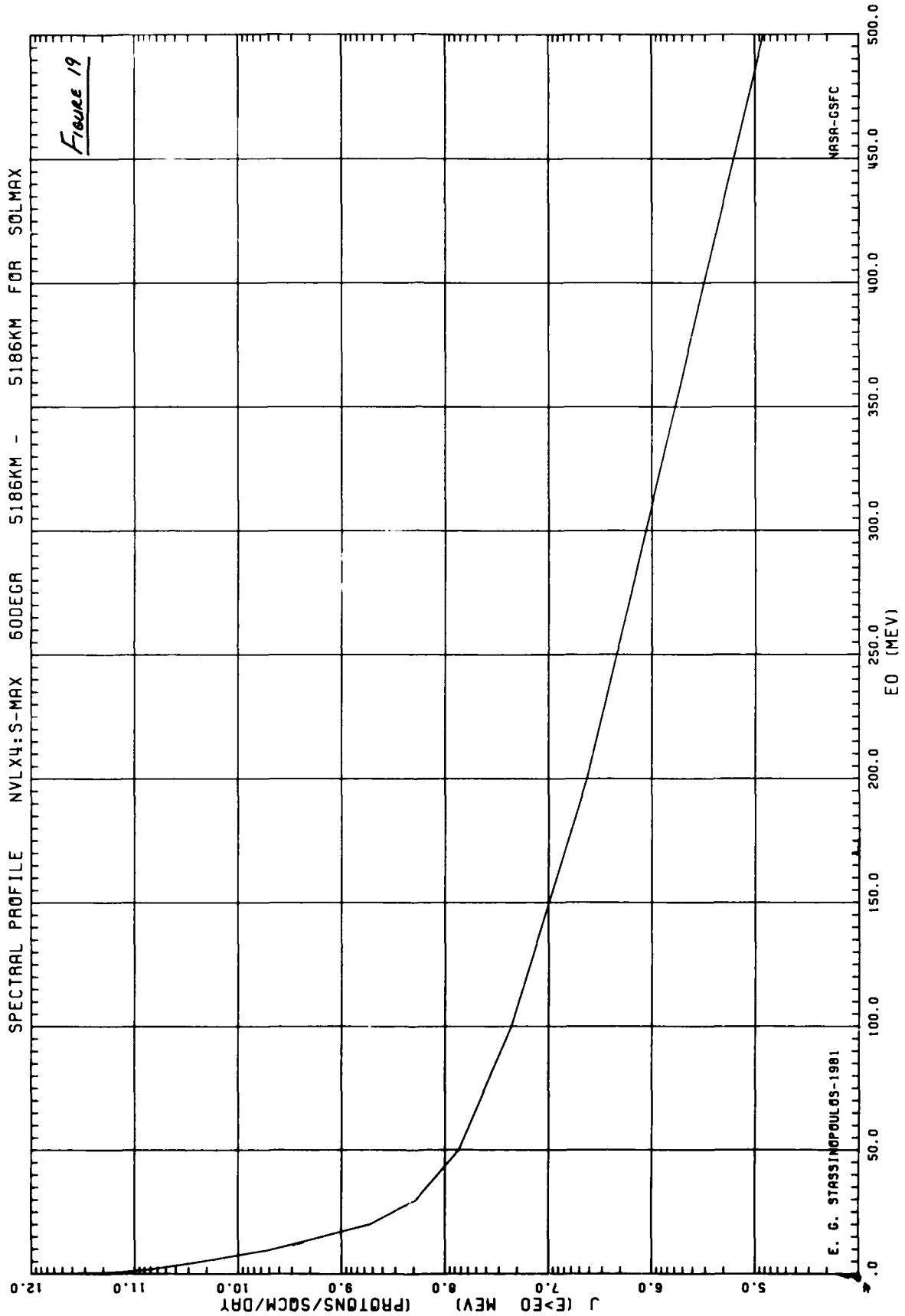
F/G 22/3

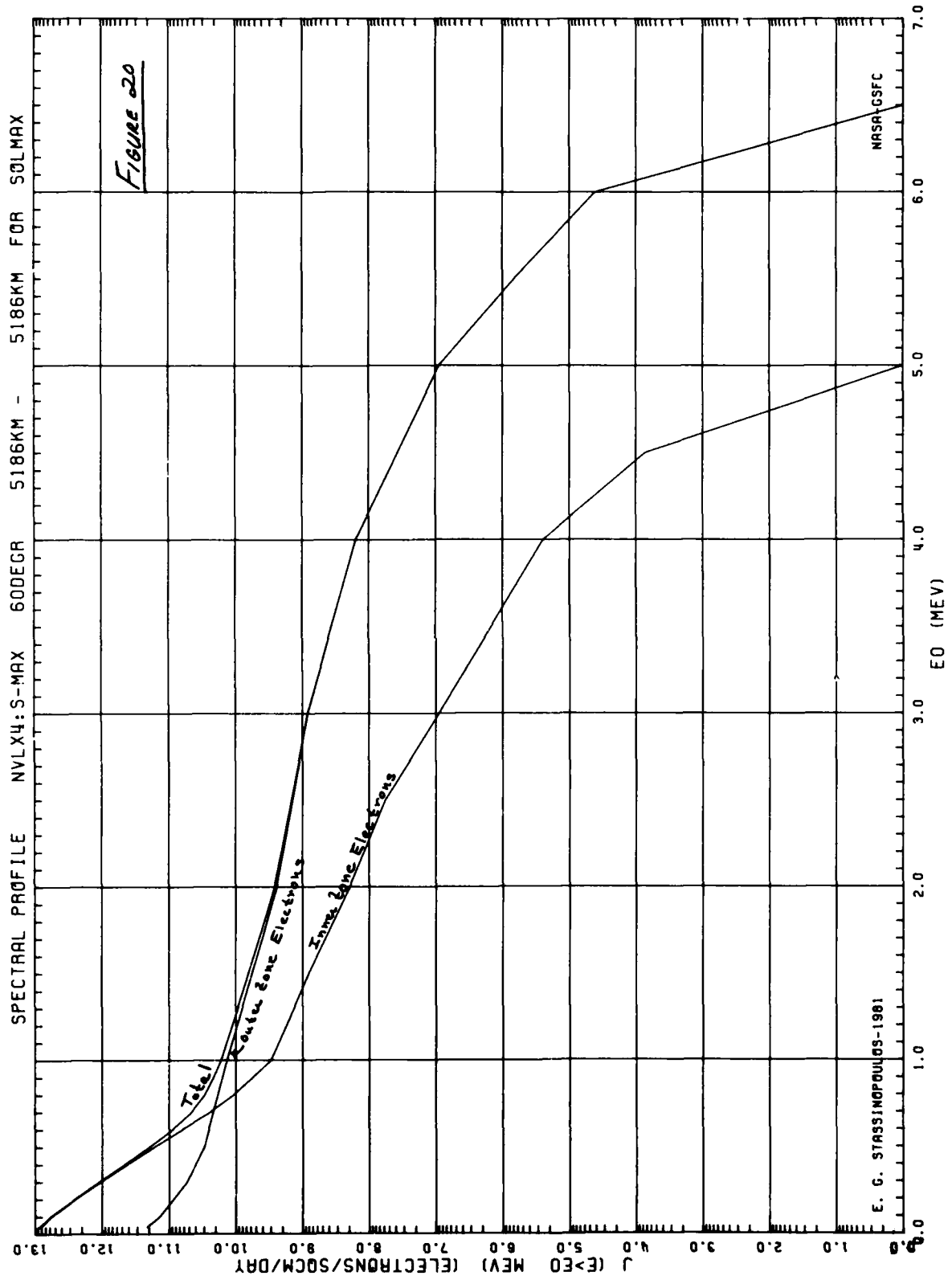
NL

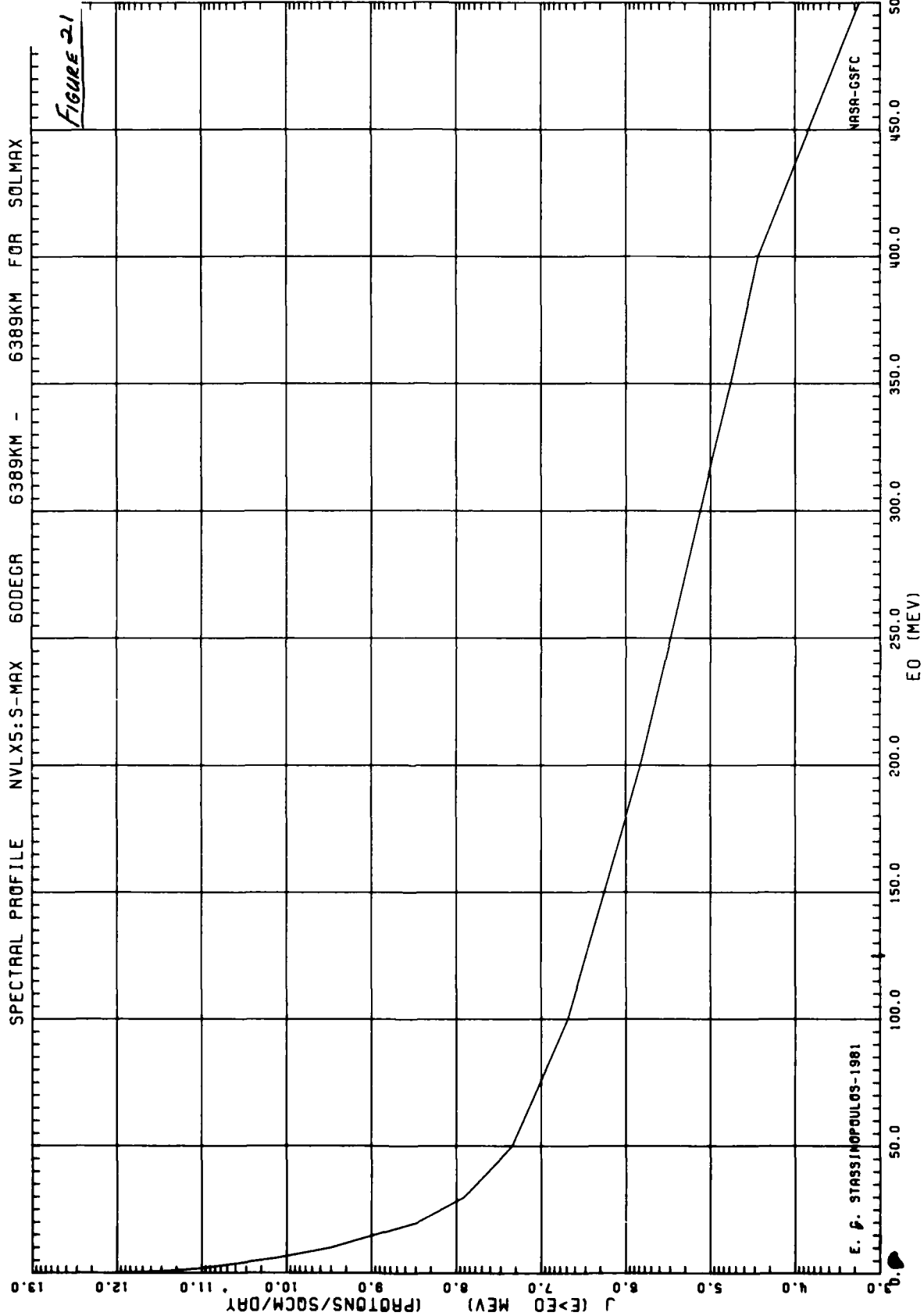


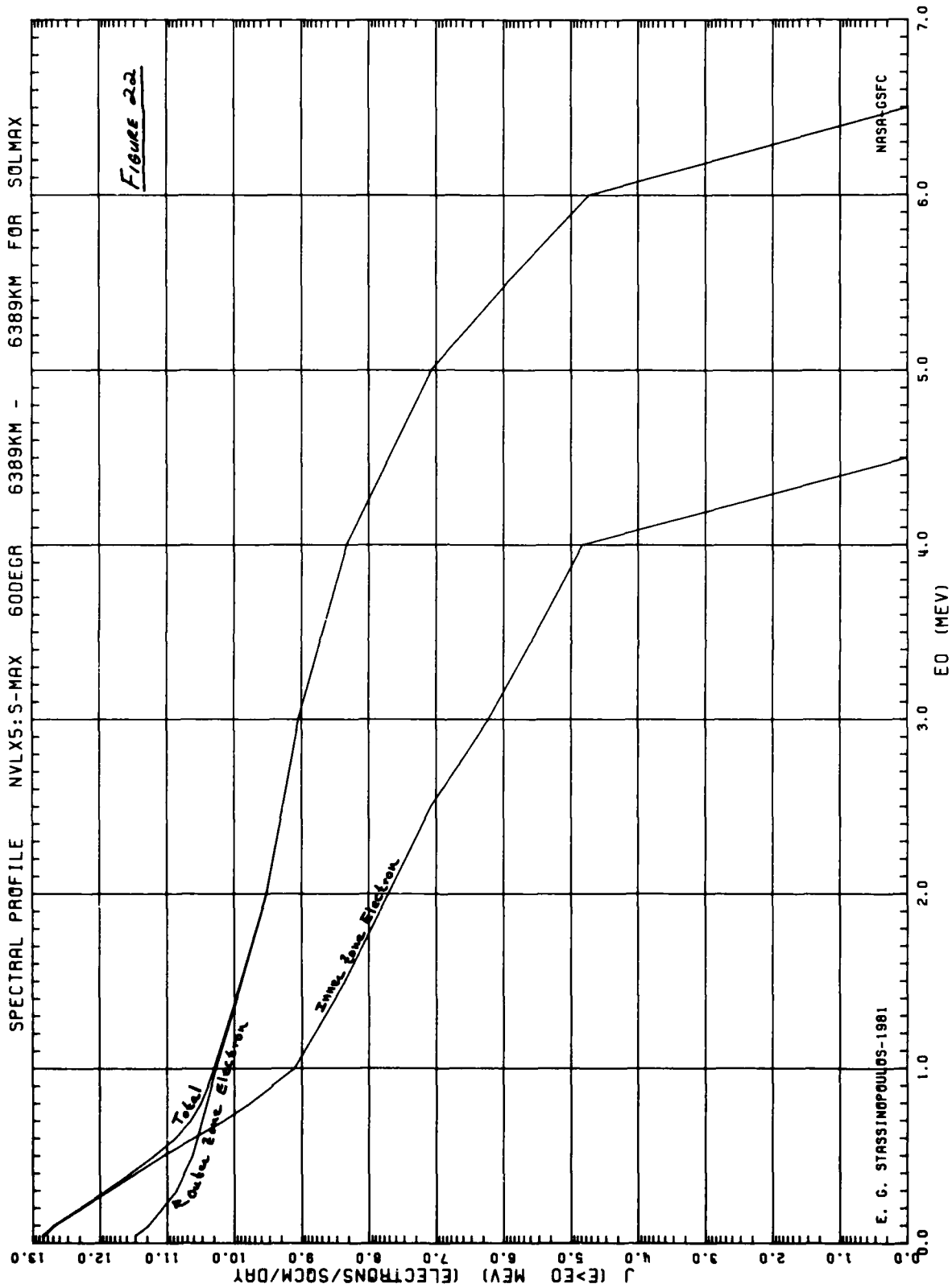


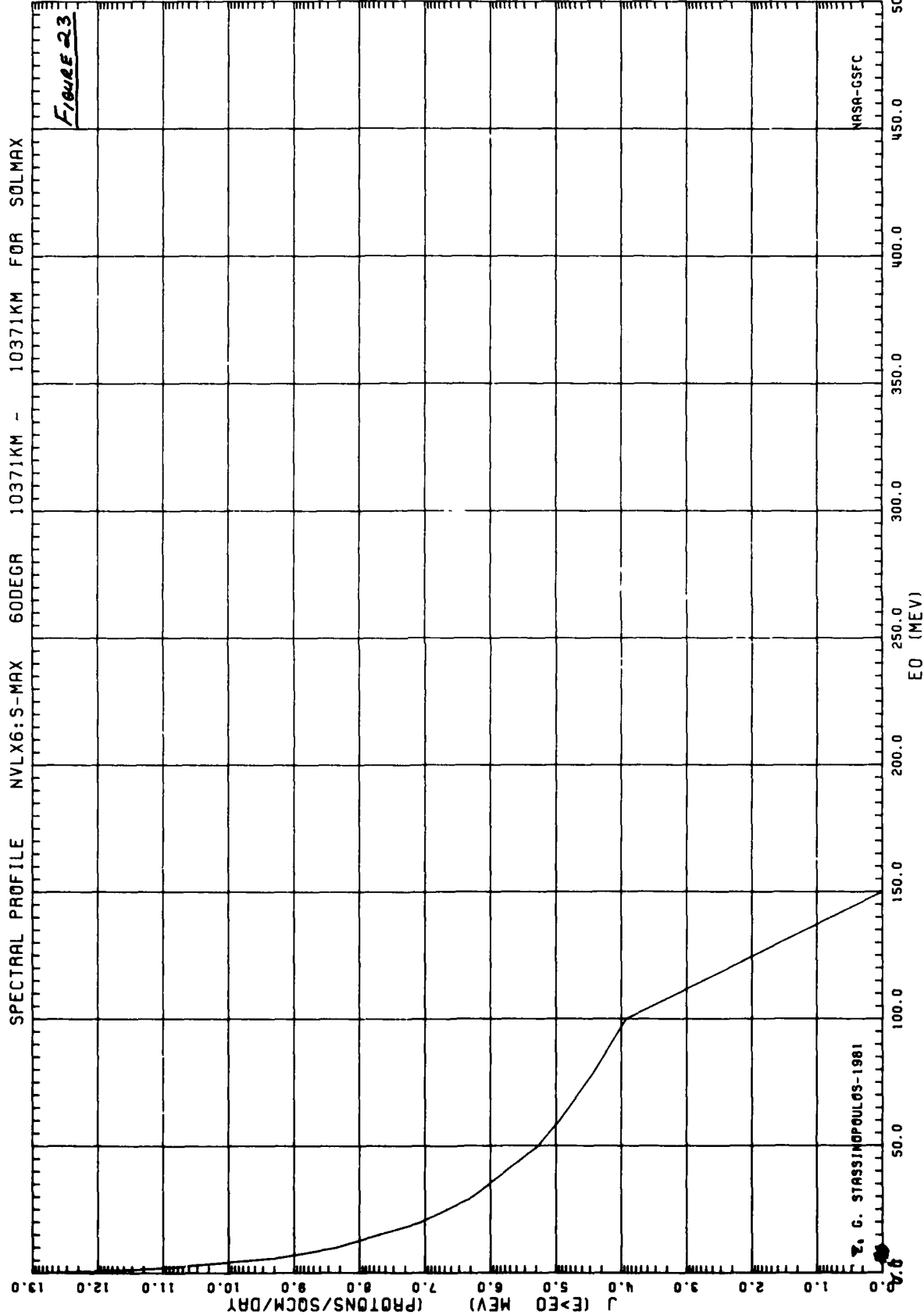
MICROCOPY RESOLUTION TEST CHART
NATIONAL BUREAU OF STANDARDS 1963-A

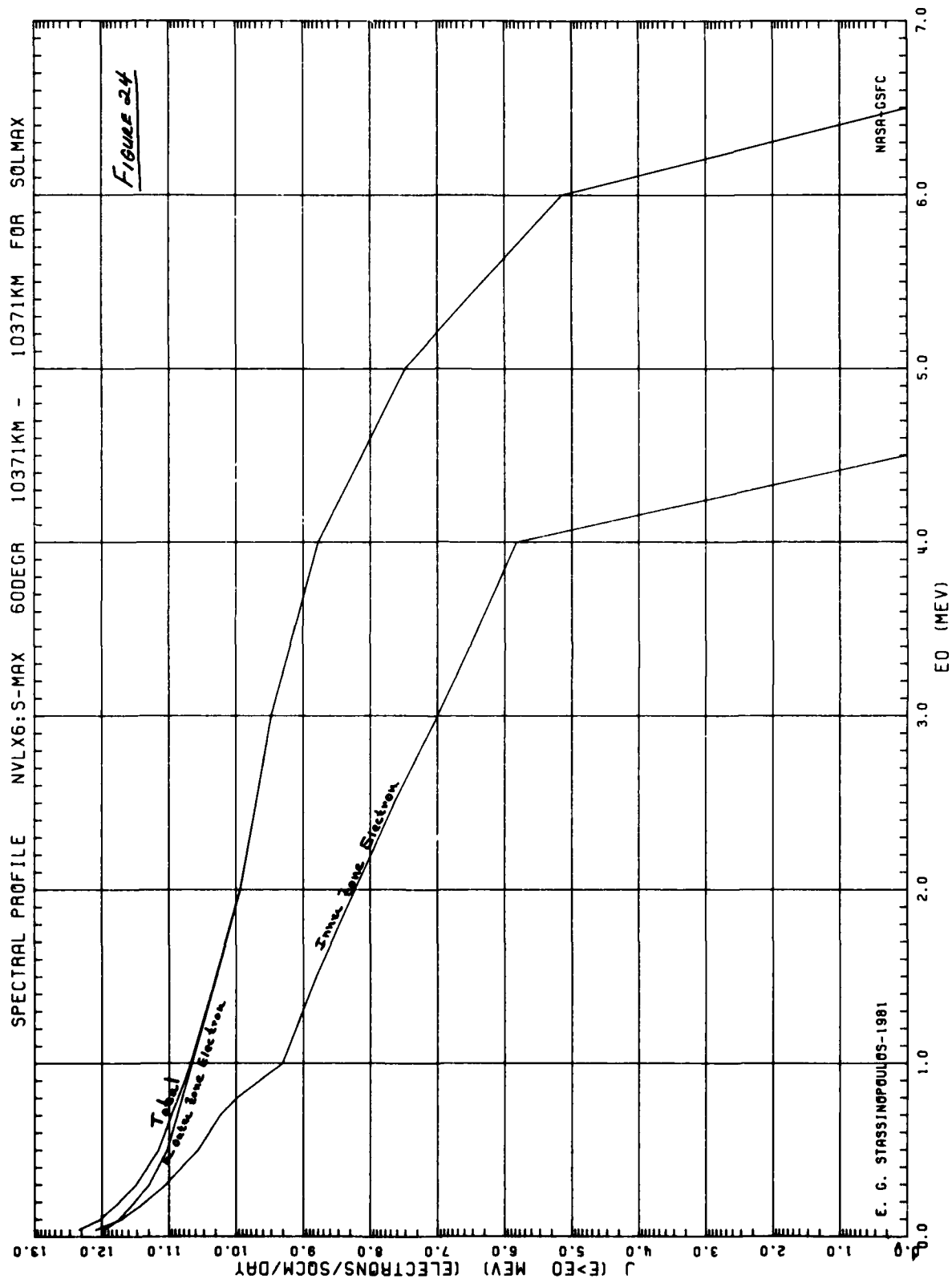


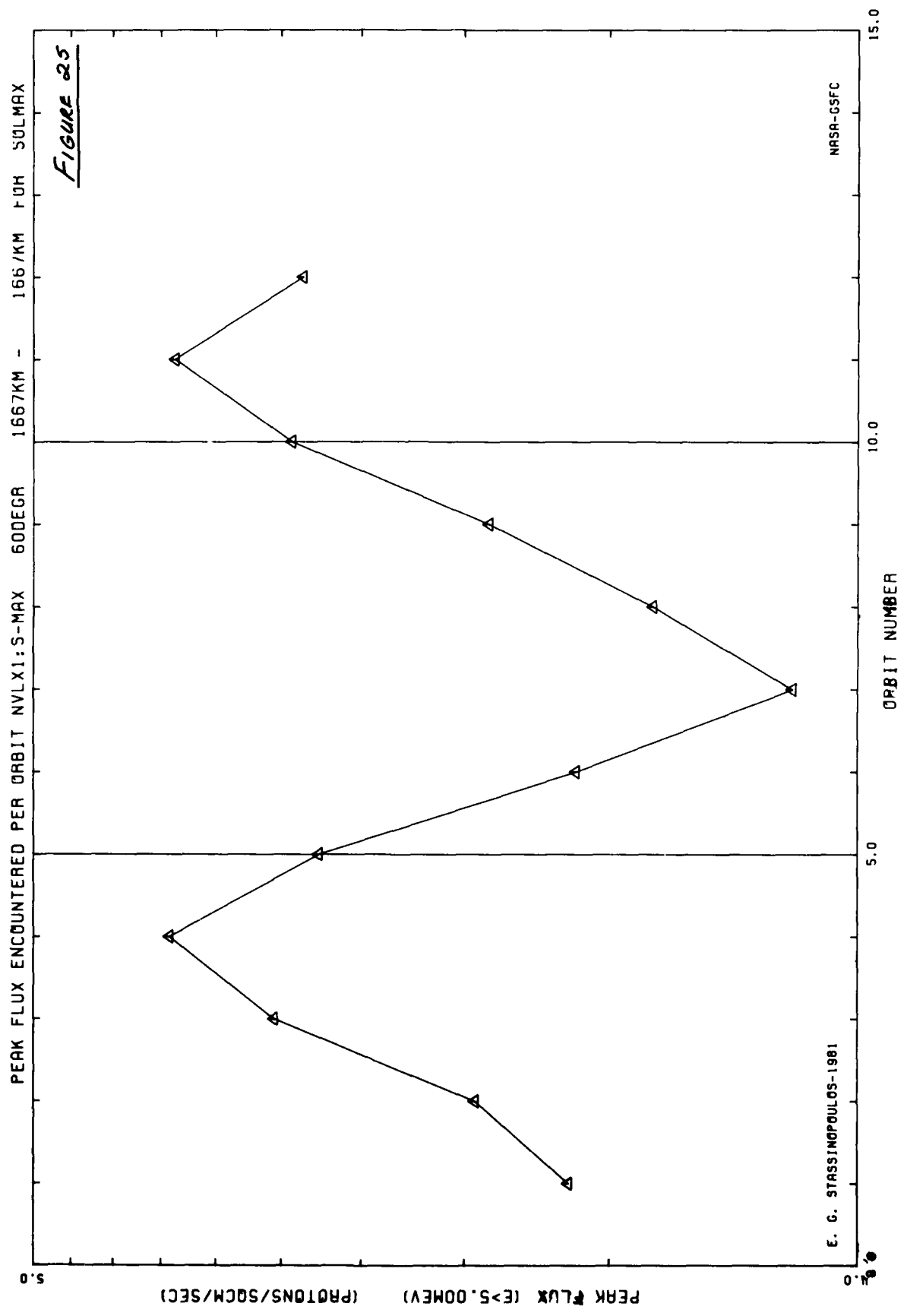


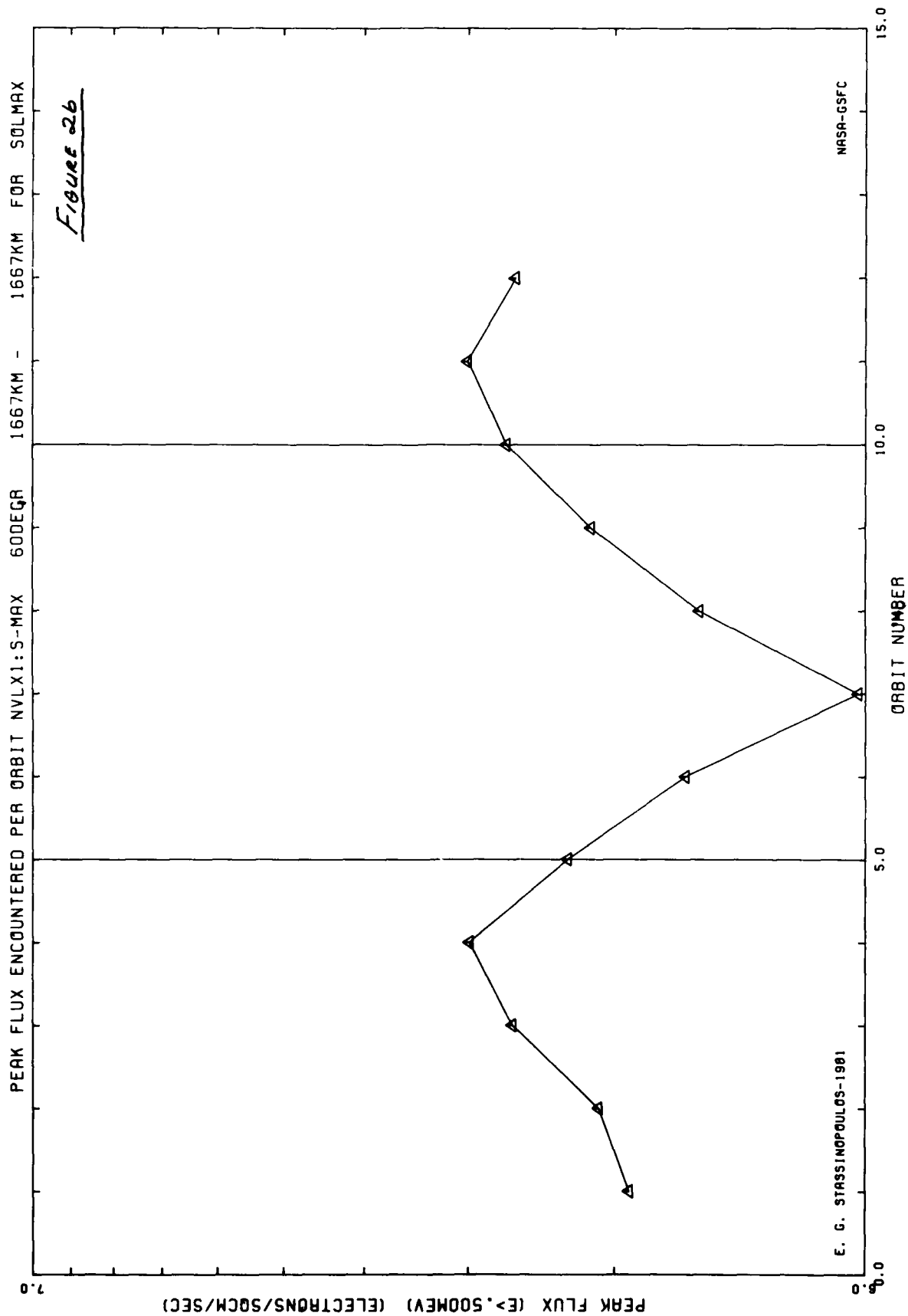


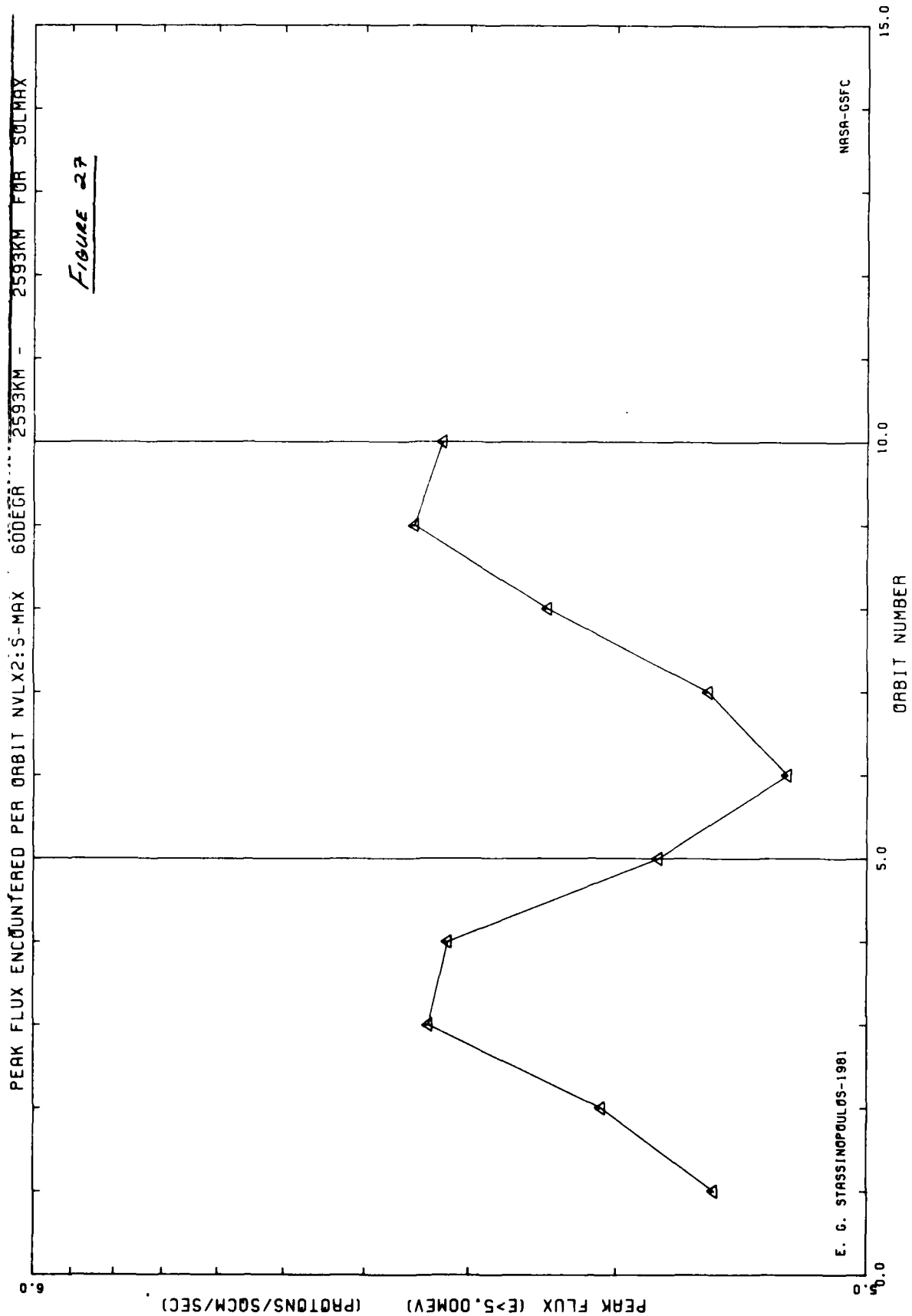


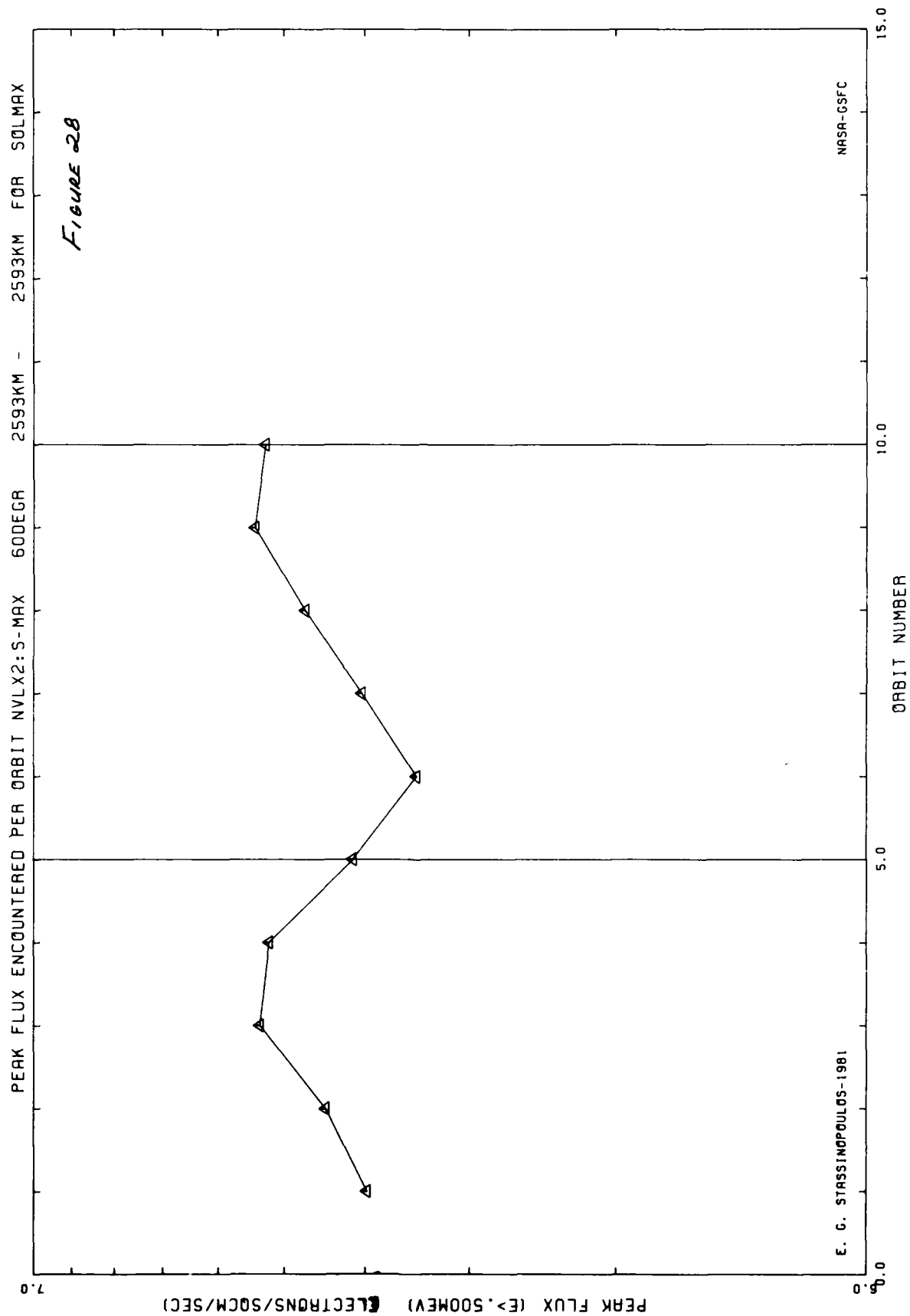


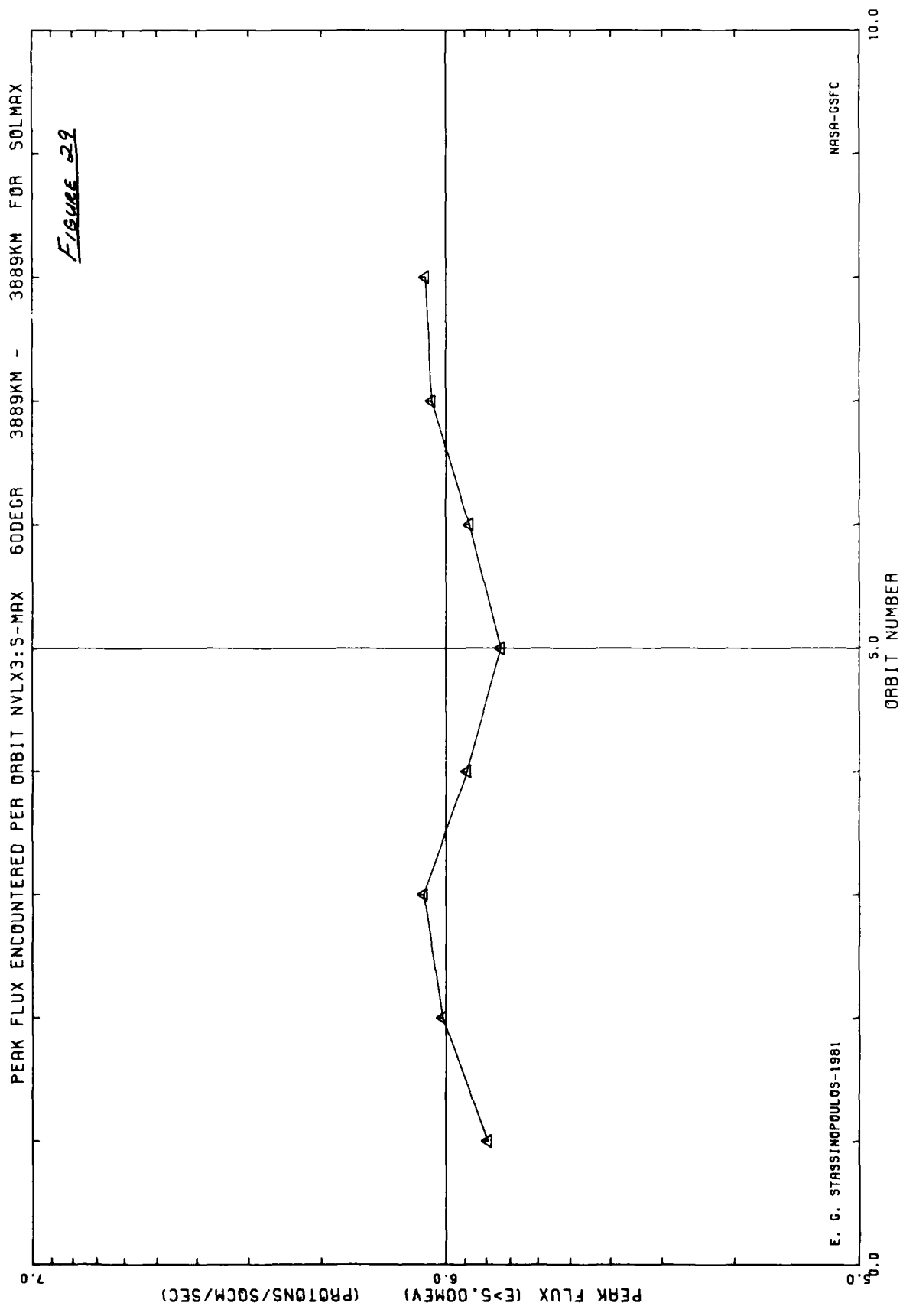


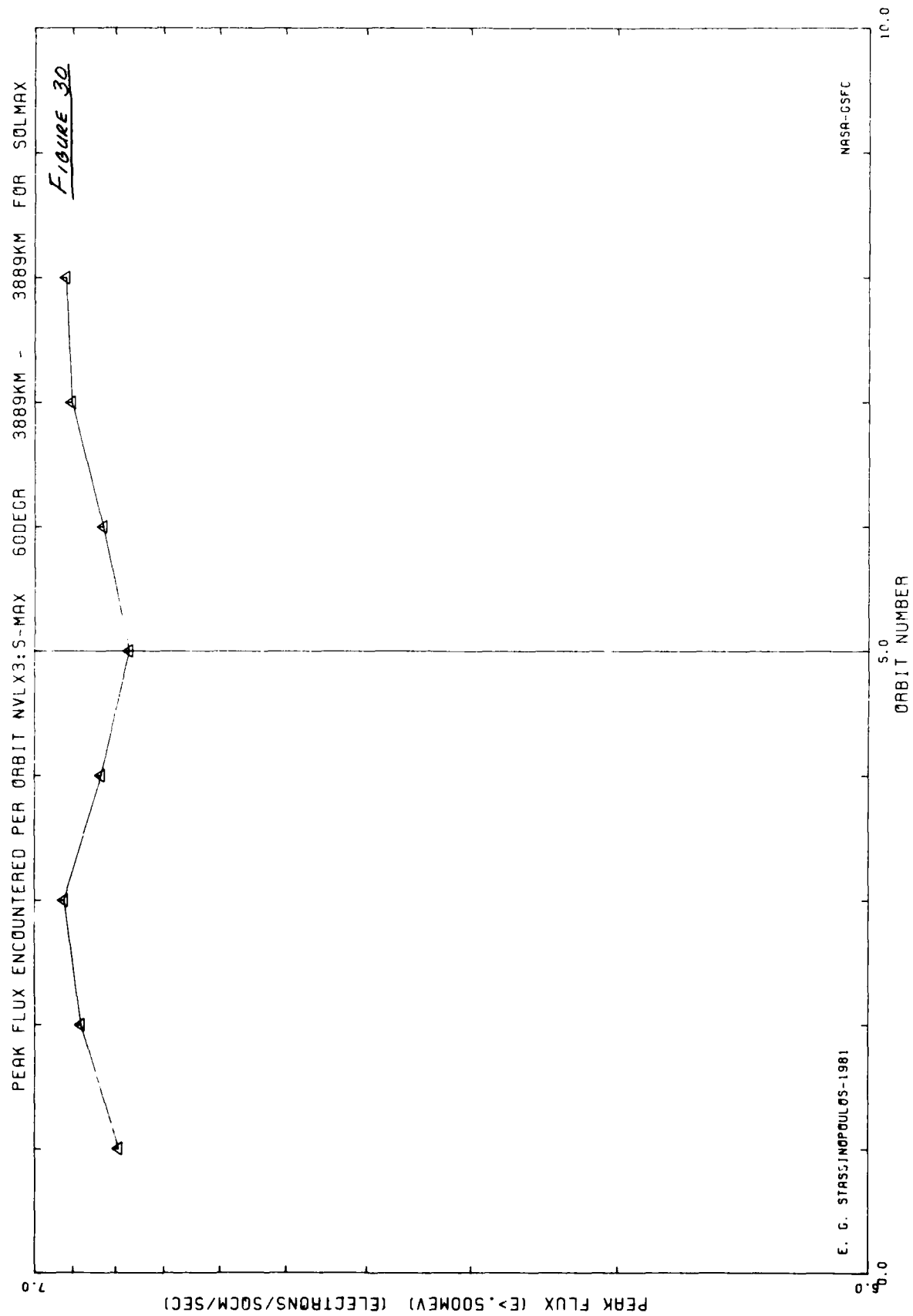


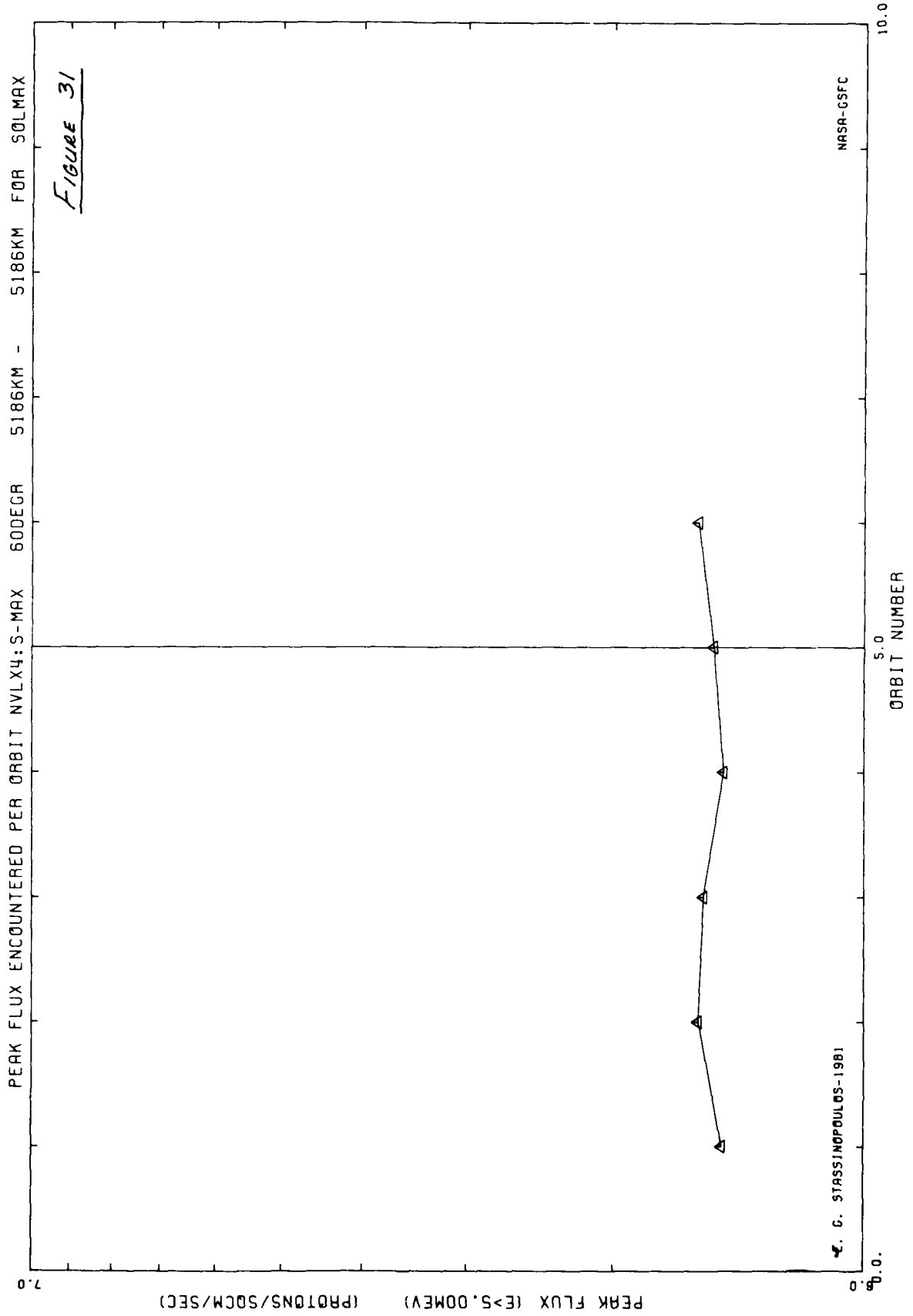












PEAK FLUX ENCOUNTERED PER ORBIT NVLX4: S-MAX 60DEGR 5186KM - 5186KM FOR SOLMAX

Figure 32

PEAK FLUX ($E > 500\text{MEV}$) (ELECTRONS/SCCM/SEC)

E. G. STASSINOPoulos-1981

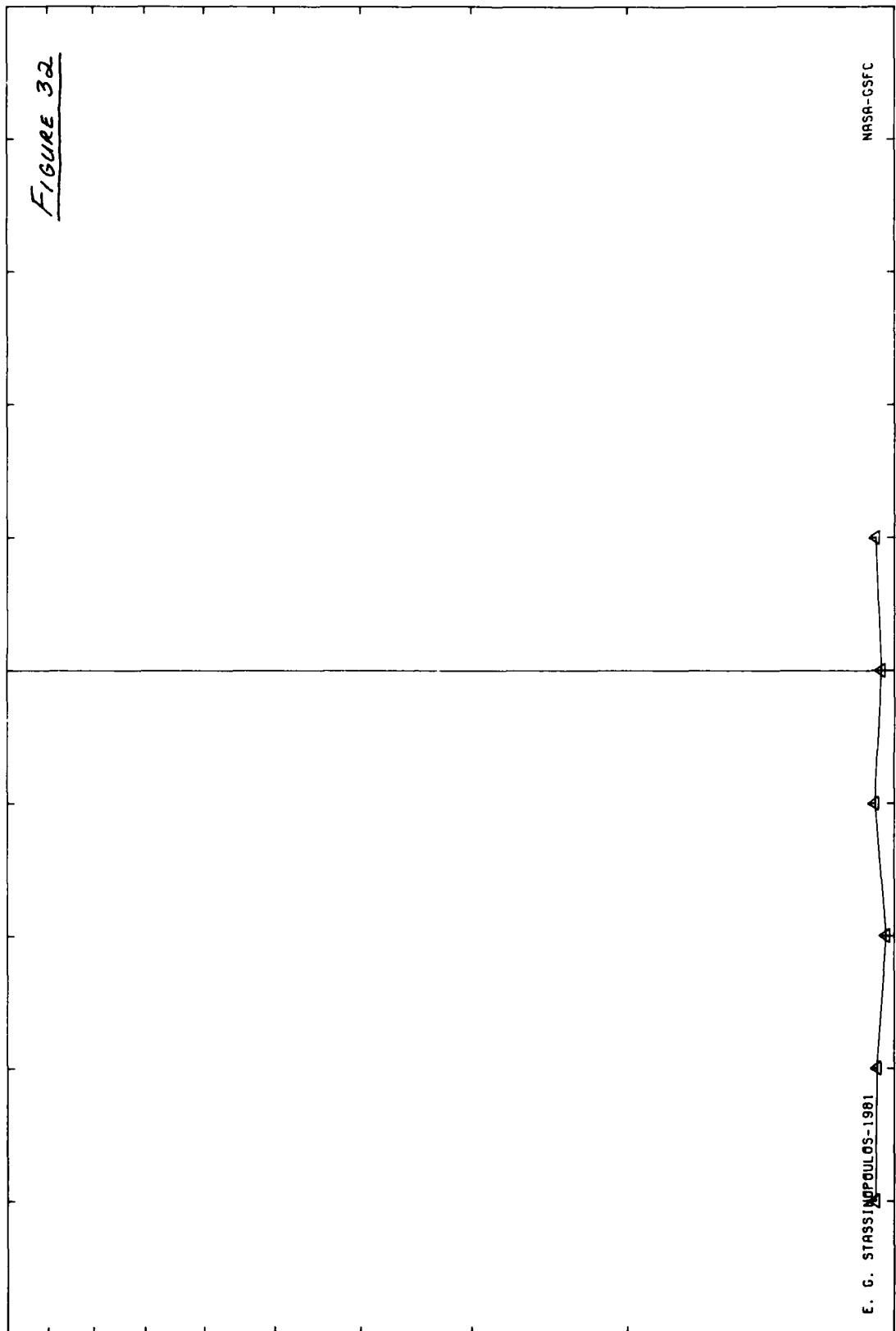
NASA-OSFC

10.0

5.0

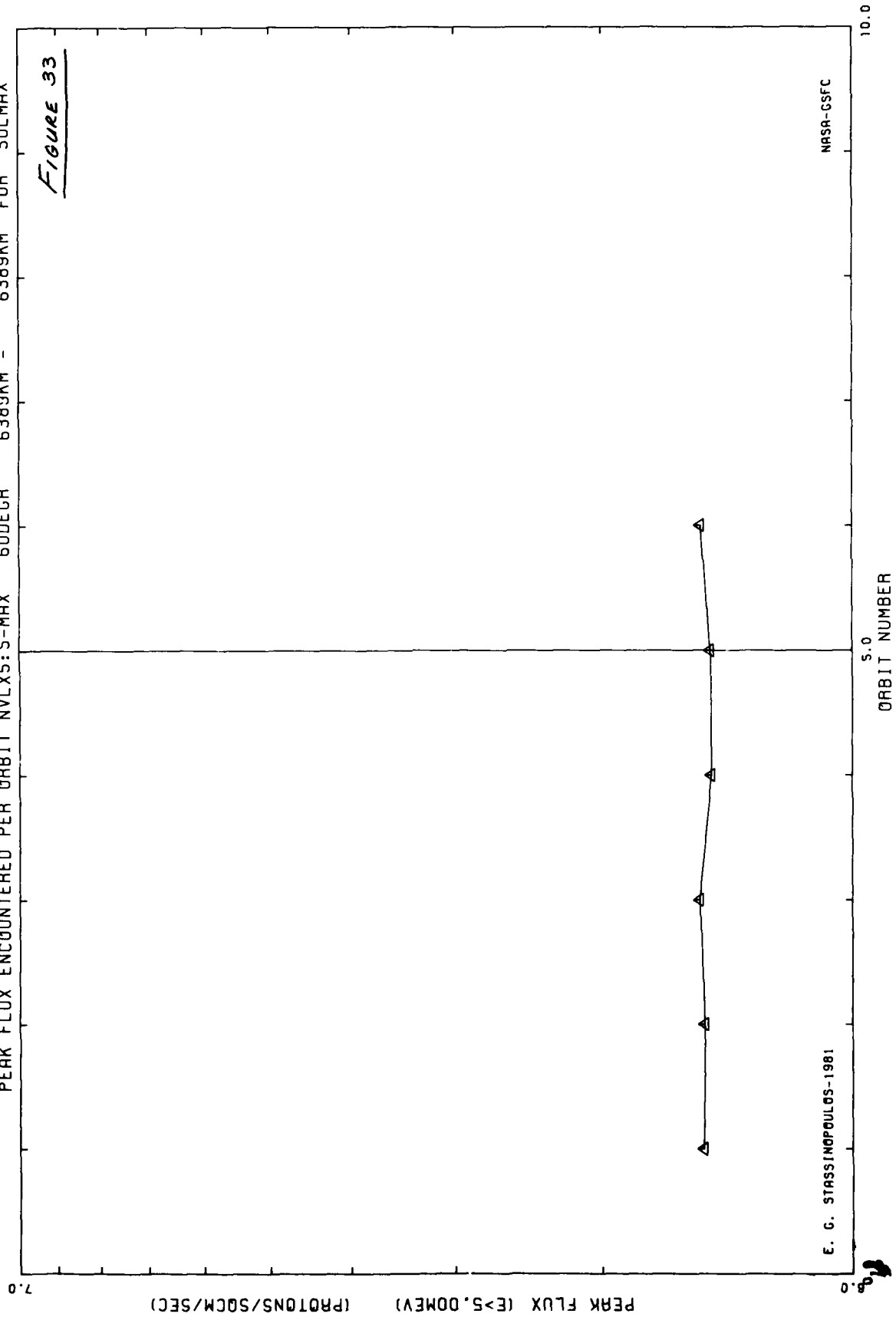
ORBIT NUMBER

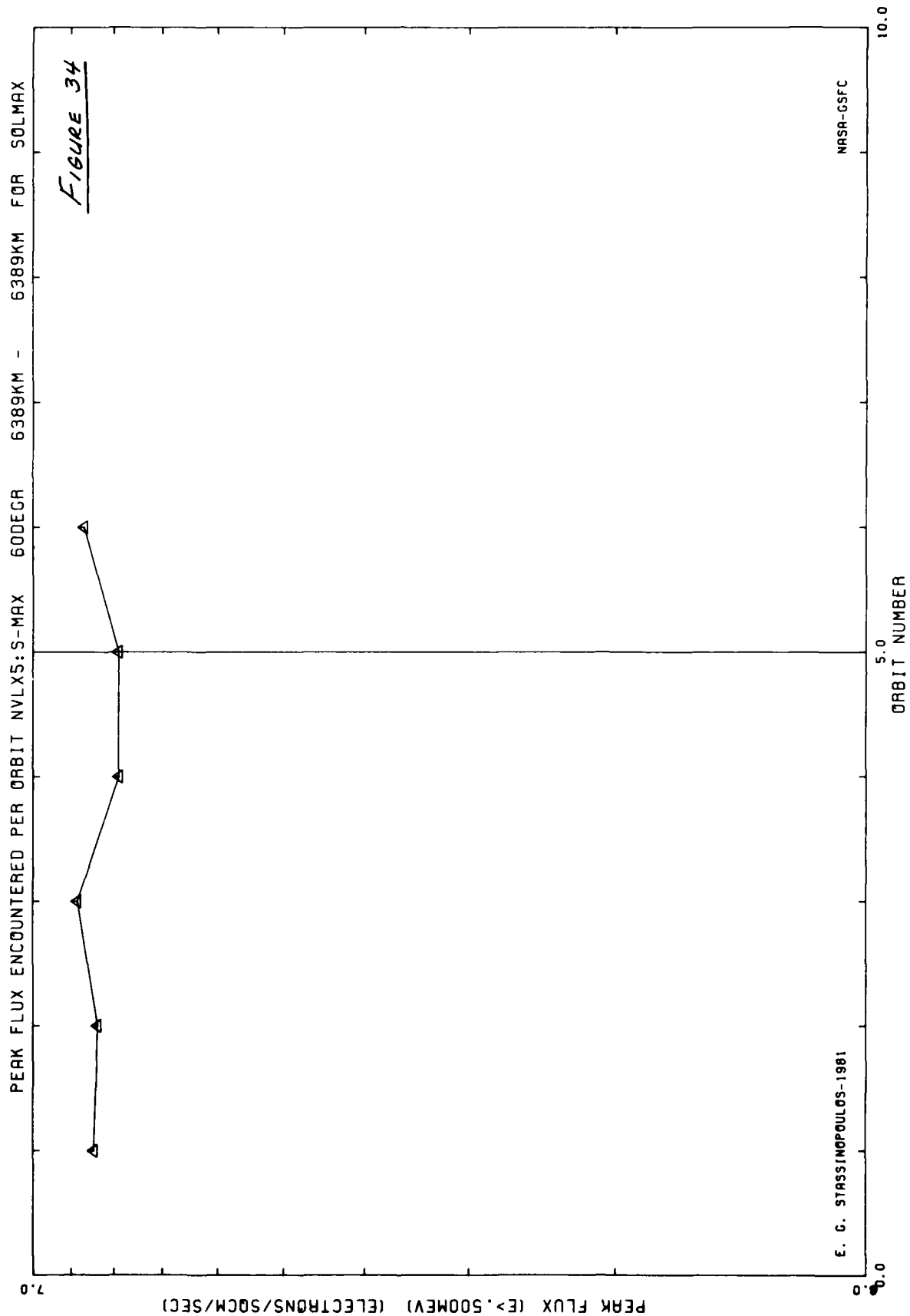
0.0



PEAK FLUX ENCOUNTERED PER ORBIT NVLX5:S-MAX 60DEGR 6389KM - 6389KM FOR SOLMAX

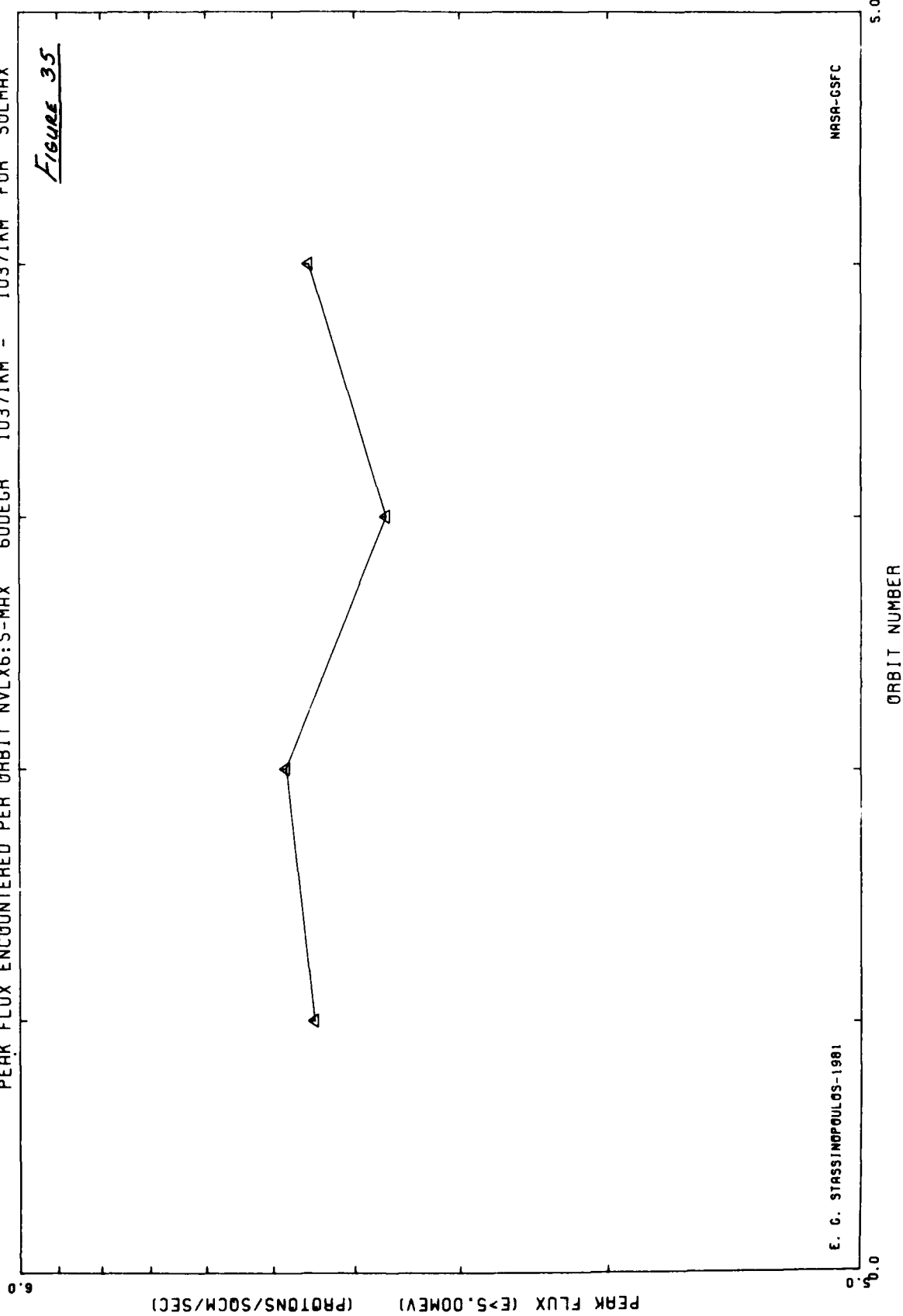
FIGURE 33





PEAK FLUX ENCOUNTERED PER ORBIT NVLX6:S-MAX 60DEGR 10371KM - 10371KM FOR SOLMAX

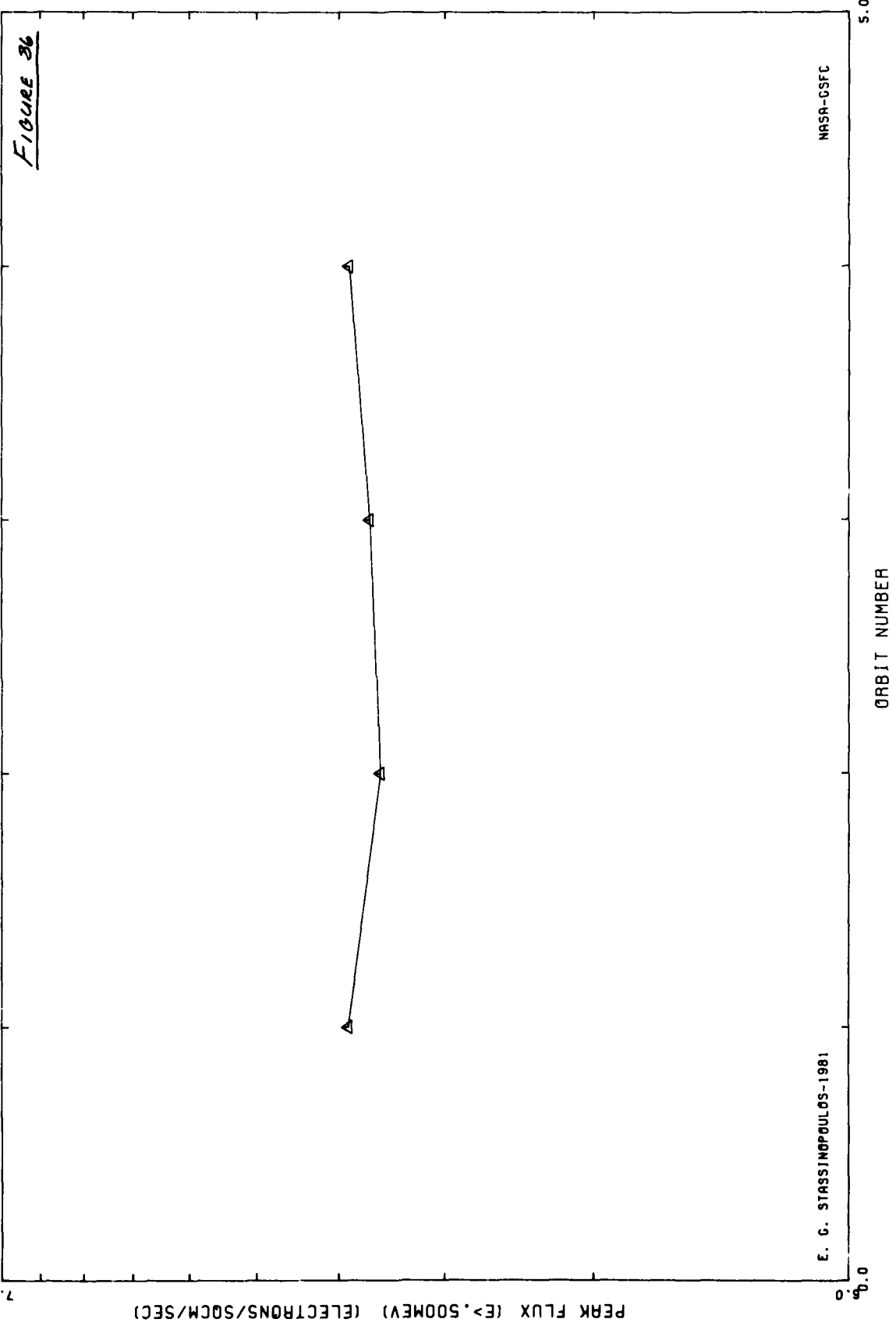
Figure 35



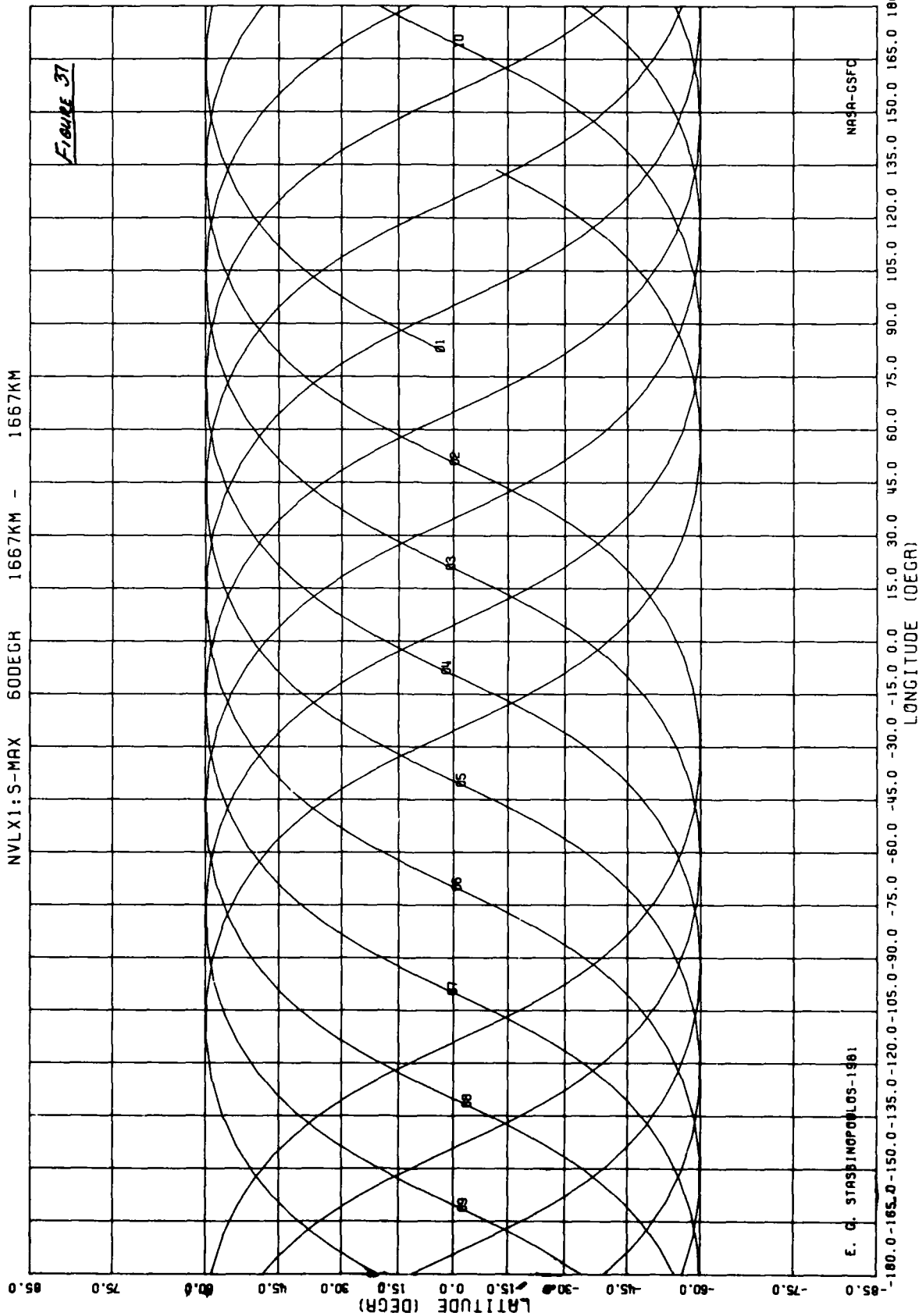
E. C. STASSINOPOULOS-1981

NASA-GSFC

PEAK FLUX ENCOUNTERED PER ORBIT NVLX6:S-MAX 60DEGR 10371KM - 10371KM FOR SOLMAX

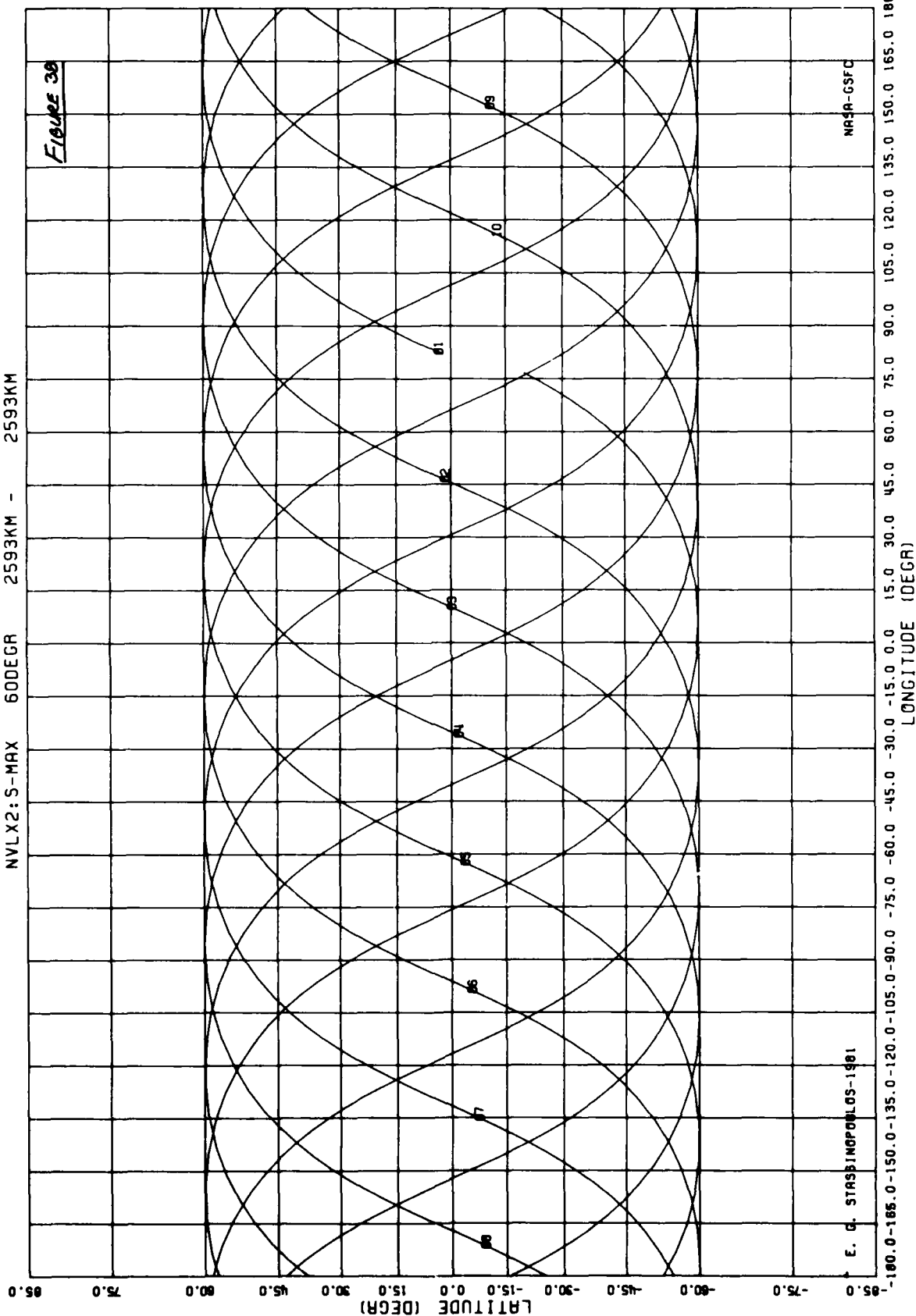


NVLX1:S-MAX 60DEGH 1667KM - 1667KM



E. C. STASINOPOLIS-1981

NVLX2:S-MAX 60DEGR 2593KM - 2593KM

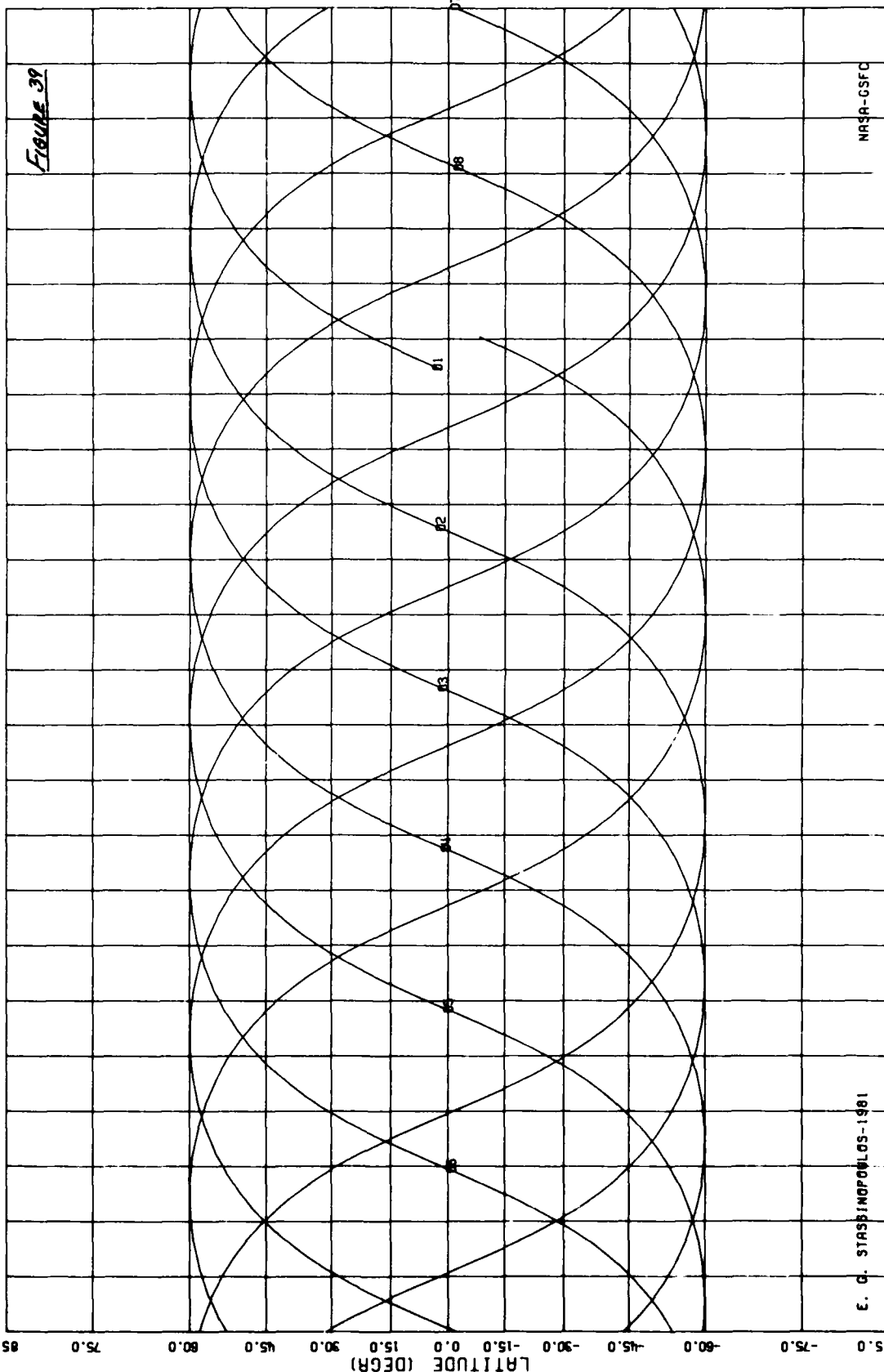


NVLX3: S-MAX 3889KM - 3889KM

60DEGR

3889KM

85.0



E. G. STASSINOPOLIS-1981

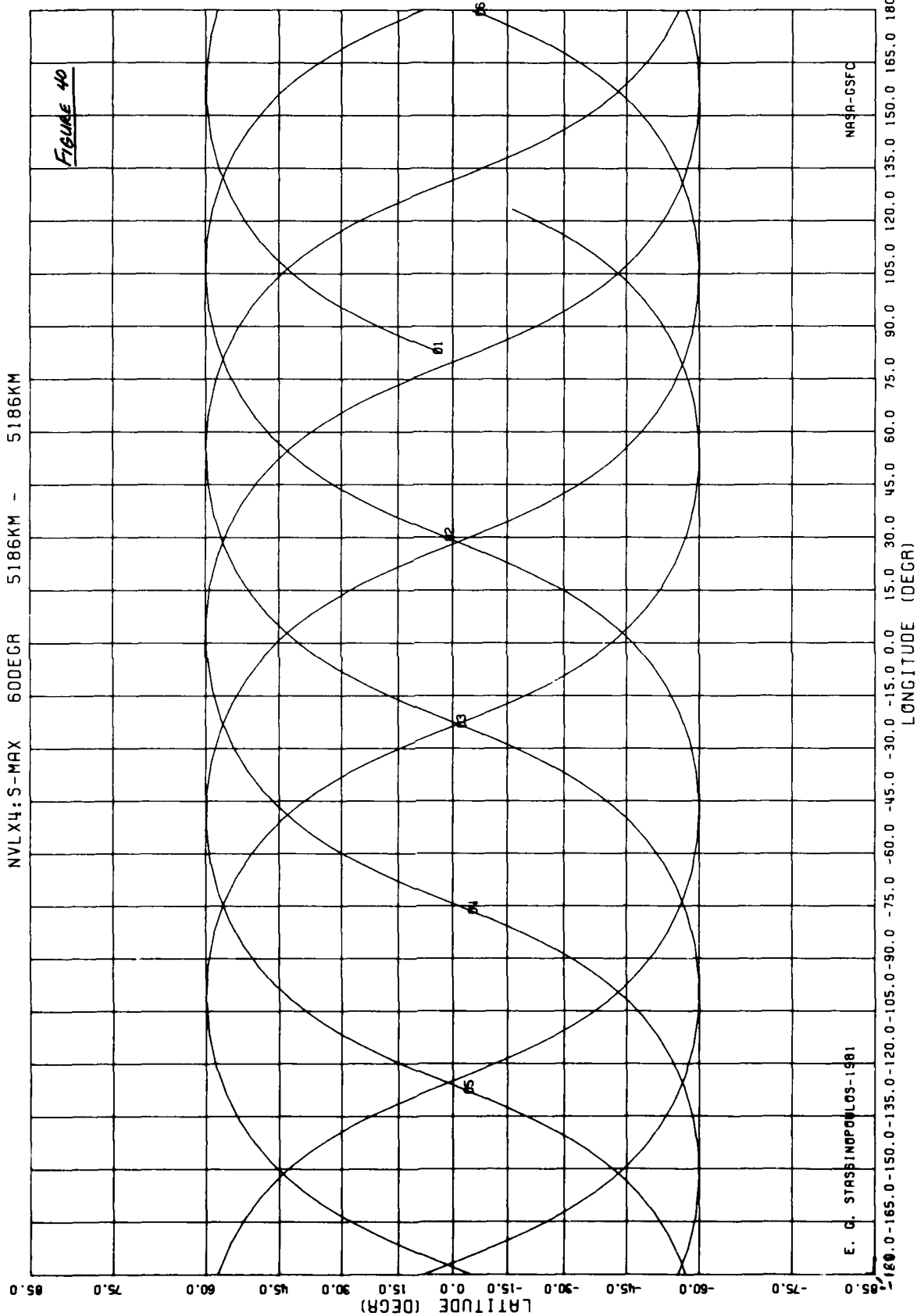
NASA-OSFC

Figure 39

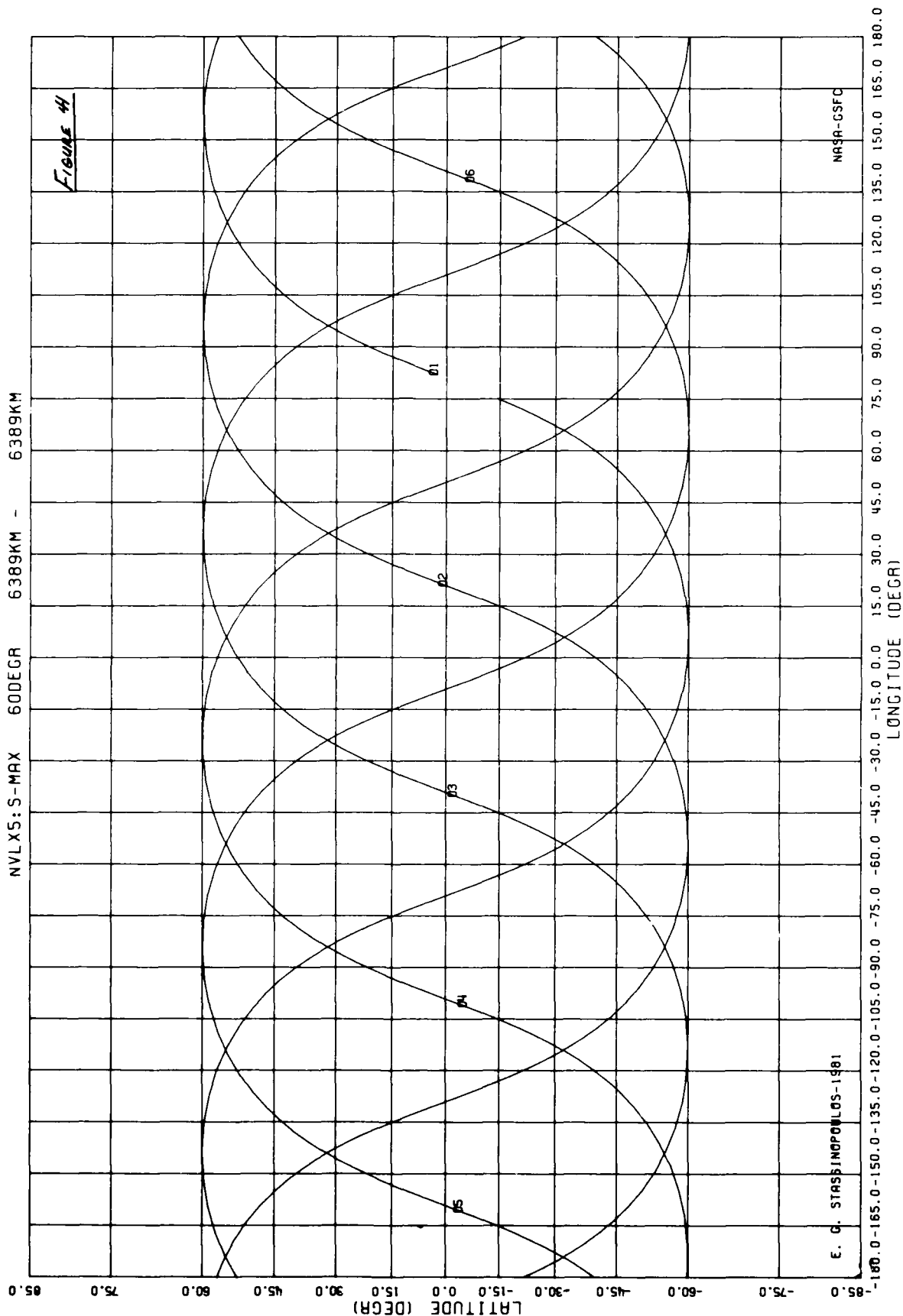
LONGITUDE (DEGR)

-180.0 -165.0 -150.0 -135.0 -120.0 -105.0 -90.0 -75.0 -60.0 -45.0 -30.0 -15.0 0.0 15.0 30.0 45.0 60.0 75.0 90.0 105.0 120.0 135.0 150.0 165.0 180.0

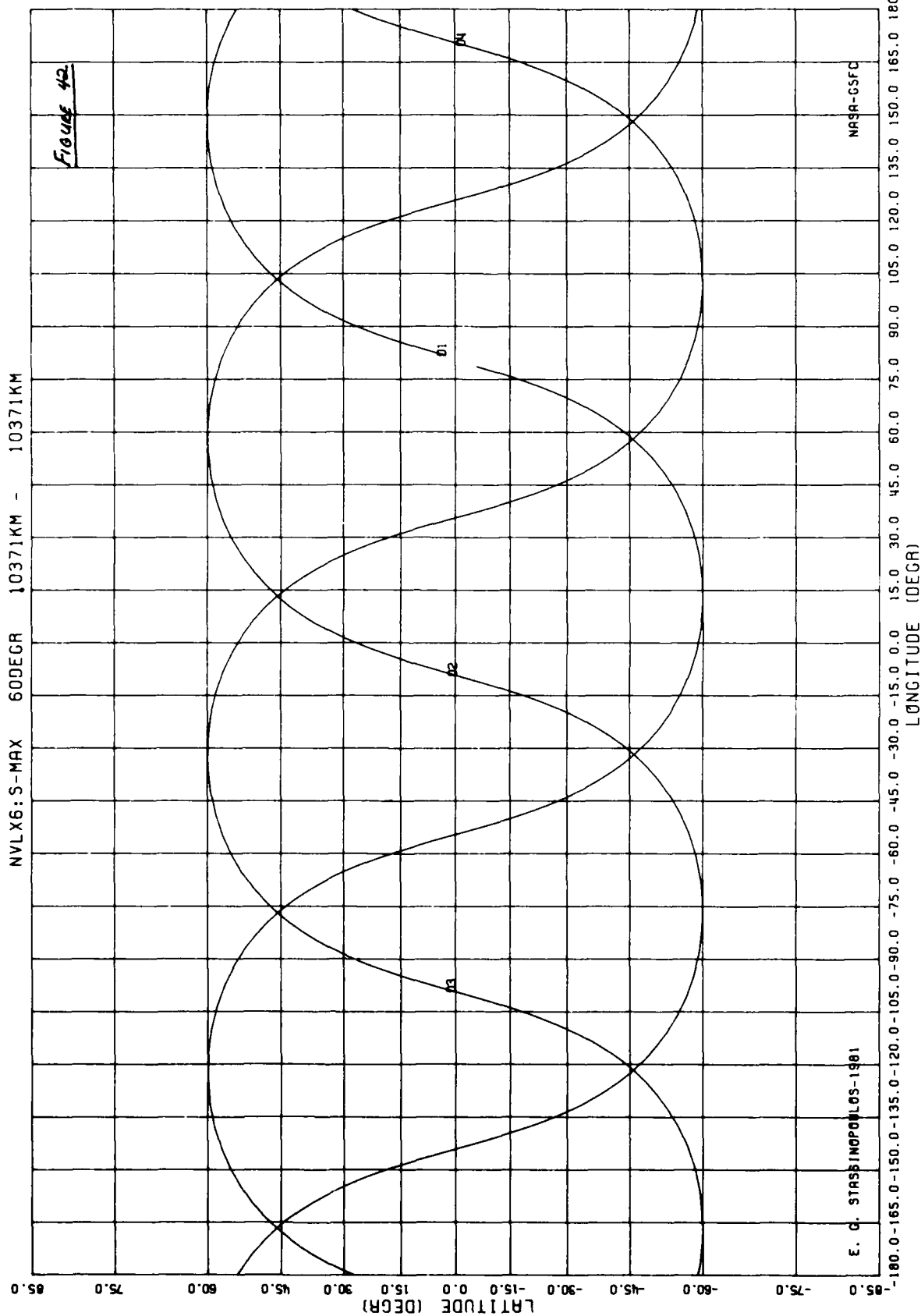
NVLX4:S-MAX 60DEGR 5186KM - 5186KM

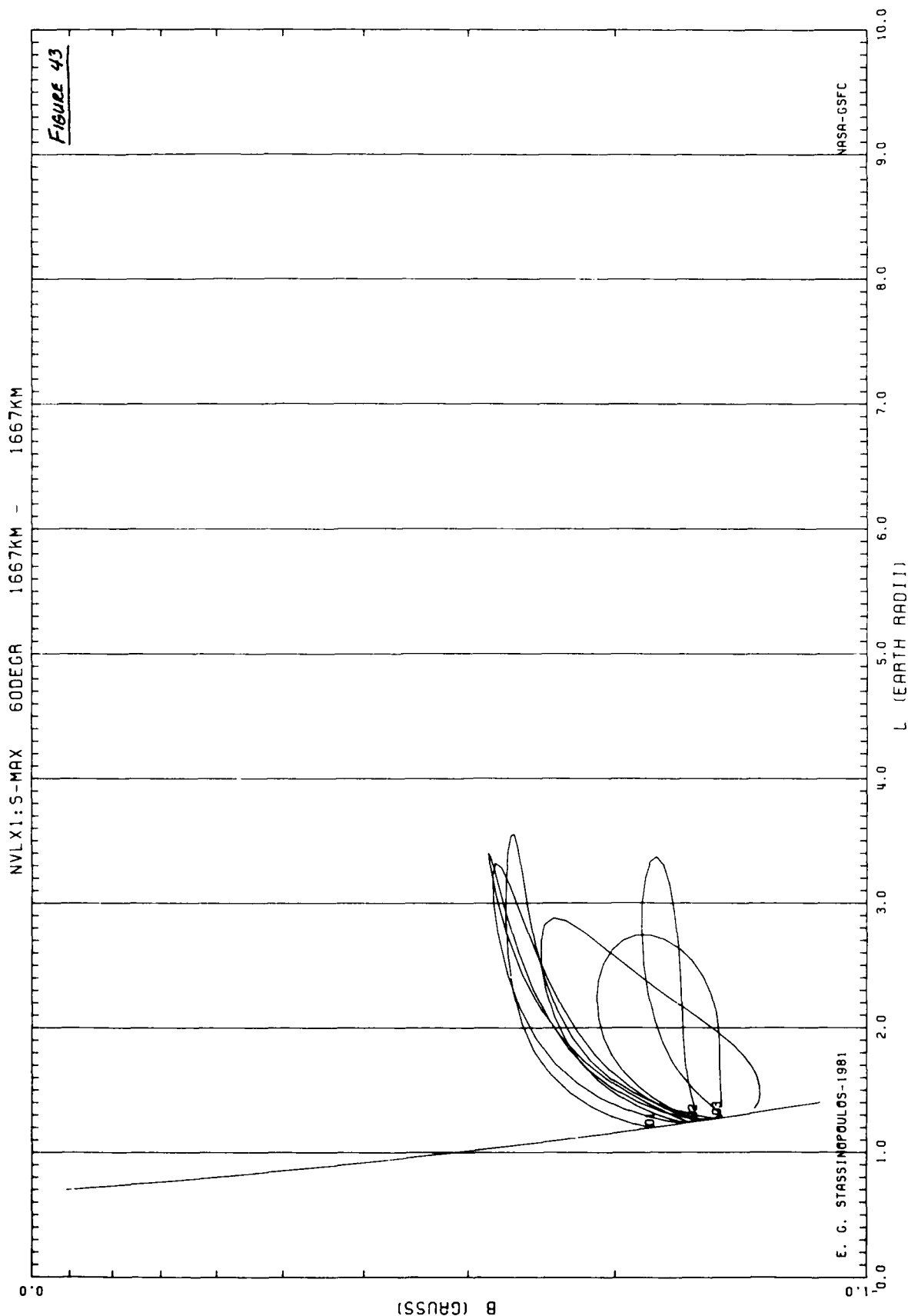


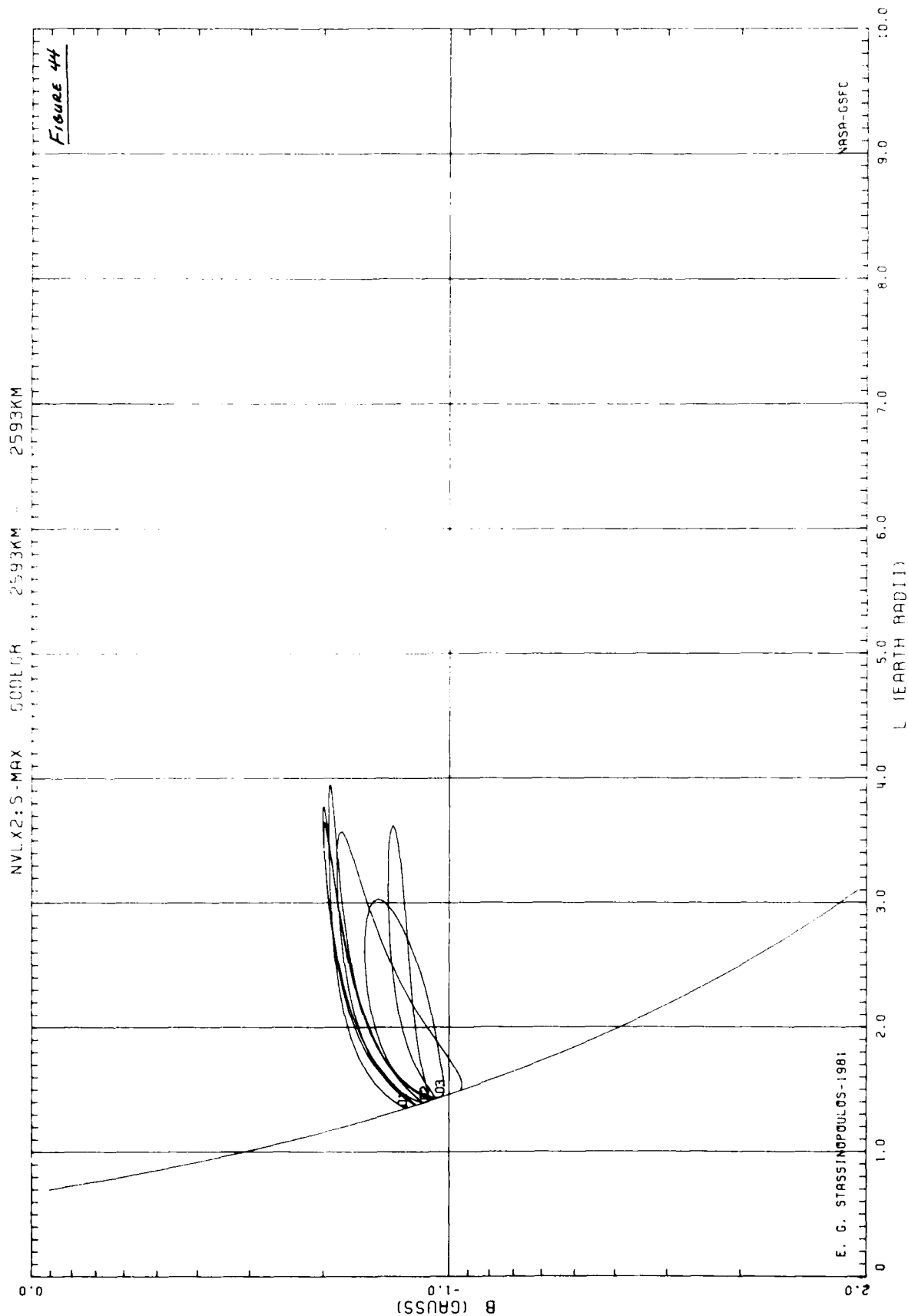
NVLX5: S-MAX 60DEGR 6389KM - 6389KM

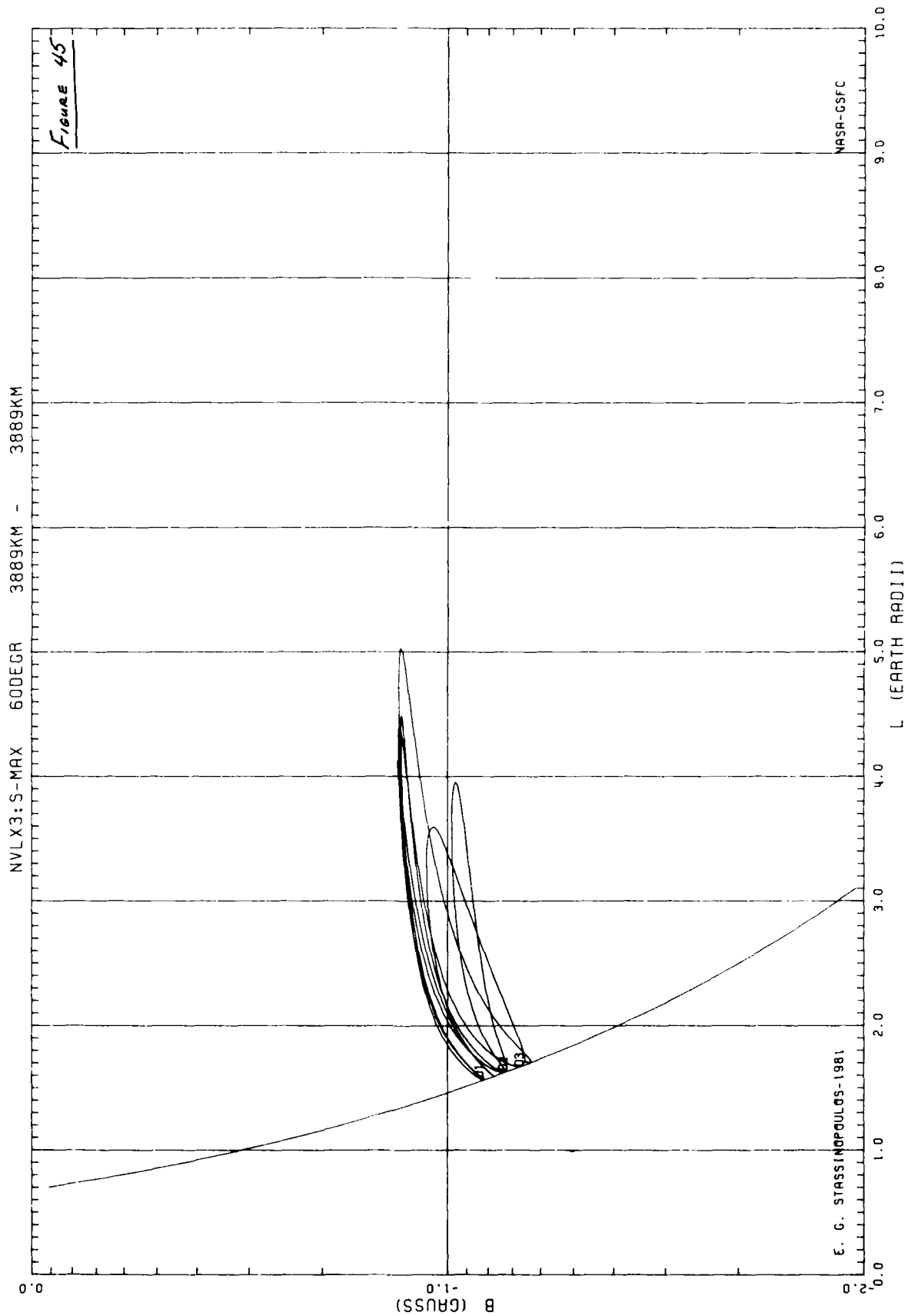


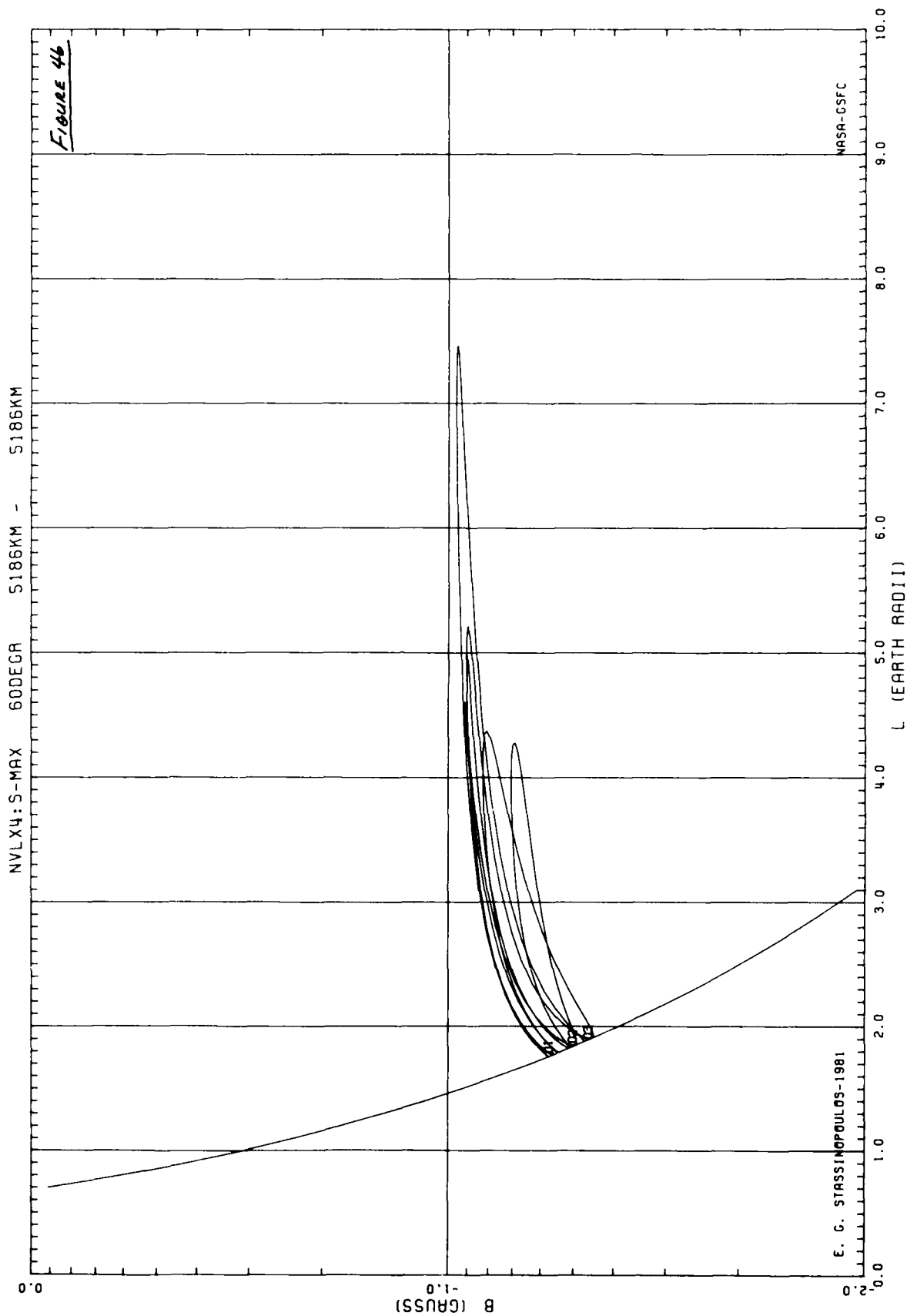
NVLX6: S-MAX 60DEGR 10371KM - 10371KM

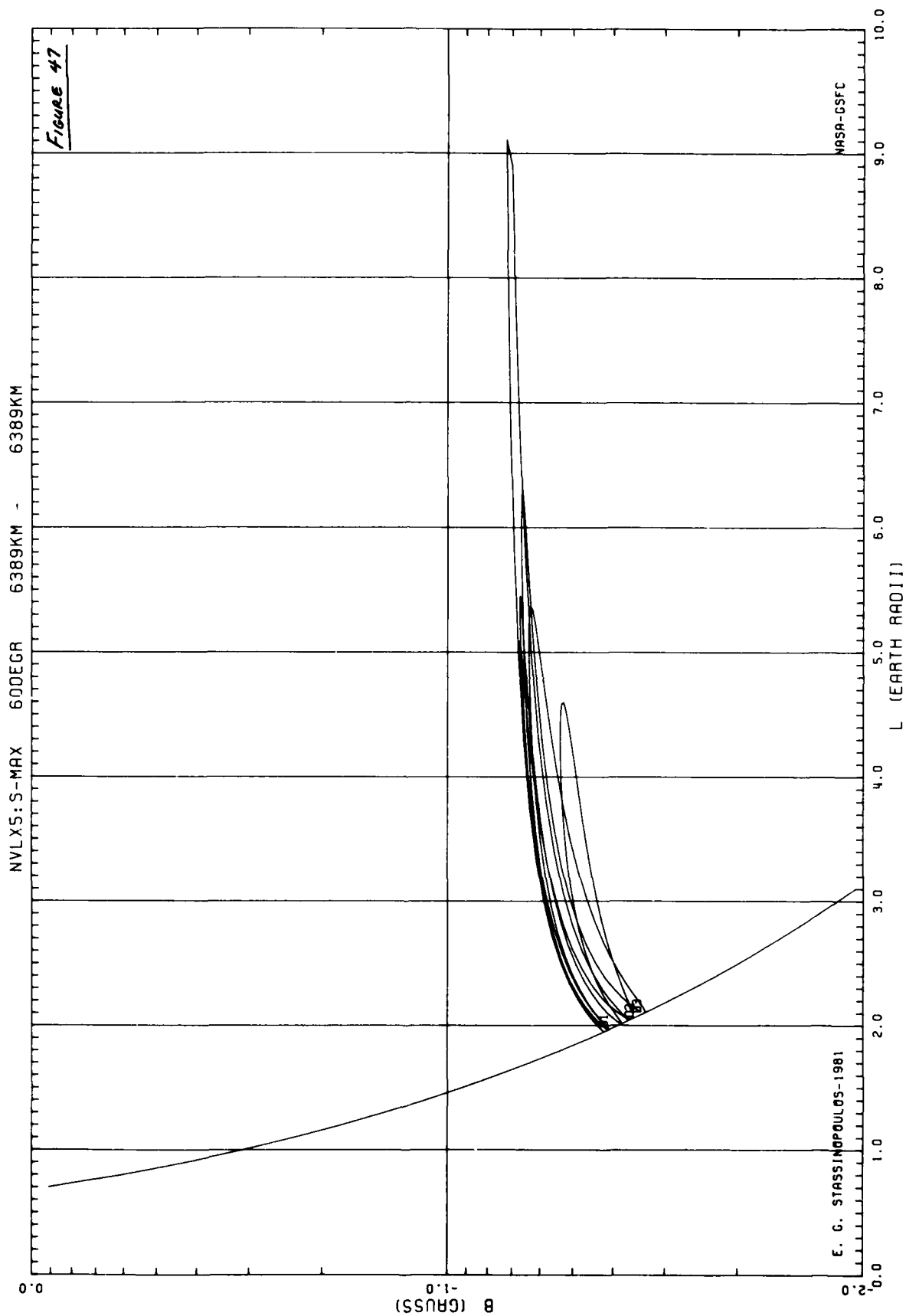












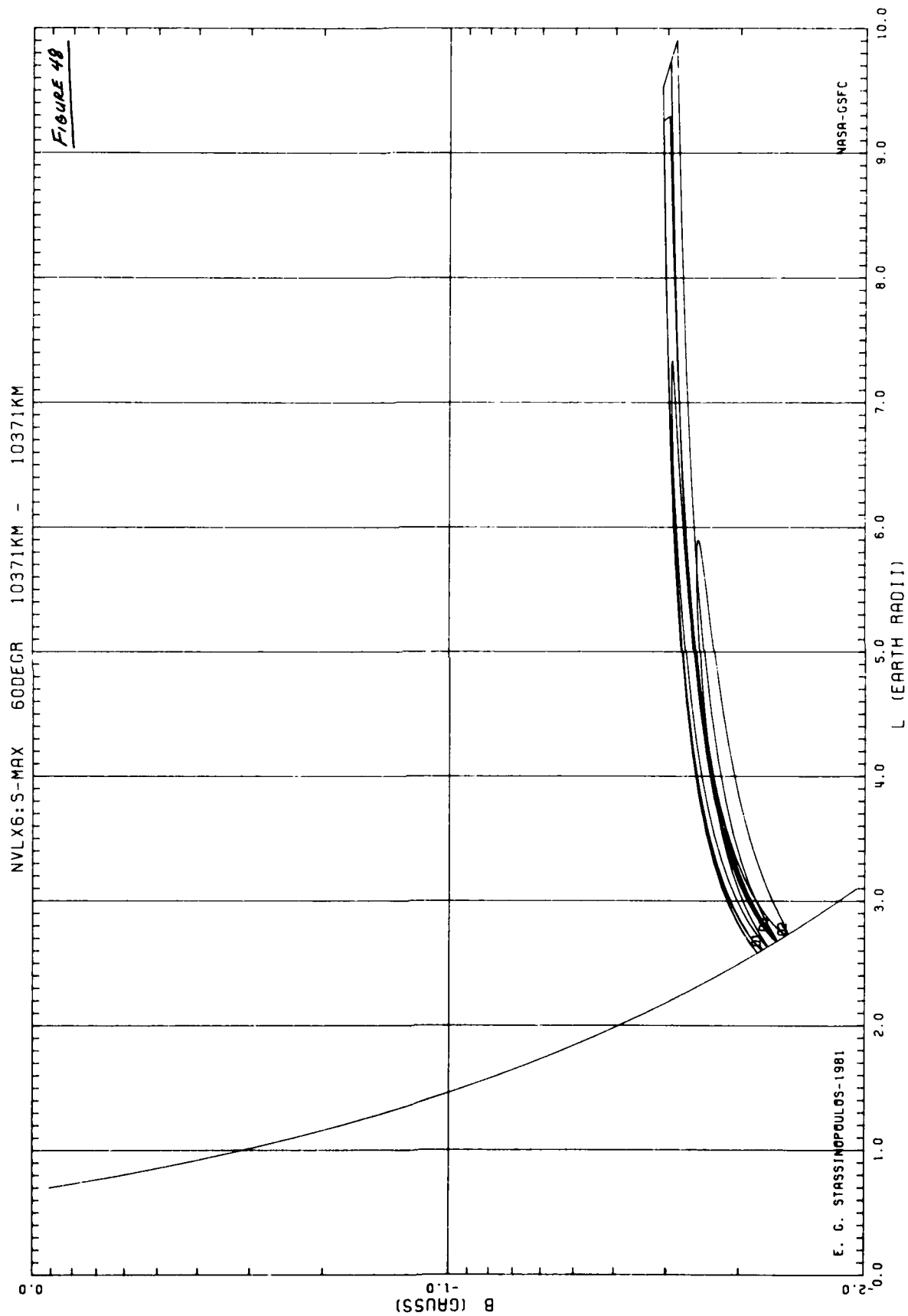


FIGURE 49

DOSE AT TRANSMISSION SURFACE OF FINITE ALUMINUM SLAB SHIELDS ** NVLX1:S-MAX

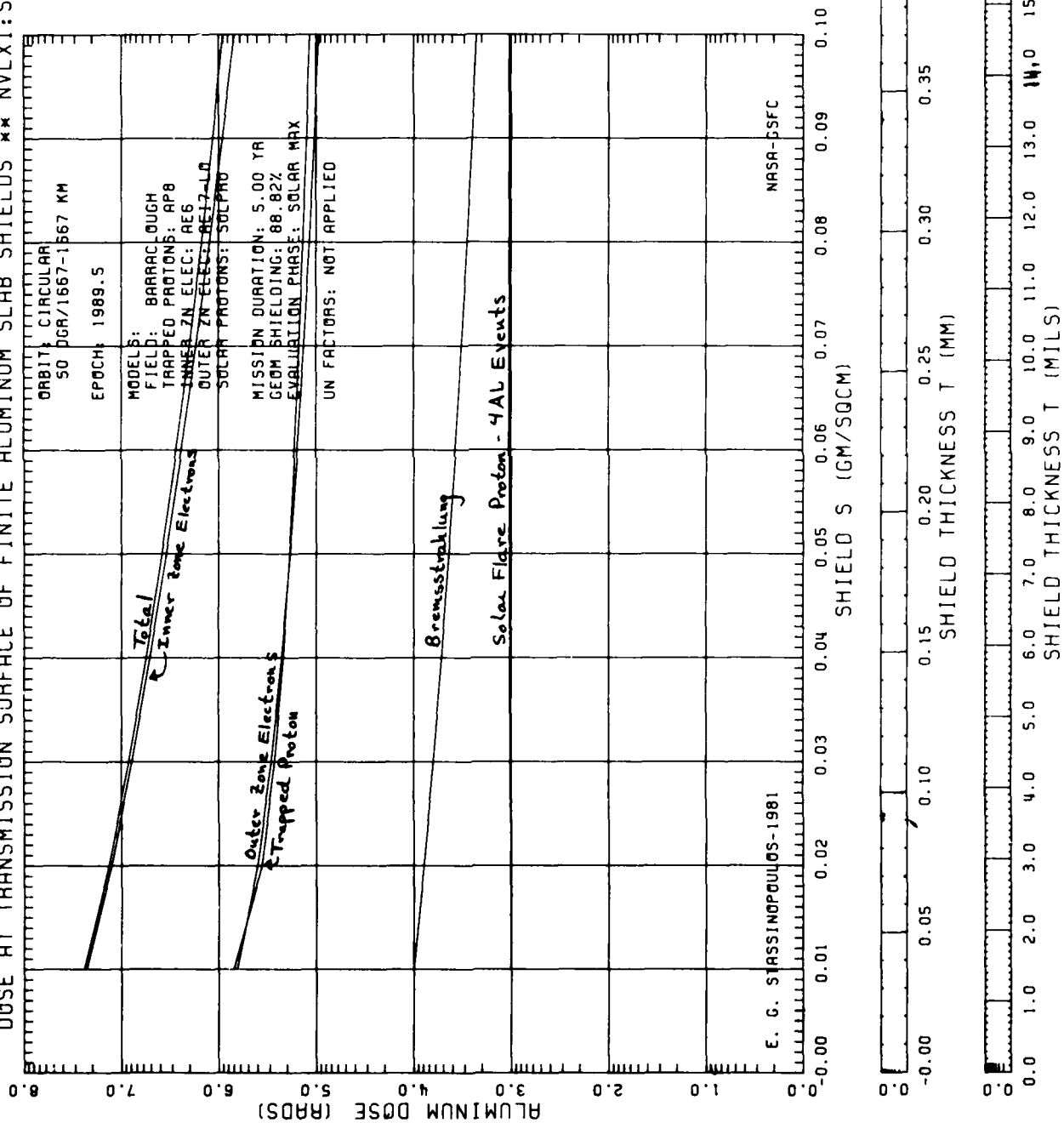


Figure 50

DOSE AT TRANSMISSION SURFACE OF FINITE ALUMINUM SLAB SHIELDS ** NVLX1:S-MAX

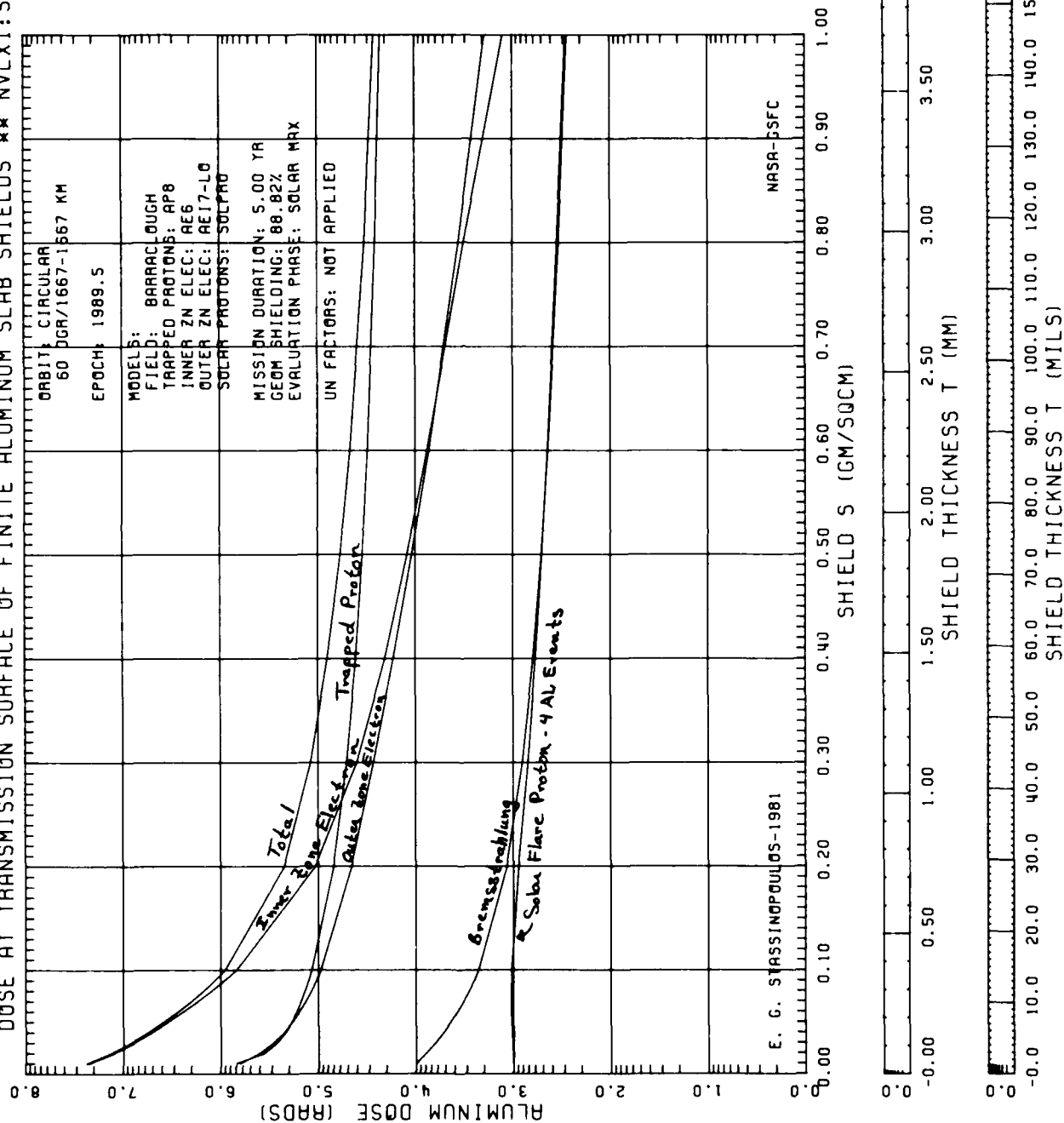


FIGURE 51

DOSE AT TRANSMISSION SURFACE OF FINITE ALUMINUM SLAB SHIELDS ** NVLX1: S-MAX

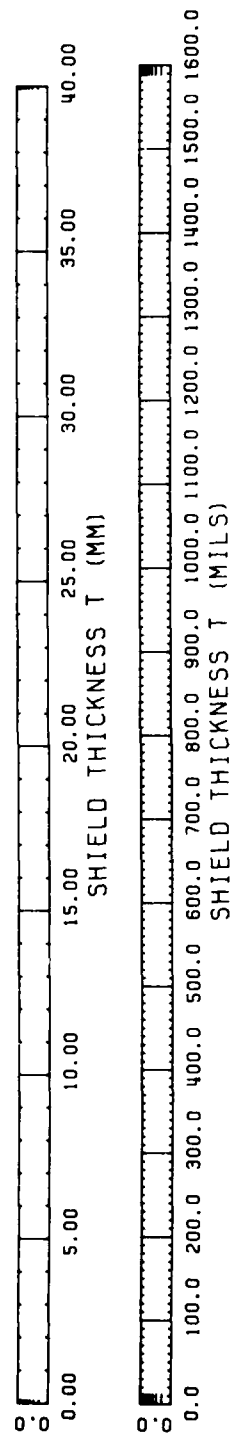
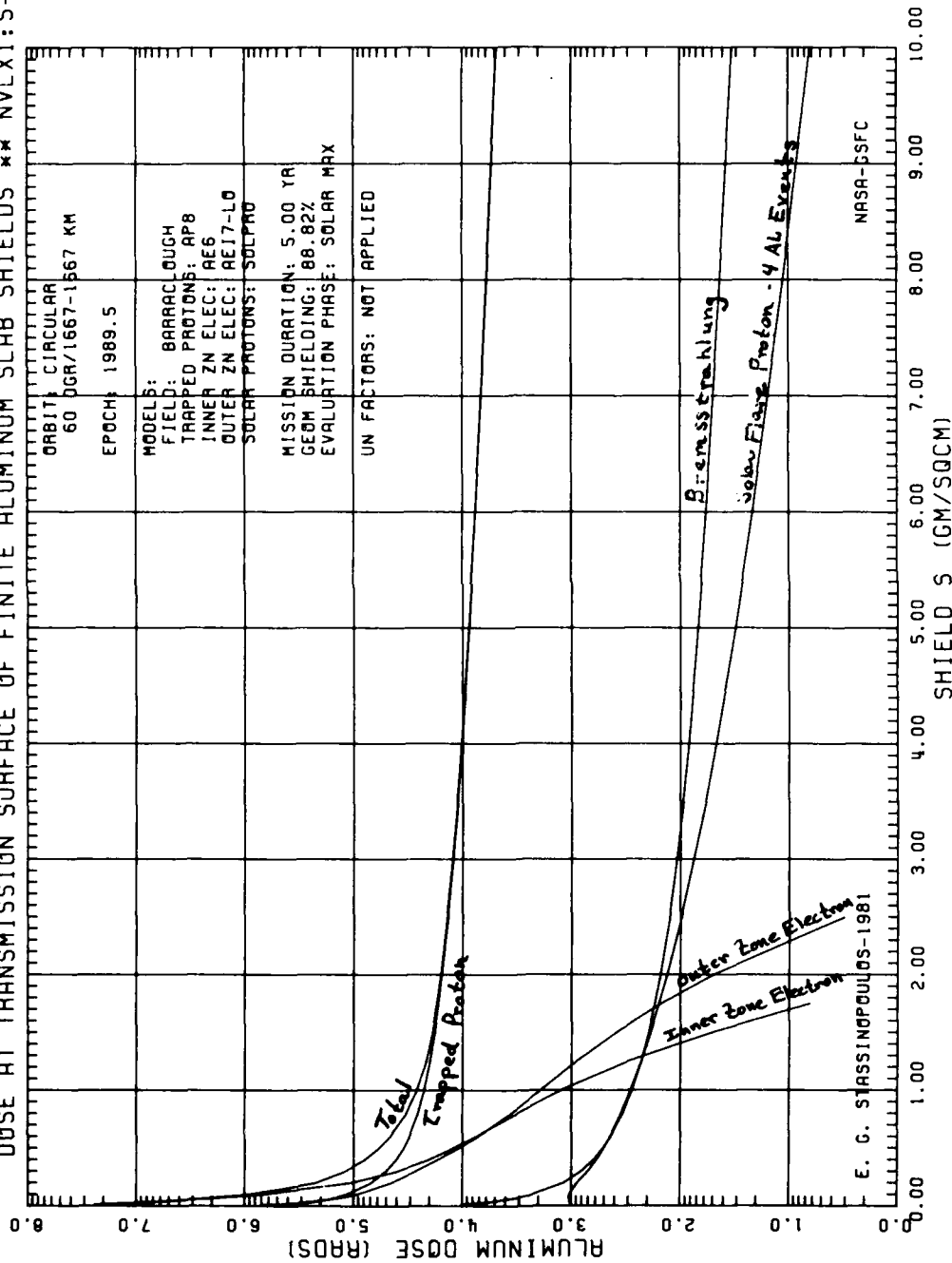


FIGURE 52

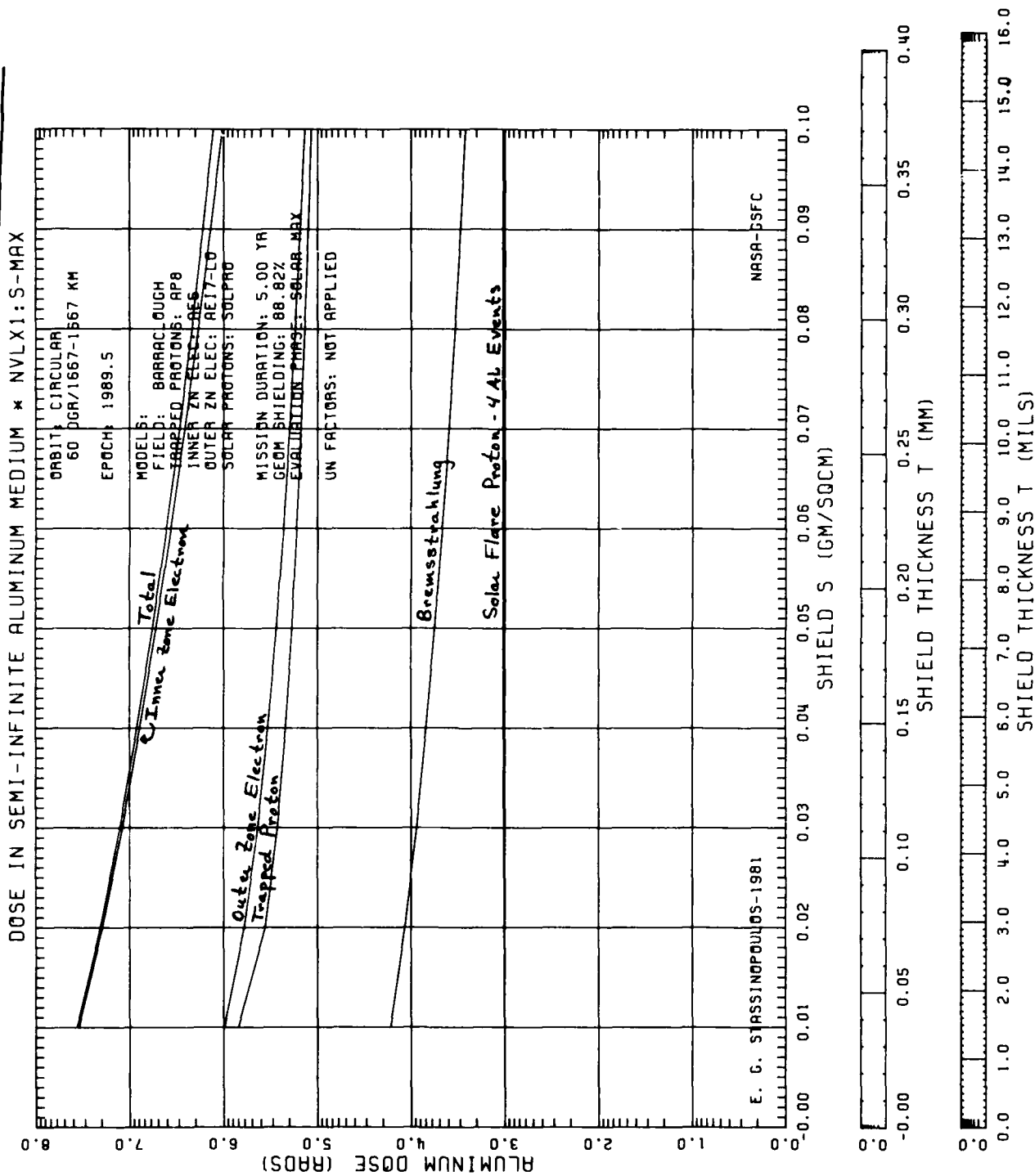


Figure 53

DOSE IN SEMI-INFINITE ALUMINUM MEDIUM * NVLX1:S-MAX

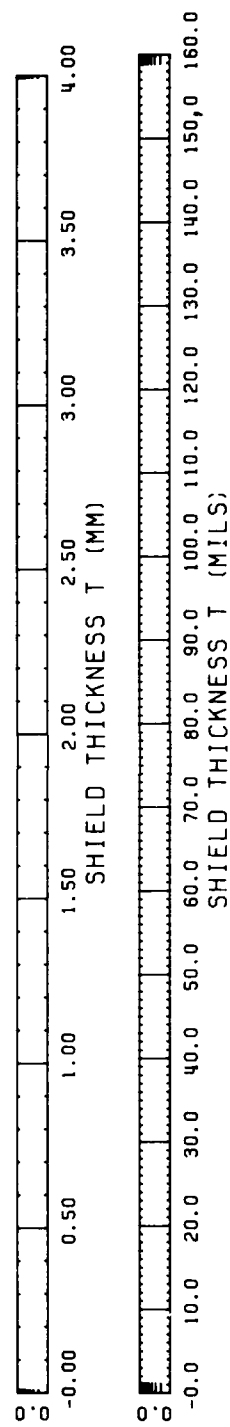
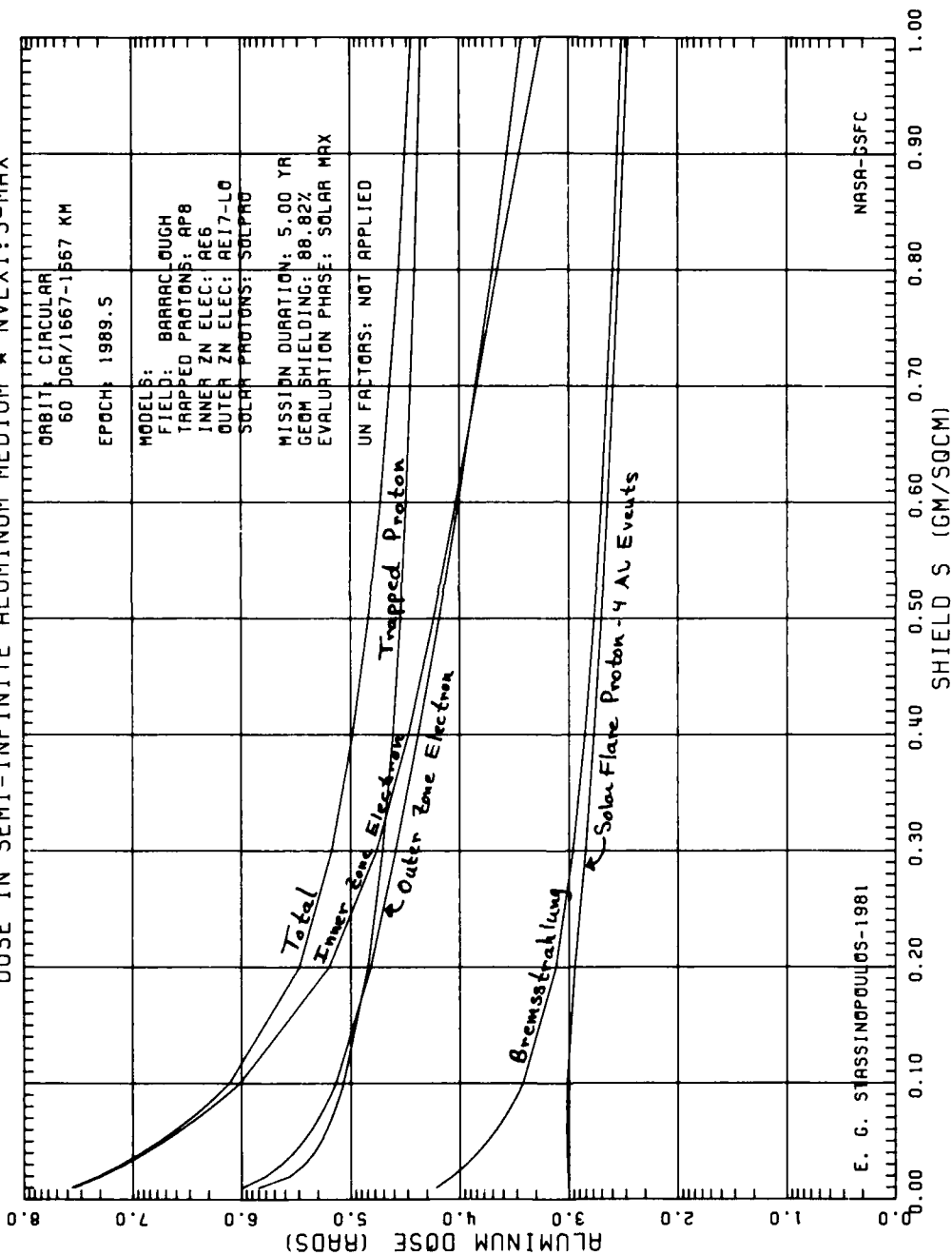
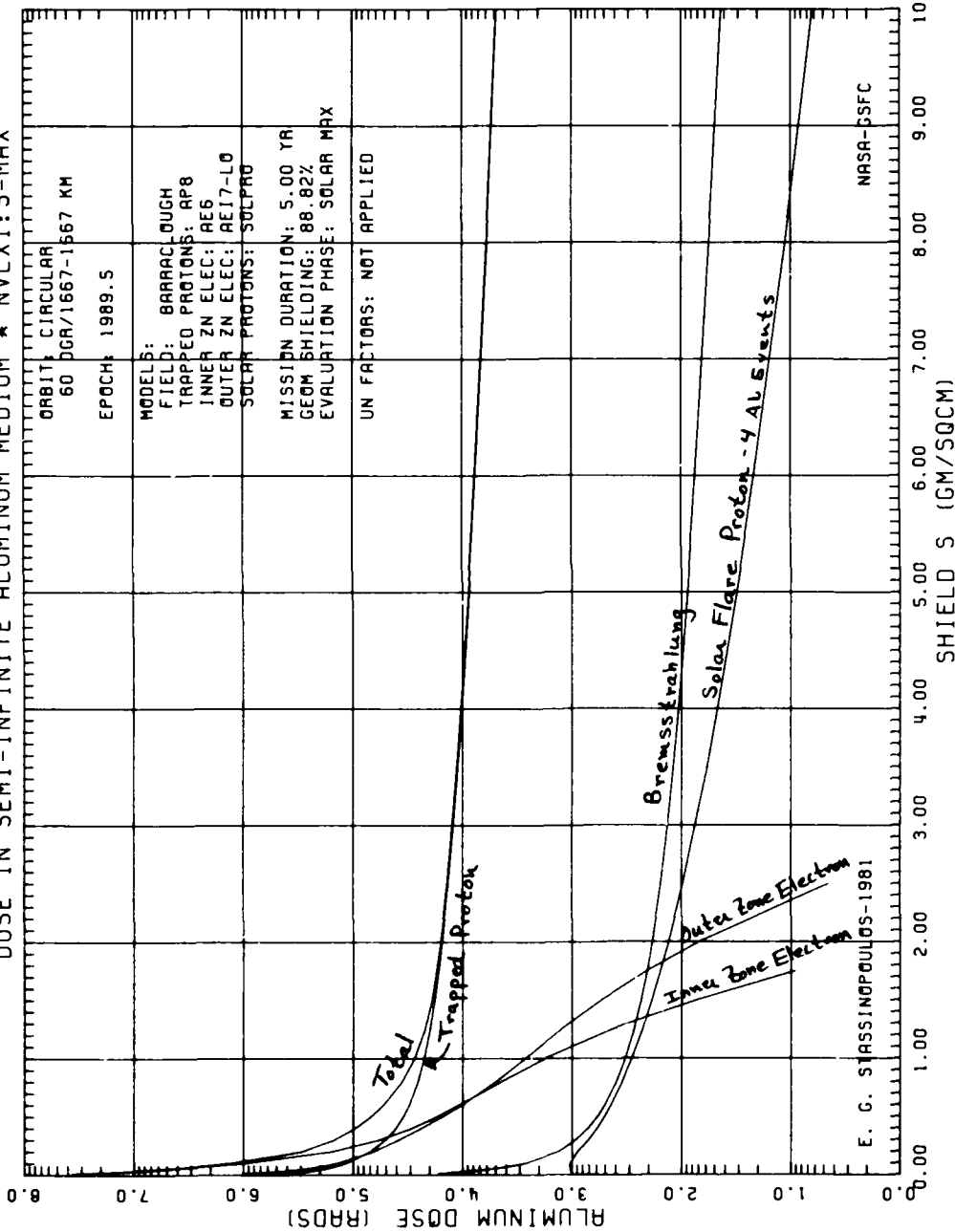


Figure 54

DOSE IN SEMI-INFINITE ALUMINUM MEDIUM * NVLX1:S-MAX



0.0 5.00 10.00 15.00 20.00 25.00 30.00 35.00 40.00

SHIELD THICKNESS T (MM)

0.0 100.0 200.0 300.0 400.0 500.0 600.0 700.0 800.0 900.0 1000.0 1100.0 1200.0 1300.0 1400.0 1500.0 1600.0

SHIELD THICKNESS T (MILS)

Figure 55

DOSE AT CENTER OF ALUMINUM SPHERES ** NVLX1:S-MAX

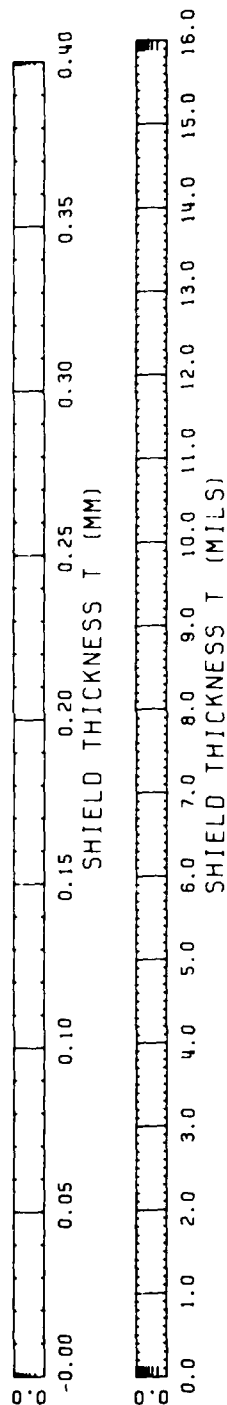
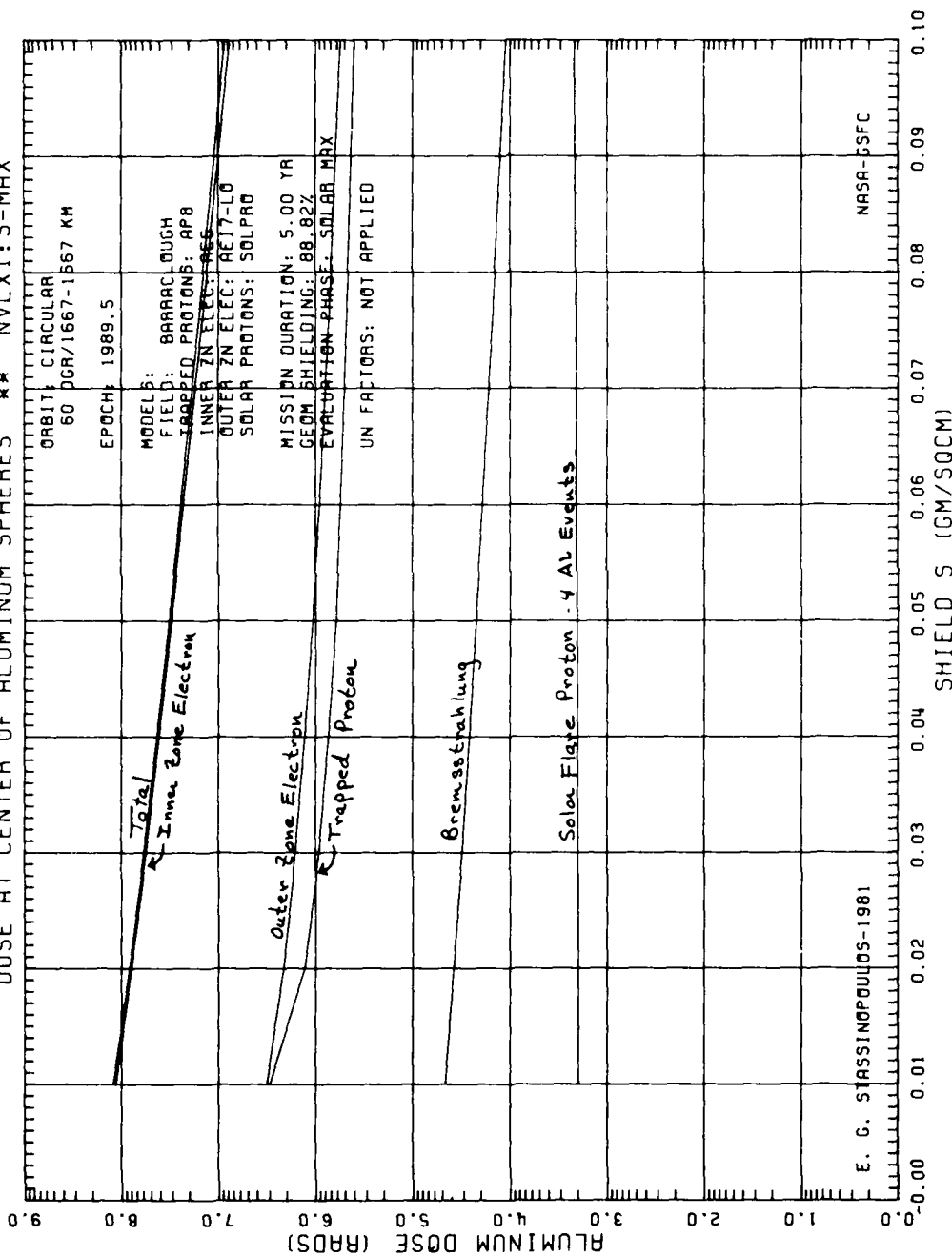


Figure 56

DOSE AT CENTER OF ALUMINUM SPHERES ** NVLX1:S-MAX

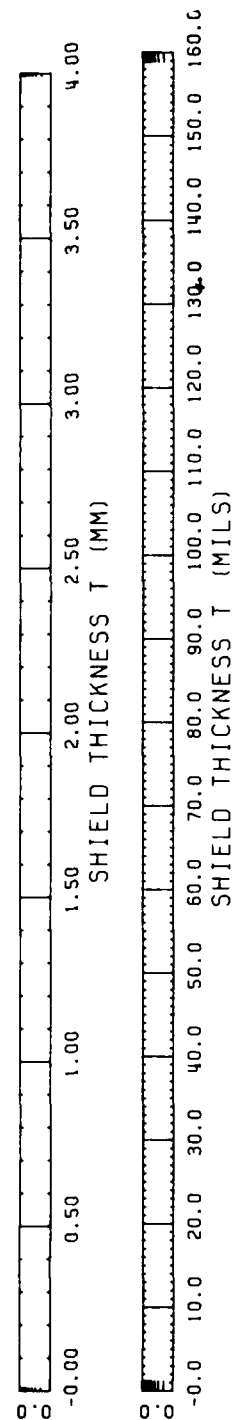
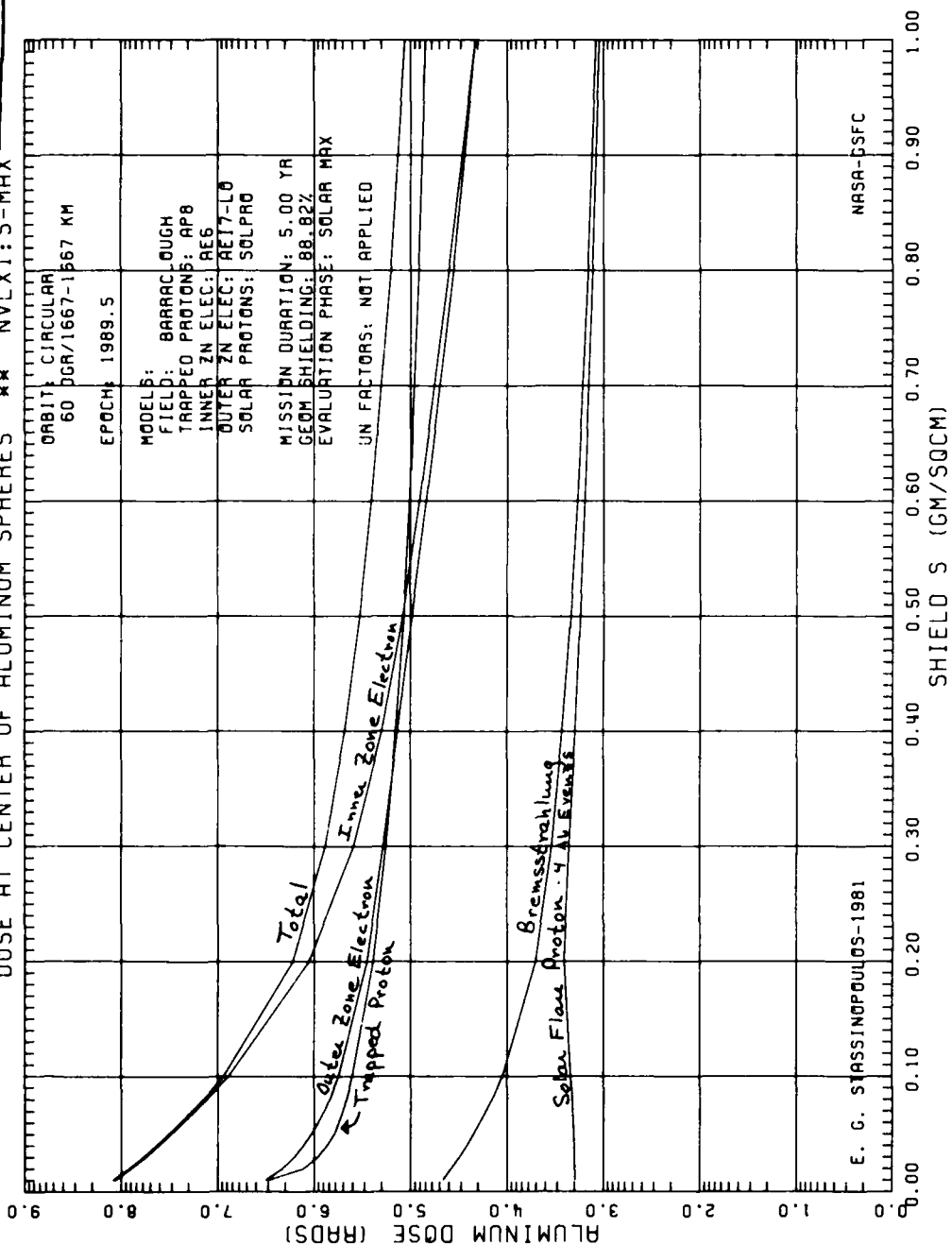
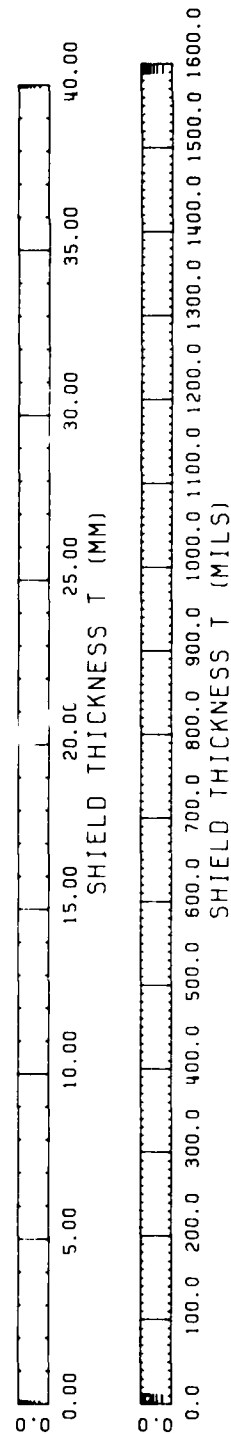
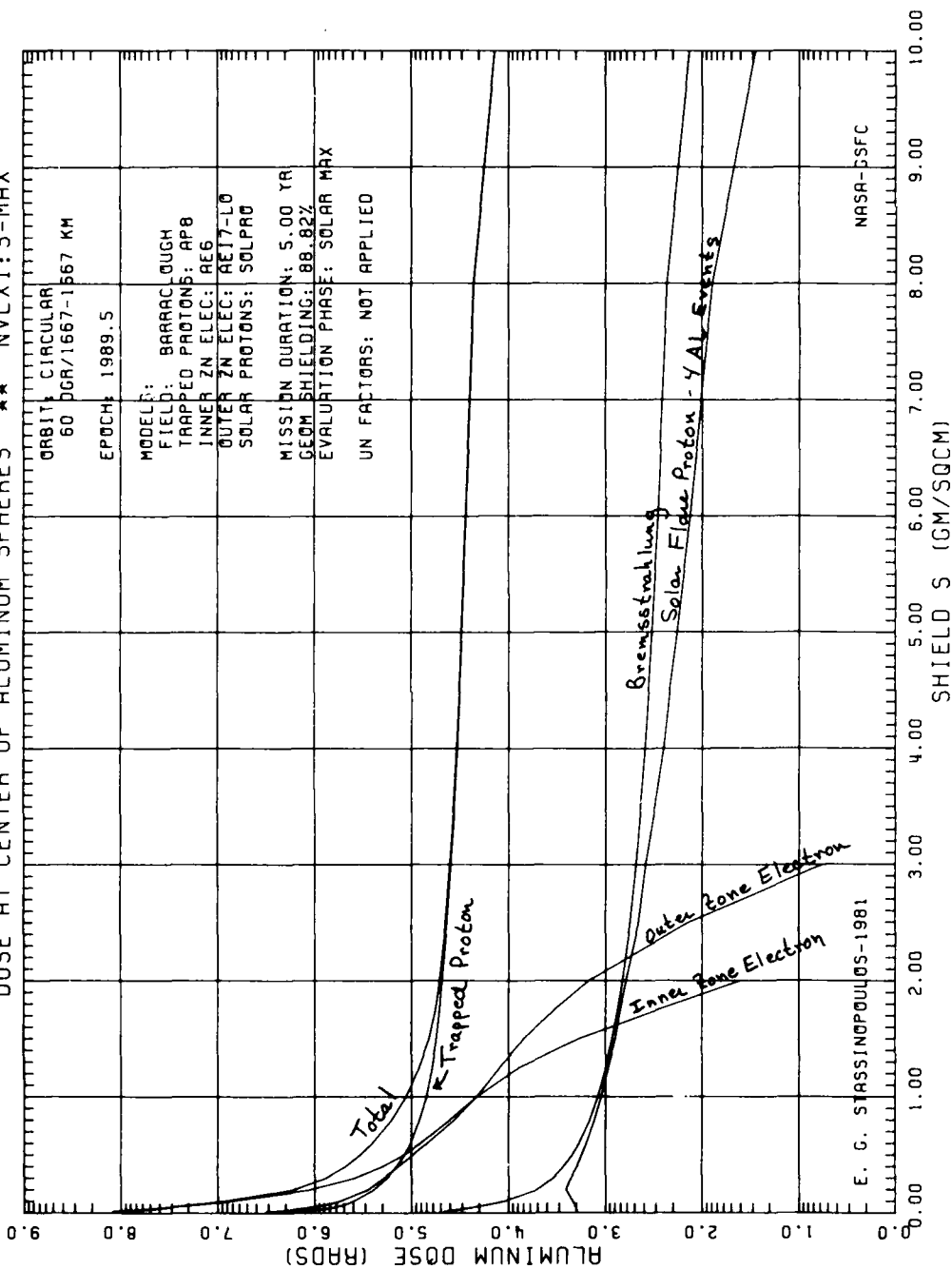


FIGURE 57

DOSE AT CENTER OF ALUMINUM SPHERES ** NVLX1:S-MAX



DOSE AT TRANSMISSION SURFACE OF FINITE ALUMINUM SLAB SHIELDS ** NVLX2:S-MAX

Figure 58

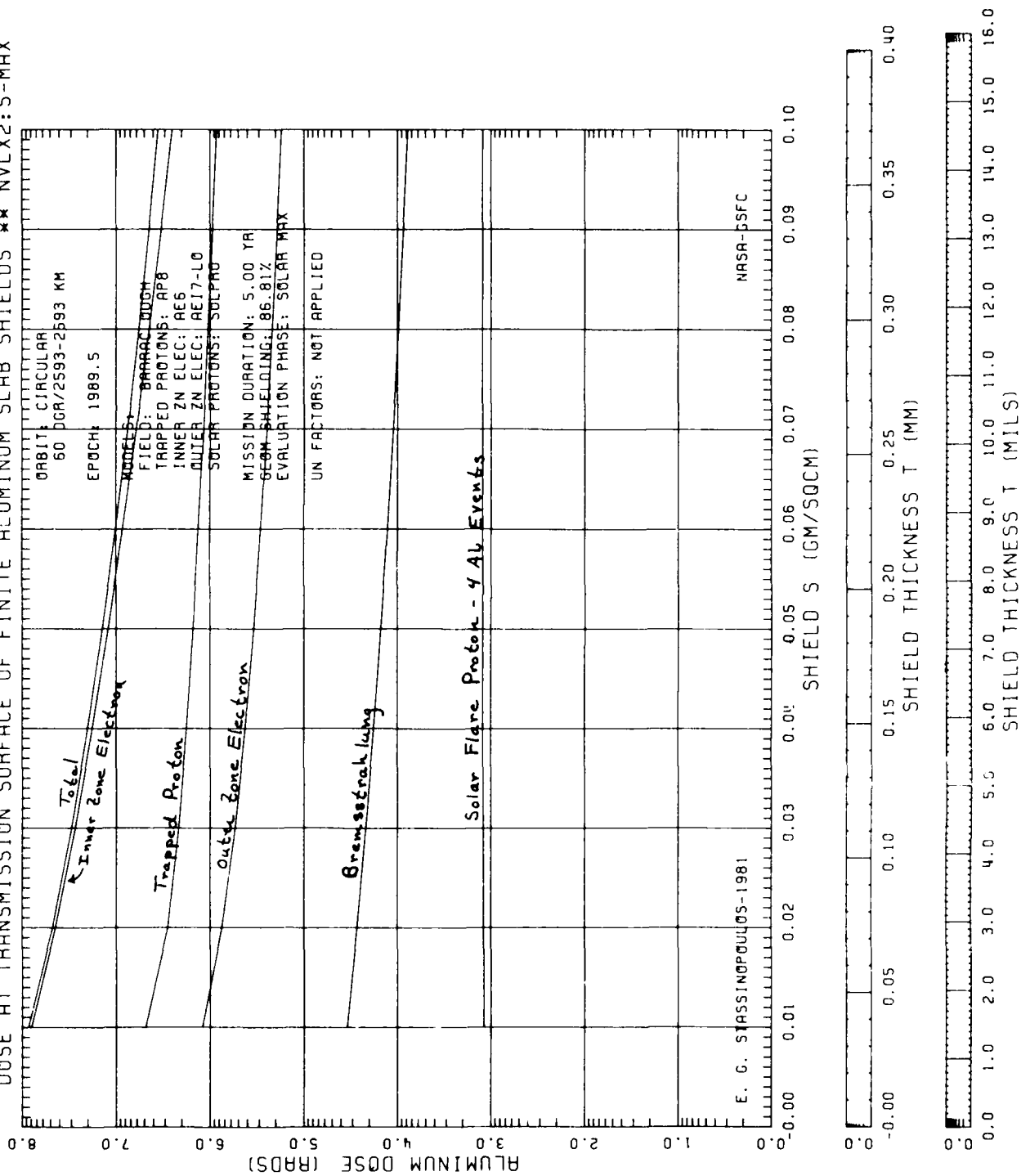


FIGURE 59

DOSE AT TRANSMISSION SURFACE OF FINITE ALUMINUM SLAB SHIELDS ** NVLX2:S-MAX

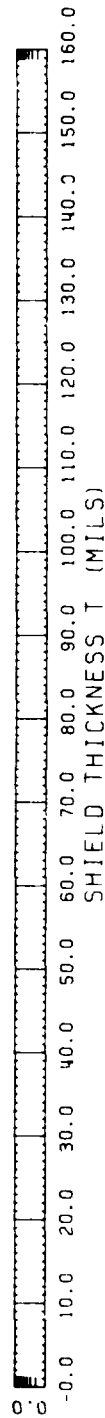
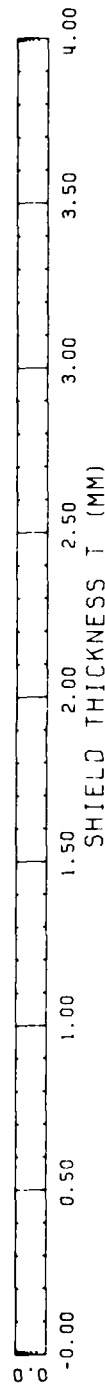
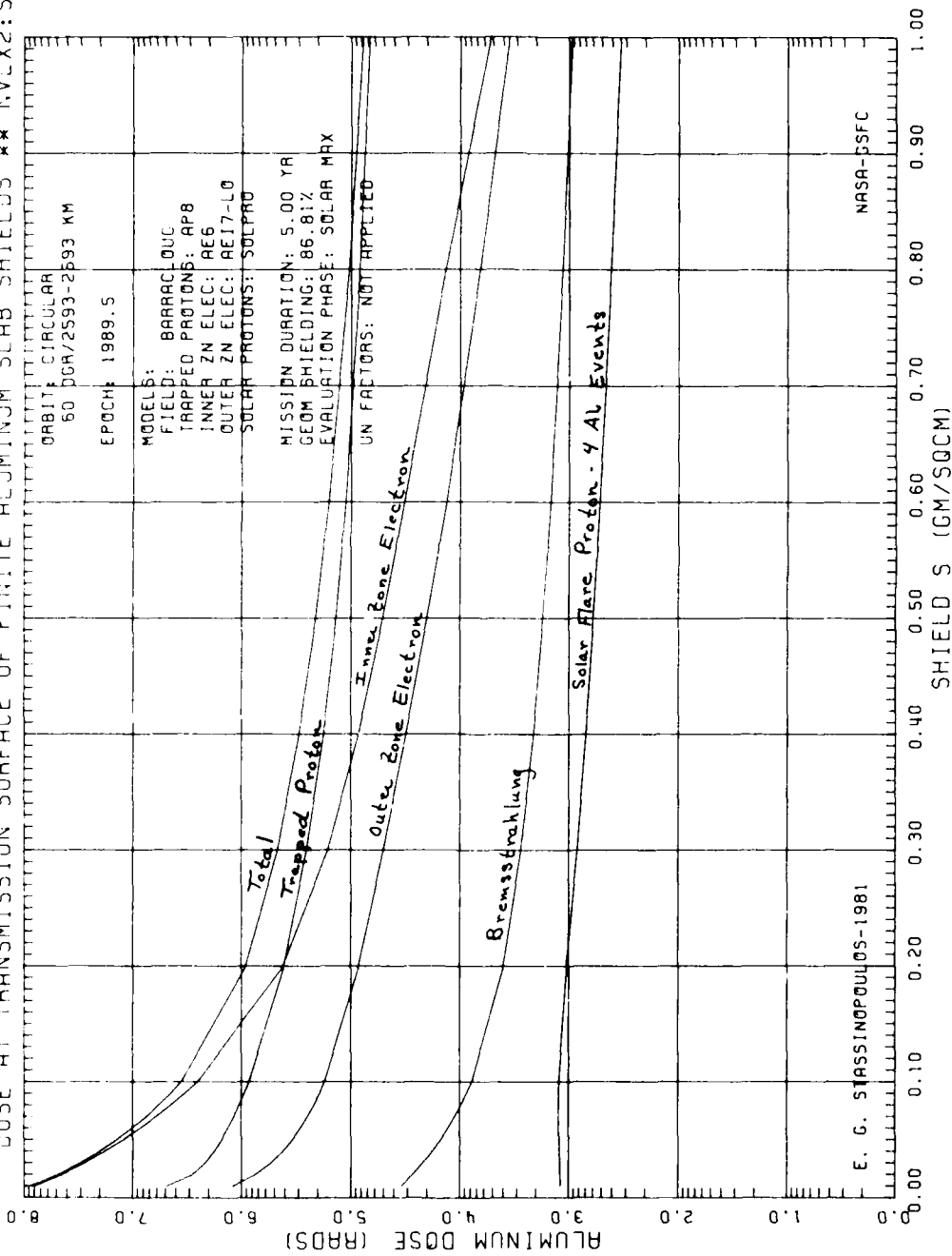


FIGURE 60

NVLX2:S-MAX

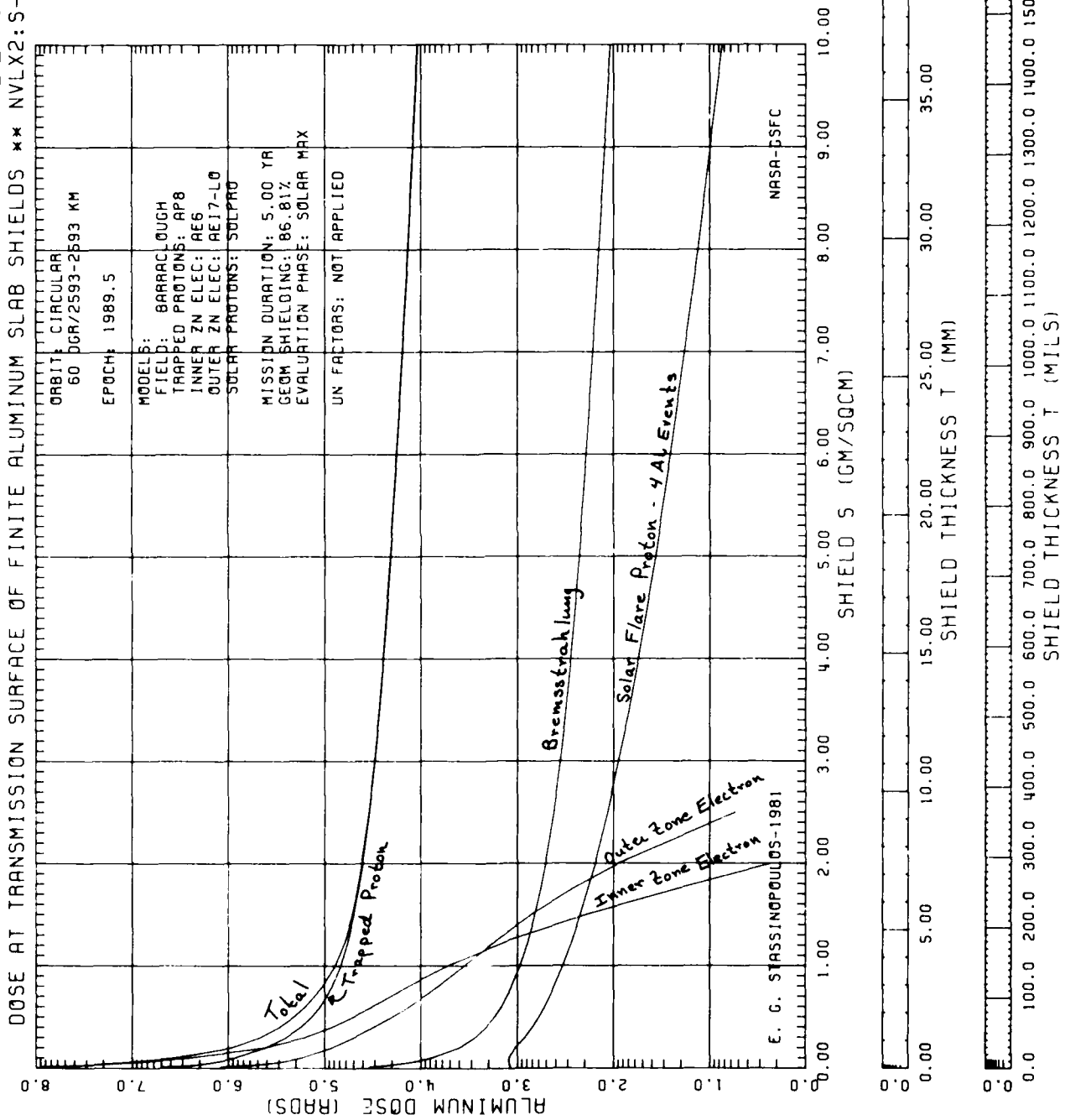


FIGURE 61

DOSE IN SEMI-INFINITE ALUMINUM MEDIUM * NVLX2:S-MAX

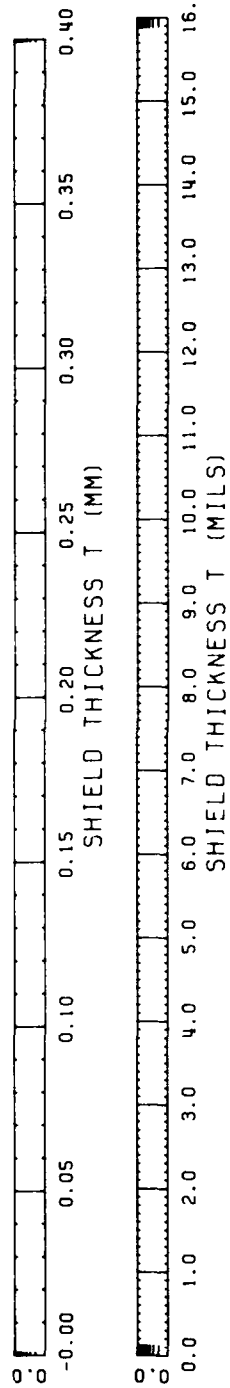
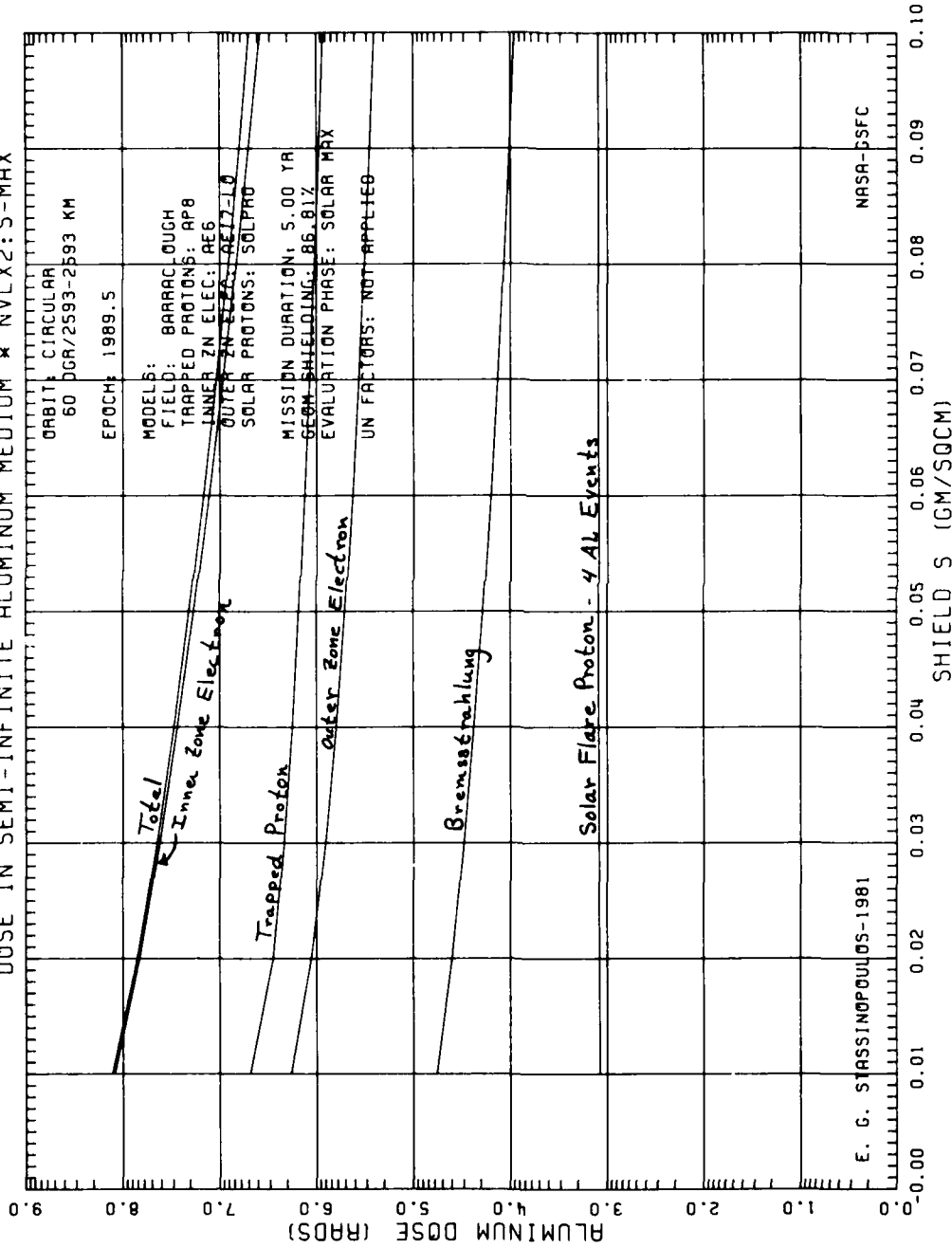


FIGURE 62

DOSE IN SEMI-INFINITE ALUMINUM MEDIUM * NVLX2:S-MAX

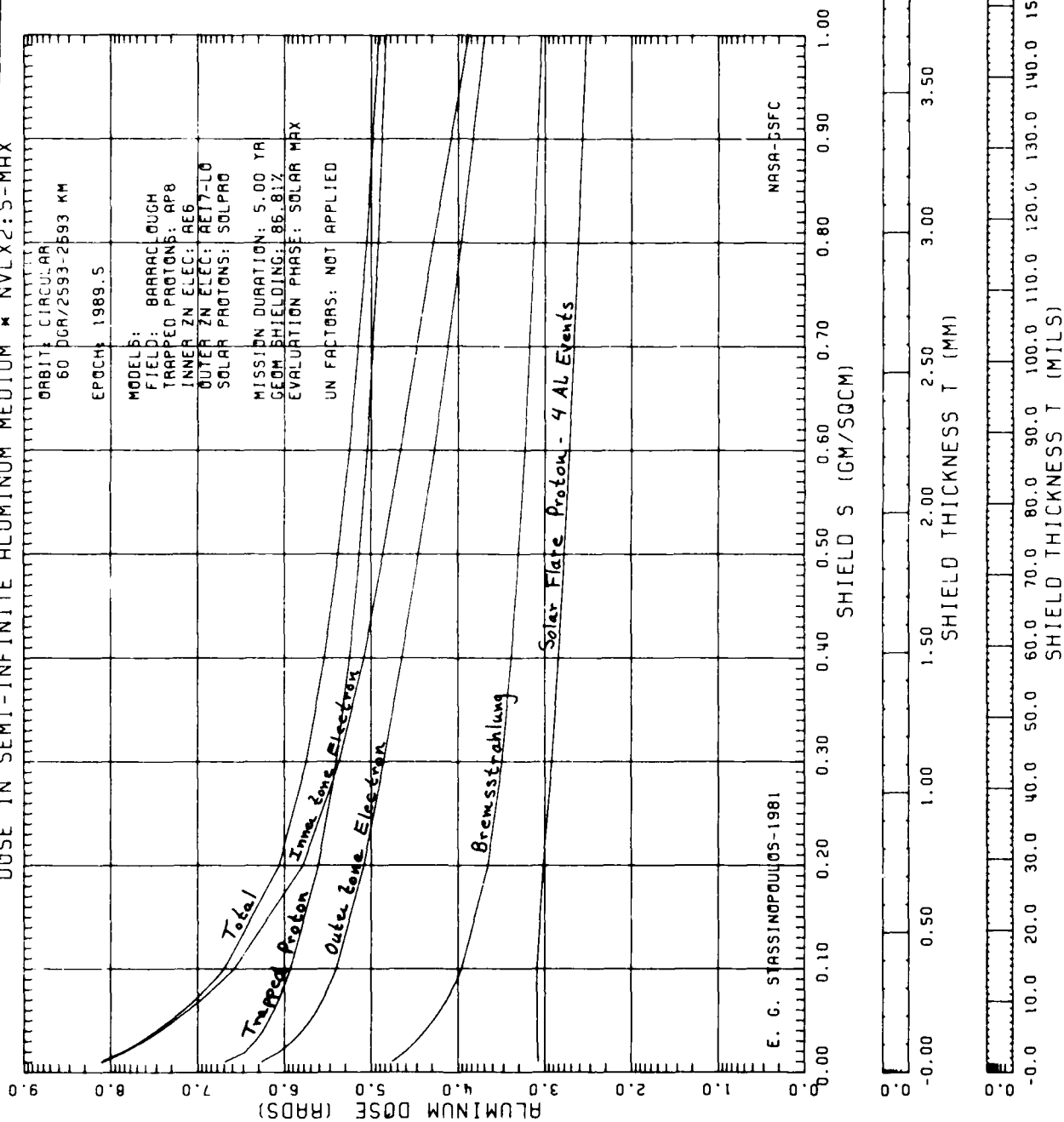


FIGURE 63

DOSE IN SEMI-INFINITE ALUMINUM MEDIUM * NVLX2:S-MAX

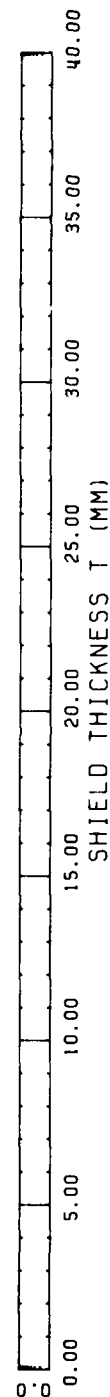
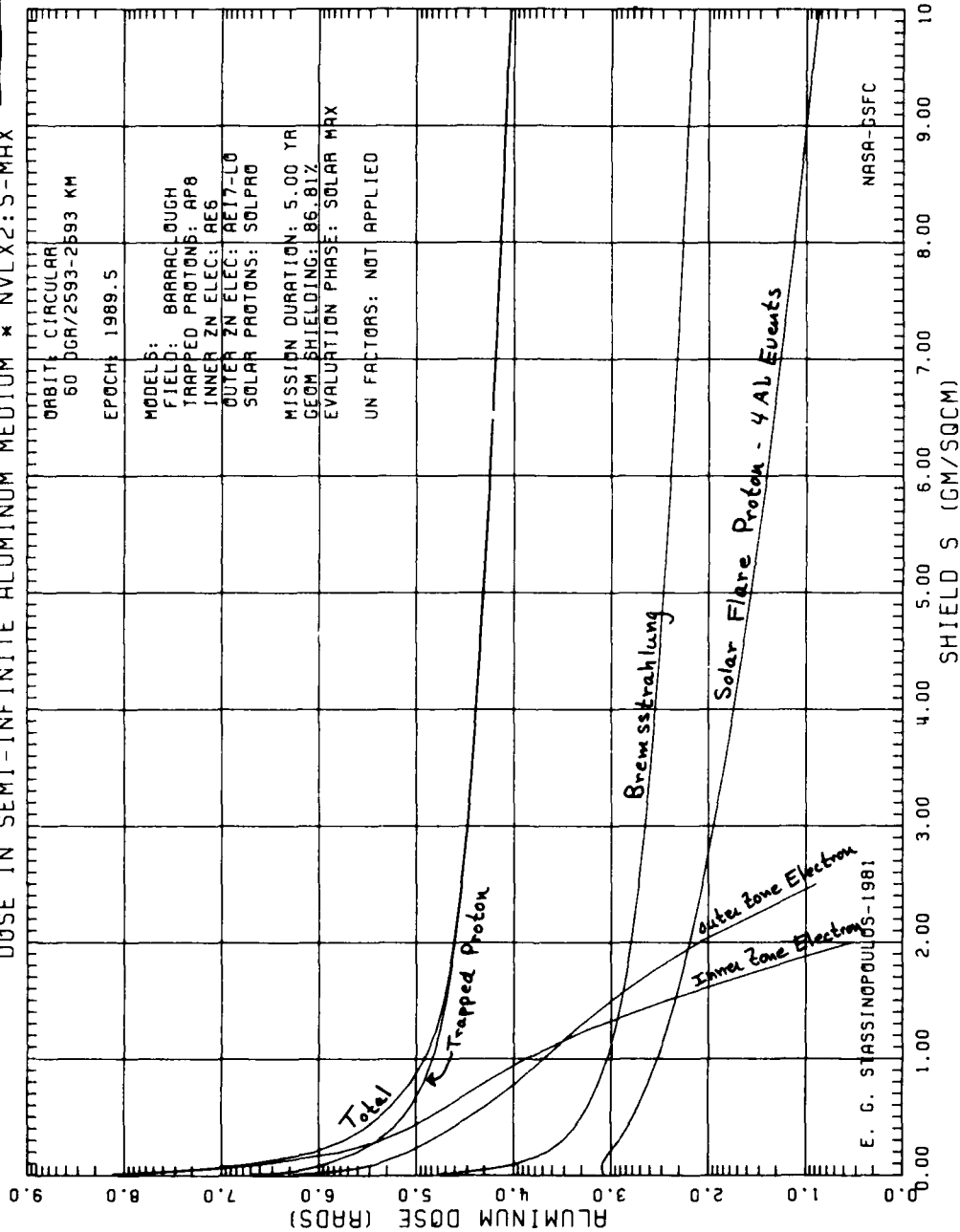


FIGURE 64

DOSE AT CENTER OF ALUMINUM SPHERES ** NVLX2:S-MAX

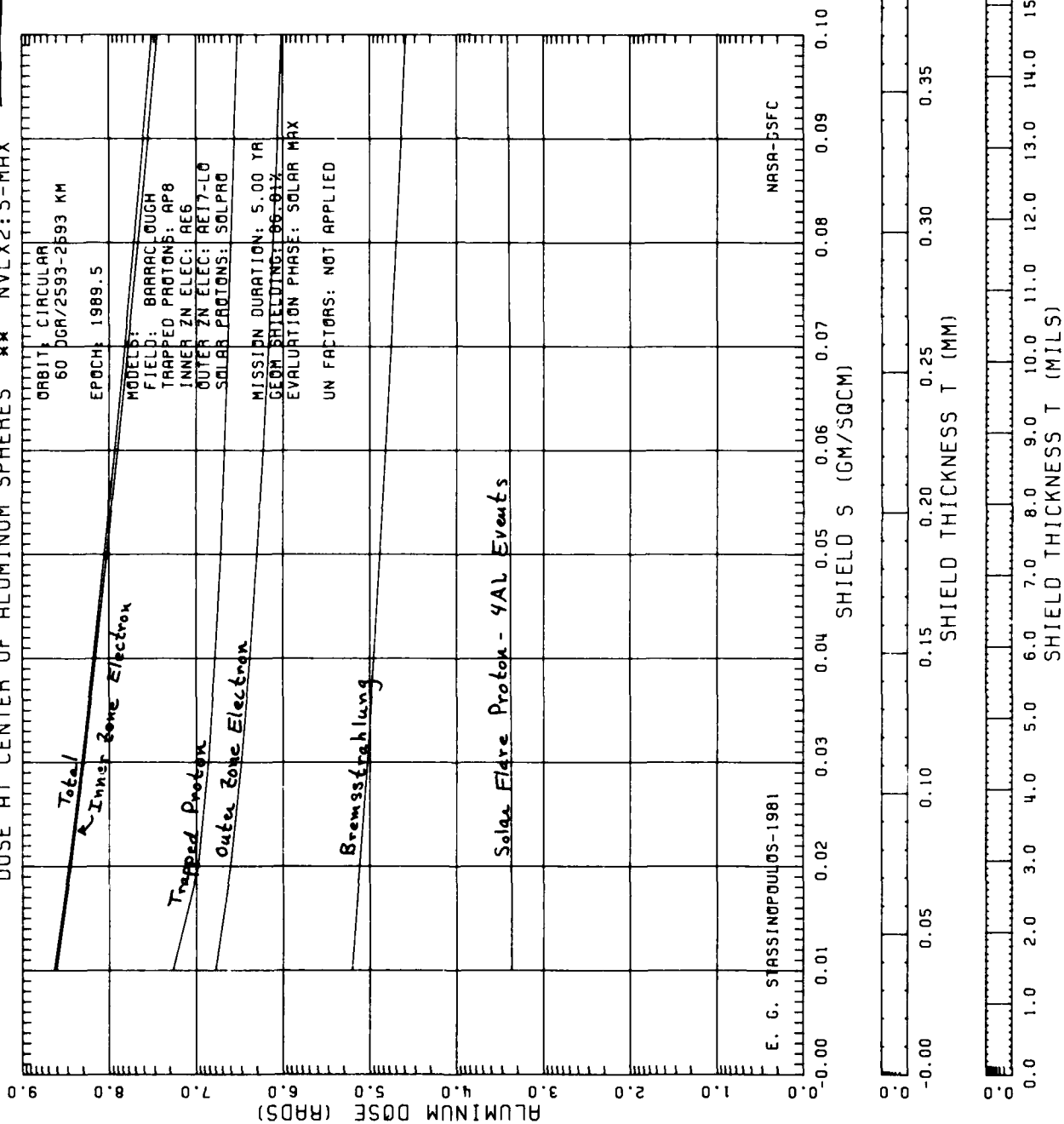


FIGURE 65

DOSE AT CENTER OF ALUMINUM SPHERES ** NVLX2:S-MAX

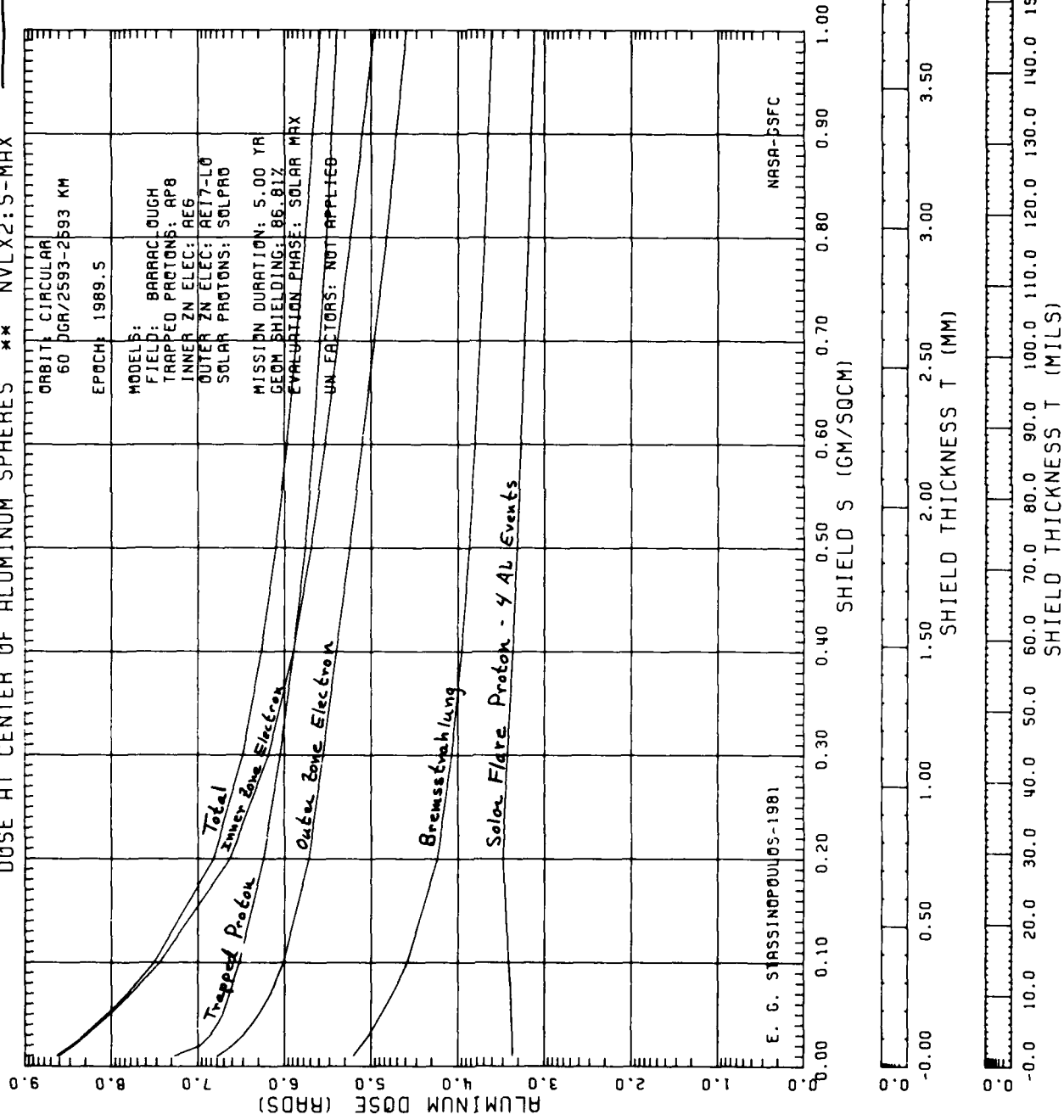


Figure 66

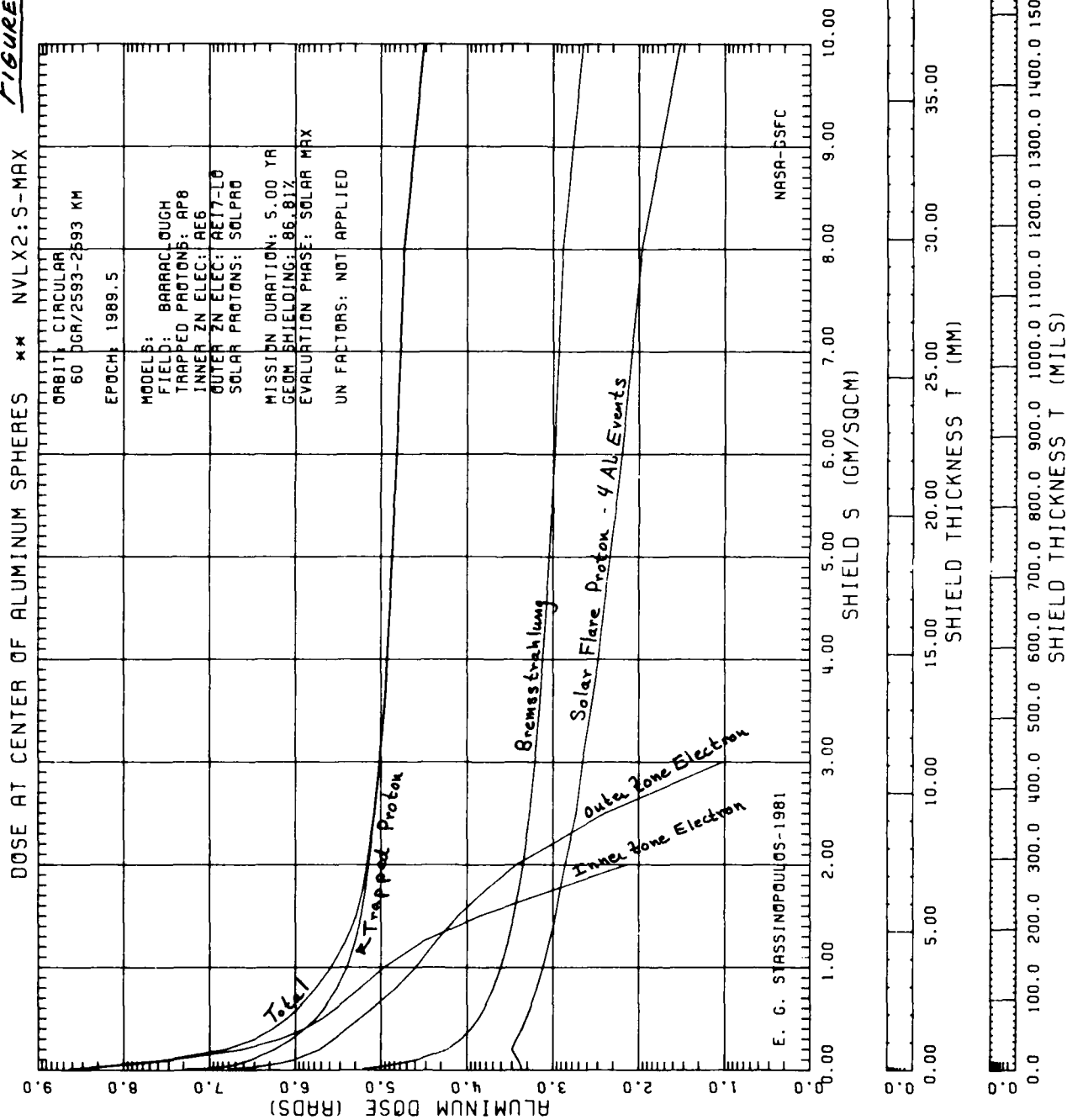


Figure 67
DOSE AT TRANSMISSION SURFACE OF FINITE ALUMINUM SLAB SHIELDS ** NVLX3:S-MAX

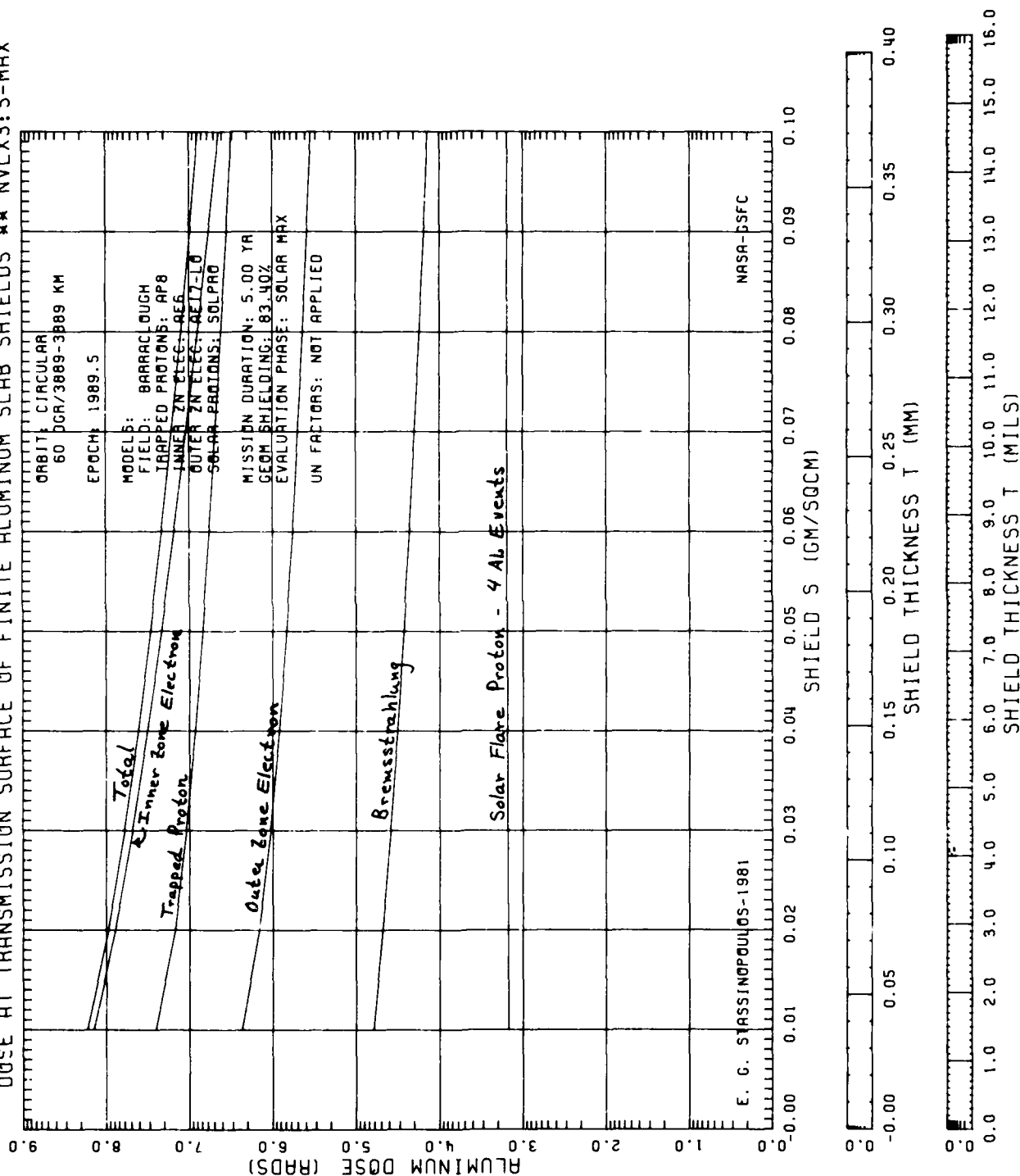


Figure 68
DOSE AT TRANSMISSION SURFACE OF FINITE ALUMINUM SLAB SHIELDS ** NVLX3: S-MAX

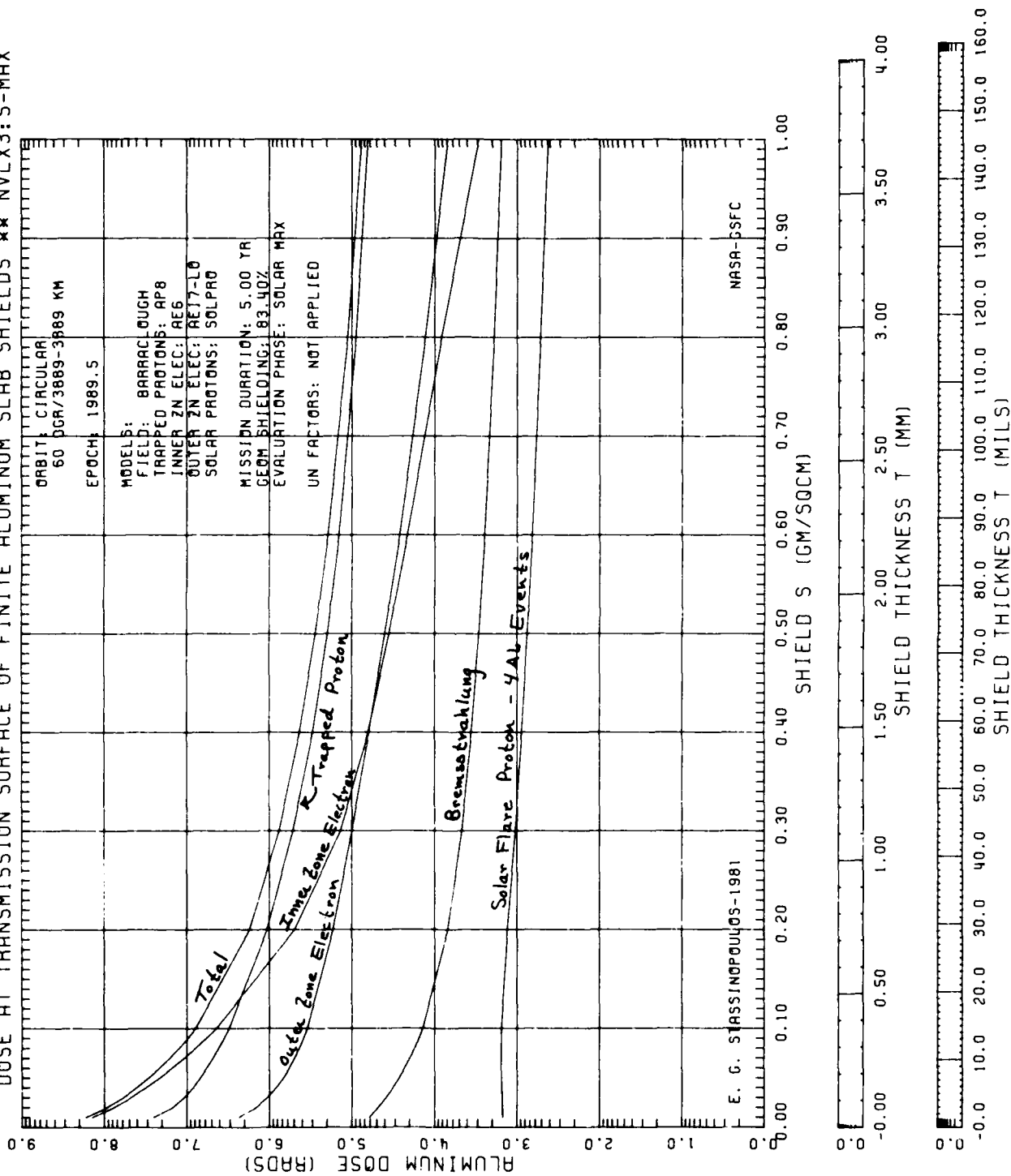


FIGURE 69

DOSE AT TRANSMISSION SURFACE OF FINITE ALUMINUM SLAB SHIELDS ** NVLX3:S-MAX

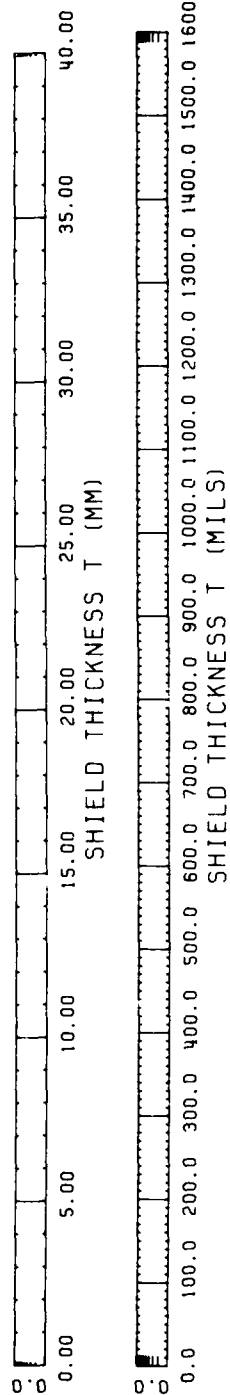
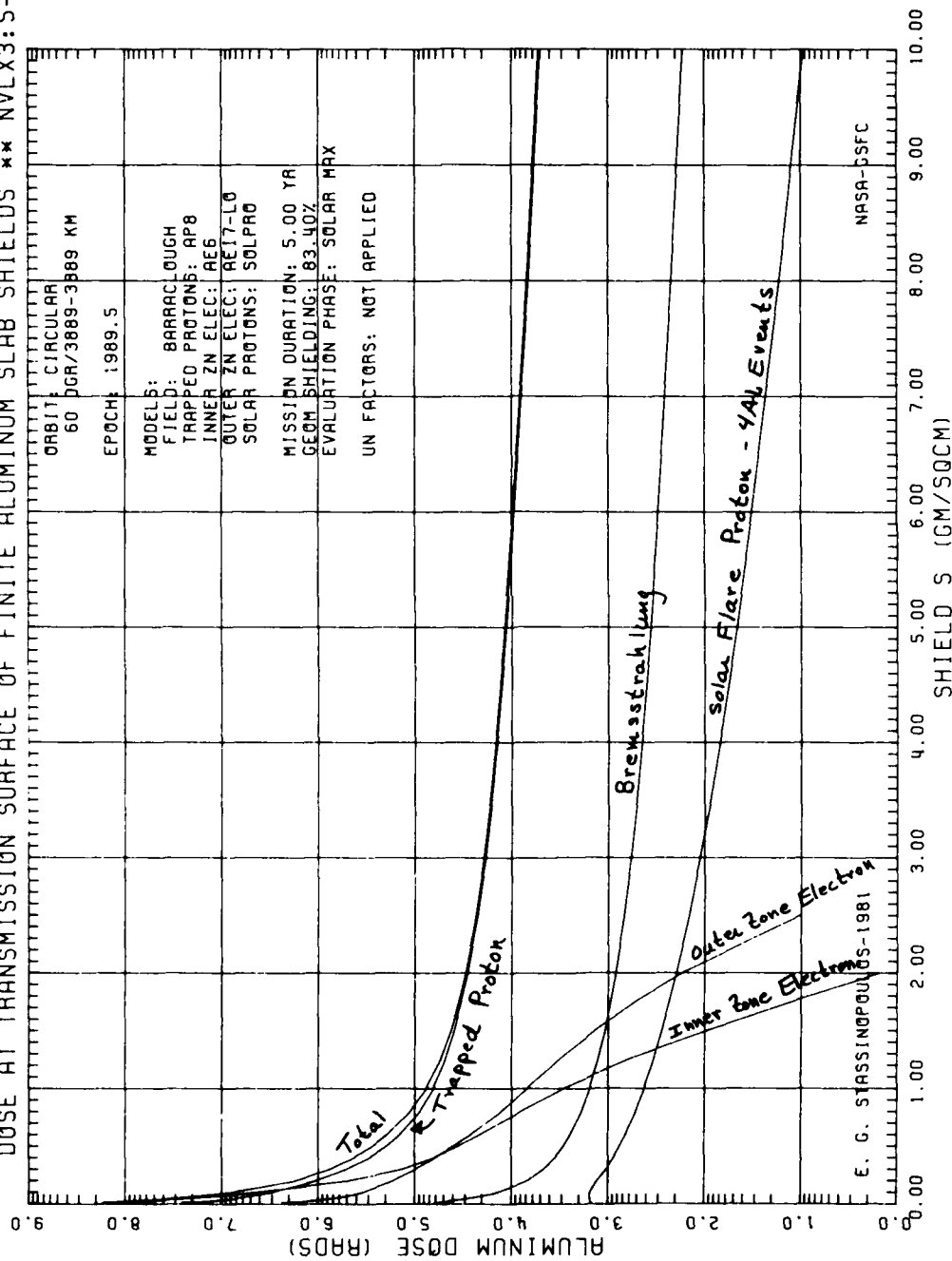


Figure 70

DOSE IN SEMI-INFINITE ALUMINUM MEDIUM * NVLX3:S-MAX

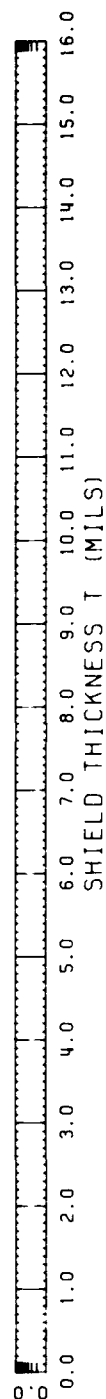
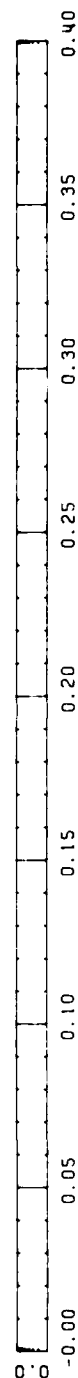
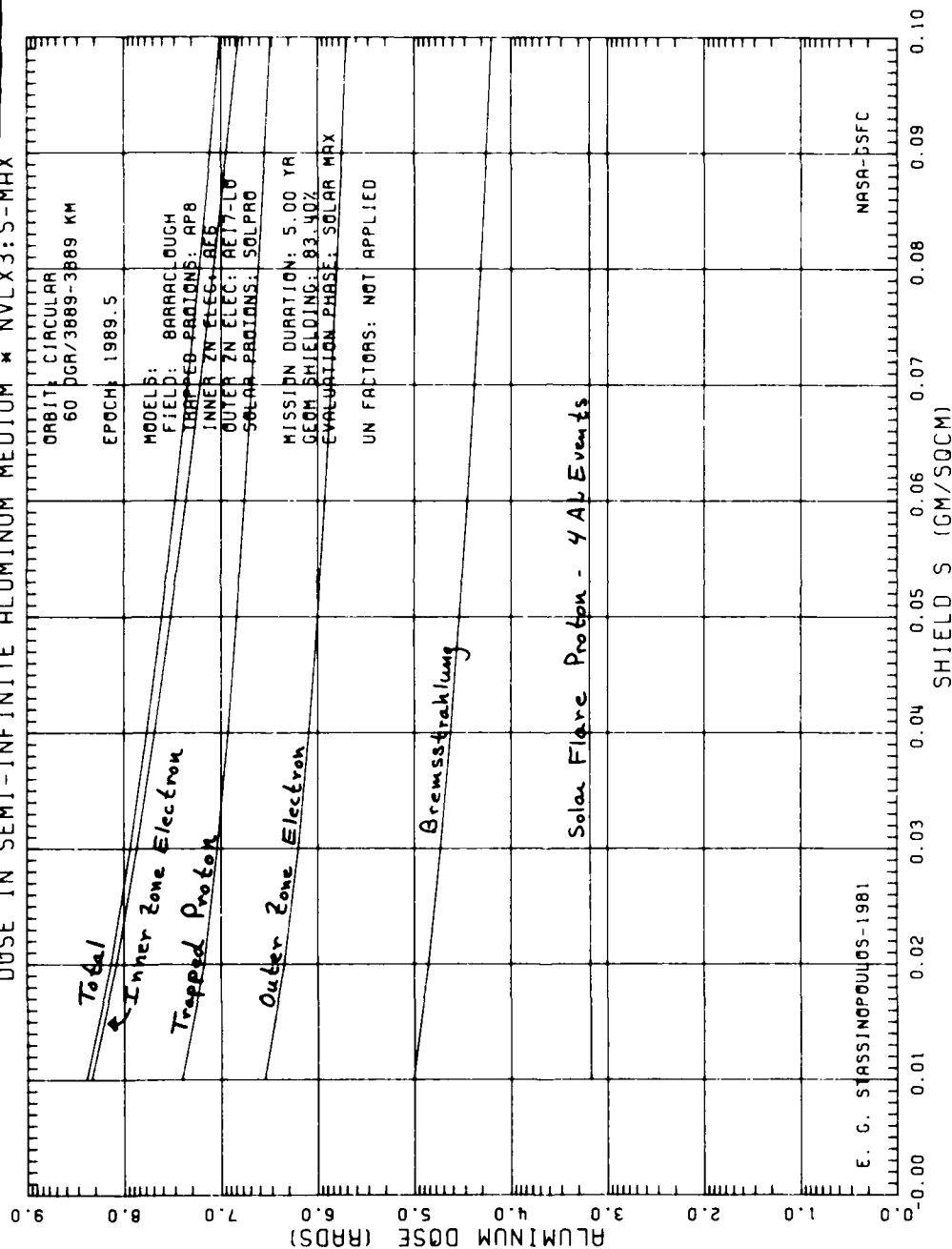


Figure 71

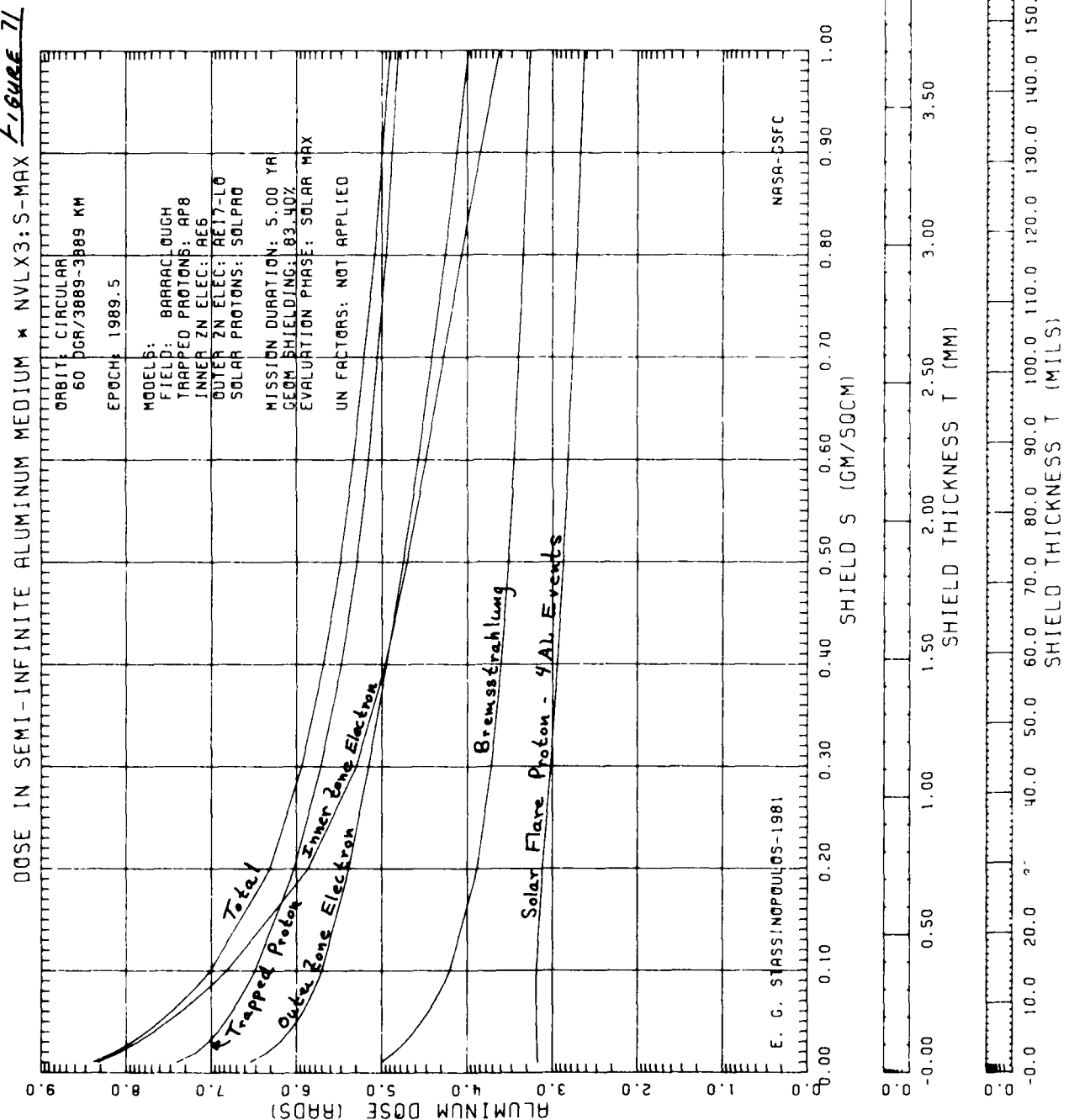


FIGURE 72

DOSE IN SEMI-INFINITE ALUMINUM MEDIUM * NVLX3:S-MAX

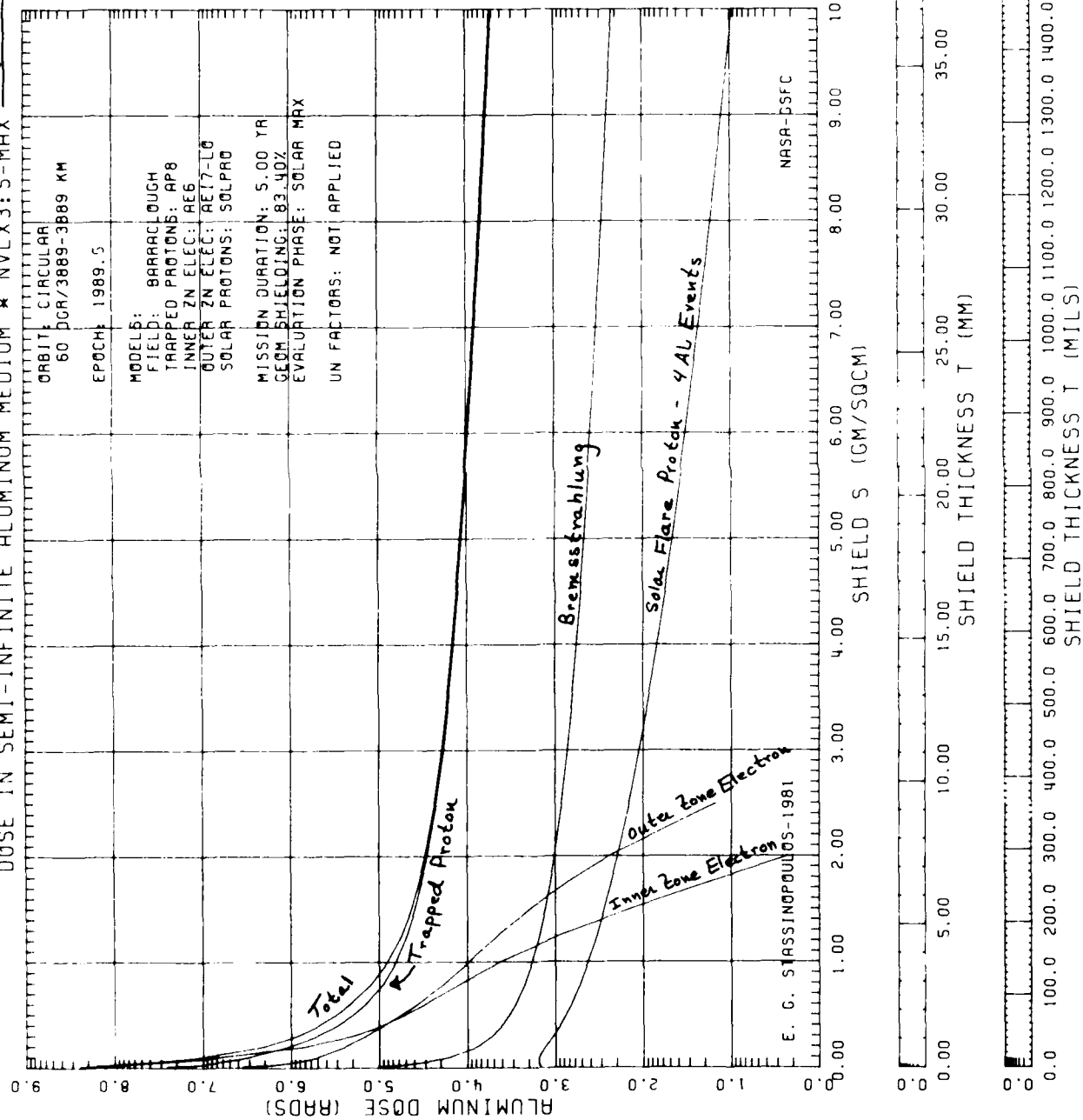


FIGURE 73

DOSE AT CENTER OF ALUMINUM SPHERES ** NVLX3: S-MAX

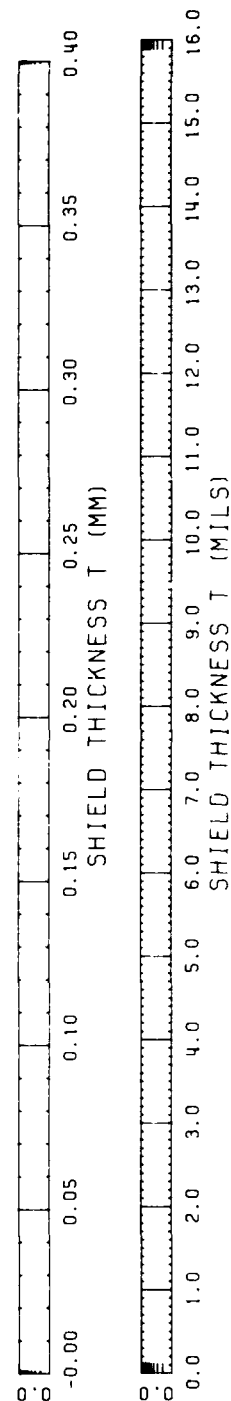
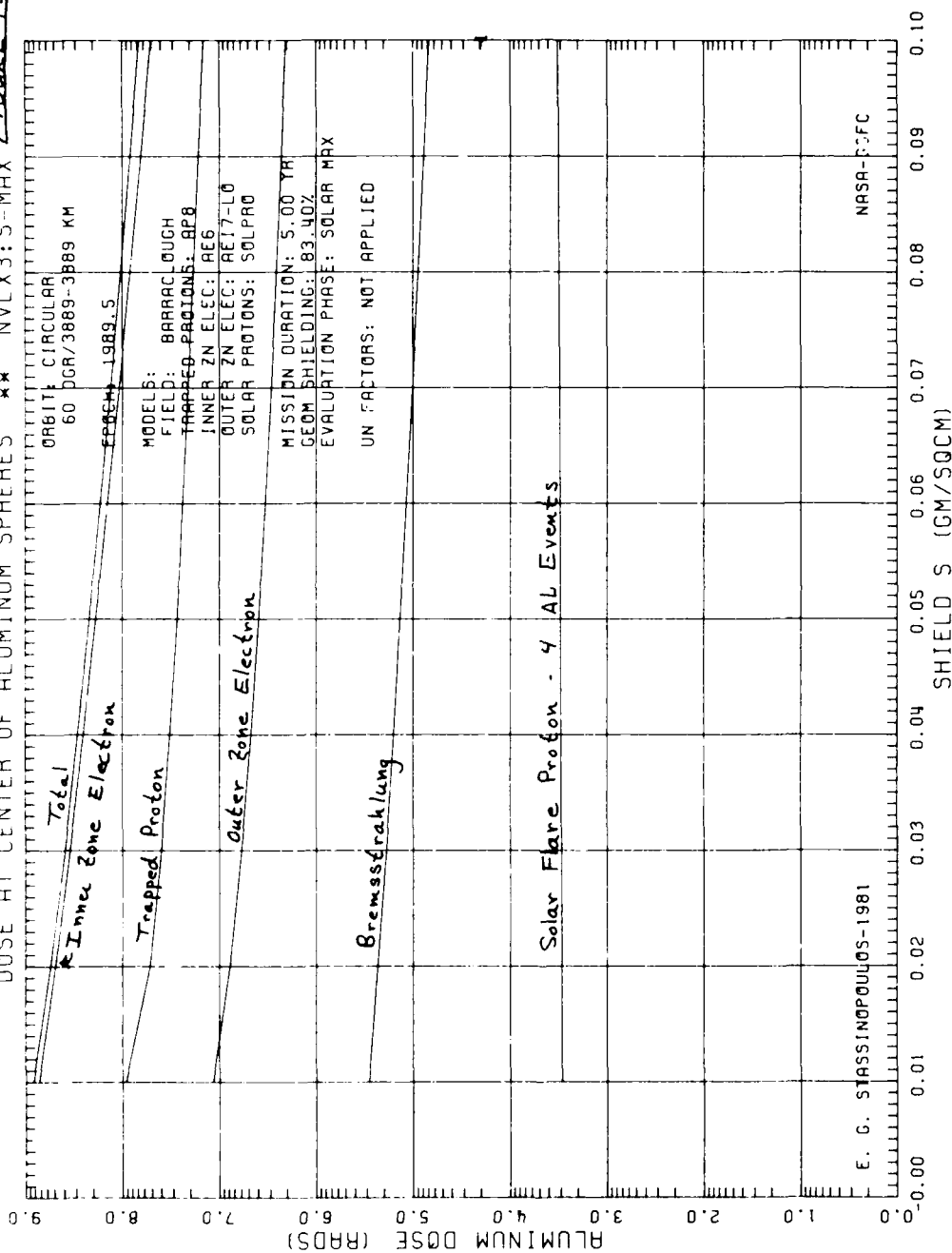


FIGURE 74

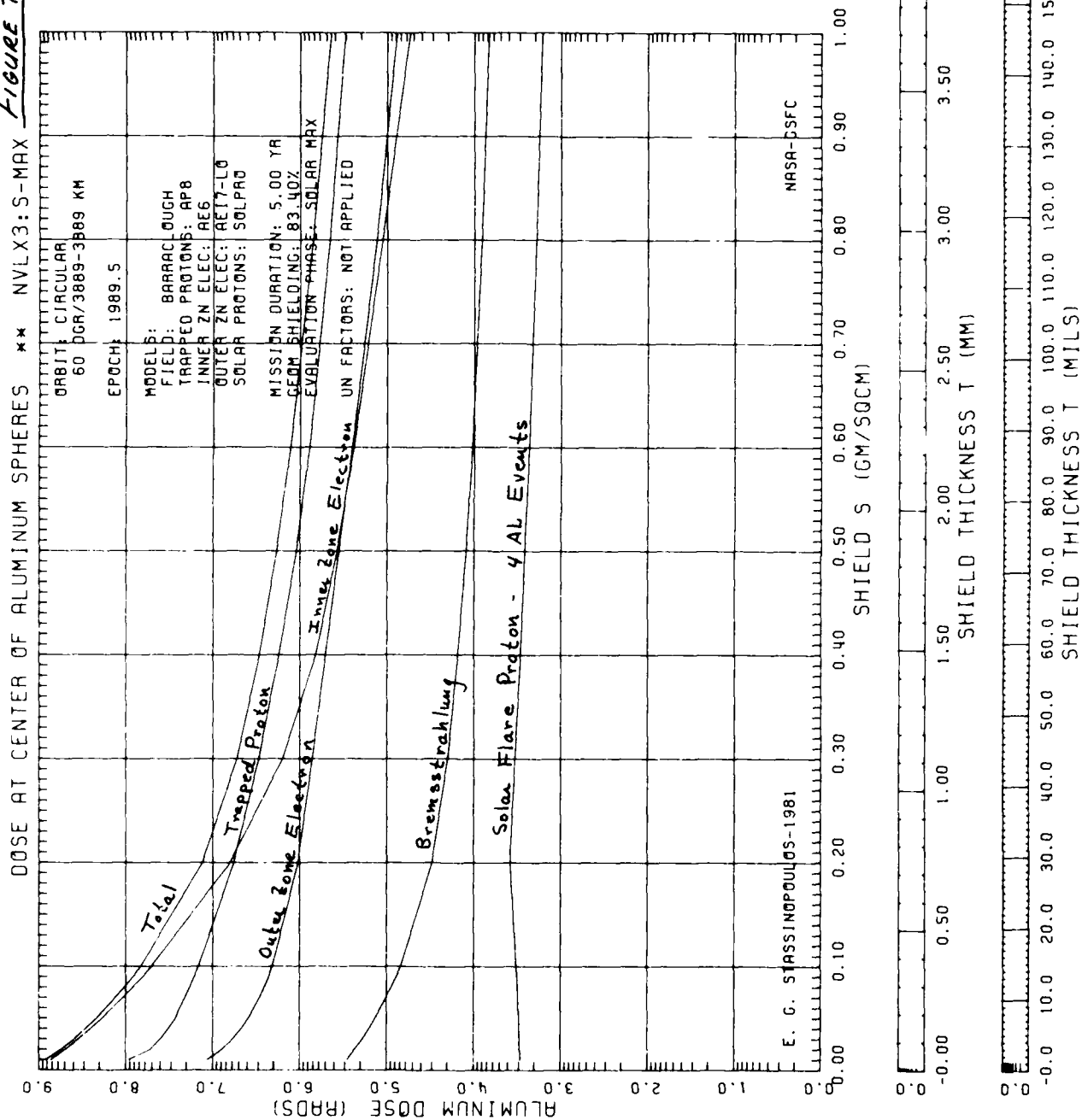


FIGURE 75

DOSE AT CENTER OF ALUMINUM SPHERES ** NVLX3:S-MAX

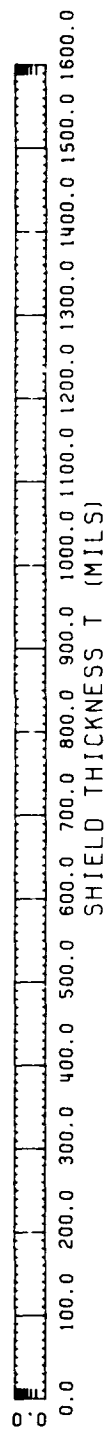
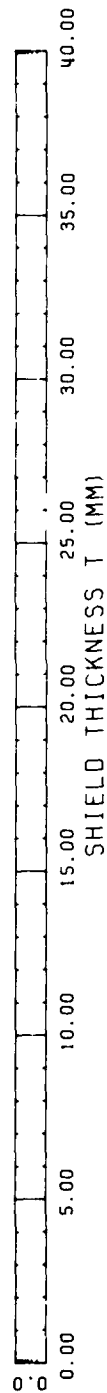
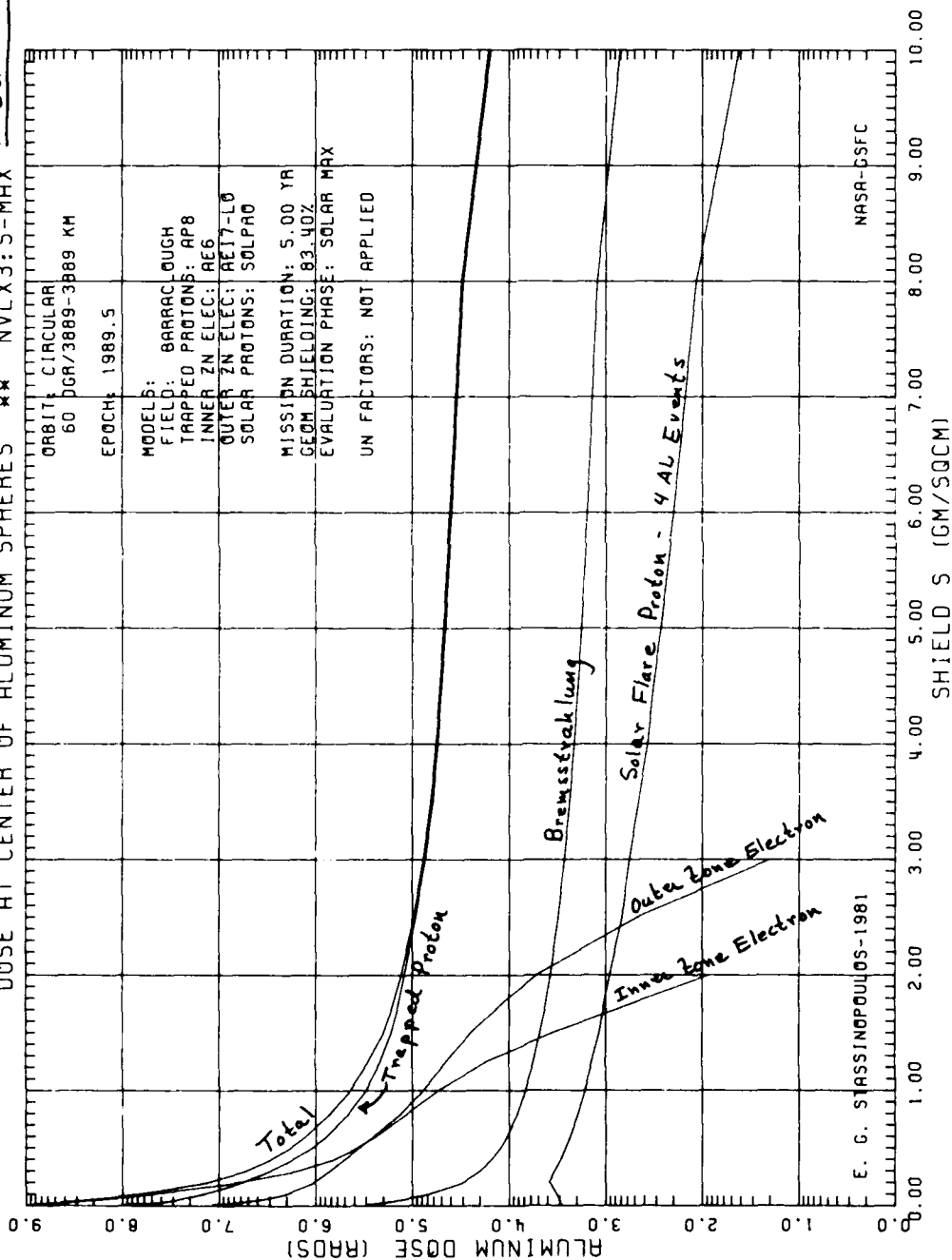


FIGURE 76

DOSE AT TRANSMISSION SURFACE OF FINITE ALUMINUM SLAB SHIELDS ** NVLX4:S-MAX

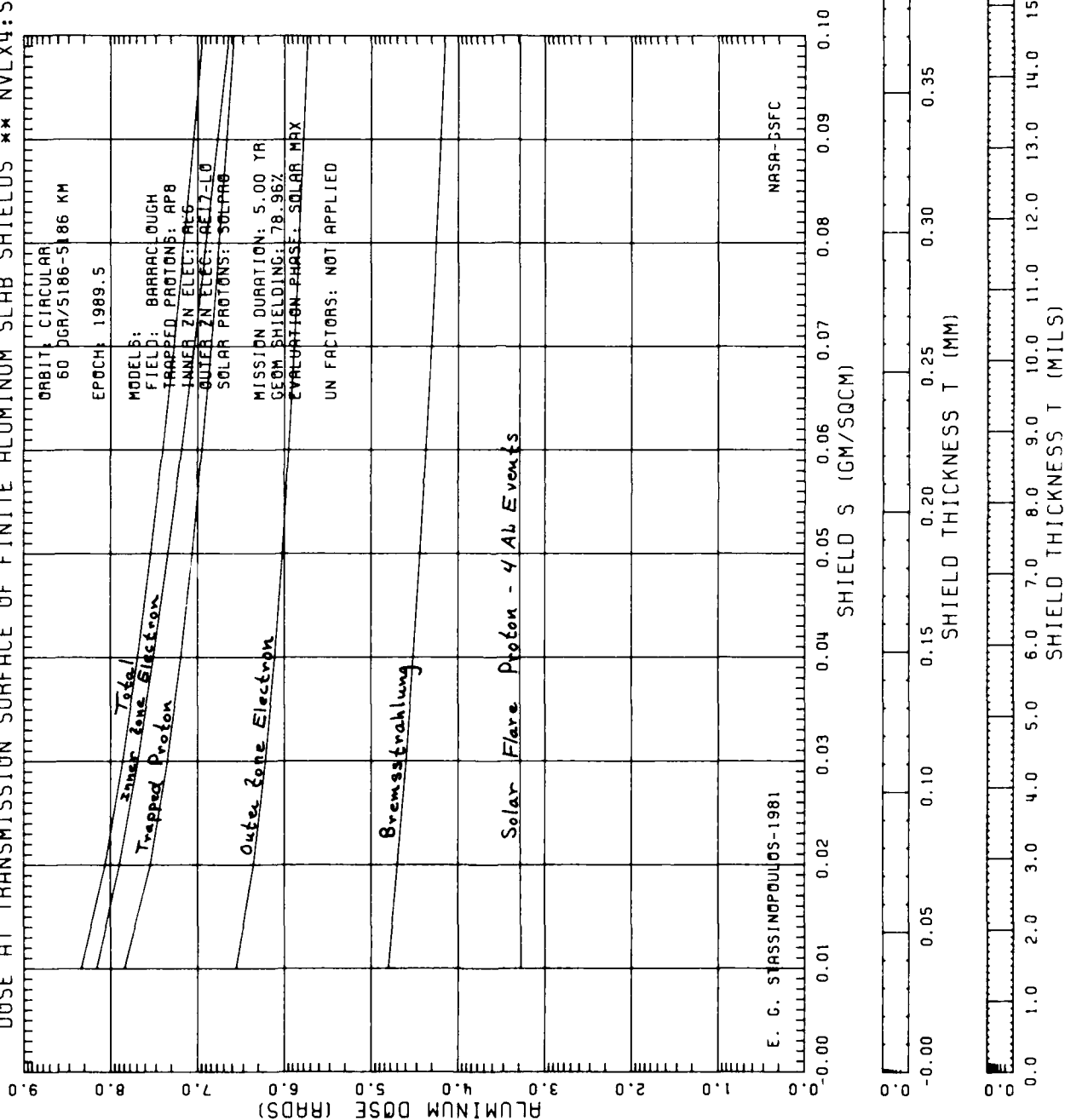


Figure 77

DOSE AT TRANSMISSION SURFACE OF FINITE ALUMINUM SLAB SHIELDS ** NVLX4: S-MAX

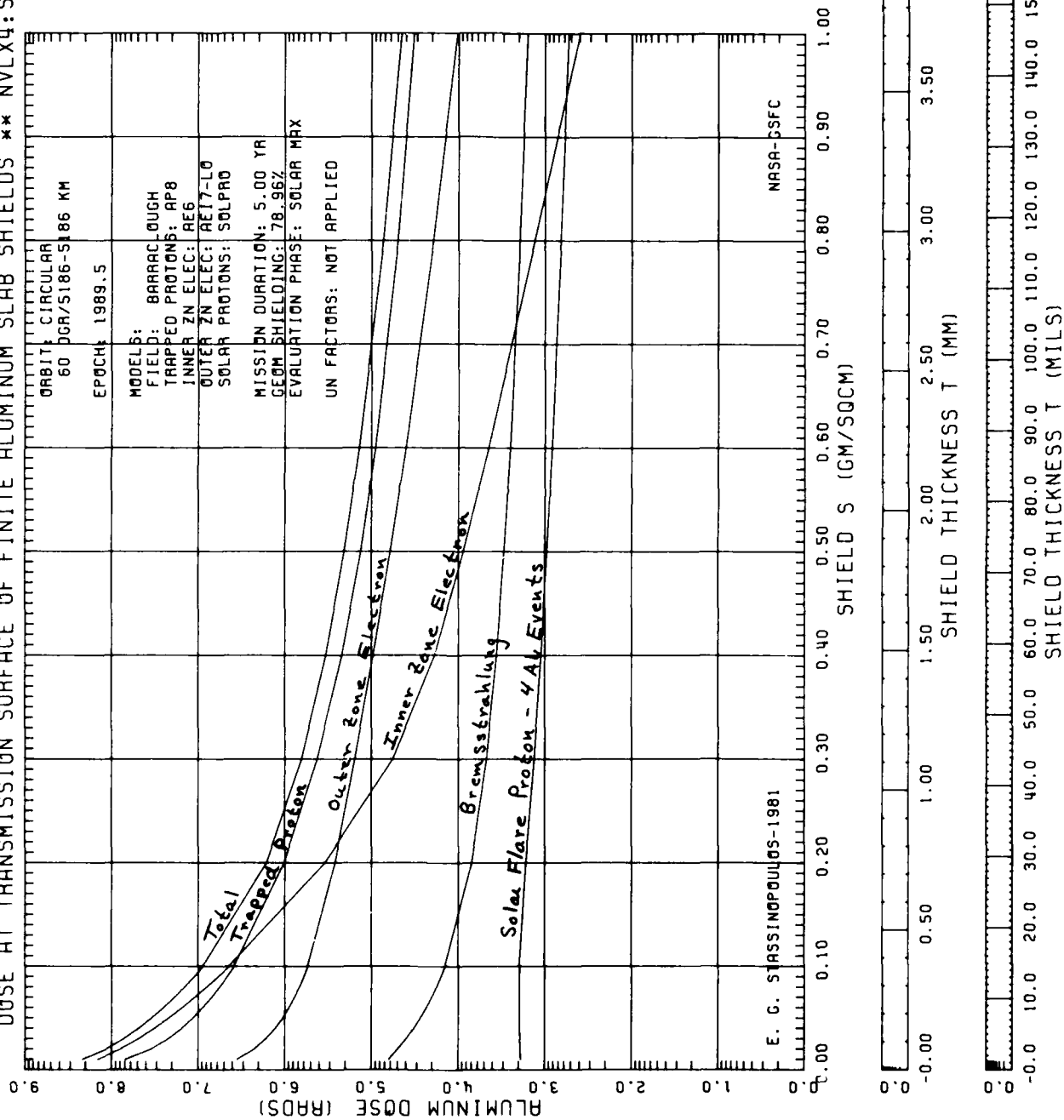


Figure 78

DOSE AT TRANSMISSION SURFACE OF FINITE ALUMINUM SLAB SHIELDS ** NVLX4: S-MAX

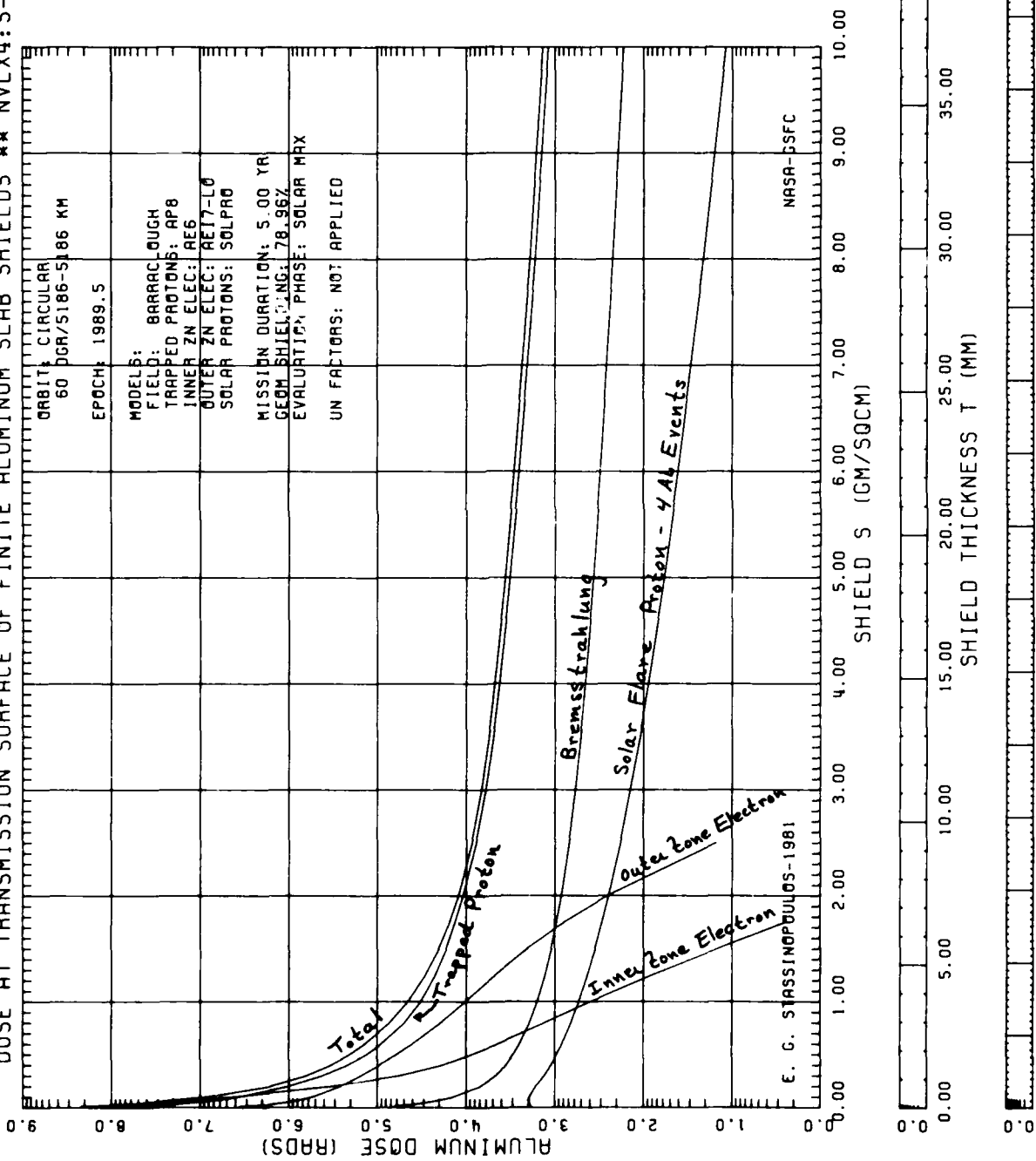


FIGURE 79

DOSE IN SEMI-INFINITE ALUMINUM MEDIUM * NVLX4:S-MAX

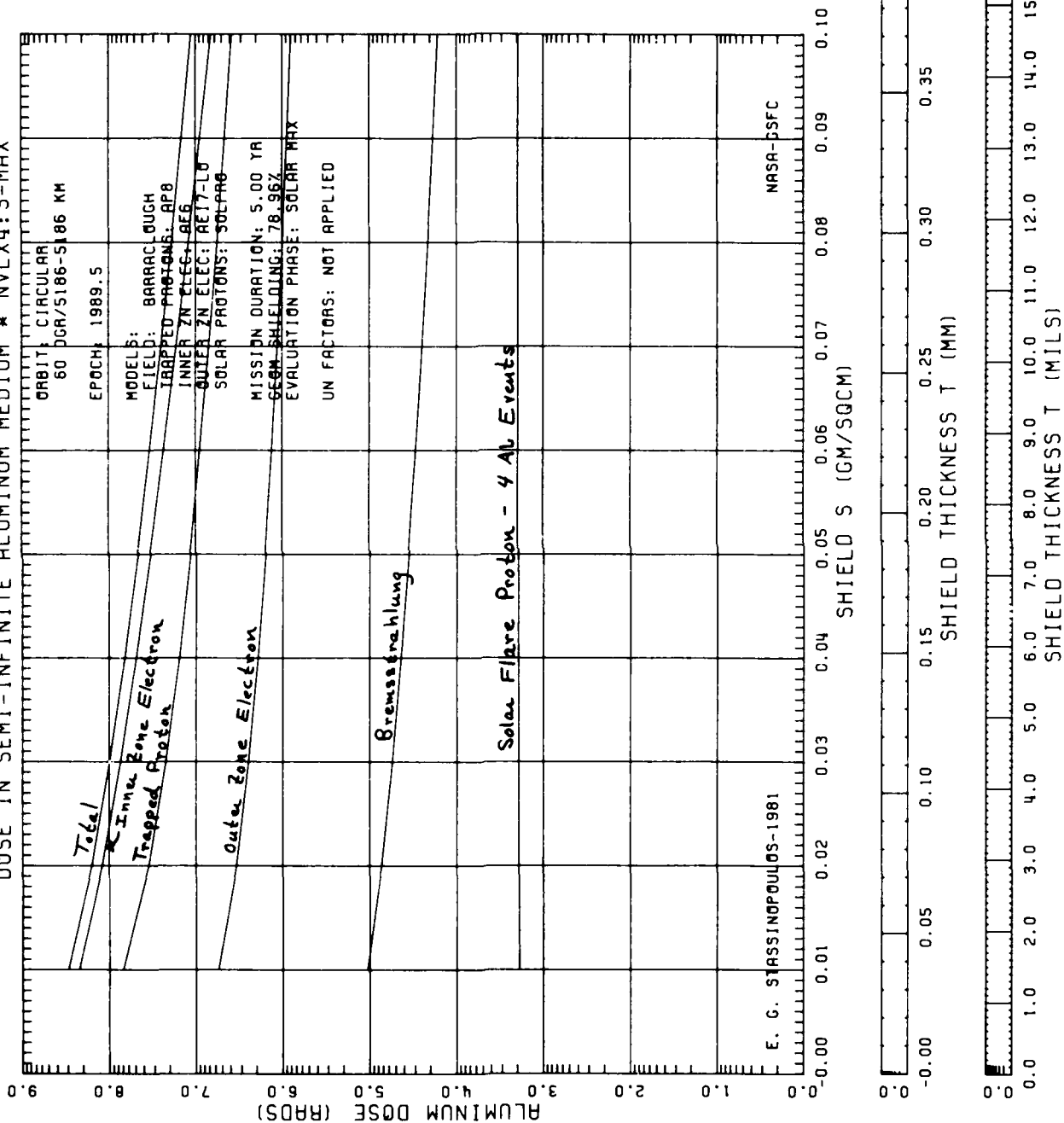


Figure 80

DOSE IN SEMI-INFINITE ALUMINUM MEDIUM * NVLX4:S-MAX

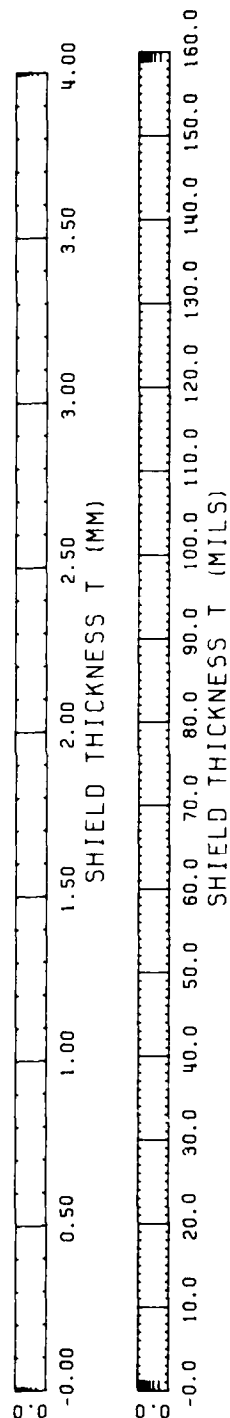
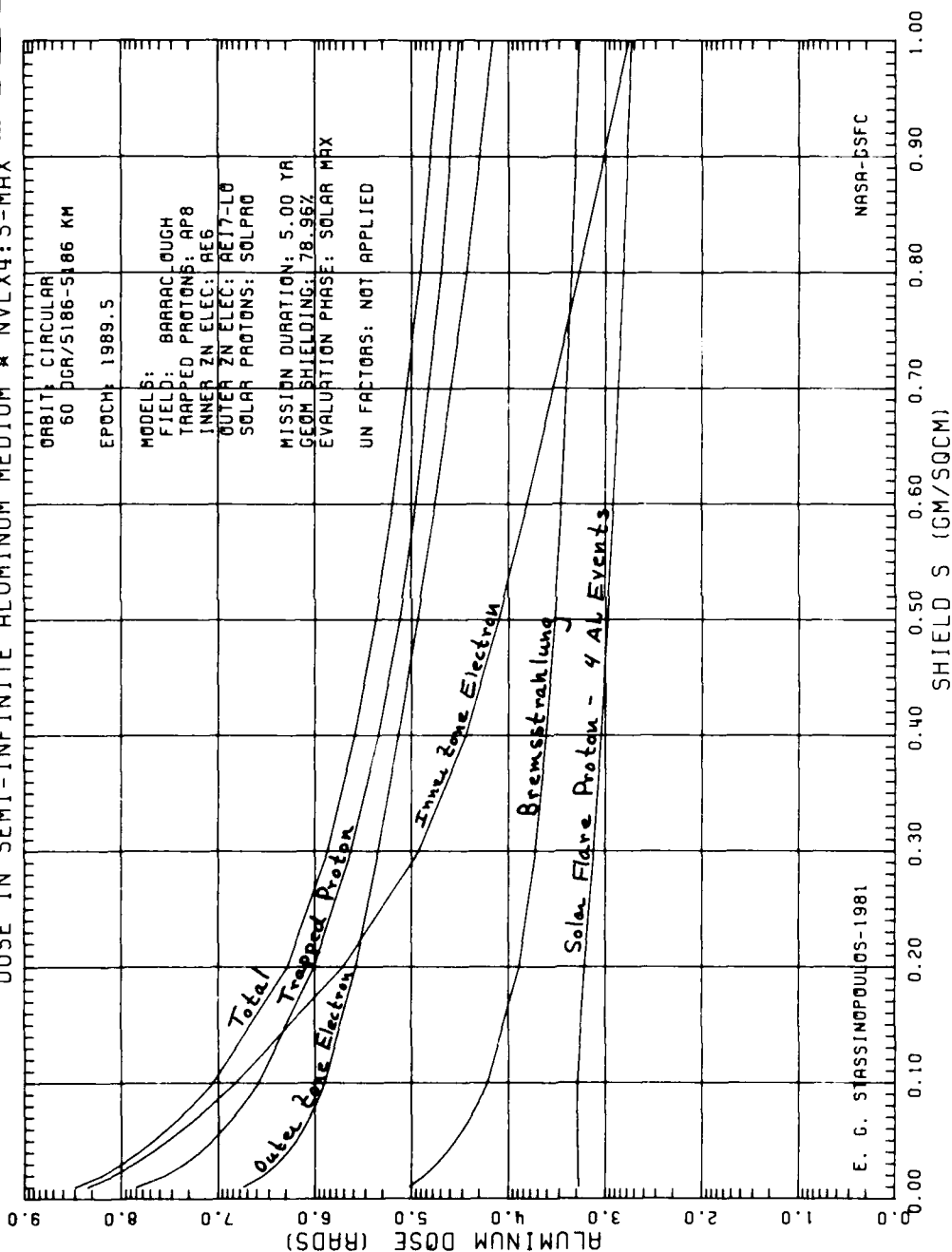
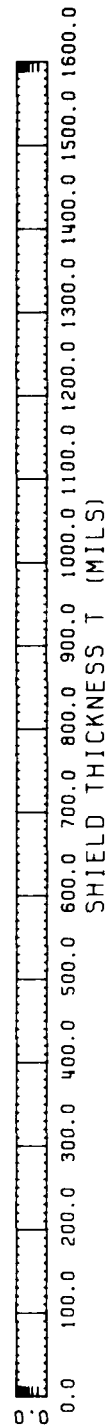
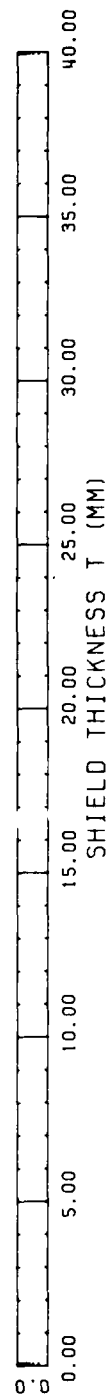
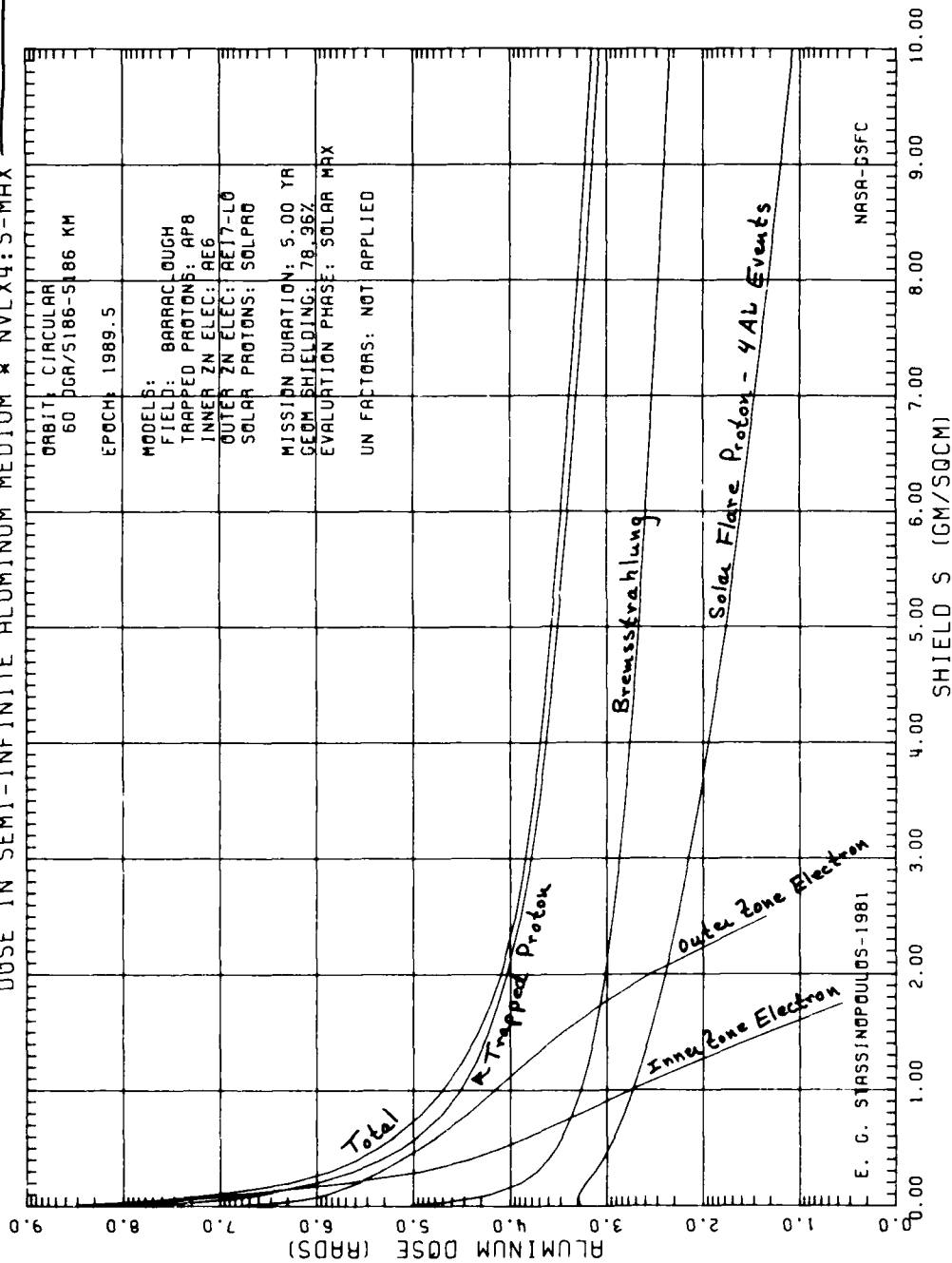


FIGURE 81

DOSE IN SEMI-INFINITE ALUMINUM MEDIUM * NVLX4:S-MAX



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FIGURE 82

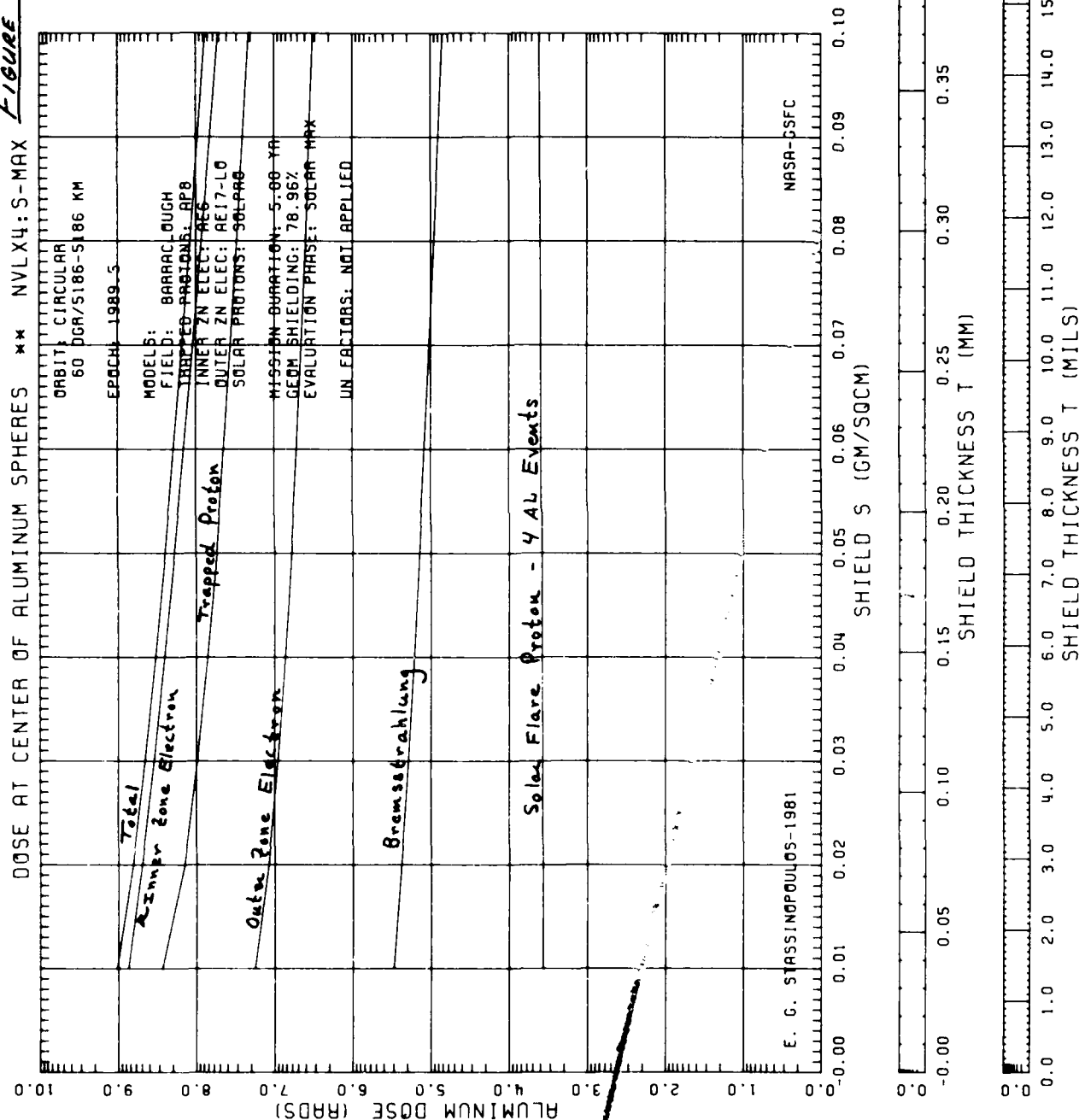
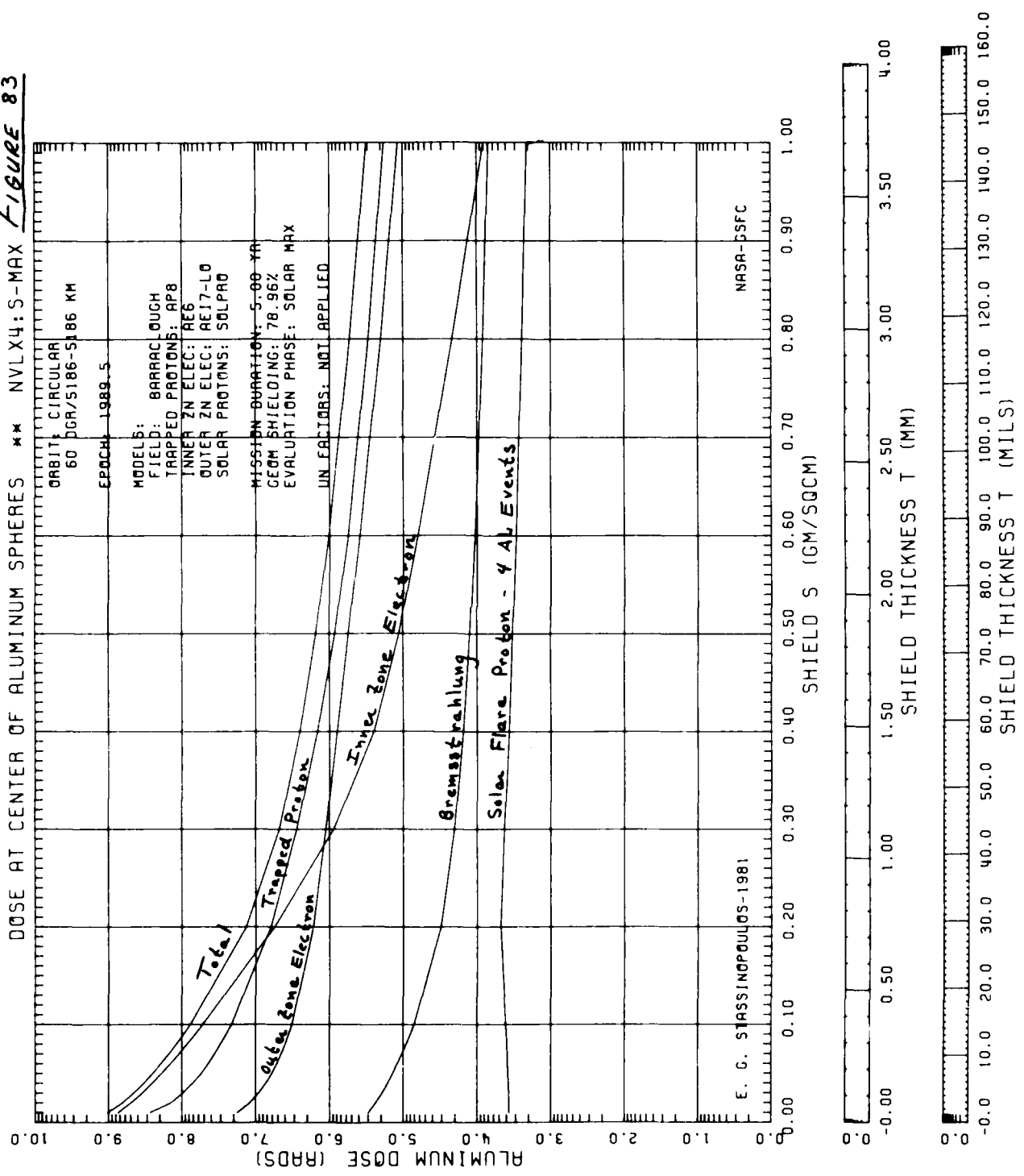


FIGURE 83



DOSE AT CENTER OF ALUMINUM SPHERES ** NVLX4: S-MAX *FIGURE 84*

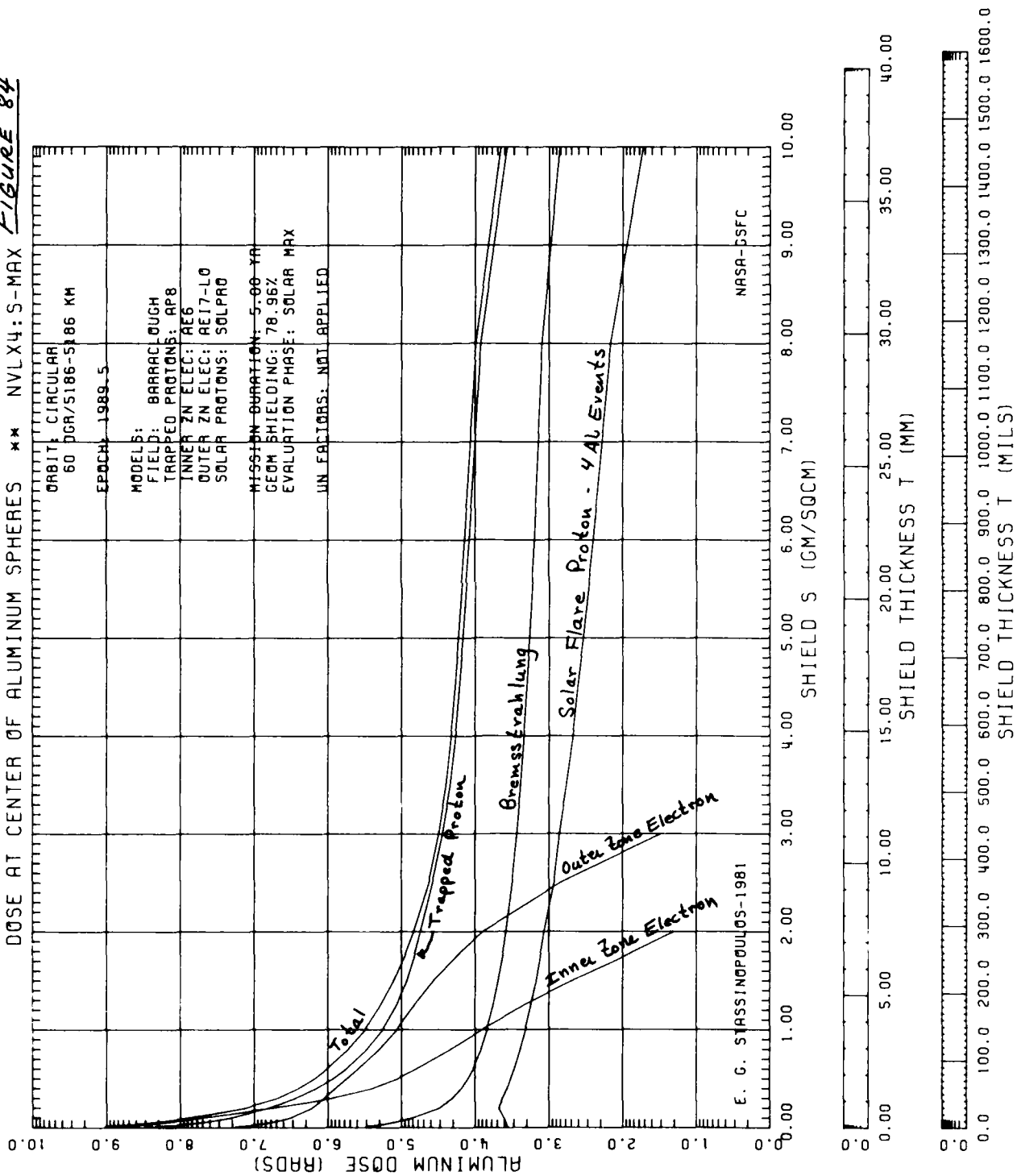


FIGURE 85

DOSE AT TRANSMISSION SURFACE OF FINITE ALUMINUM SLAB SHIELDS ** NVLX5:S-MAX

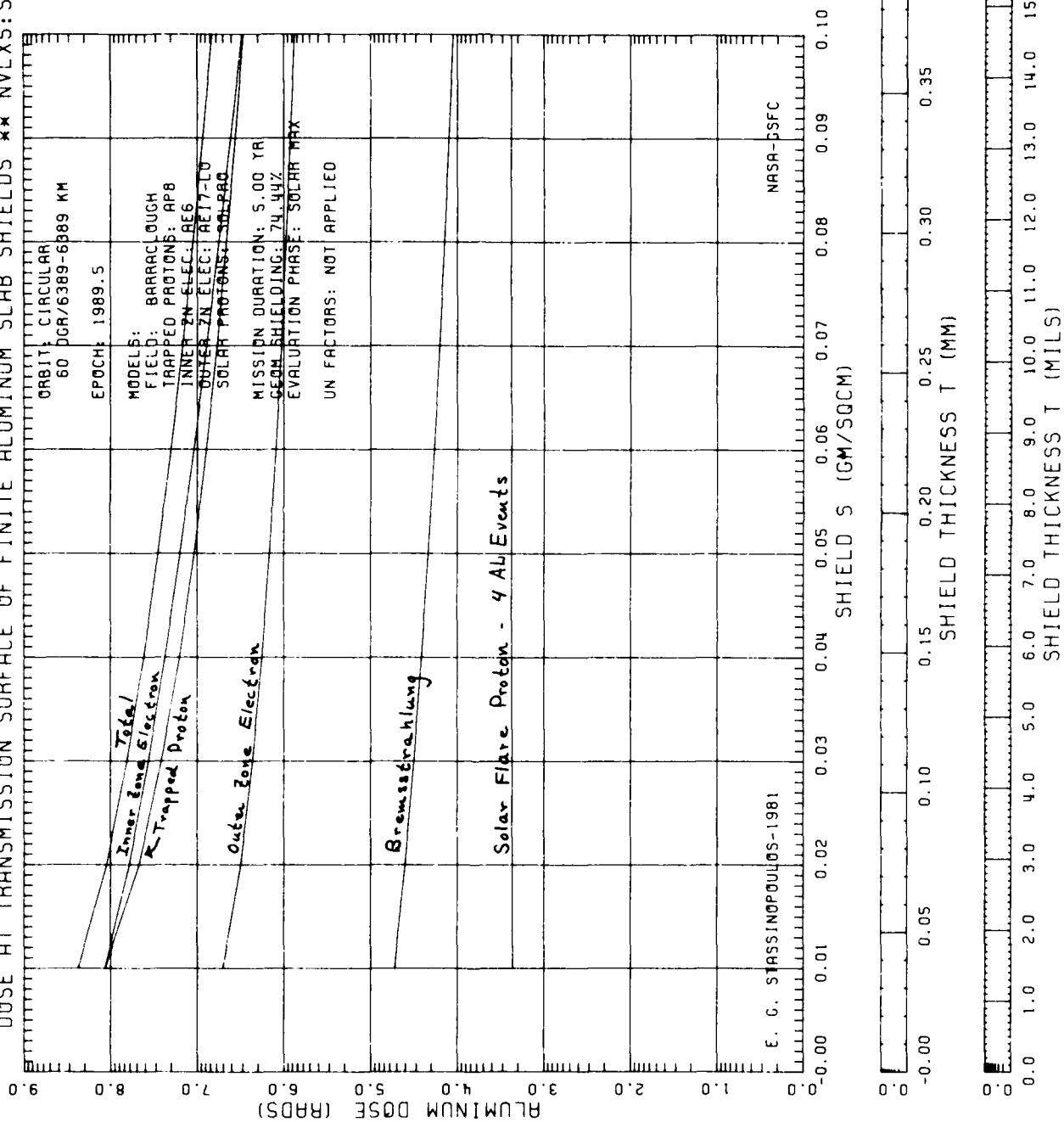


FIGURE 86

DOSE AT TRANSMISSION SURFACE OF FINITE ALUMINUM SLAB SHIELDS ** NVLX5:S-MAX

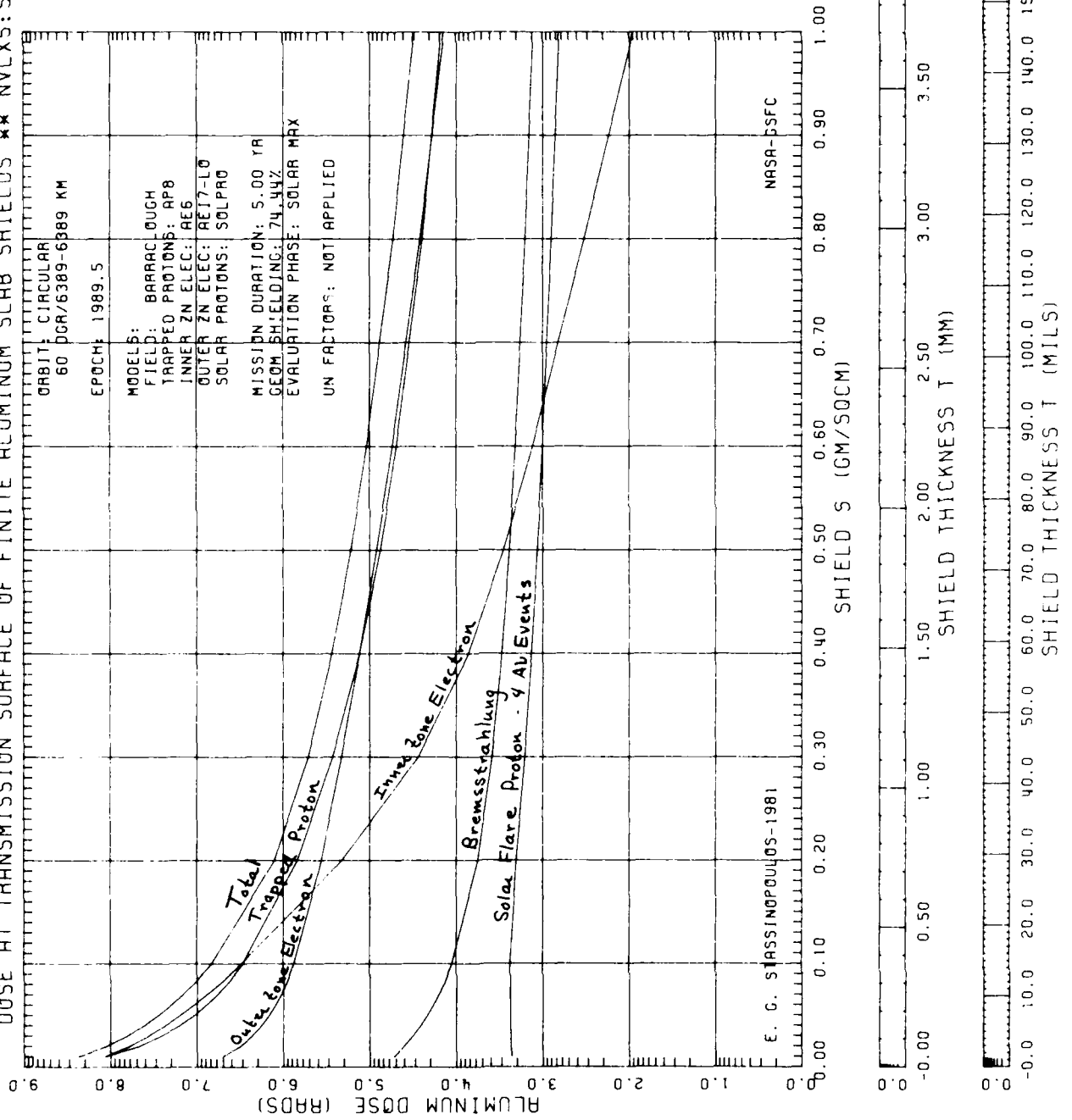


Figure 87

DOSE AT TRANSMISSION SURFACE OF FINITE ALUMINUM SLAB SHIELDS ** NVLX5: S-MAX

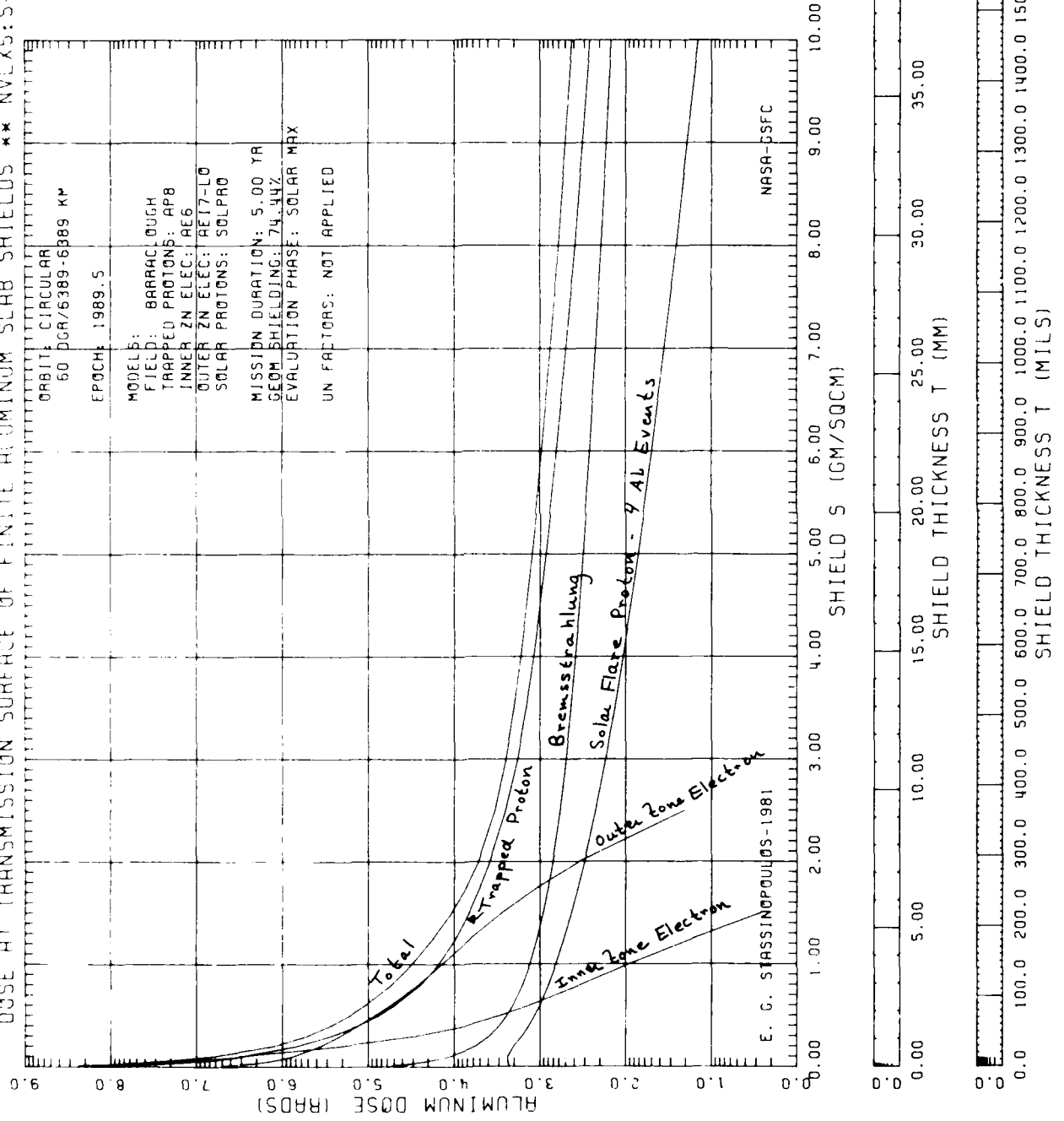


FIGURE 88

DOSE IN SEMI-INFINITE ALUMINUM MEDIUM * NVLX5:S-MAX

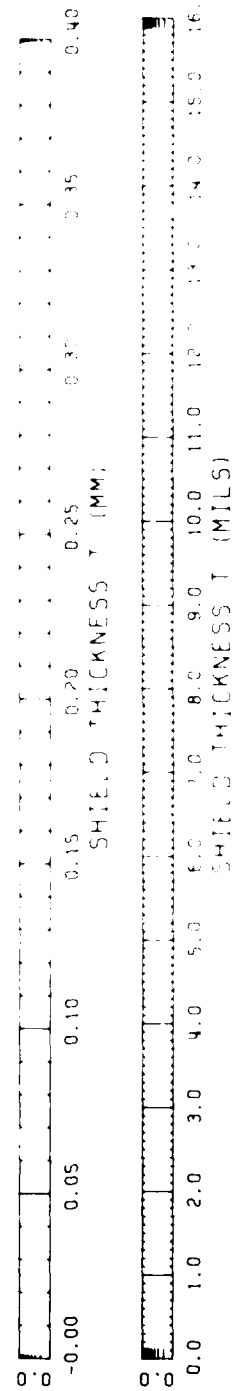
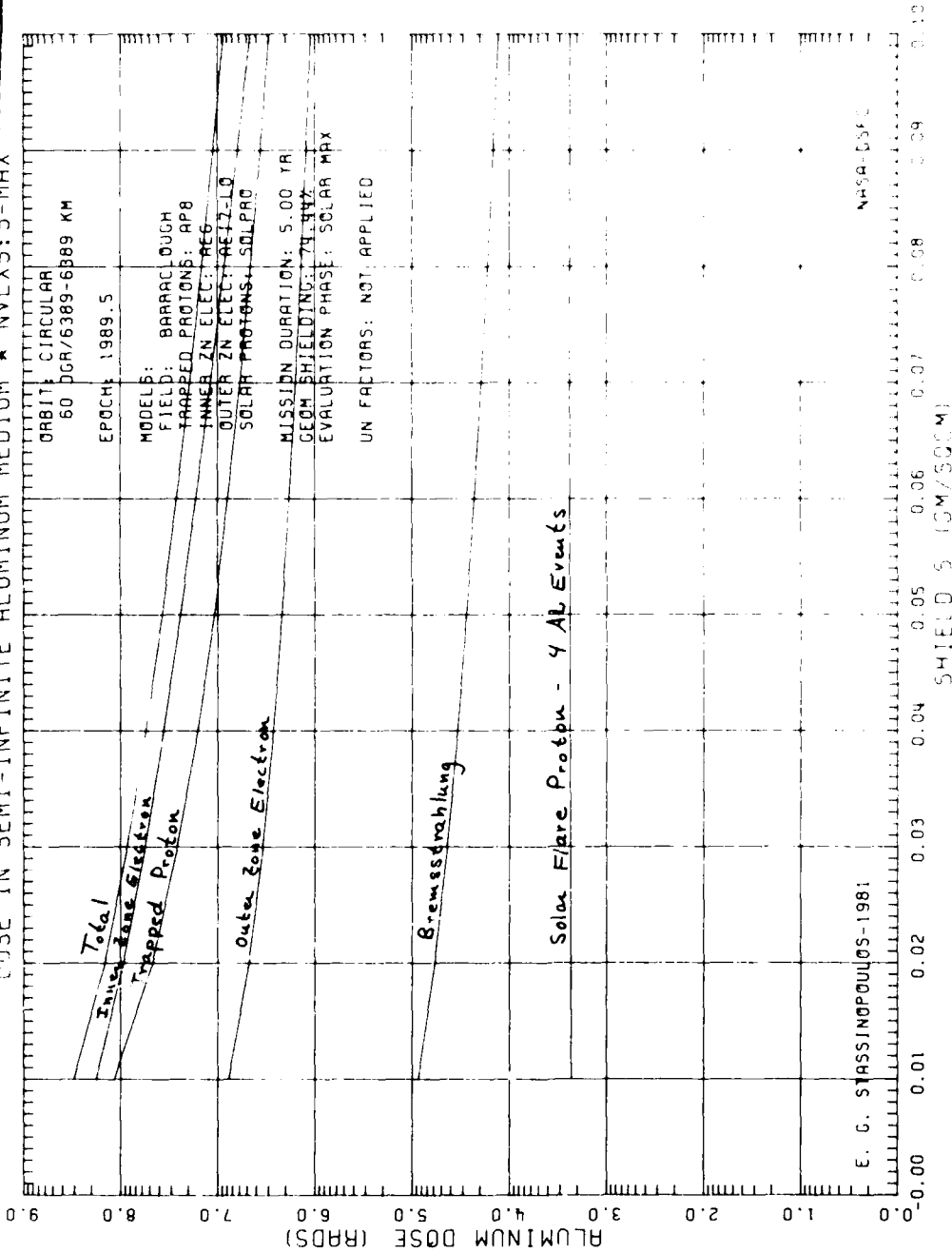


FIGURE 89

DOSE IN SEMI-INFINITE ALUMINUM MEDIUM * NVLX5: S MAX

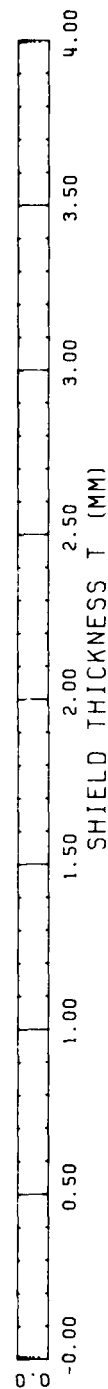
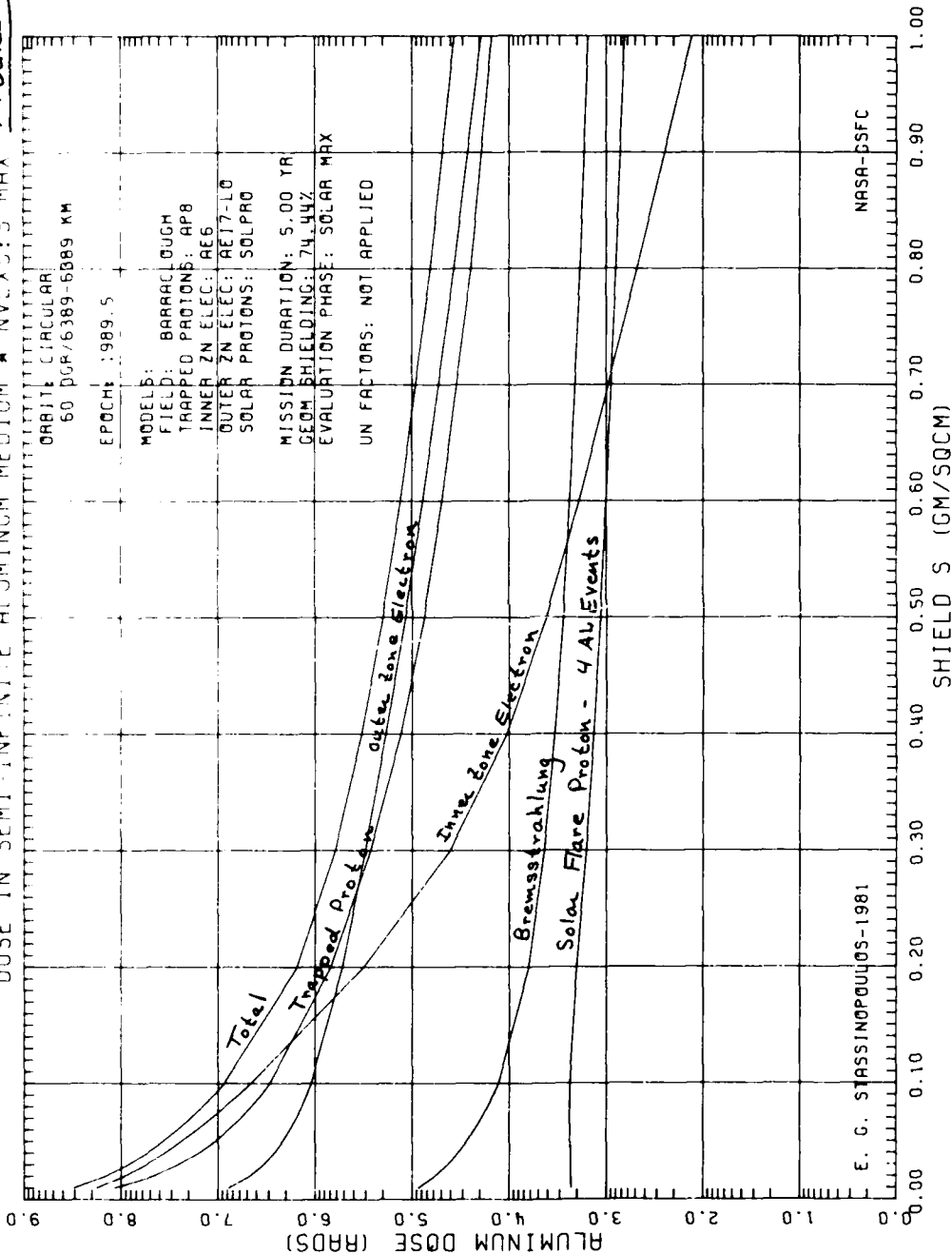
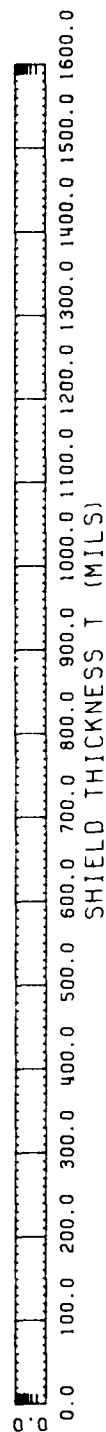
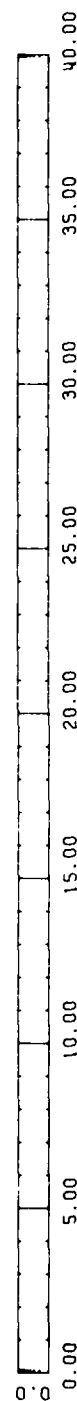
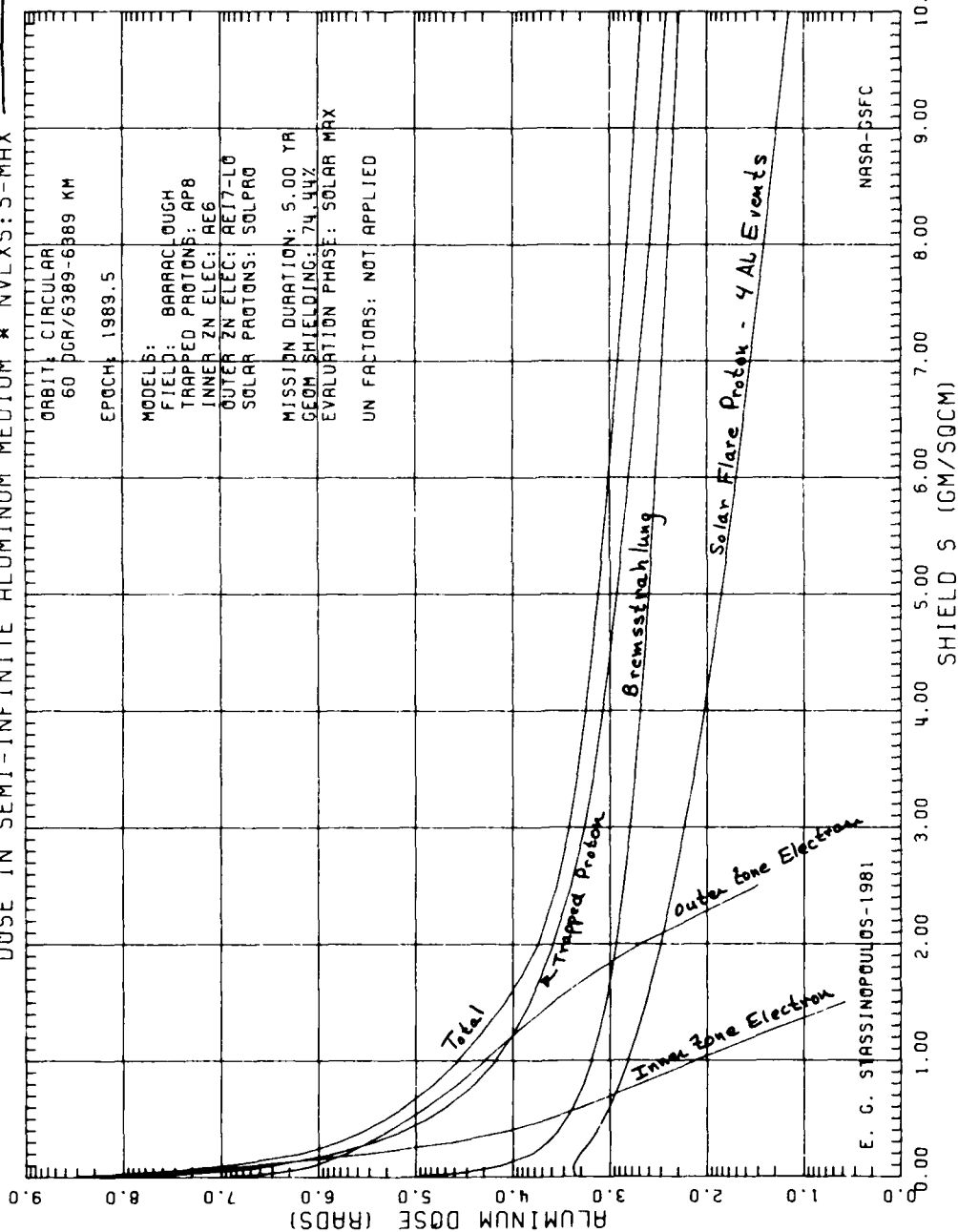


Figure 90

DOSE IN SEMI-INFINITE ALUMINUM MEDIUM * NVLX5: S-MAX



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Figure 91

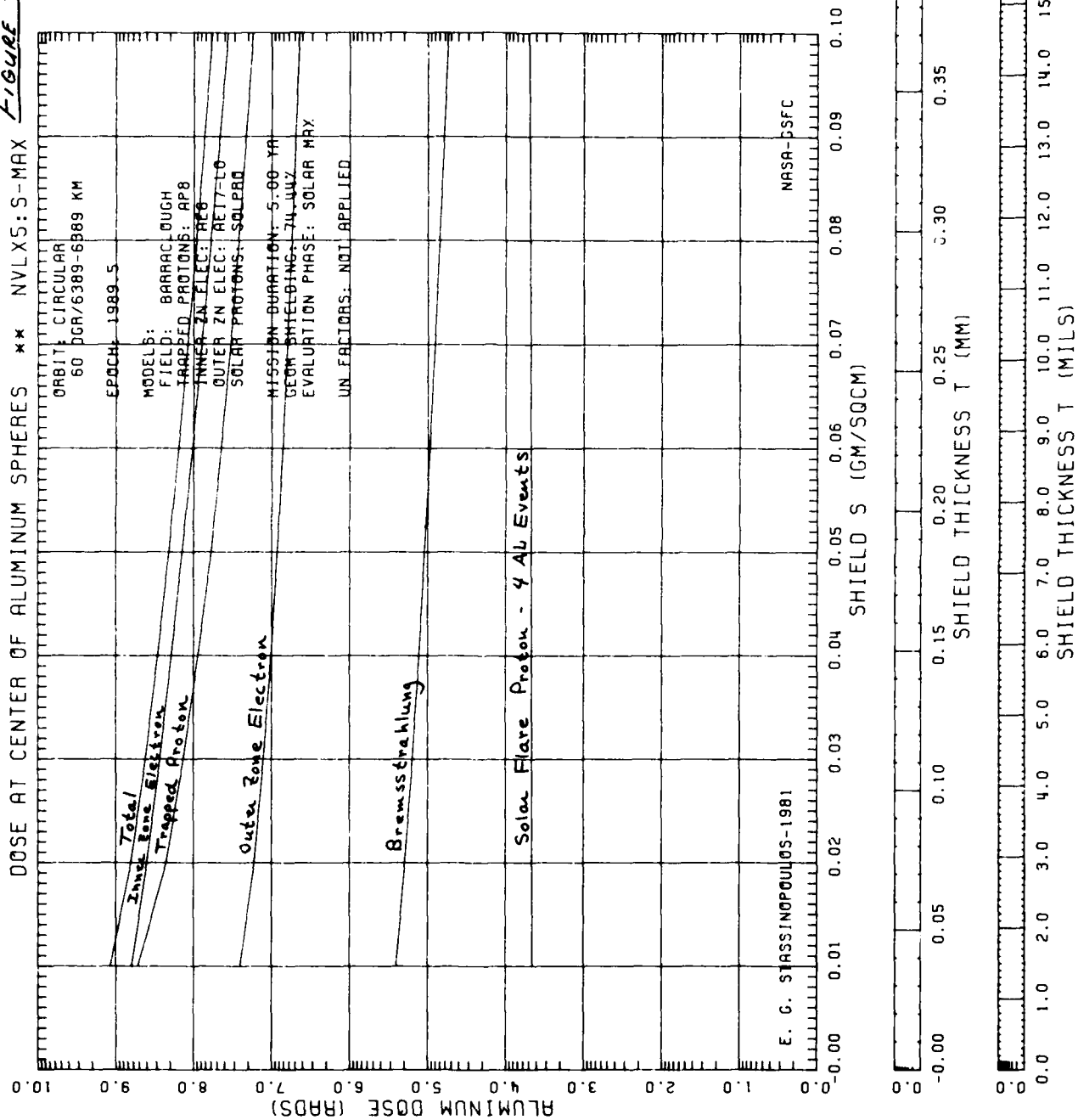


FIGURE 92

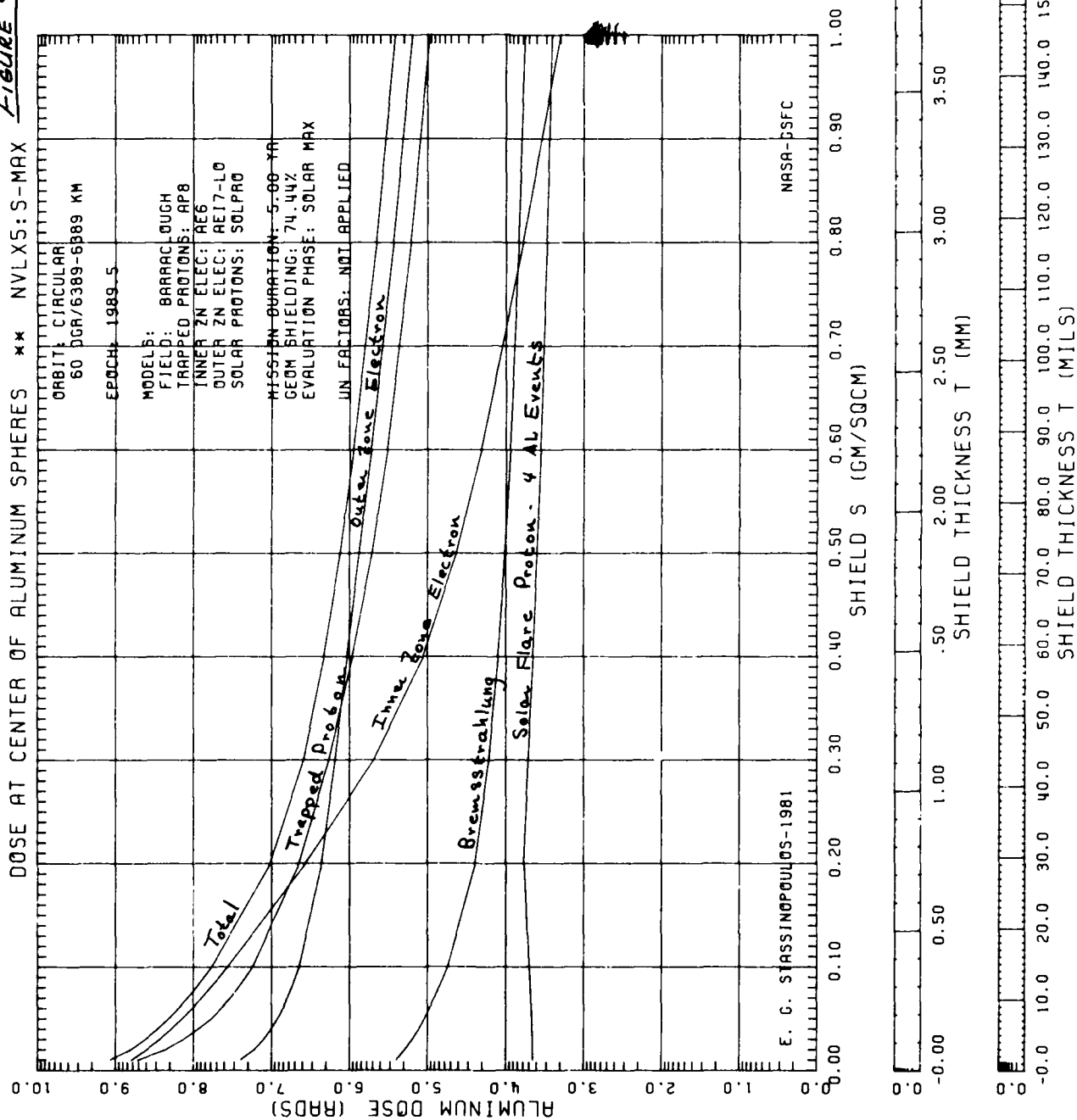


Figure 93

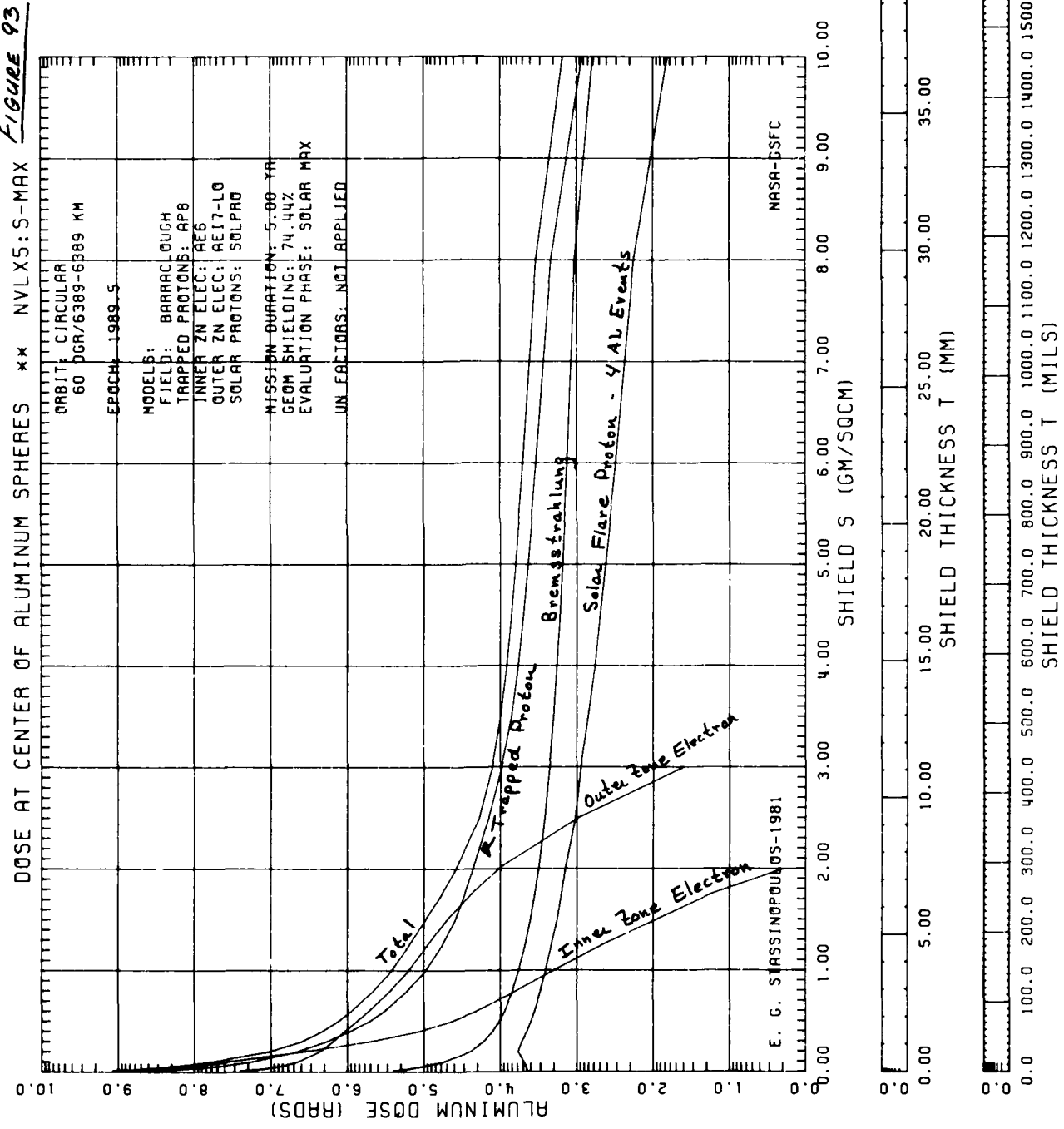


Figure 94

DOSE AT TRANSMISSION SURFACE OF FINITE ALUMINUM SLAB SHIELDS ** NVLX6:S-MAX

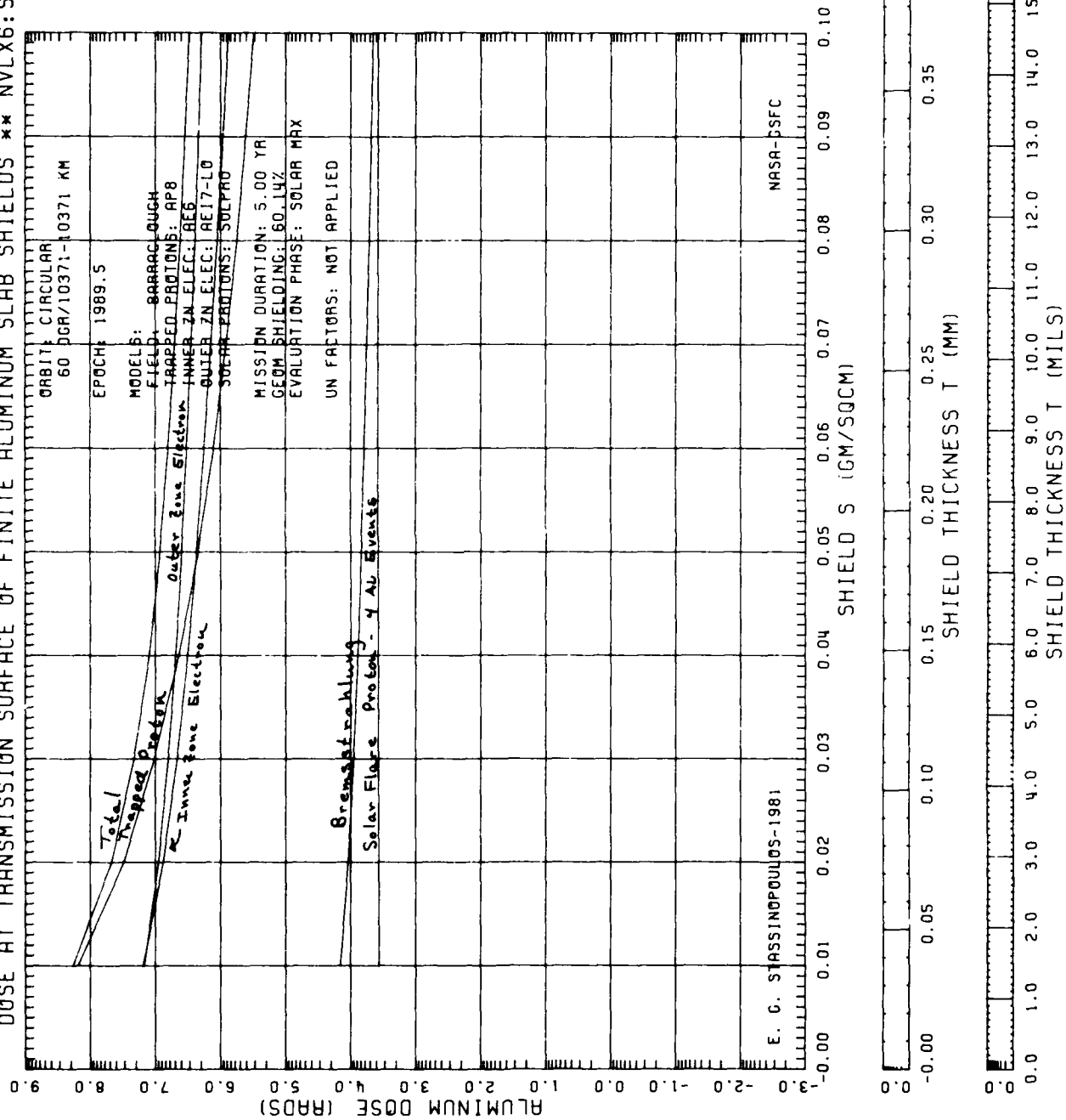


Figure 95

DOSE AT TRANSMISSION SURFACE OF FINITE ALUMINUM SLAB SHIELDS ** NVLX6: S-MAX

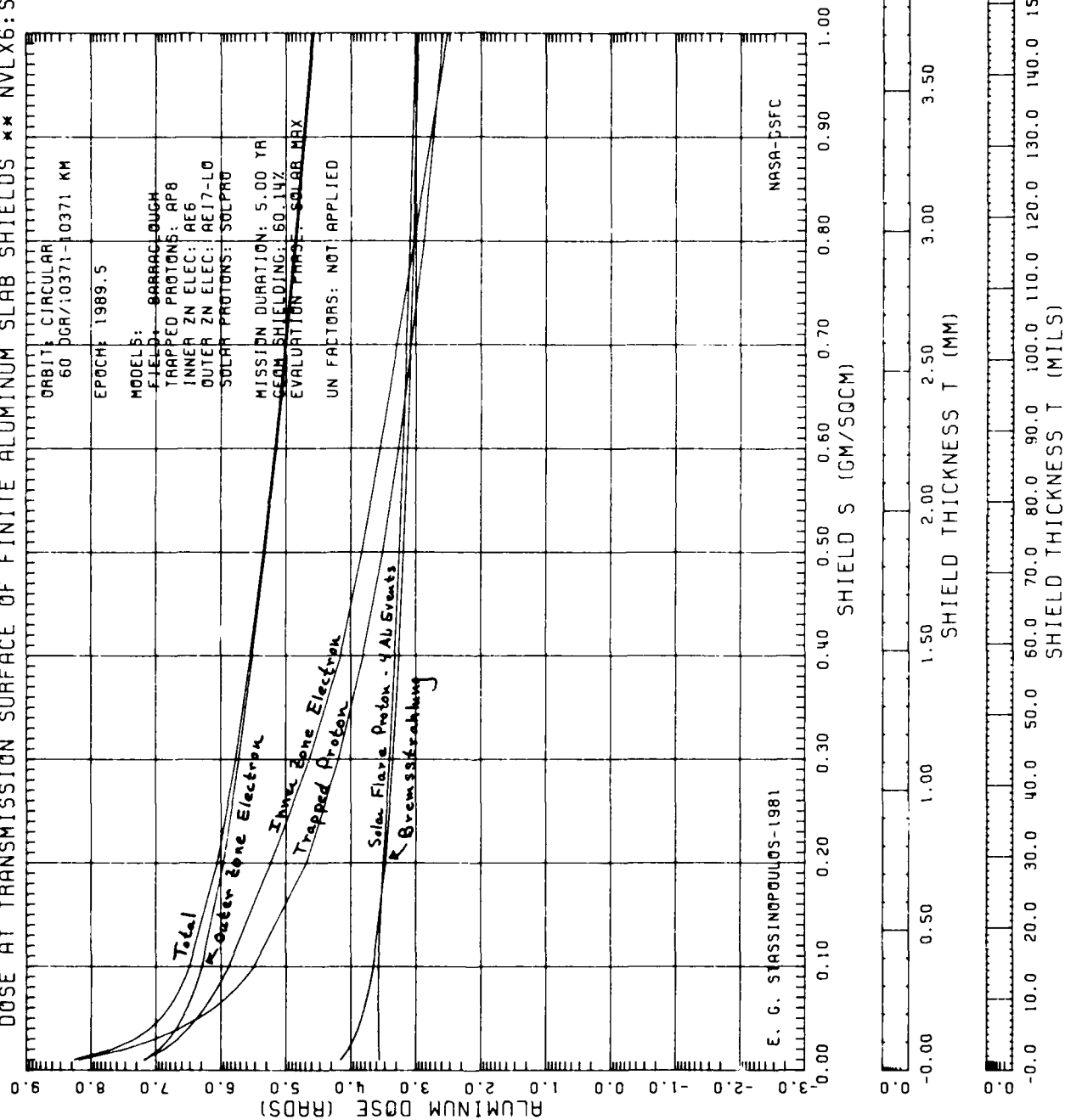


Figure 96

DOSE AT TRANSMISSION SURFACE OF FINITE ALUMINUM SLAB SHIELDS ** NVLX6: S-MAX

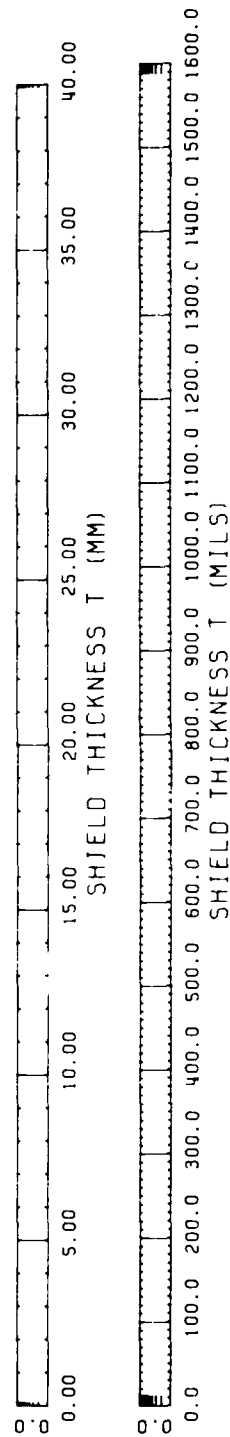
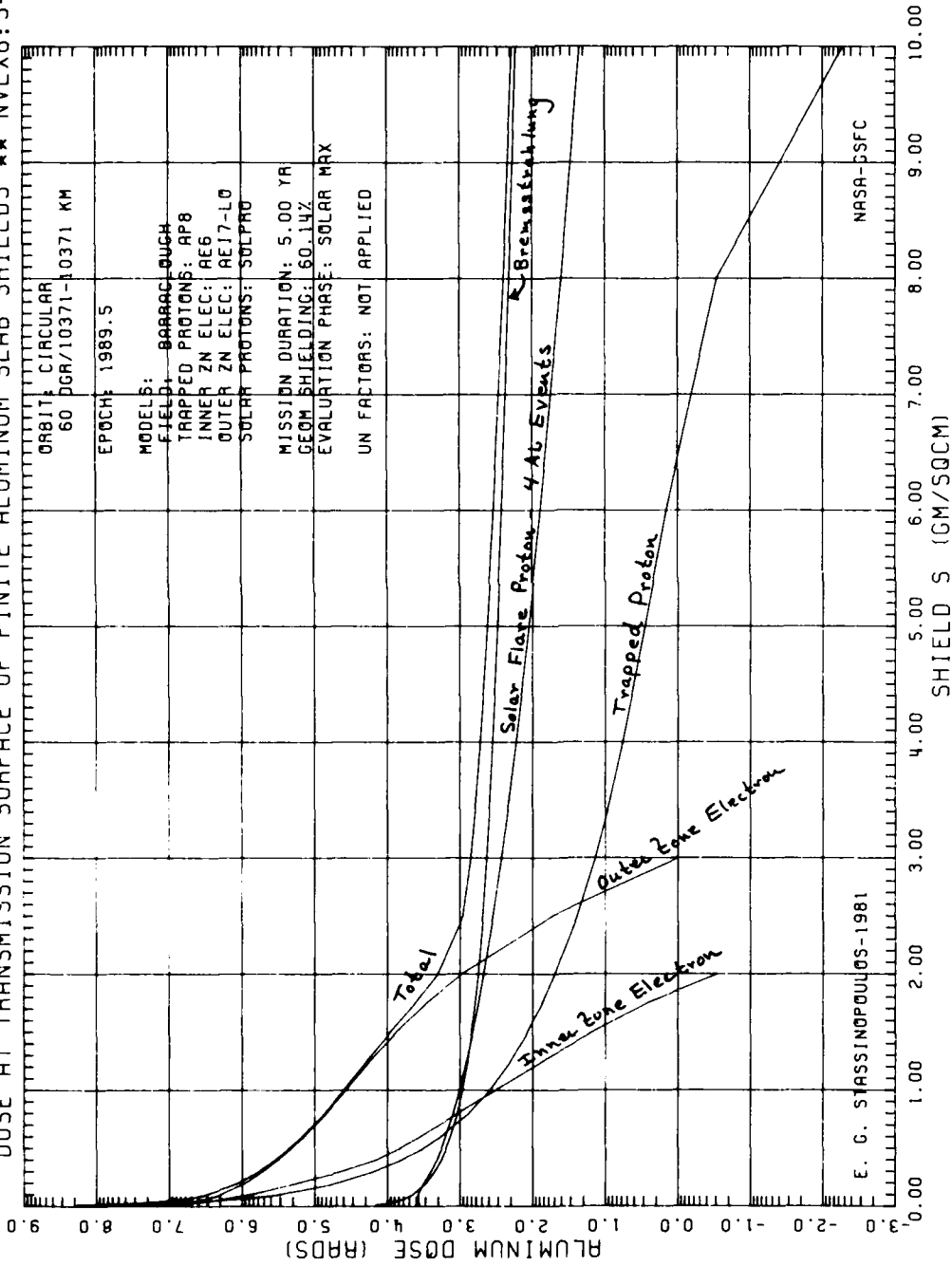
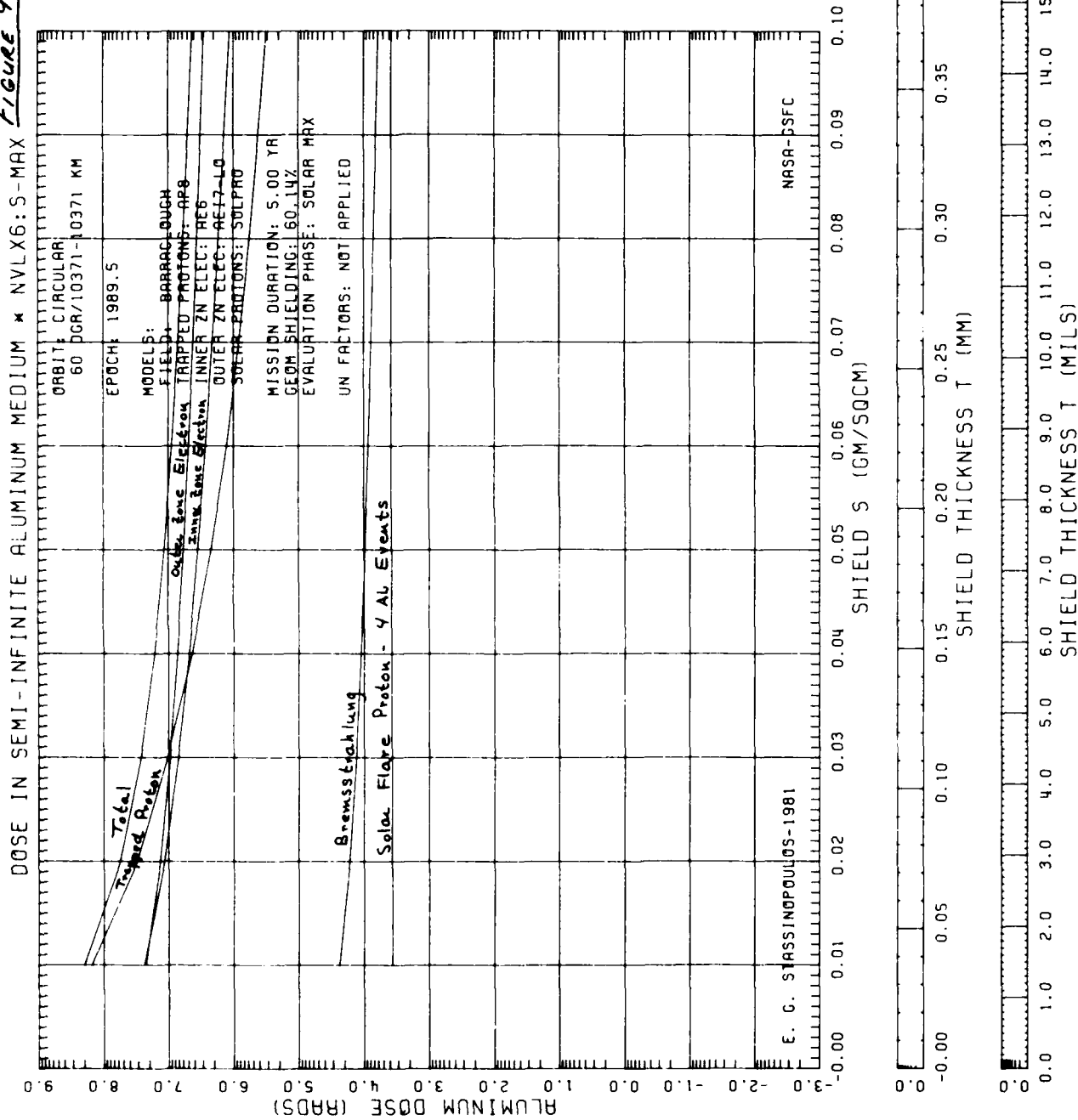


Figure 97



DOSE IN SEMI-INFINITE ALUMINUM MEDIUM * NVLX6:S-MAX **FIGURE 98**

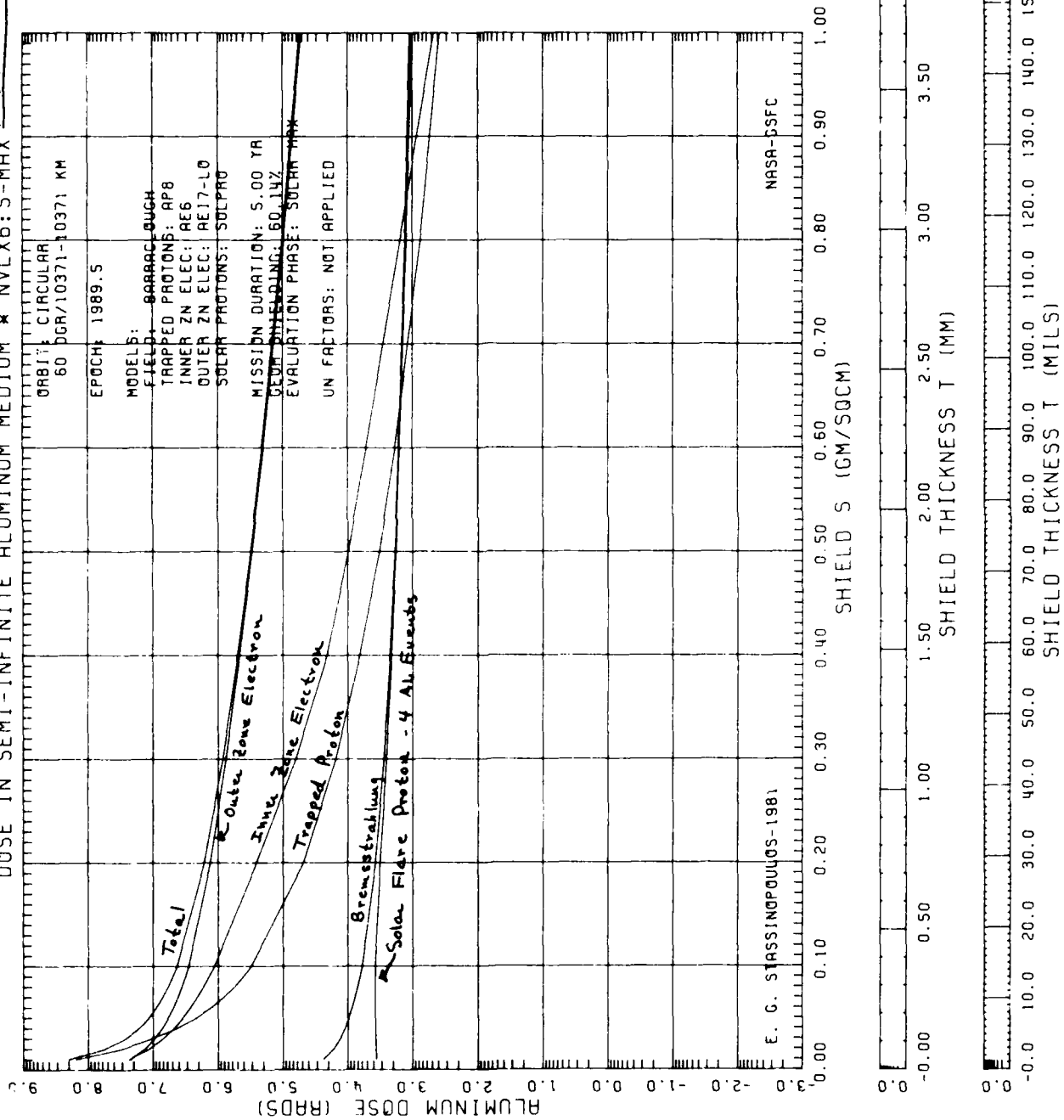
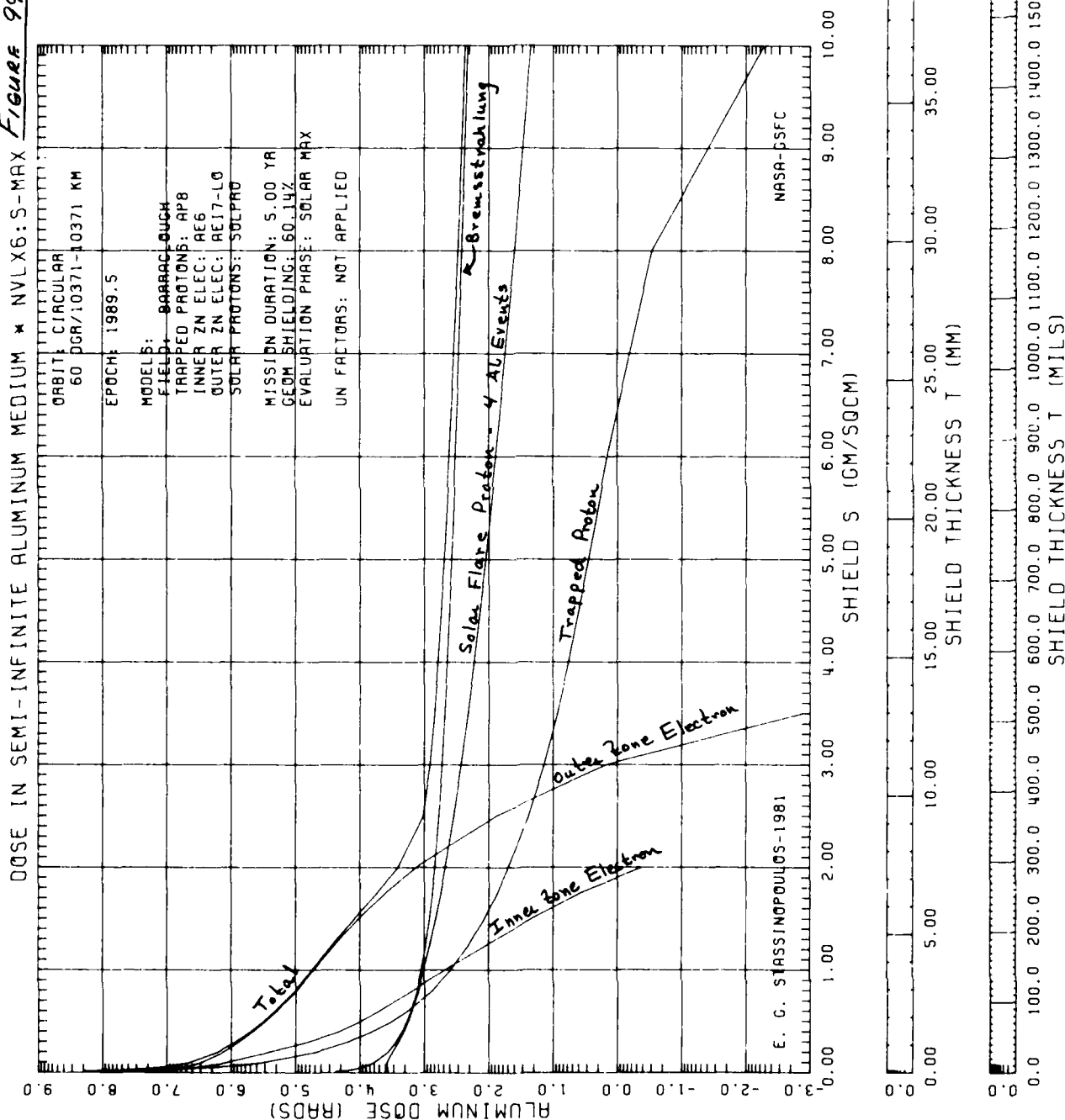


Figure 99



DOSE AT CENTER OF ALUMINUM SPHERES ** NVLX6:S-MAX Figure 100

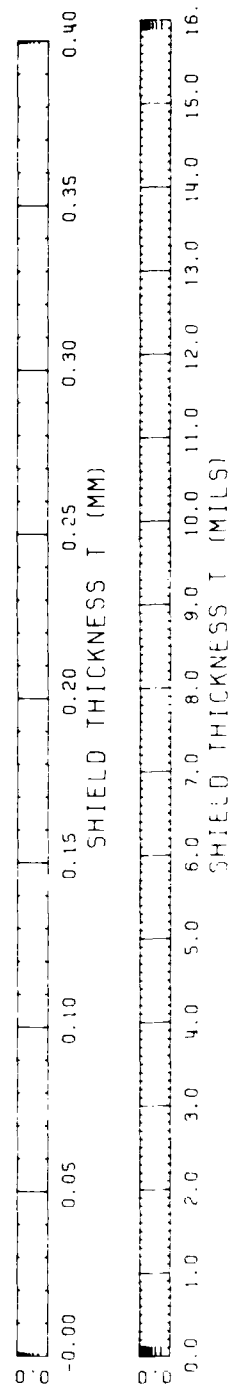
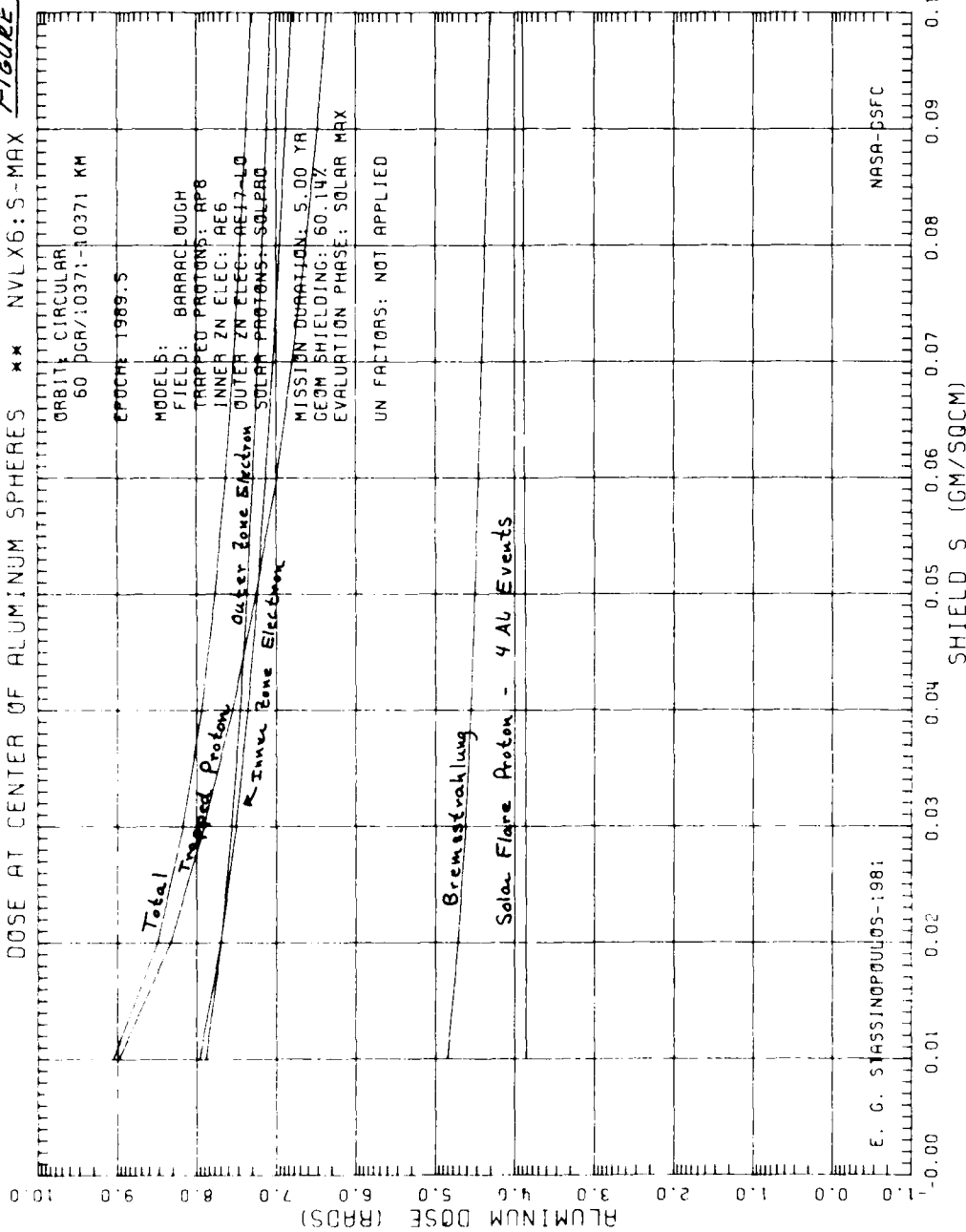


FIGURE 101

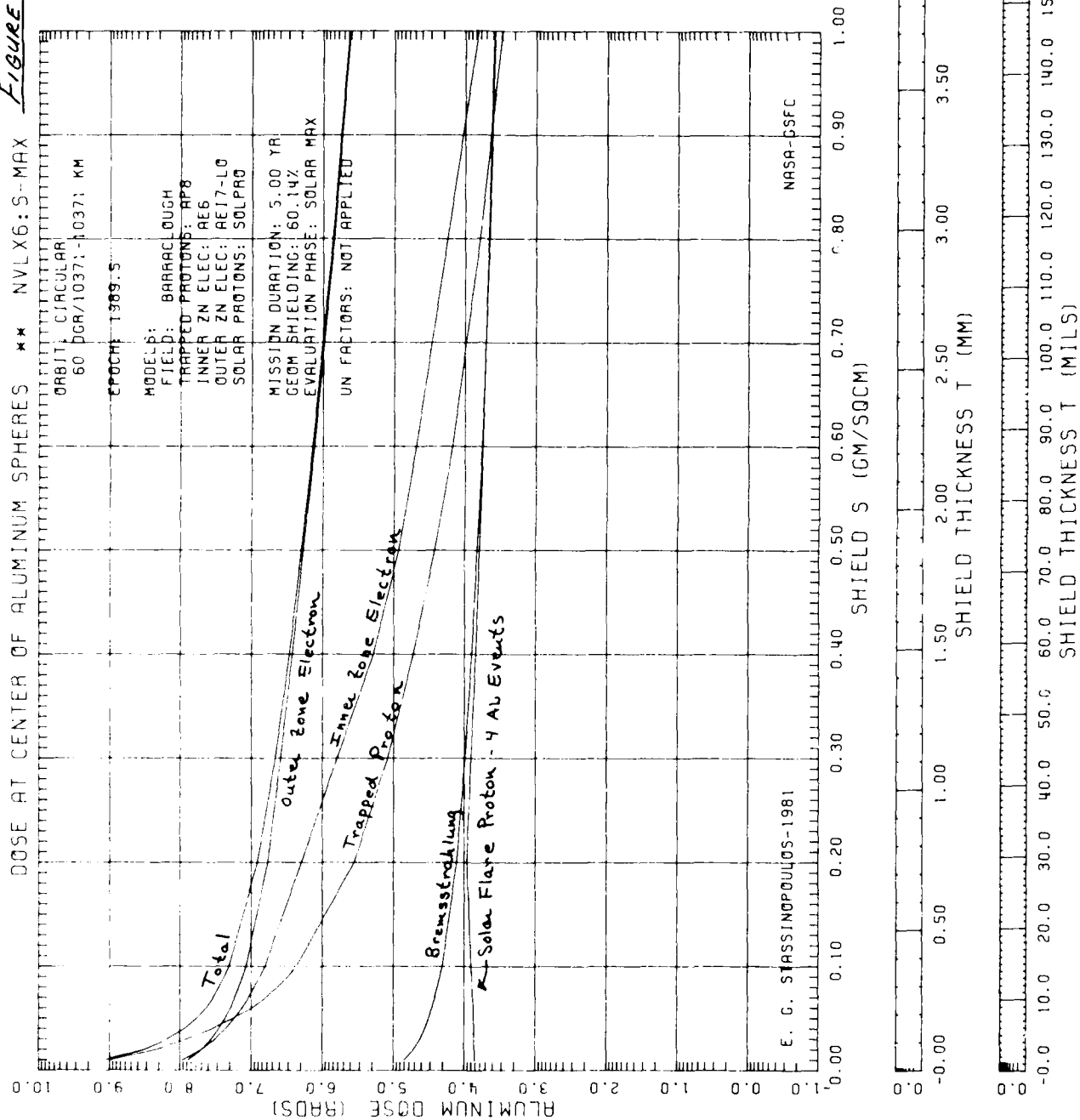
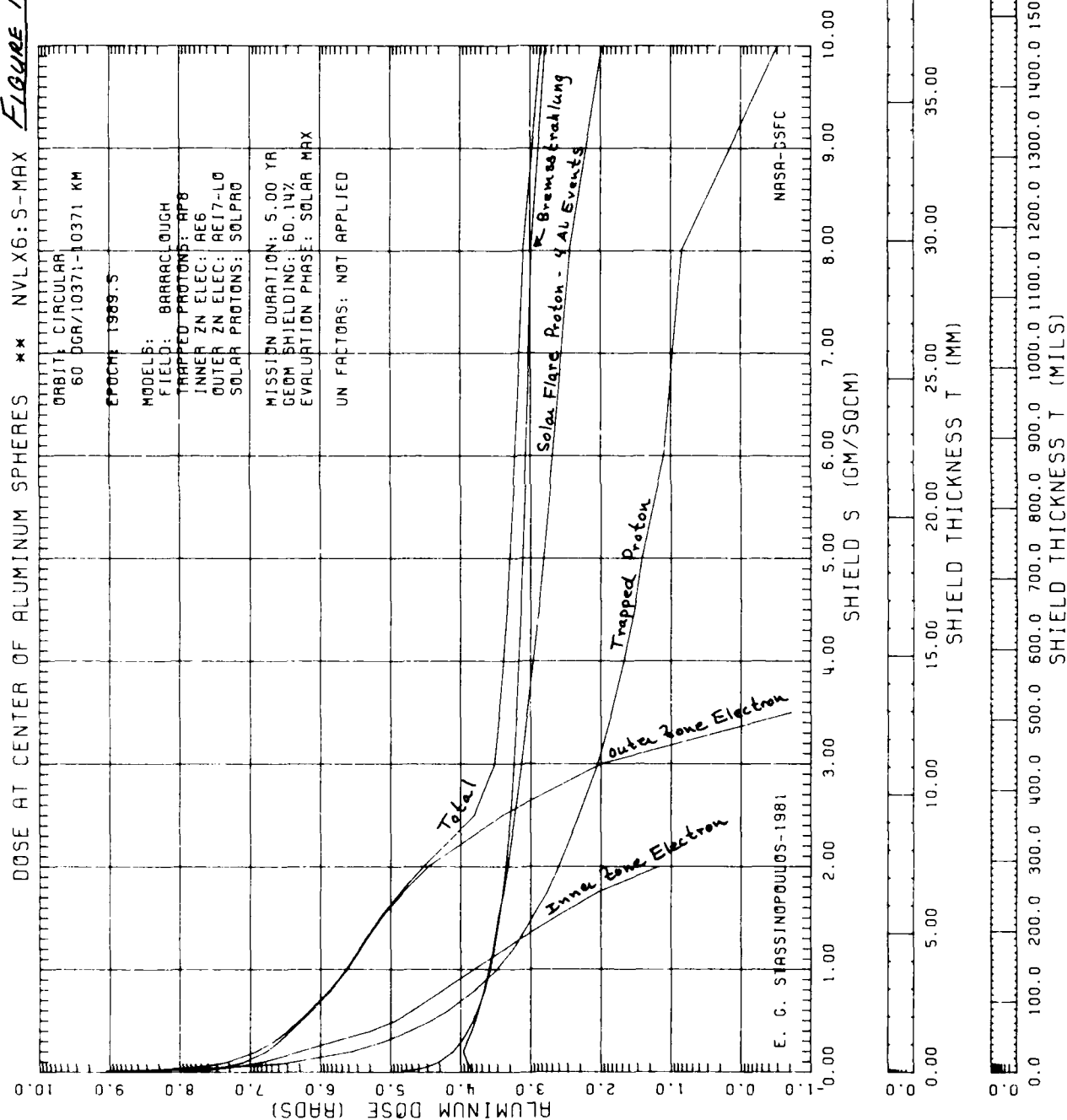
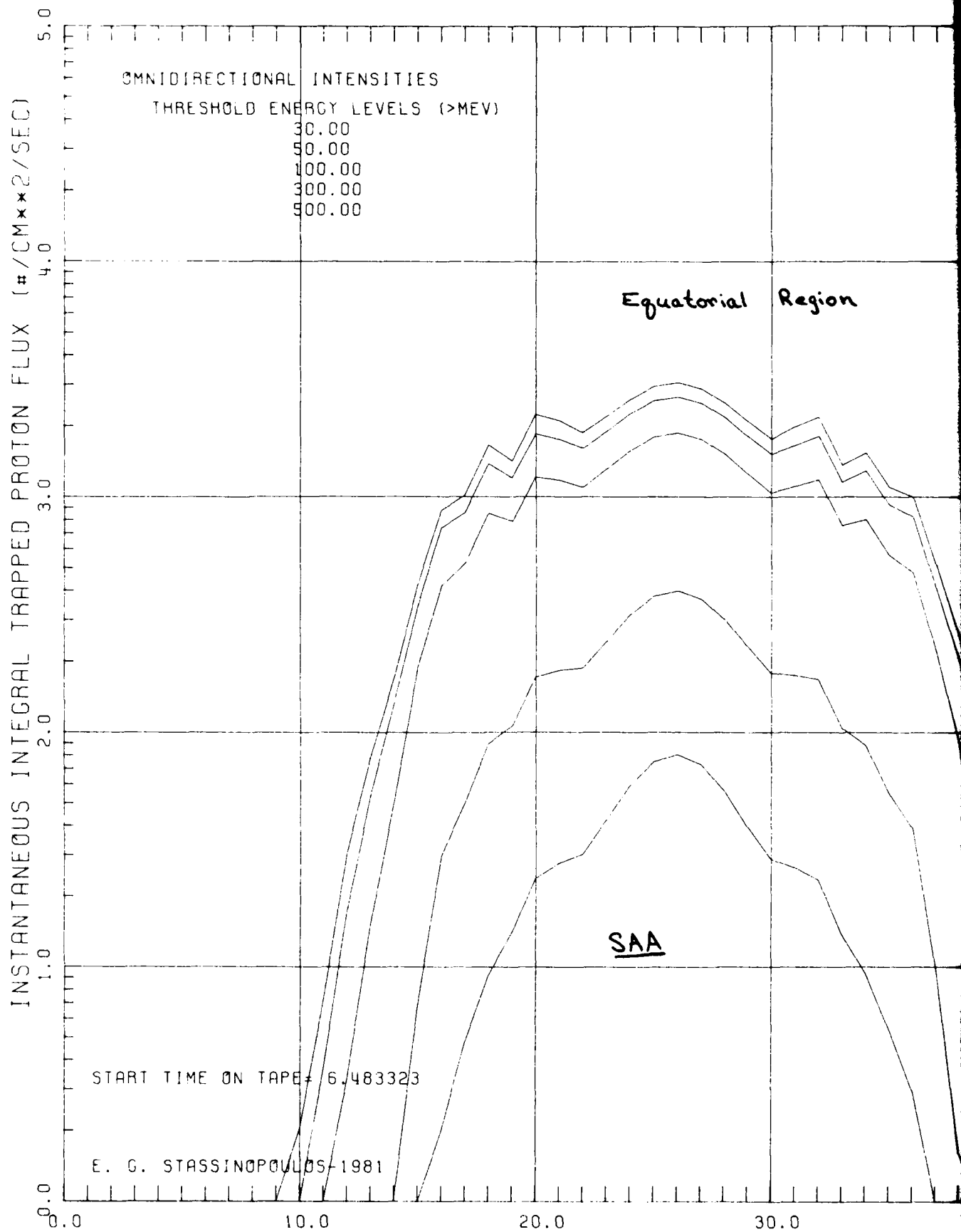


FIGURE 102





Region

30.0

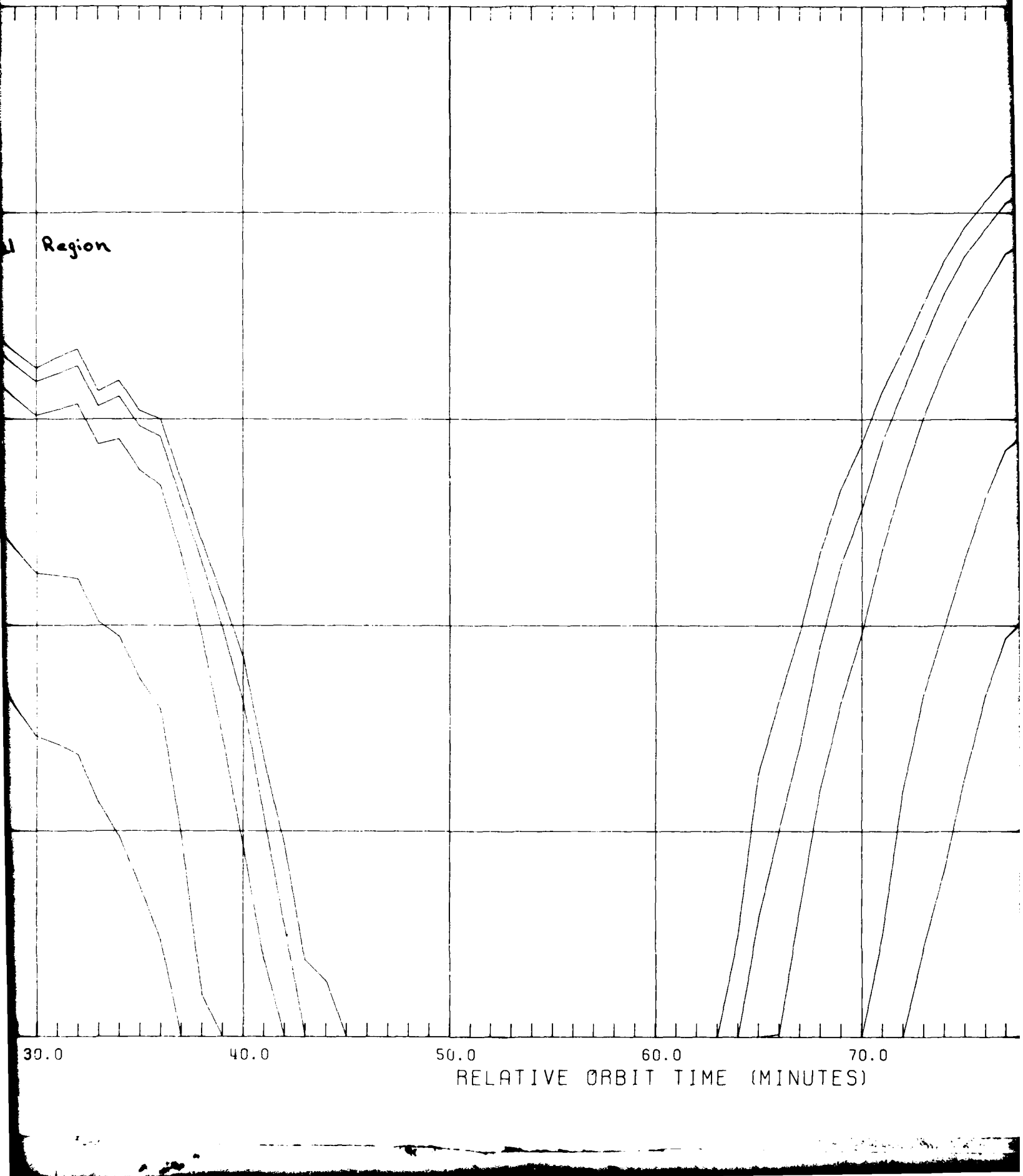
40.0

50.0

60.0

70.0

RELATIVE ORBIT TIME (MINUTES)



Equatorial Region

ORBIT: NAVELEX 1
60 DGR/1667-1667 KM

EPOCH: 1989.5

MODELS:
FIELD: BARR/75
TRAPPED PROTONS: AP8
INNER ZN ELEC: AE6
OUTER ZN ELEC: AE17-L
MISSION DURATION: 60.00
EVALUATION PHASE: SOLAR

UN FACTORS: NOT APPLIED

SAA

STOP TIME ON TAPE= 8.441

NAS

70.0

80.0

90.0

100.0

110.0

(MINUTES)

Figure 103

ORBIT: NAVELEX 1
60 DGR/1667-1567 KM

EPOCH: 1989.5

MODELS:

FIELD: BARR/75

TRAPPED PROTONS: AP8

INNER ZN ELEC: AE6

OUTER ZN ELEC: AE17-L0

MISSION DURATION: 60.00 MO

EVALUATION PHASE: SOLAR MAX

UN FACTORS: NOT APPLIED

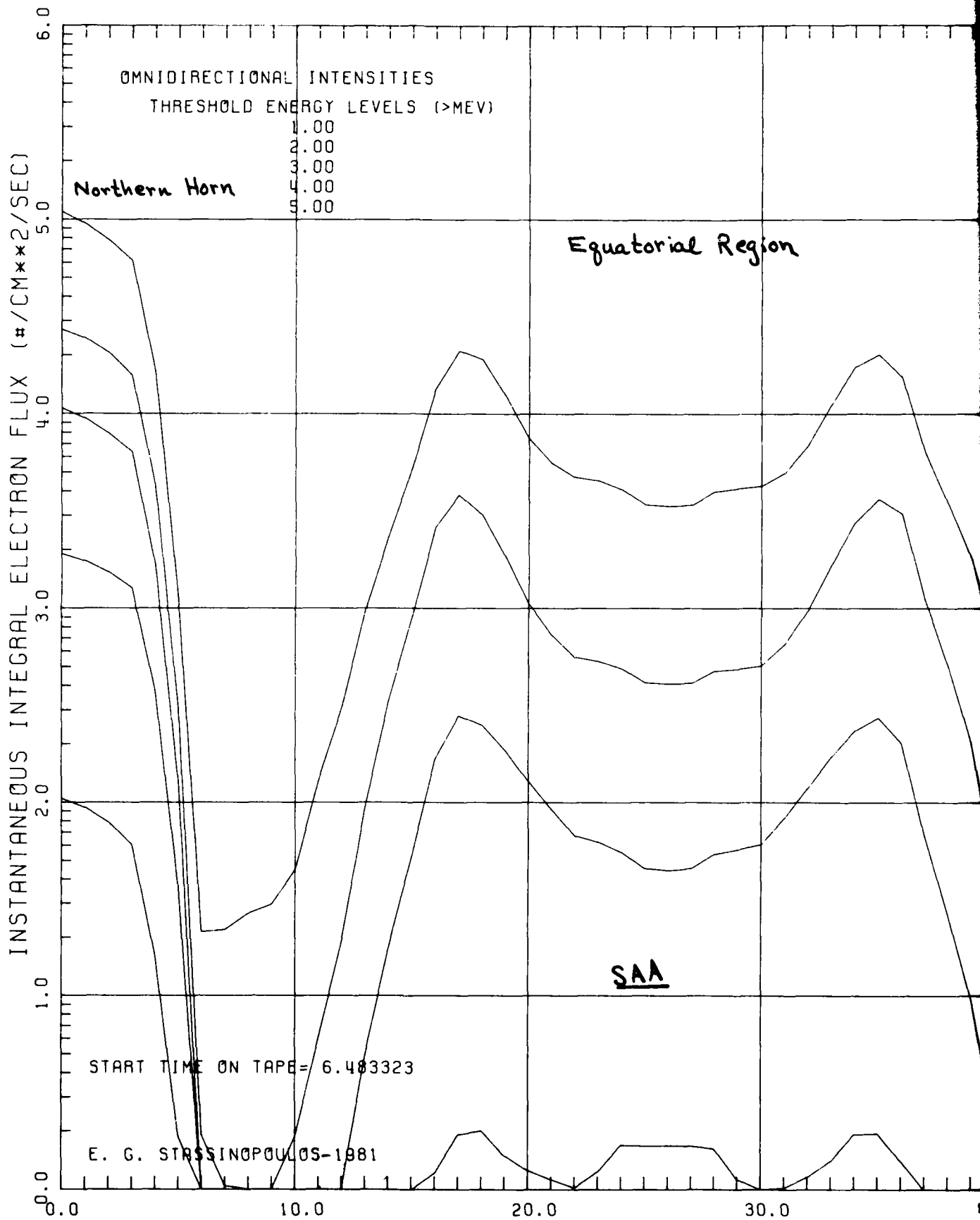
STOP TIME ON TAPE= 8.449993

NASA-GSFC

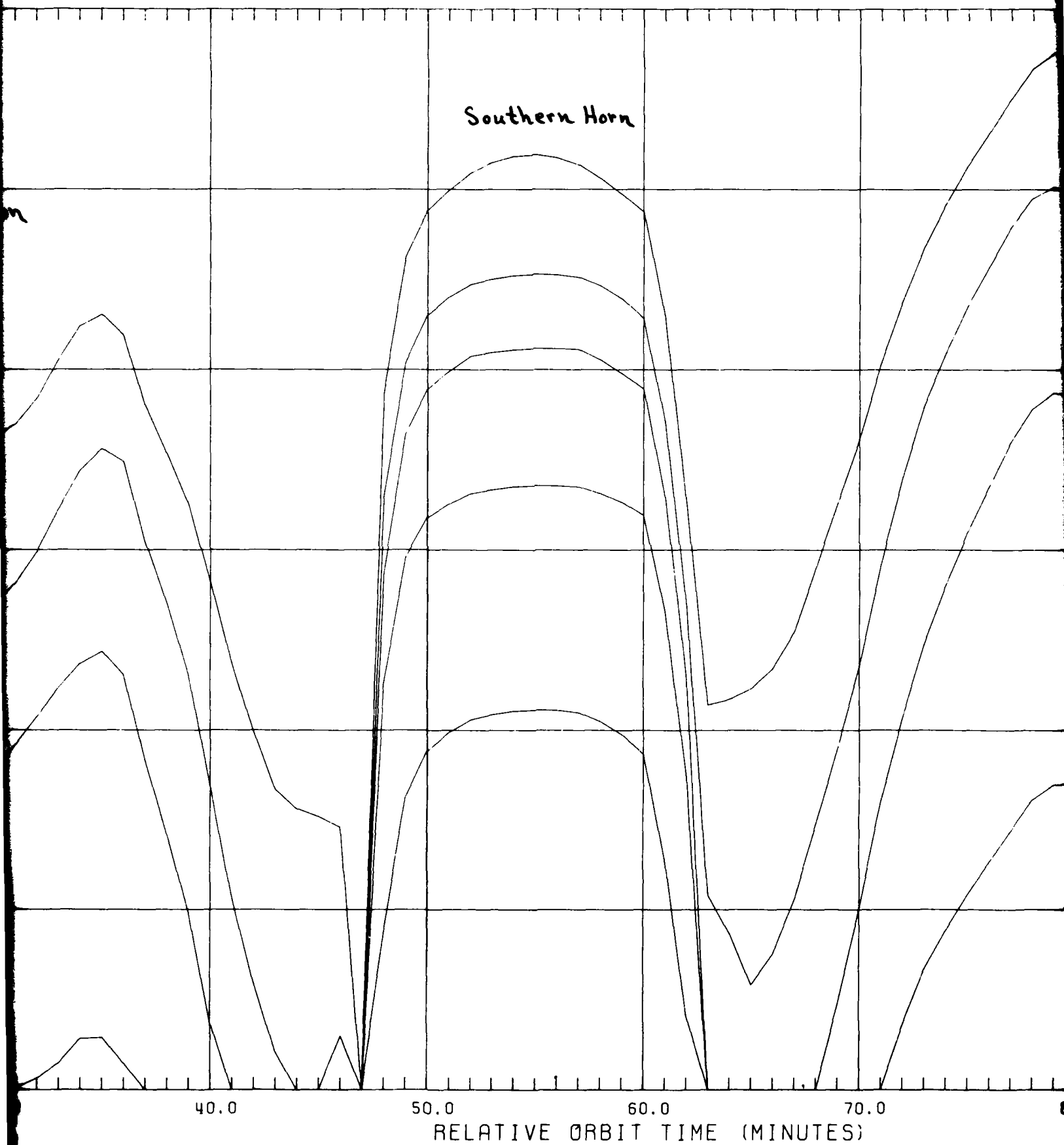
100.0

110.0

120.0



Southern Horn



3

Equatorial Region

Figure

ORBIT: NAVELEX 1
60 DGR/1667-1667 KM

EPOCH: 1989.5

North

MODELS:
FIELD: BARR/75
TRAPPED PROTONS: AP8
INNER ZN ELEC: AE6
OUTER ZN ELEC: AE17-L0
MISSION DURATION: 60.00 MO
EVALUATION PHASE: SOLAR MAX

UN FACTORS: NOT APPLIED

SAA

STOP TIME ON TAPE= 8.449993

NASA-GS

70.0 80.0 90.0 100.0 110.0

UTES)

Figure 104

ORBIT: NAVELEX 1
60 DGR/1667-1667 KM

EPOCH: 1989.5

Northern Horn

MODELS:

FIELD: BARR/75

TRAPPED PROTONS: AP8

INNER ZN ELEC: AE6

OUTER ZN ELEC: AE17-L0

MISSION DURATION: 60.00 MO

EVALUATION PHASE: SOLAR MAX

UN FACTORS: NOT APPLIED

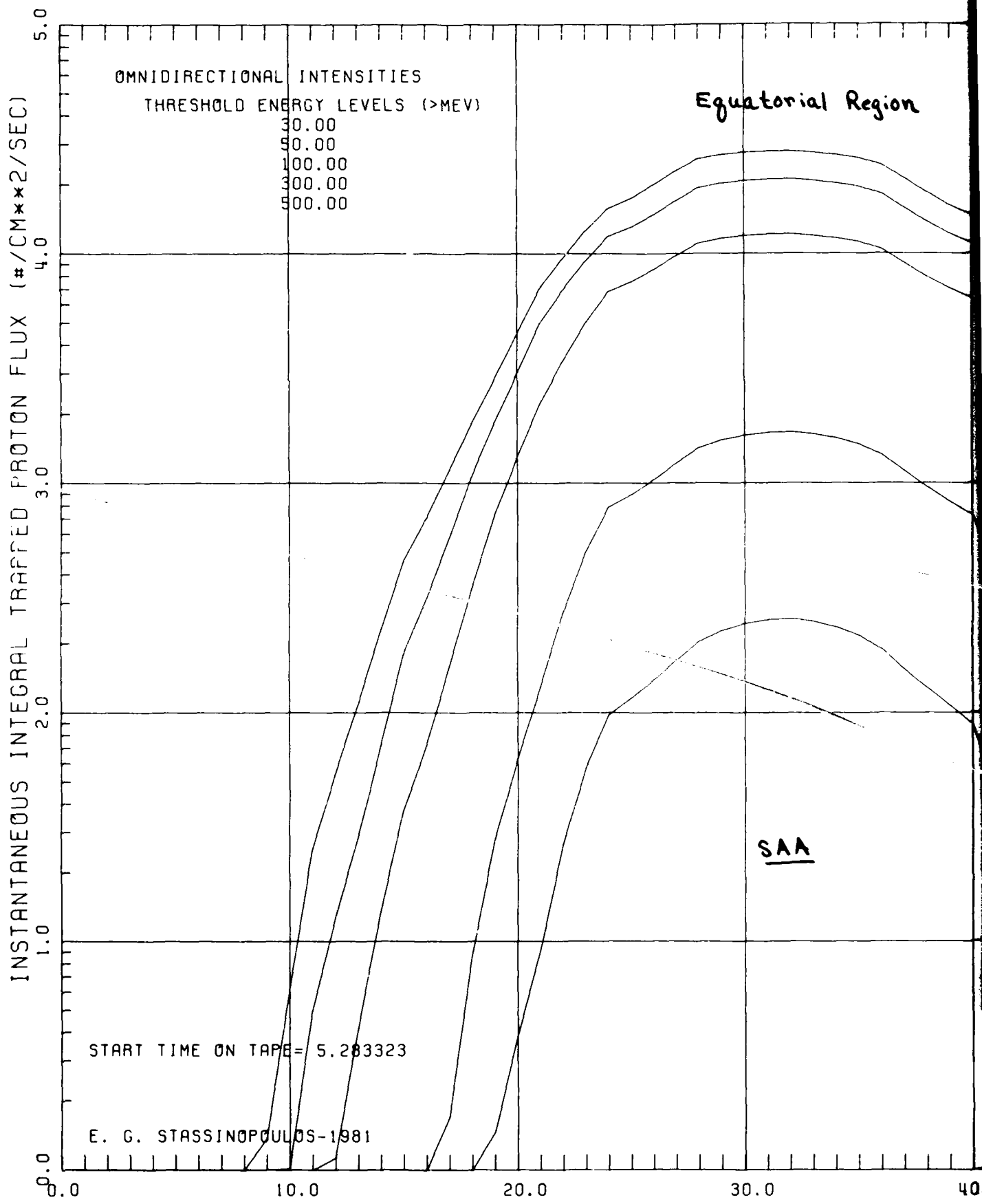
STOP TIME ON TAPE = 8.449993

NASA-GSFC

100.0

110.0

120.0



Equatorial Region

SAA

ME (MINUTES)

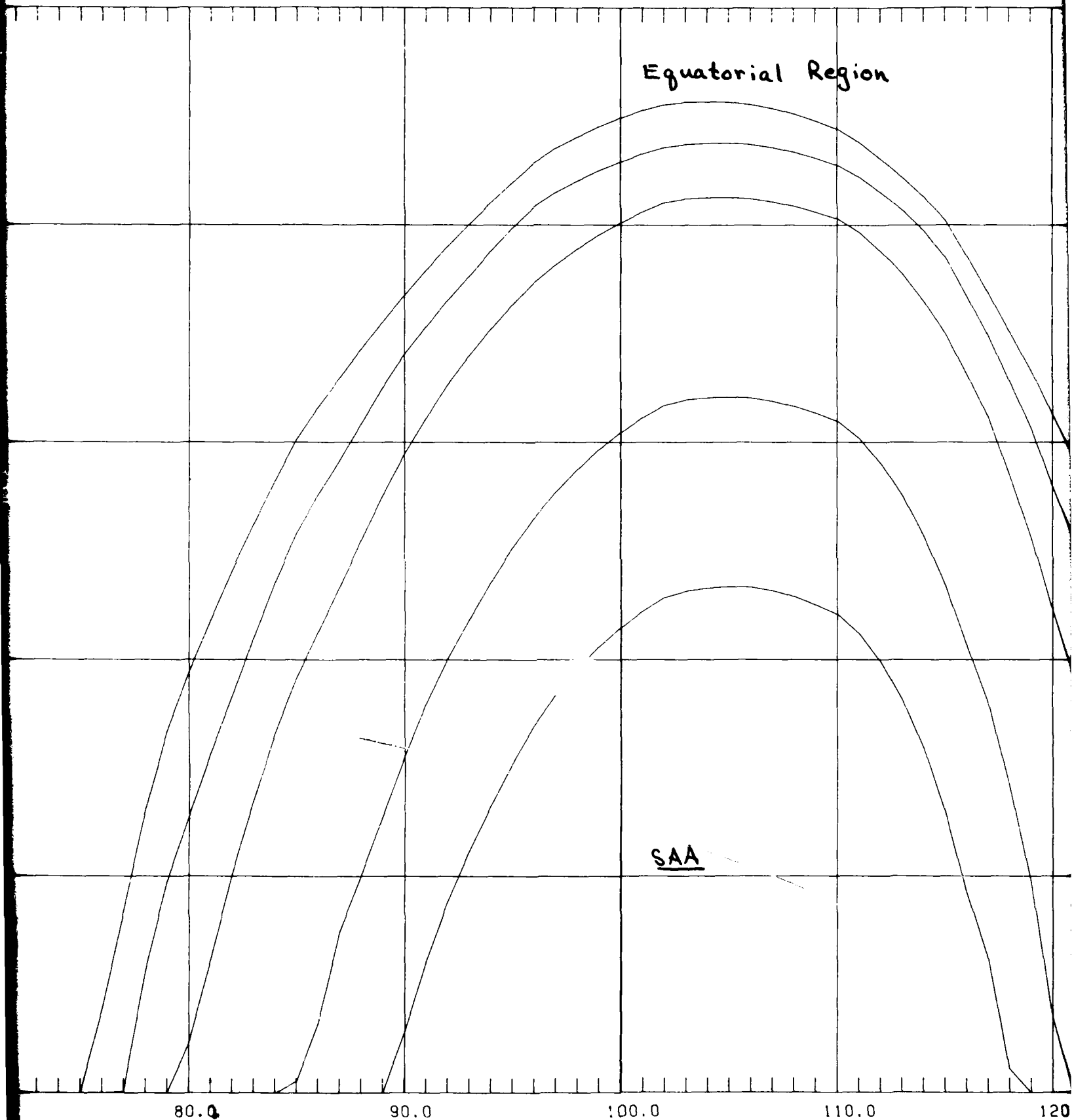
80.0

90.0

100.0

110.0

120.0



region

Figure 105

ORBIT: NAVELEX 2
60 DGR/2593-2593 KM

EPOCH: 1989.5

MODELS:

FIELD: BARR/75

TRAPPED PROTONS: AP8

INNER ZN ELEC: AEG

OUTER ZN ELEC: AE17 L0

MISSION DURATION: 60.00 MO

EVALUATION PHASE: SOLAR MAX

UN FACTORS: NOT APPLIED

STOP TIME ON TAPE= 7.616656

NASA-GSFC

10.0 120.0 130.0 140.0

AD-A141 849

ORBITAL RADIATION STUDY FOR INCLINED CIRCULAR
TRAJECTORIES(U) NATIONAL AERONAUTICS AND SPACE
ADMINISTRATION GREENBELT MD GO.. E G STASSINOPOULOS

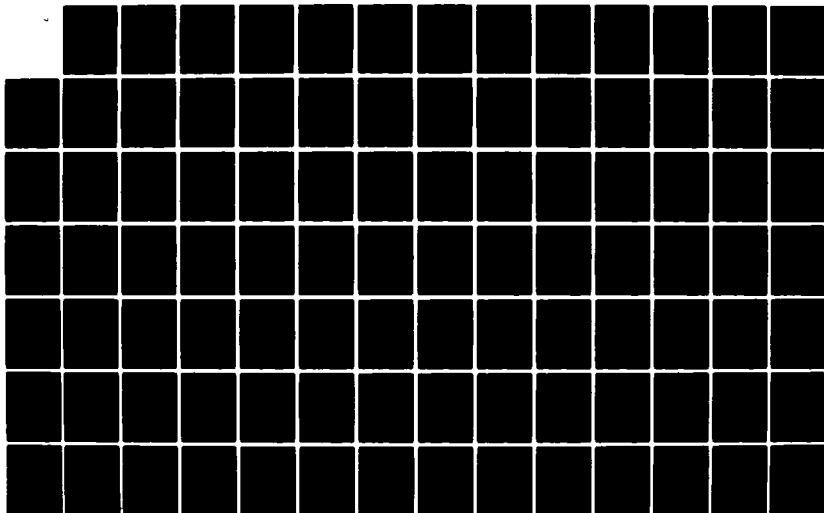
UNCLASSIFIED

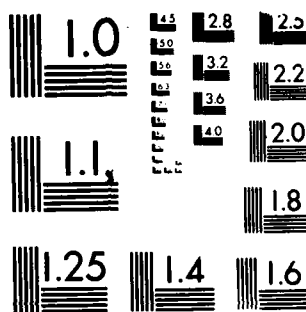
NOV 81 NASA-GSFC-X-601-81-28

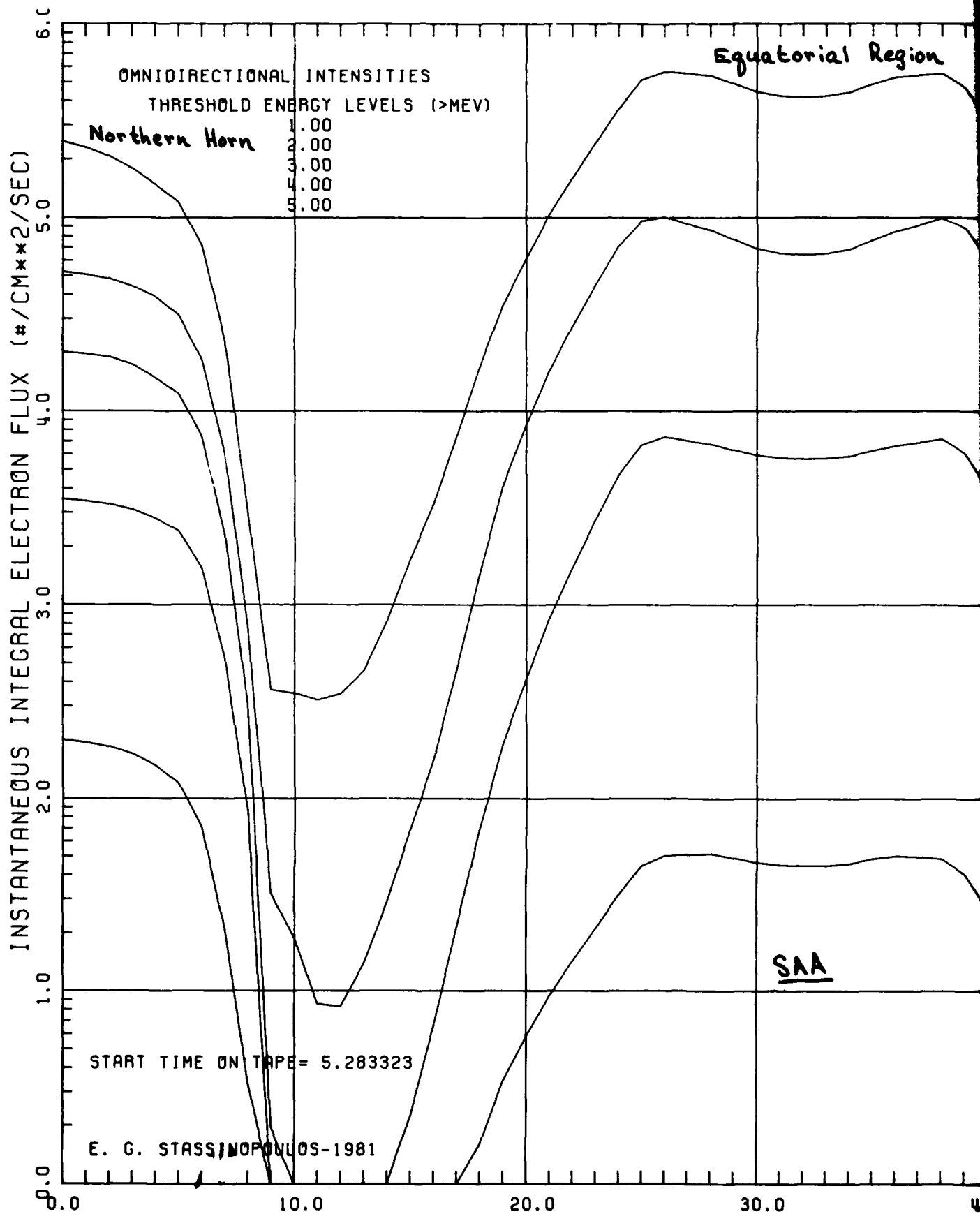
F/G 22/3

NL

3/5



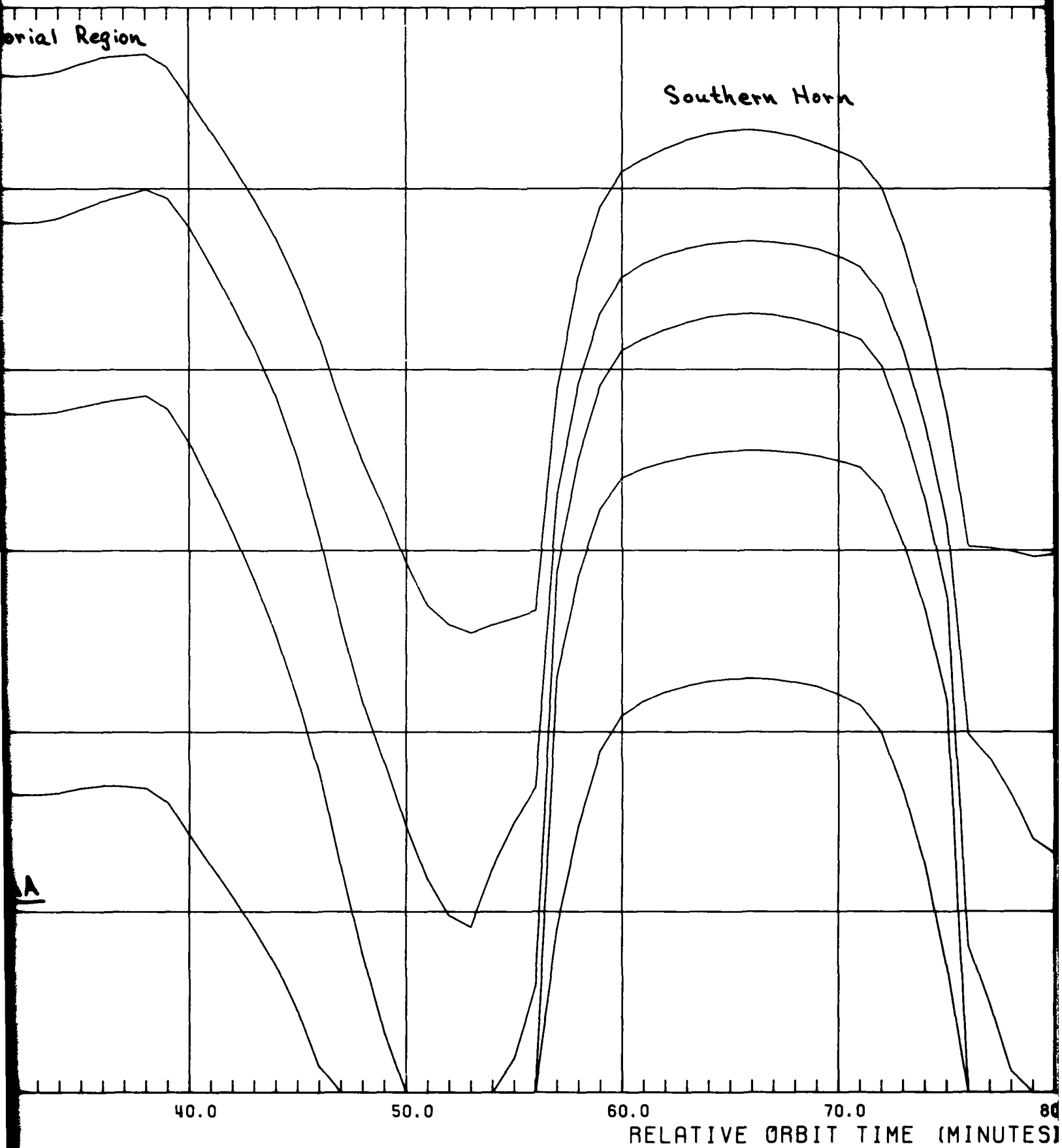




2

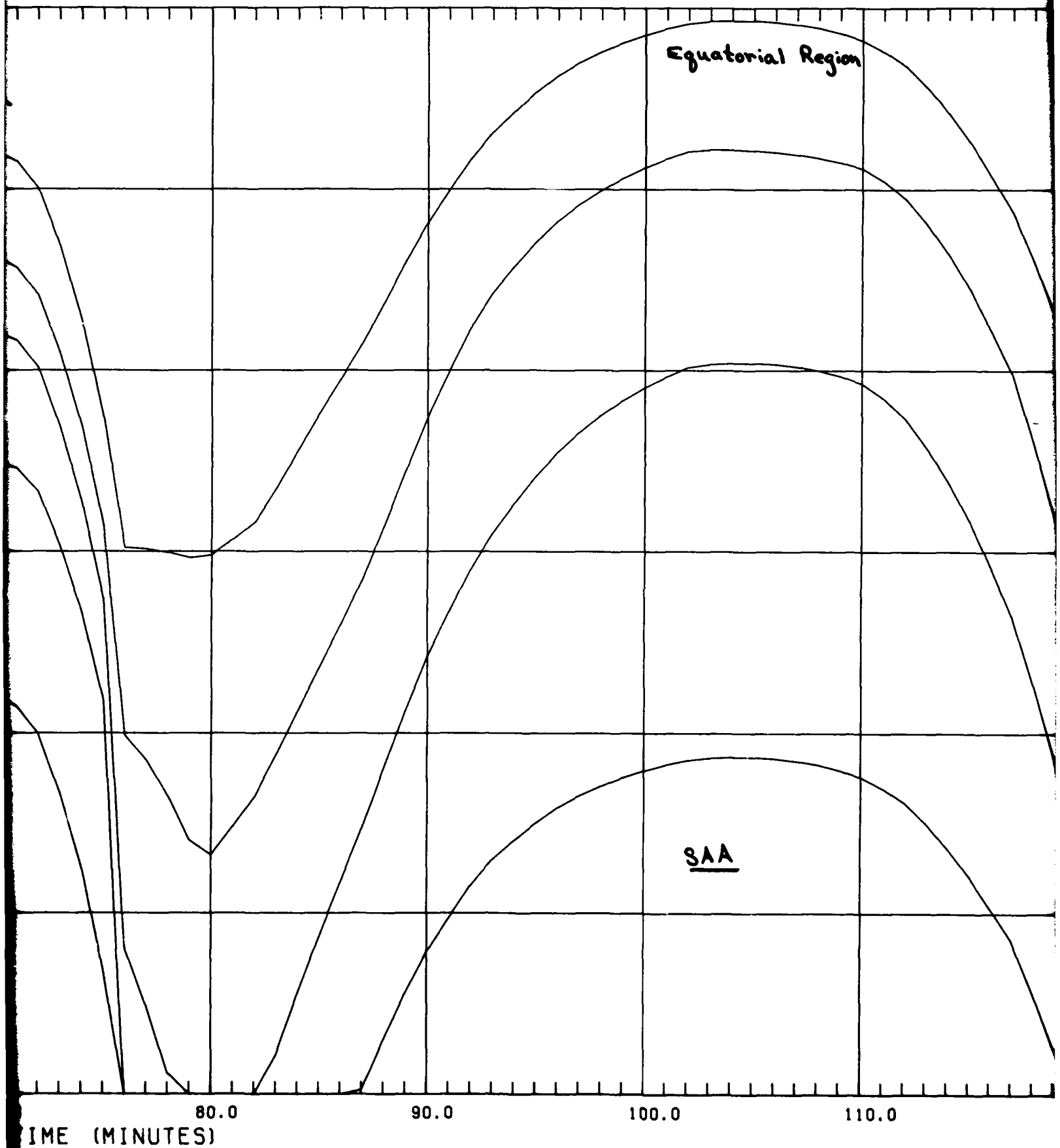
terial Region

Southern Horn

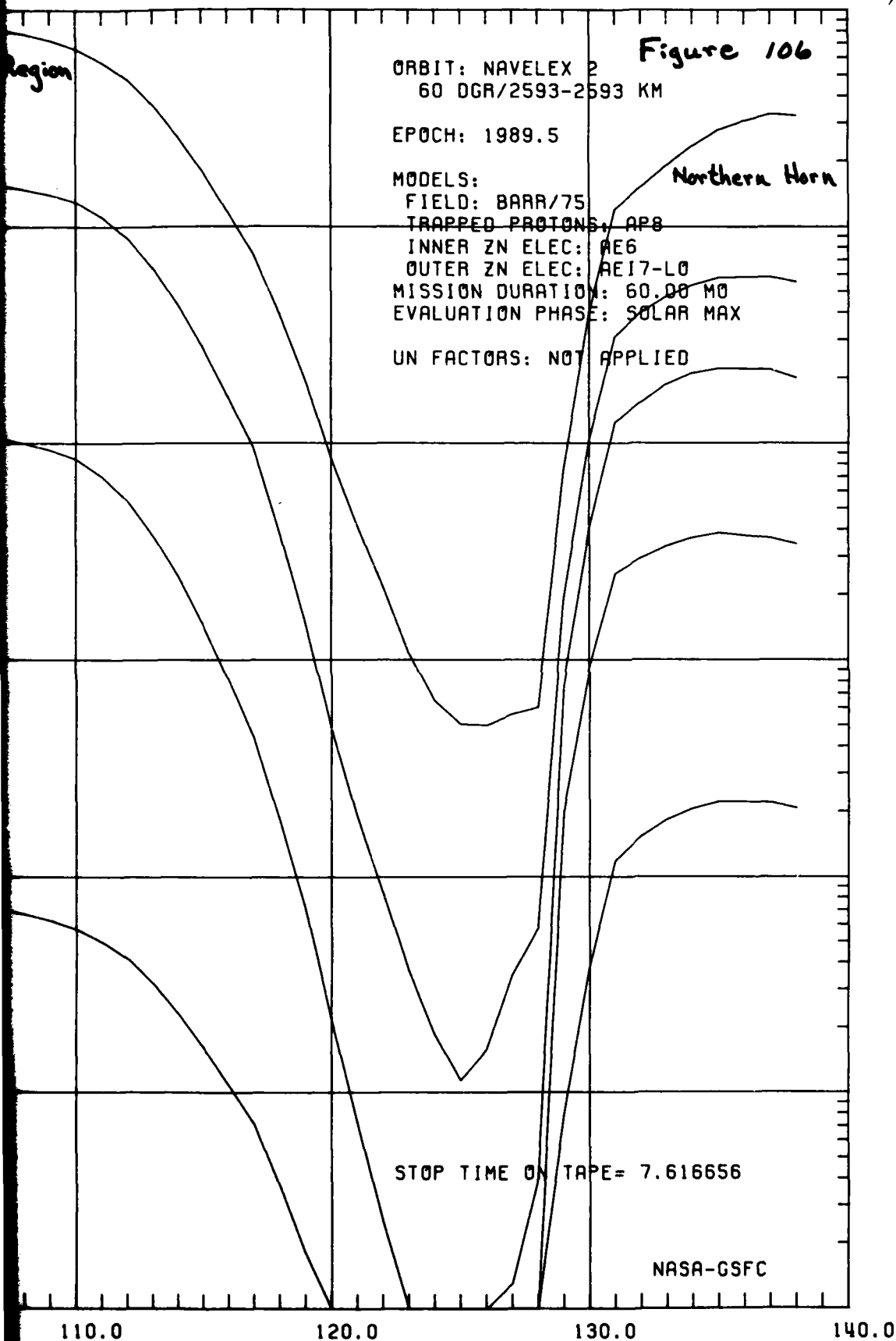


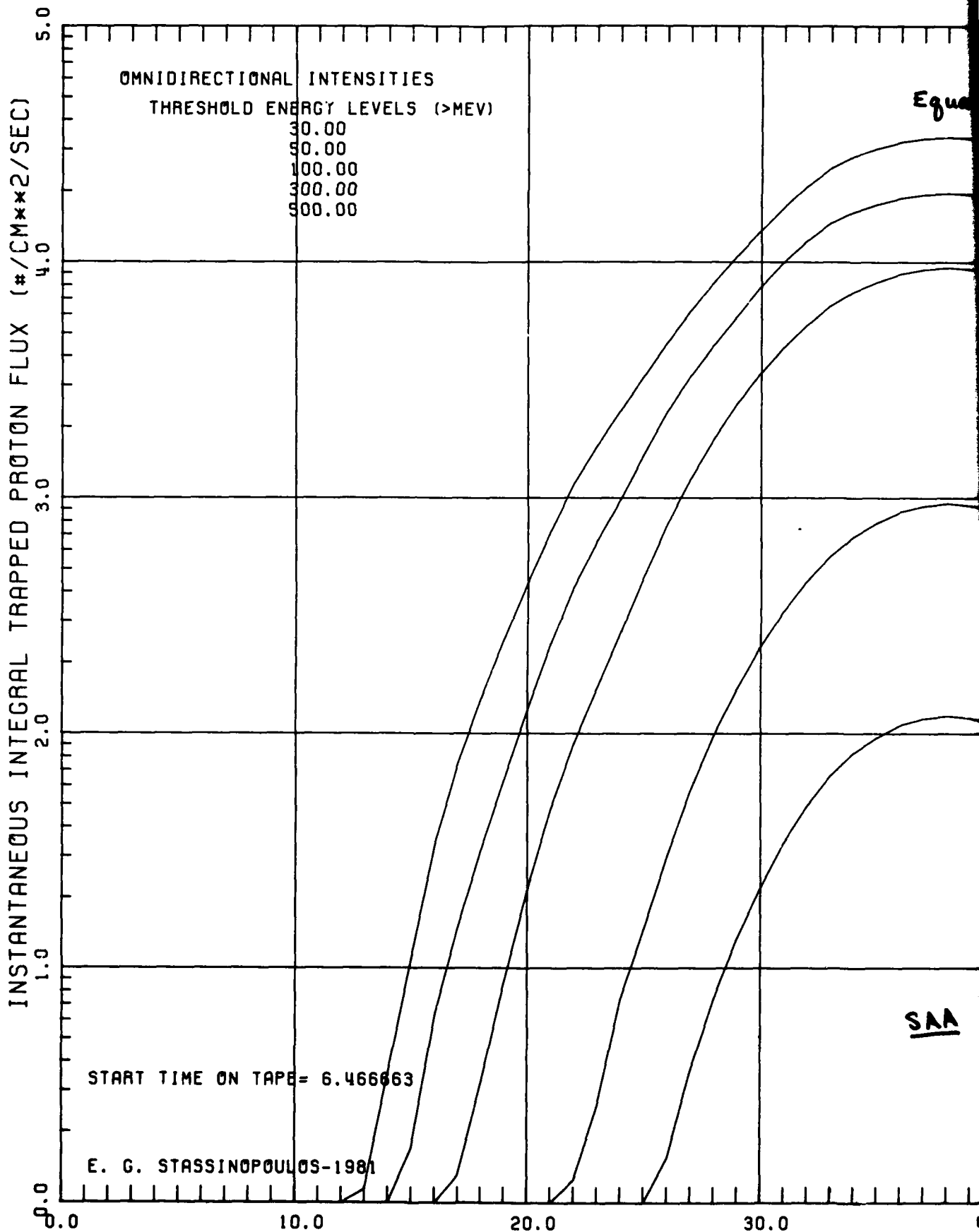
RELATIVE ORBIT TIME (MINUTES)

3



4





1
2

Equatorial Region

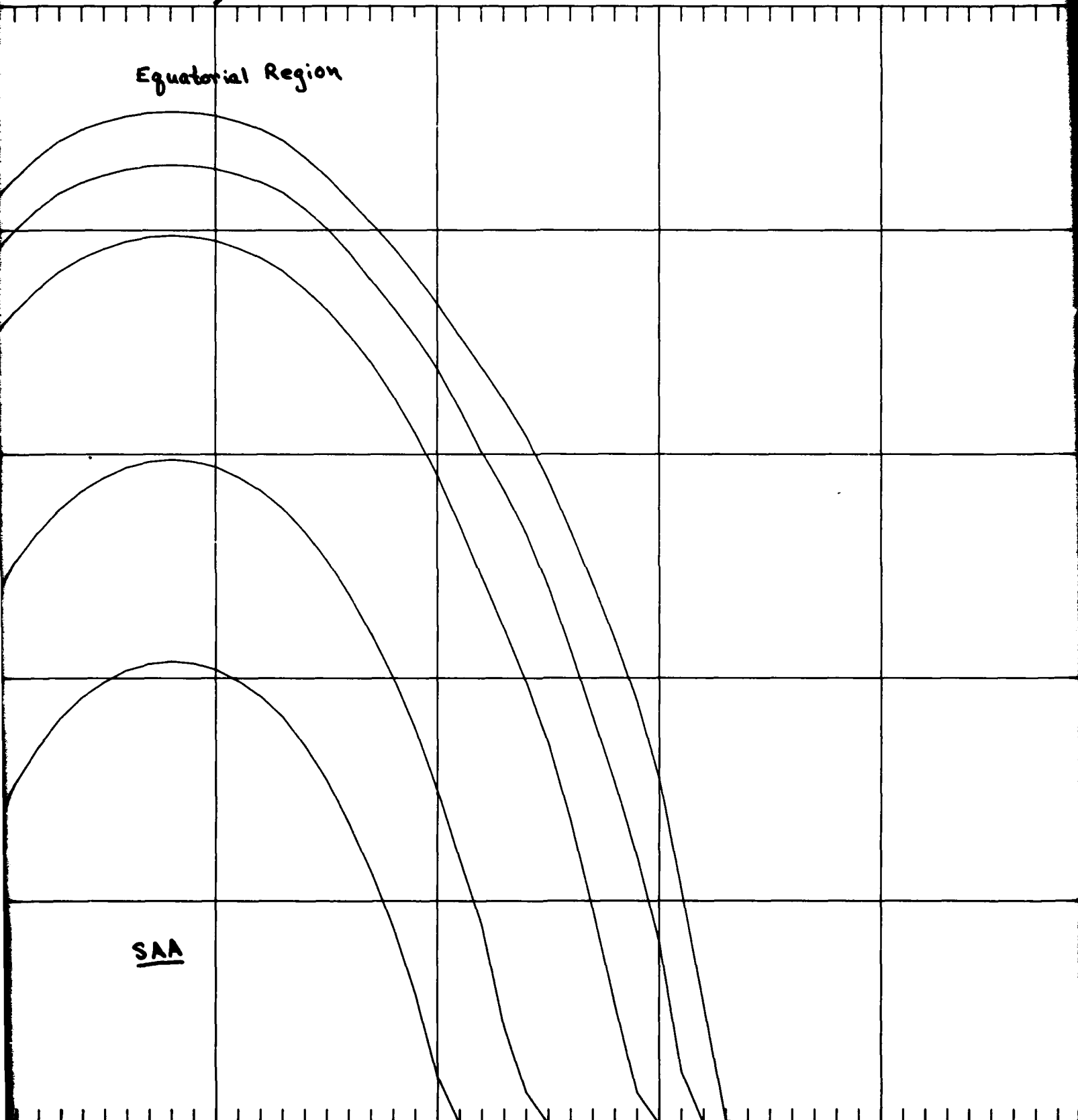
SAA

40.0

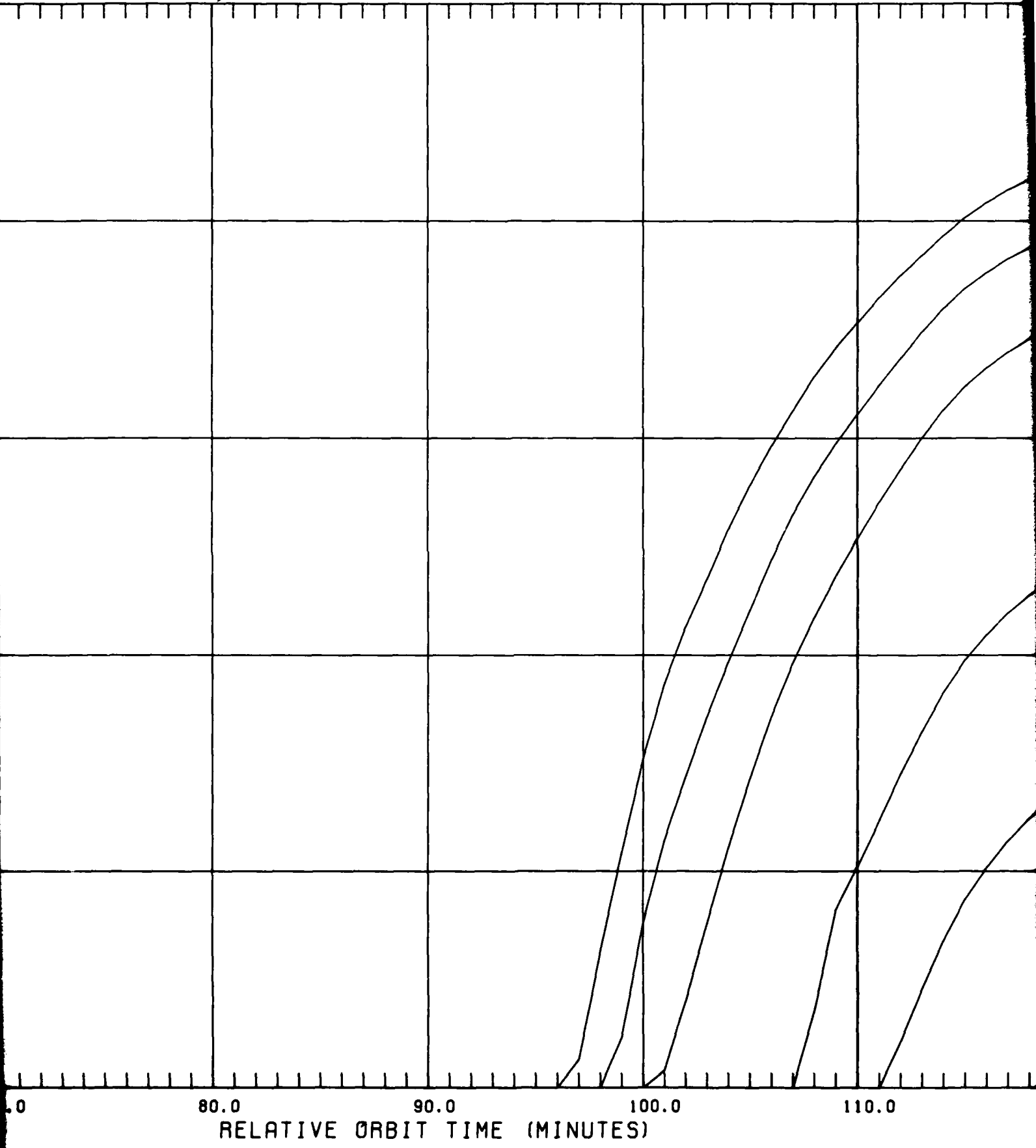
50.0

60.0

70.0



3



'4

Equatorial Region

SAA

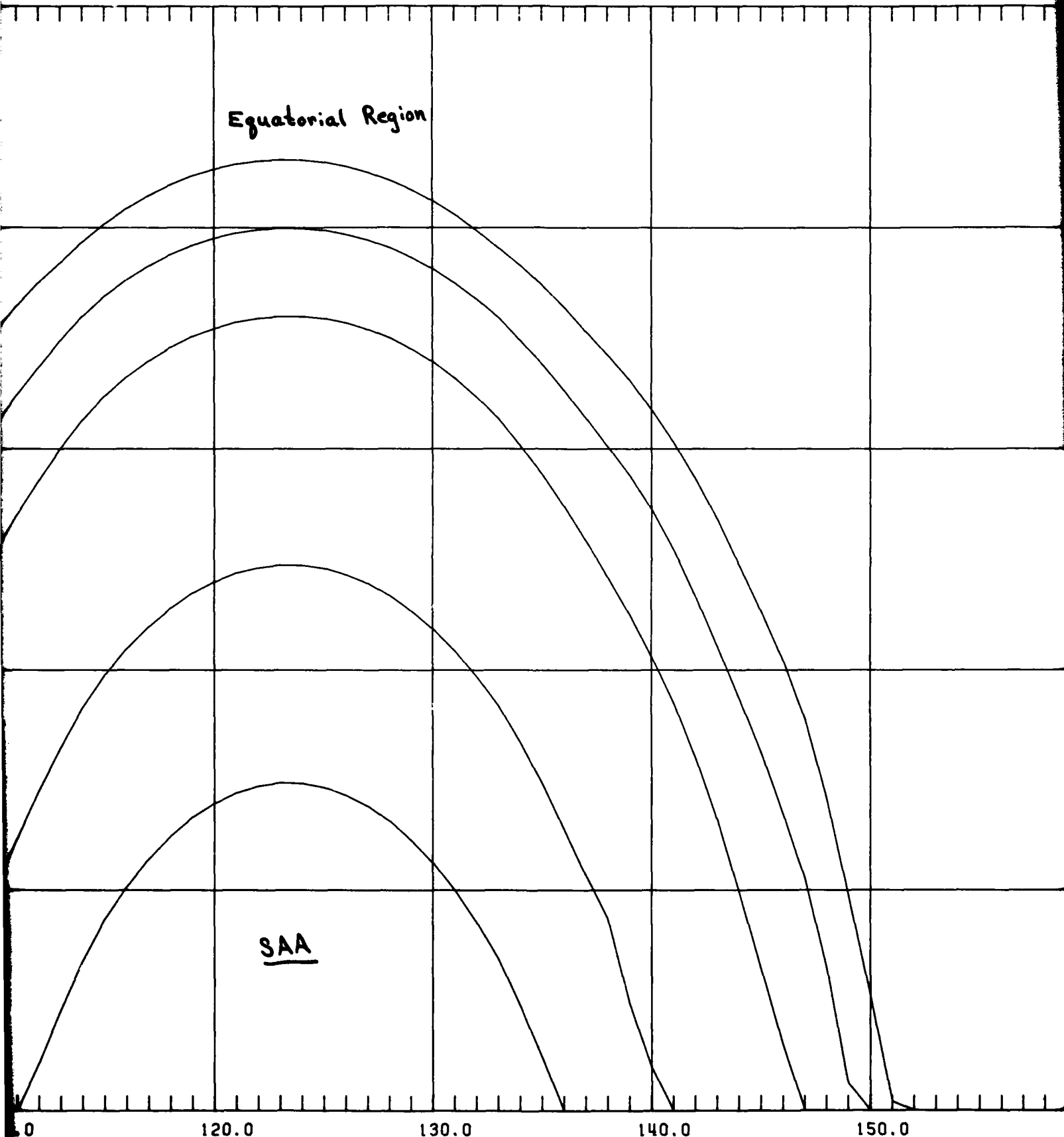


Figure 107

ORBIT: NAVELEX 3
60 DGR/3889-3889 KM

EPOCH: 1989.5

MODELS:

FIELD: BARR/75

TRAPPED PROTONS: AP8

INNER ZN ELEC: AE6

OUTER ZN ELEC: AE17 L8

MISSION DURATION: 60.00 MO

EVALUATION PHASE: SOLAR MAX

UN FACTORS: NOT APPLIED

STOP TIME ON TAPE= 9.316662

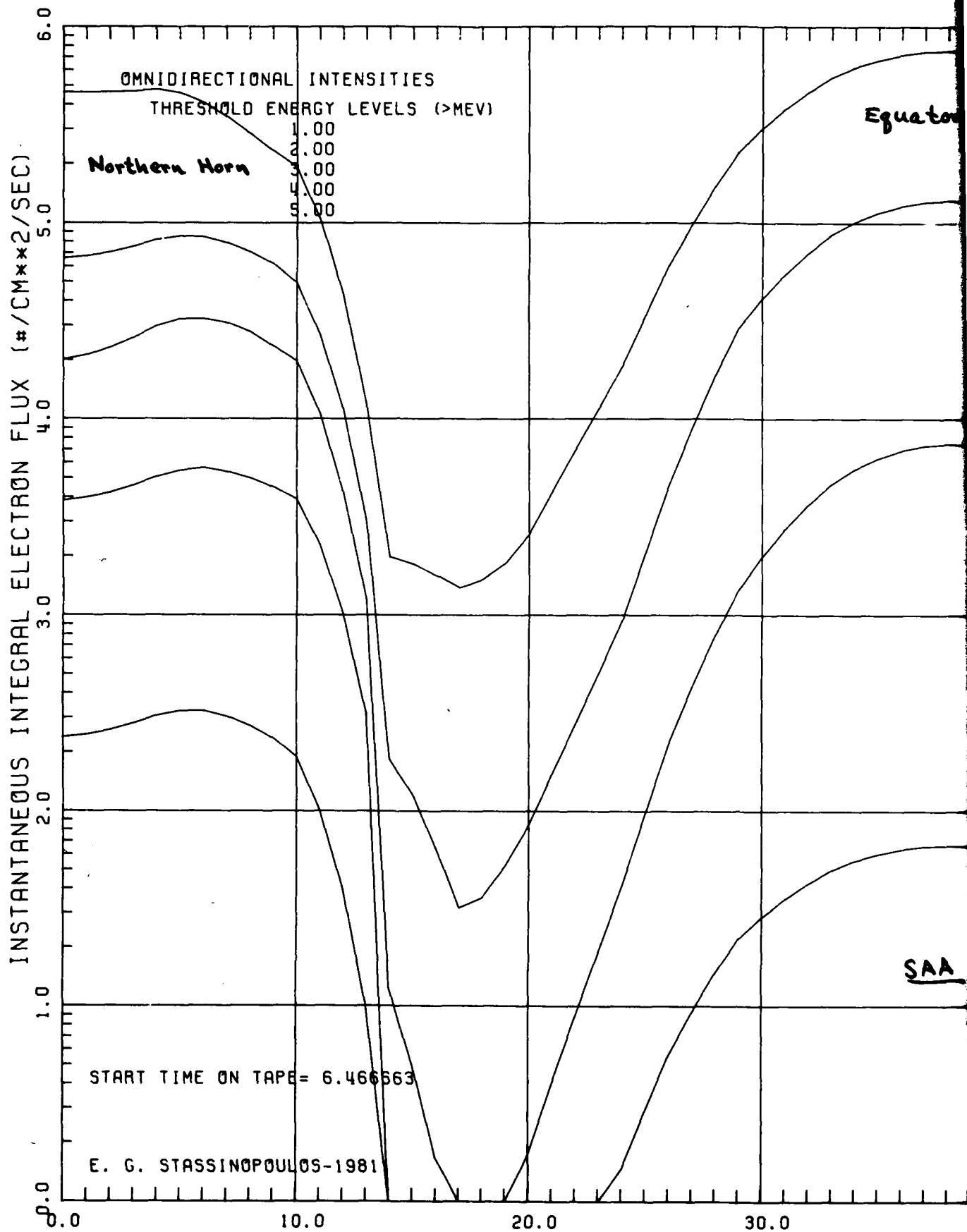
NASA-GSFC

150.0

160.0

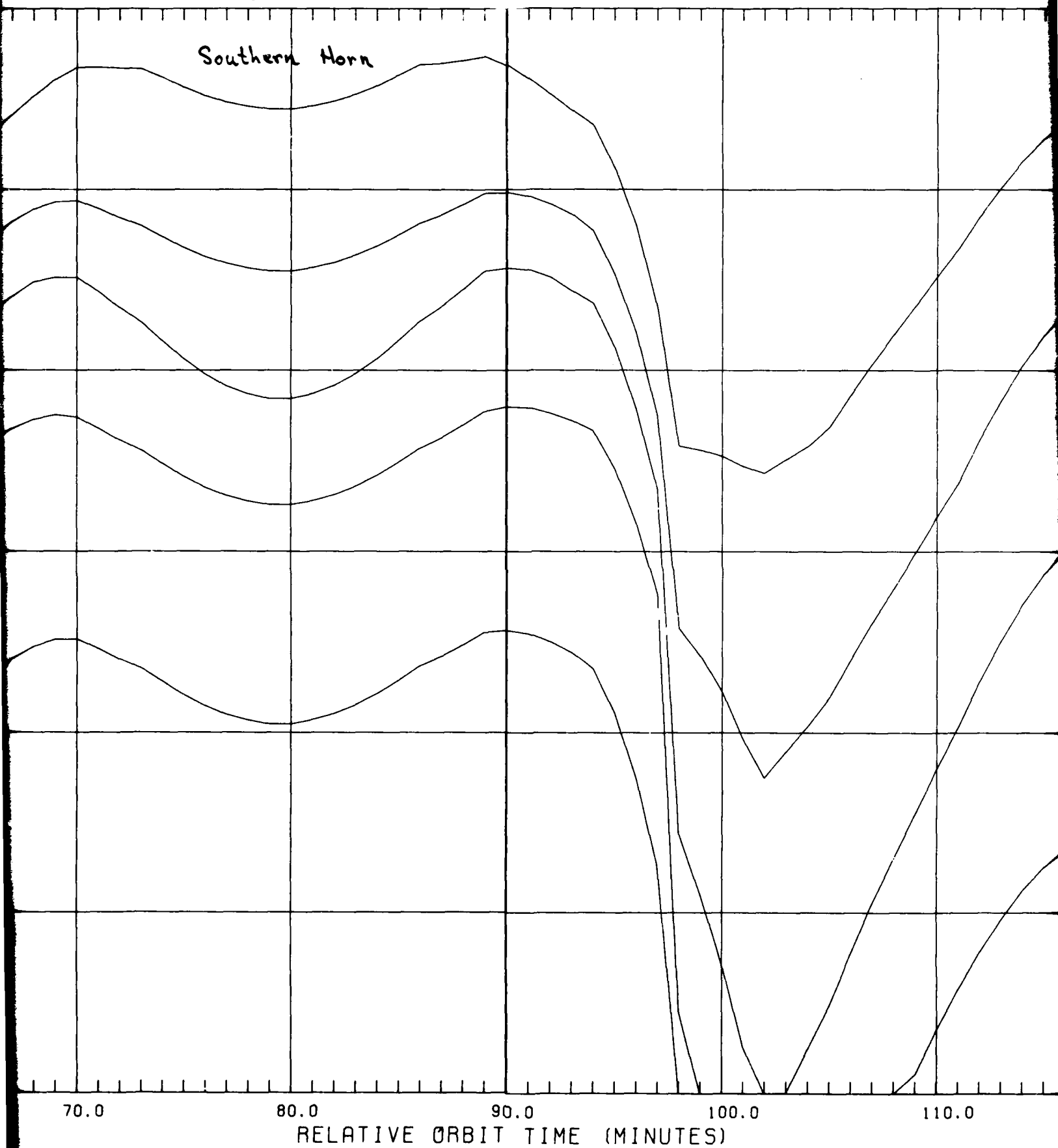
170.0

180.0



13

Southern Horn



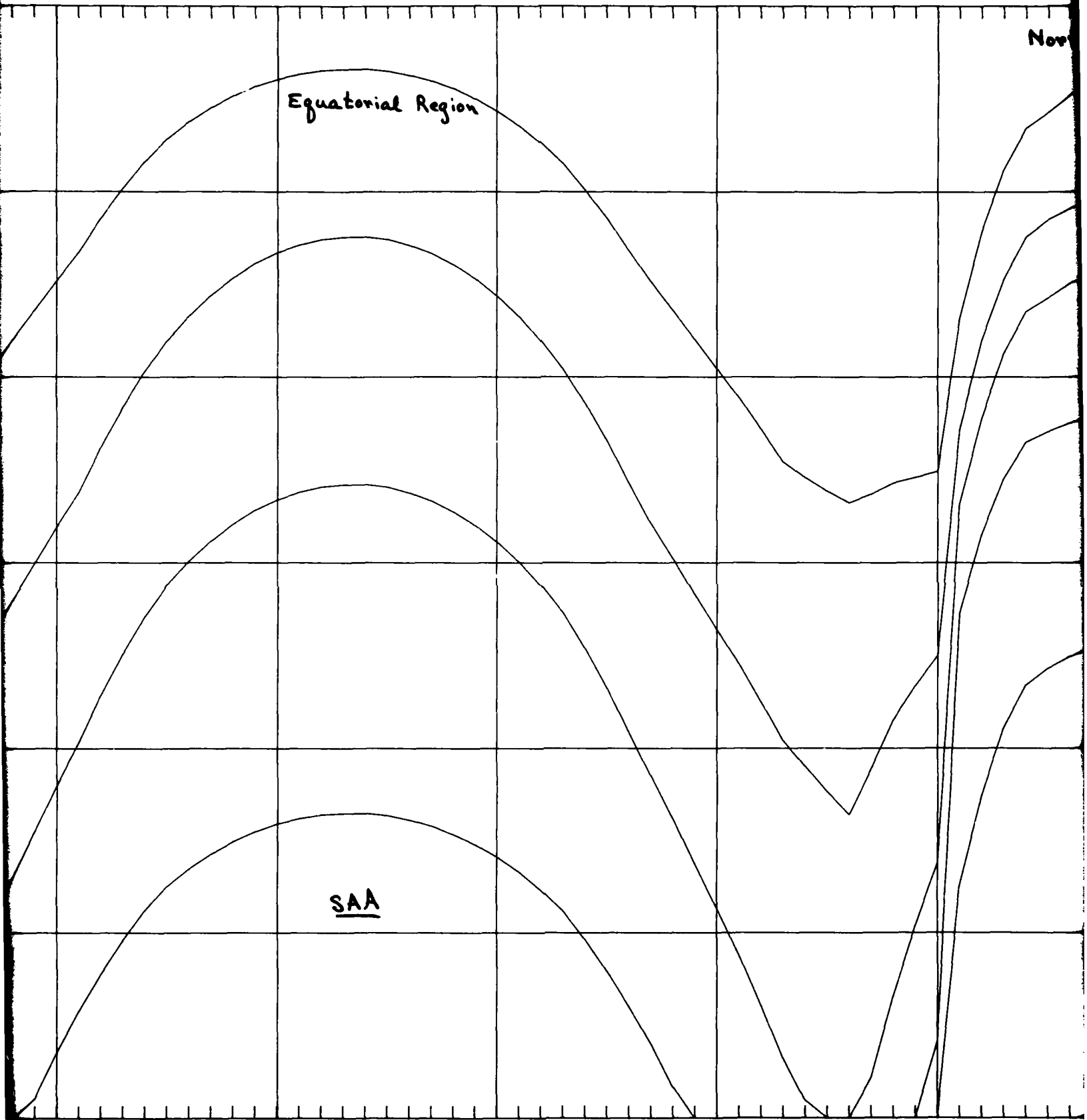
1 4

Nov

Equatorial Region

SAA

110.0 120.0 130.0 140.0 150.0



5

Northern Horn

Figure 108

ORBIT: NAVELEX B
60 DGR/3889-3889 KM

EPOCH: 1989.5

MODELS:

FIELD: BARR/75

TRAPPED PROTONS: AP8

INNER ZN ELEC: AE6

OUTER ZN ELEC: AE17-L0

MISSION DURATION: 60.00 MO

EVALUATION PHASE: SOLAR MAX

UN FACTORS: NOT APPLIED

STOP TIME ON TAPE= 9.316662

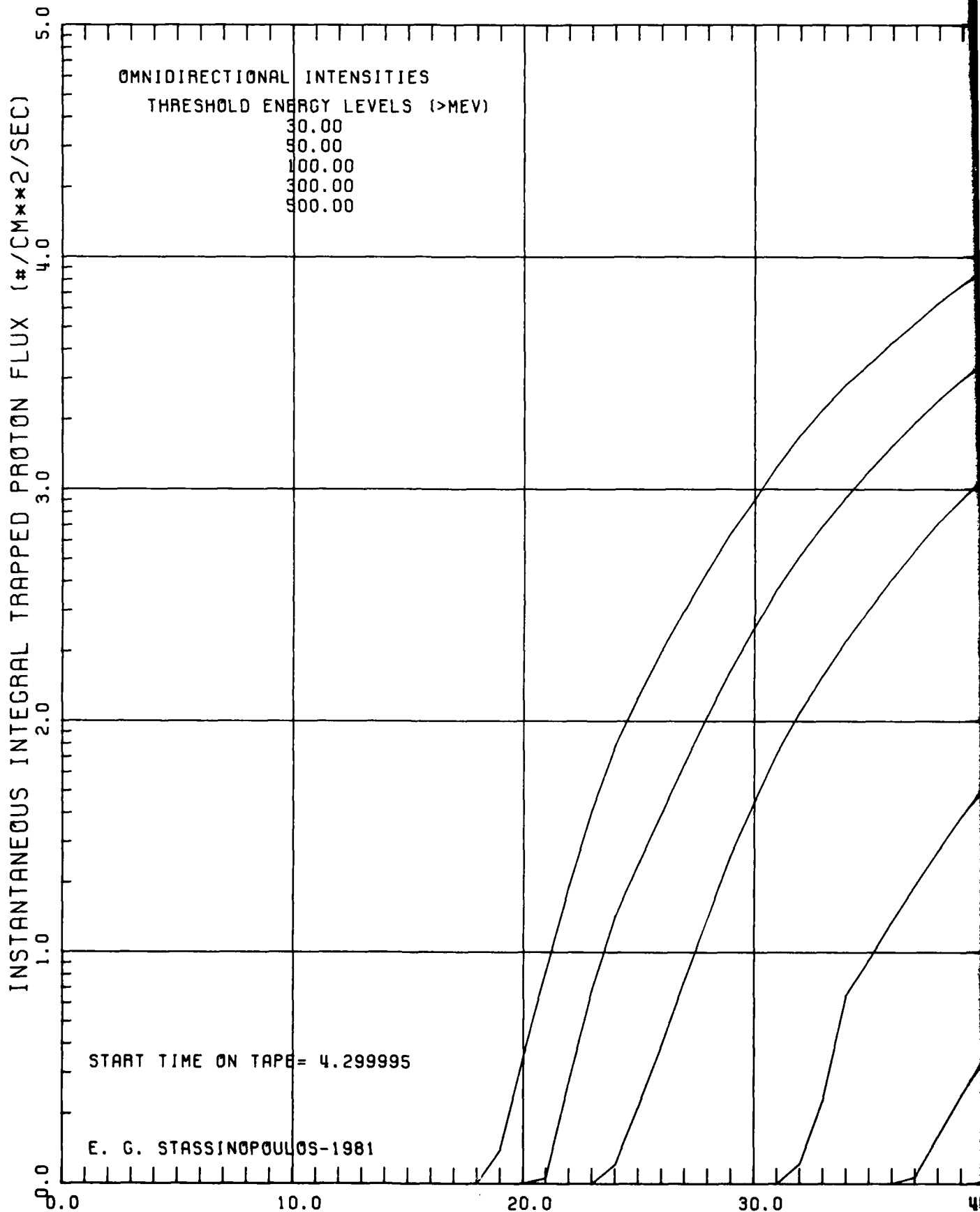
NASA-GSFC

150.0

160.0

170.0

180.0



2

Equatorial Region

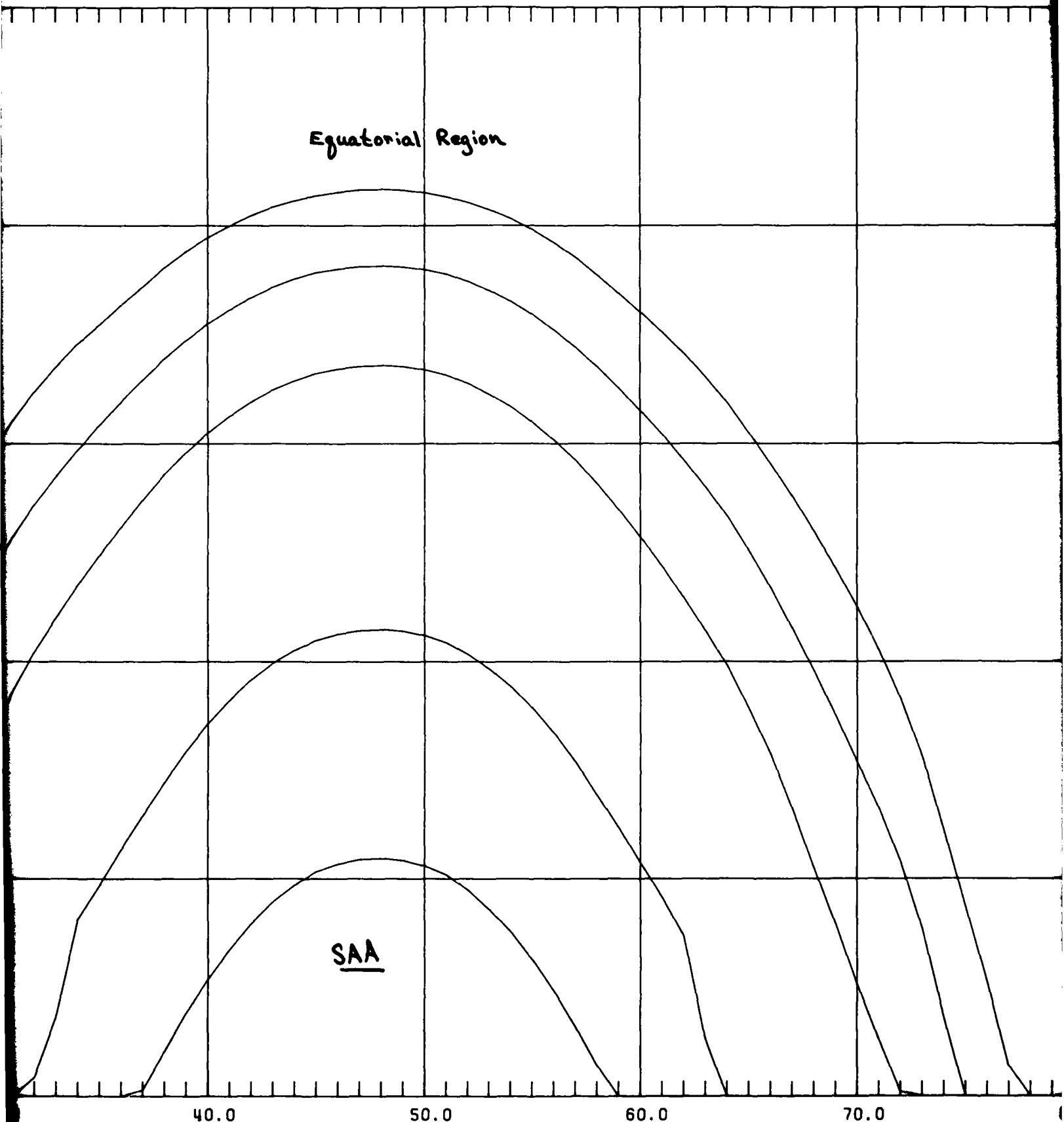
SAA

40.0

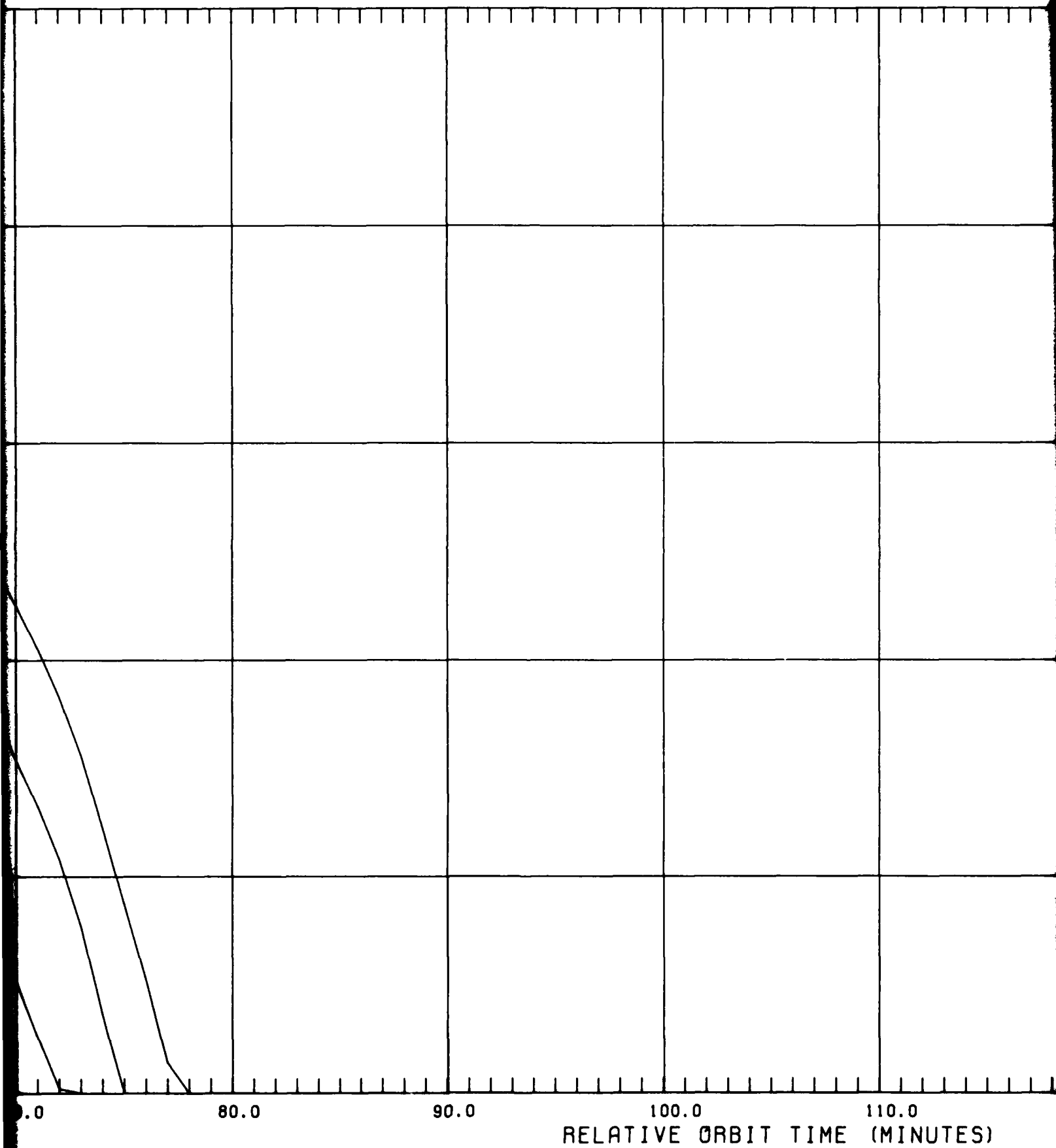
50.0

60.0

70.0



3'



4'

Equatorial

SAA

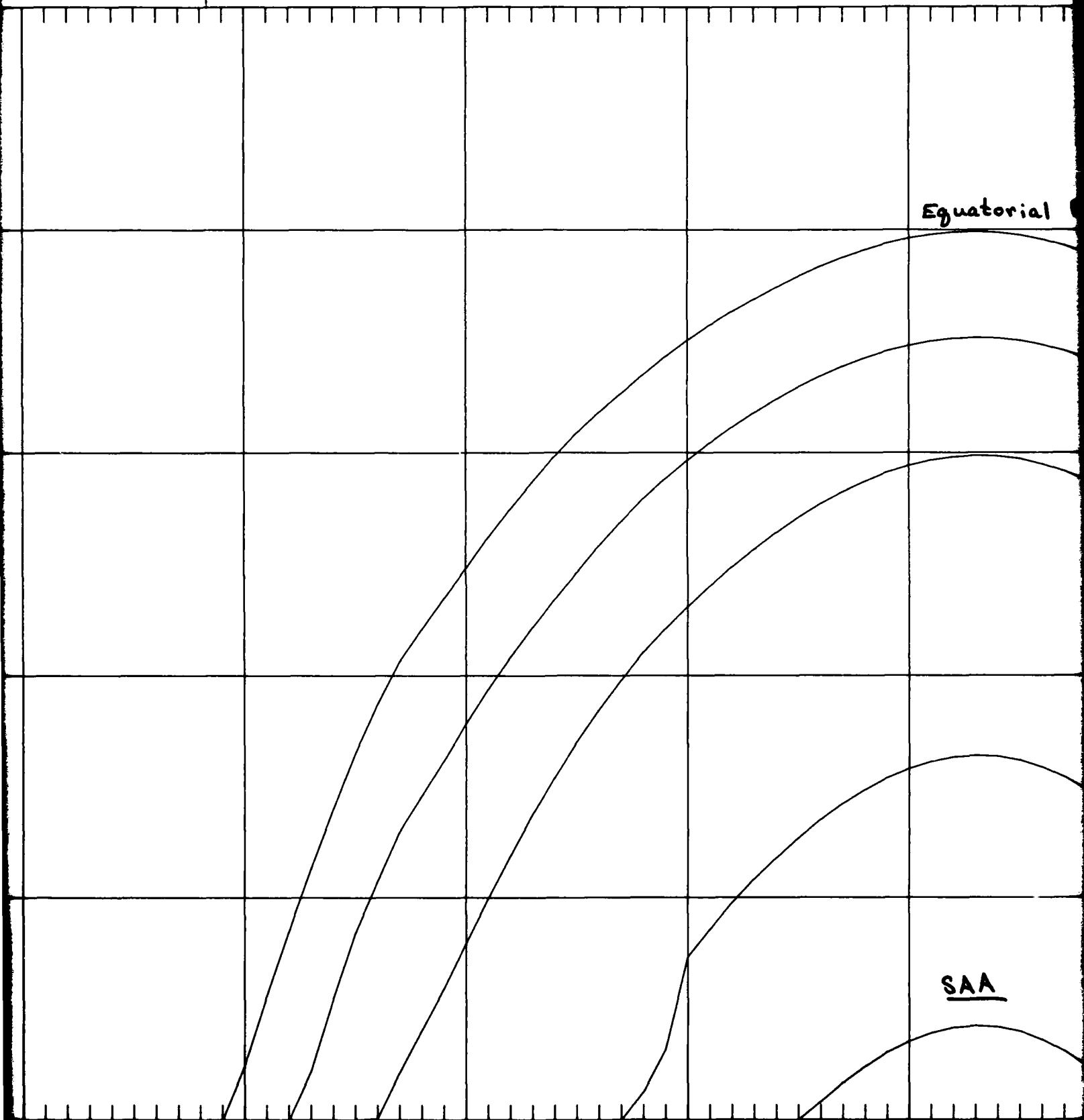
10.0
(MINUTES)

120.0

130.0

140.0

150.0



5.

Equatorial Region

ORBIT: NA
60 DGR/S

EPOCH: 196

MODELS:
FIELD: BA
TRAPPED P
INNER ZN
OUTER ZN
MISSION DU
EVALUATION

UN FACTORS

SAA

STOP TIME

150.0

160.0

170.0

180.0

190.0

Figure 109

ORBIT: NAVELEX 4
60 DGR/5186-5186 KM

EPOCH: 1989.5

MODELS:

FIELD: BARR/75

TRAPPED PROTONS: AP8

INNER ZN ELEC: AE6

OUTER ZN ELEC: AE17 L0

MISSION DURATION: 60.00 MO

EVALUATION PHASE: SOLAR MAX

UN FACTORS: NOT APPLIED

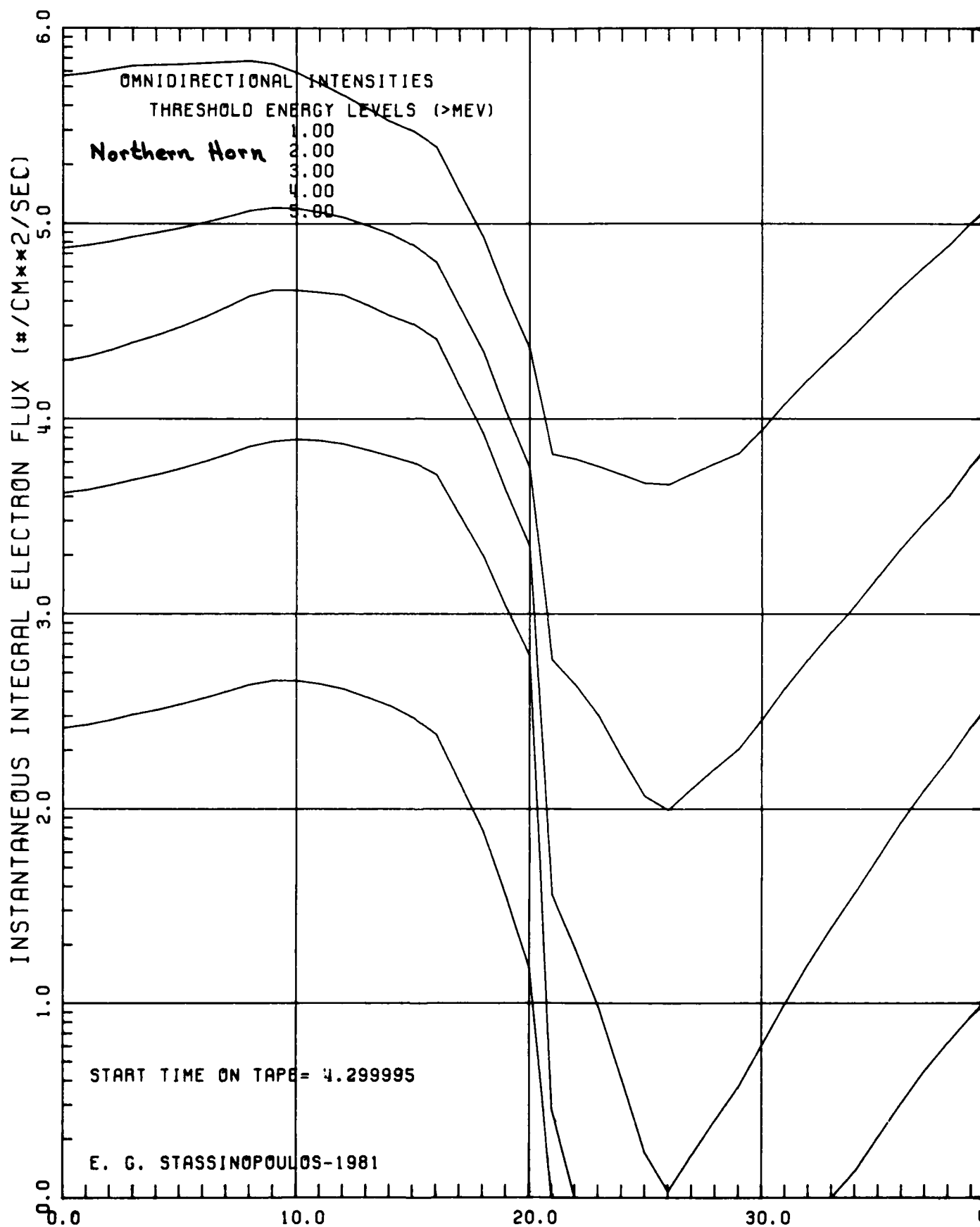
STOP TIME ON TAPE= 7.716662

NASA-GSFC

190.0

200.0

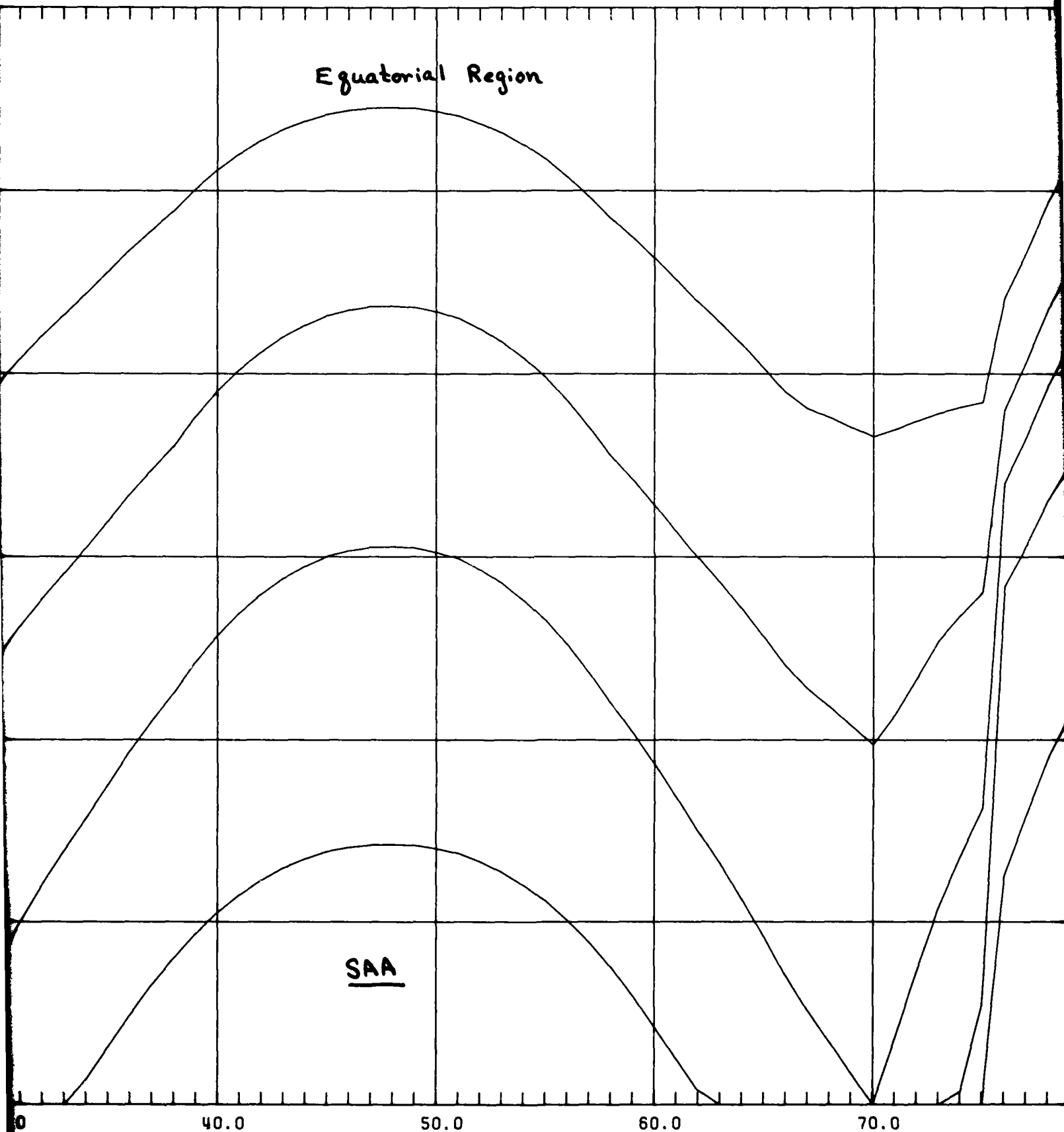
210.0



2

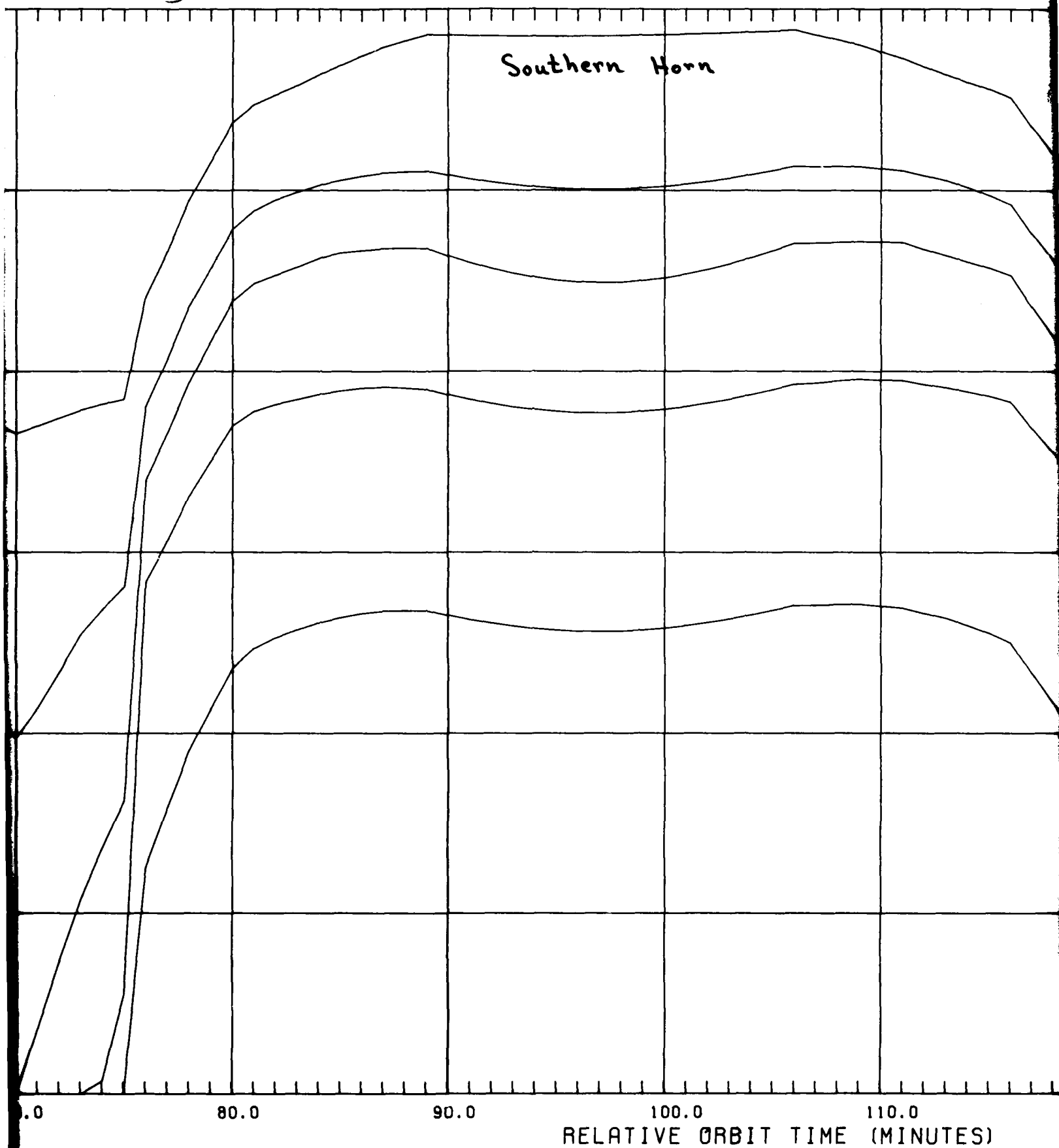
Equatorial Region

SAA

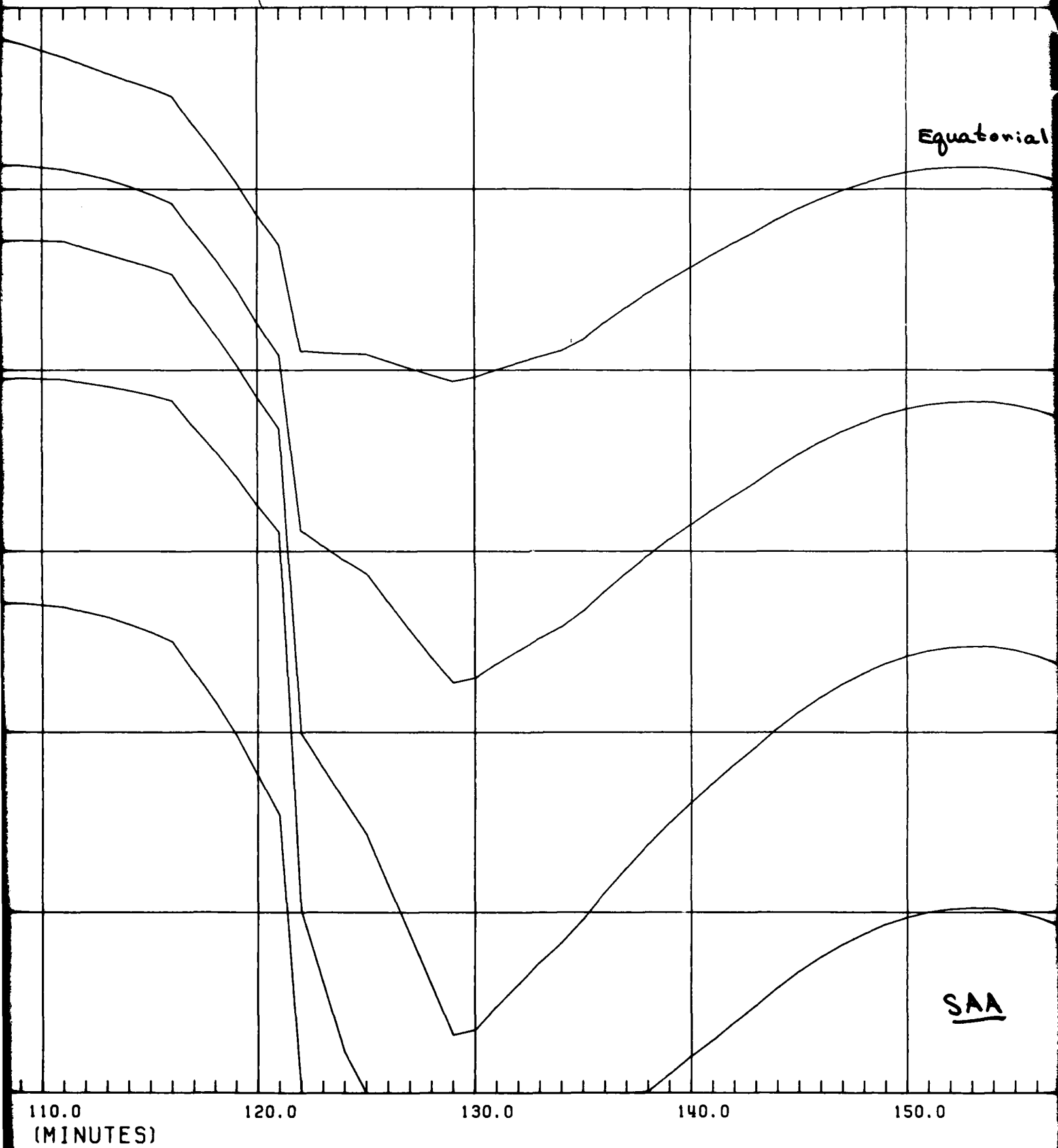


3'

Southern Horn



4



5'

Equatorial Region

Northern Horn

ORBIT: N
60 DGR

EPOCH: 1

MODELS:

FIELD:

TRAPPED

INNER Z

OUTER Z

MISSION

EVALUATI

UN FACTOR

SAA

STOP TIME

150.0

160.0

170.0

180.0

190.0

Northern Hemisphere

Figure 110

ORBIT: NAVELEX 4
60 DGR/5186-5186 KM

EPOCH: 1989.5

MODELS:

FIELD: BARR/75

TRAPPED PROTONS: AP8

INNER ZN ELEC: AE6

OUTER ZN ELEC: AE17-L0

MISSION DURATION: 60.00 MO

EVALUATION PHASE: SOLAR MAX

UN FACTORS: NOT APPLIED

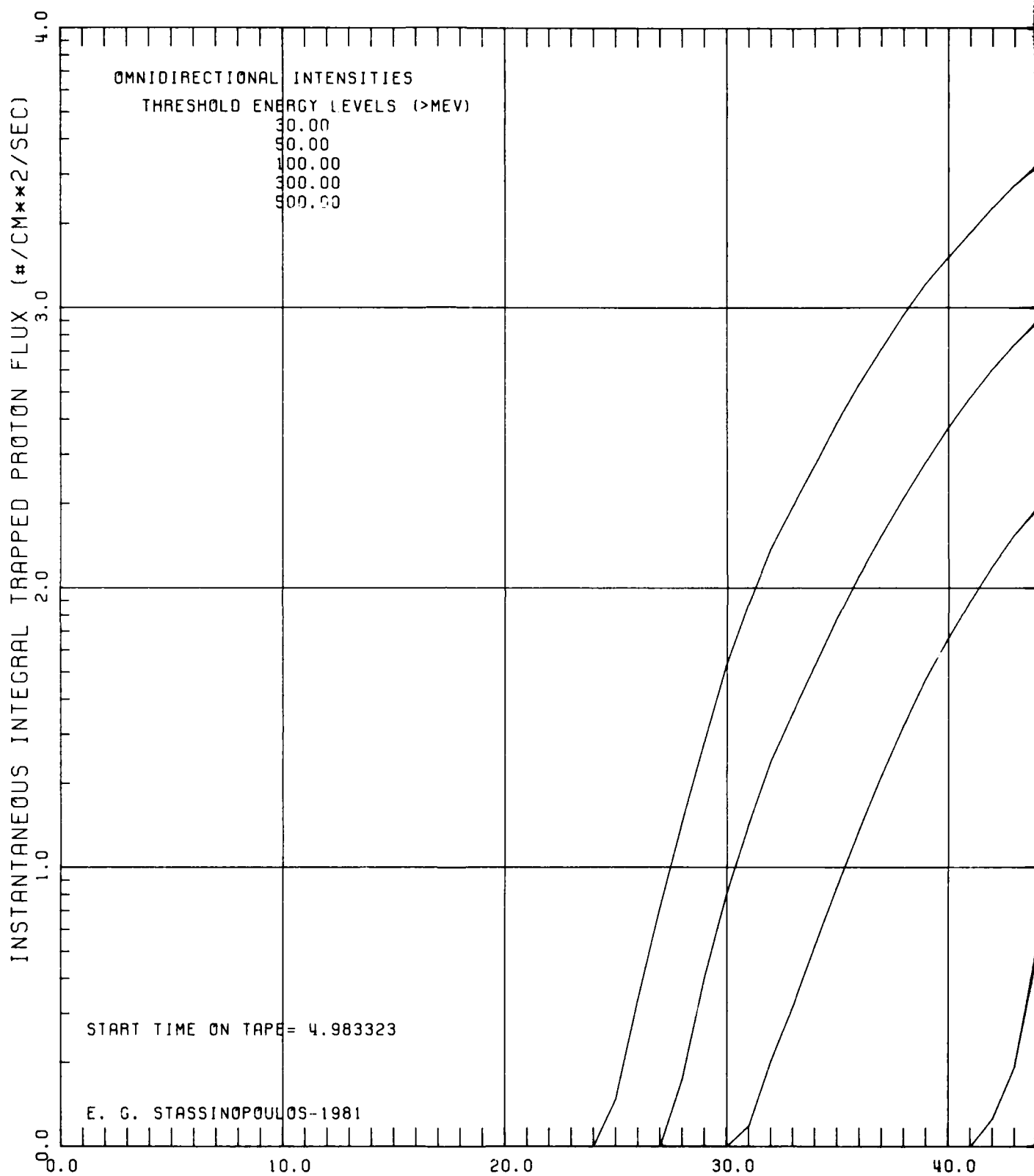
STOP TIME ON TAPE= 7.716662

NASA-GSFC

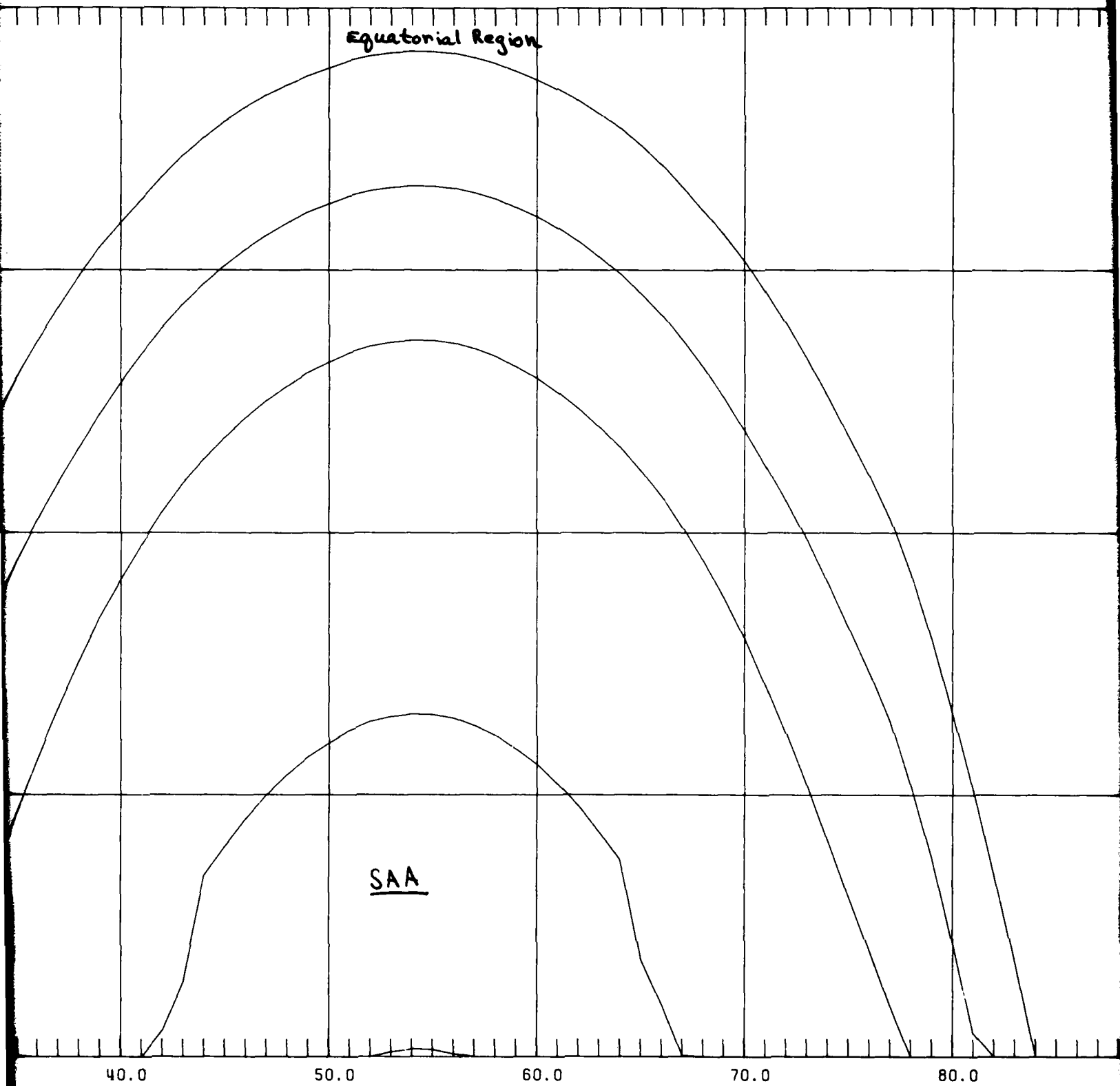
190.0

200.0

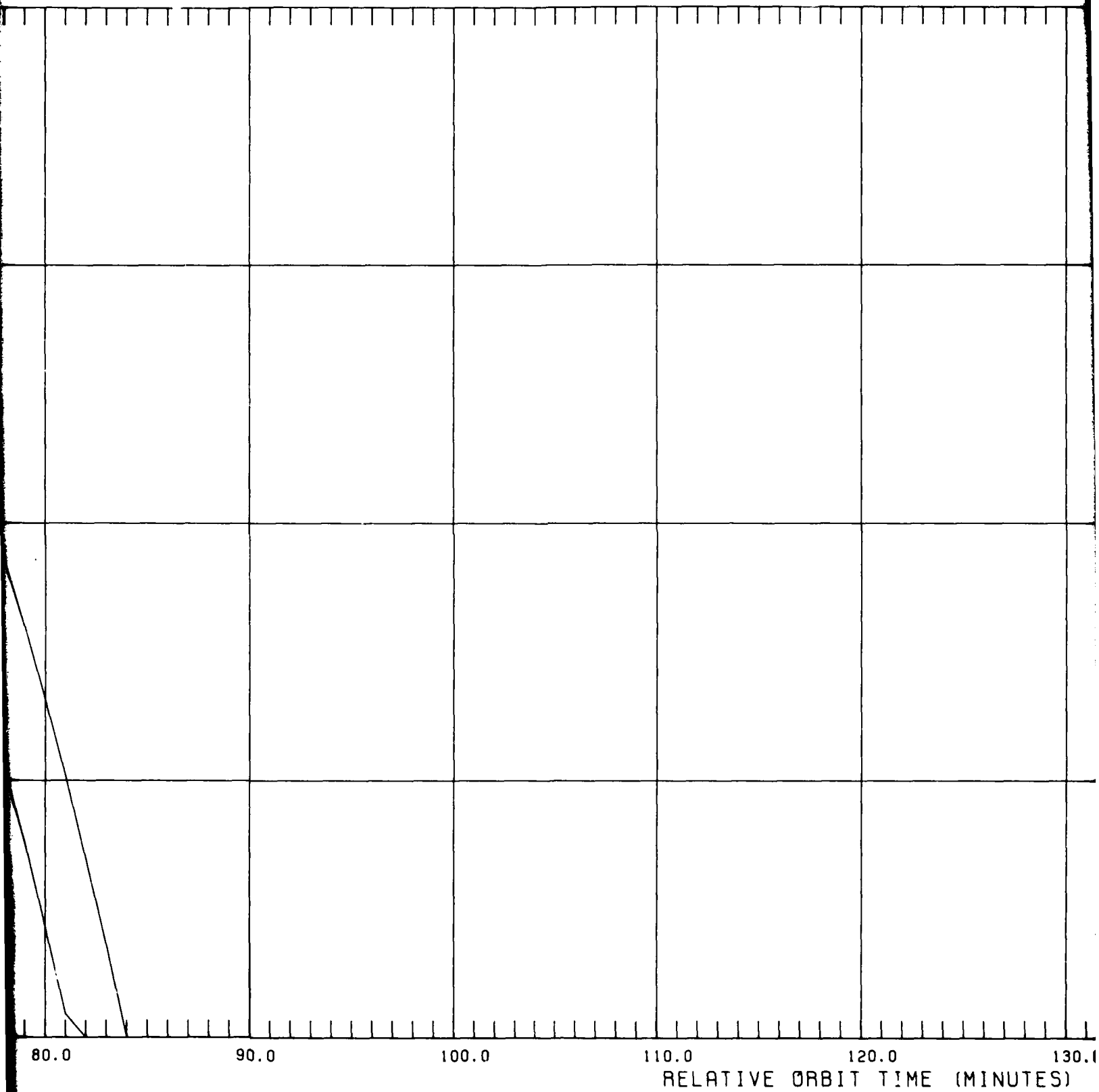
210.0



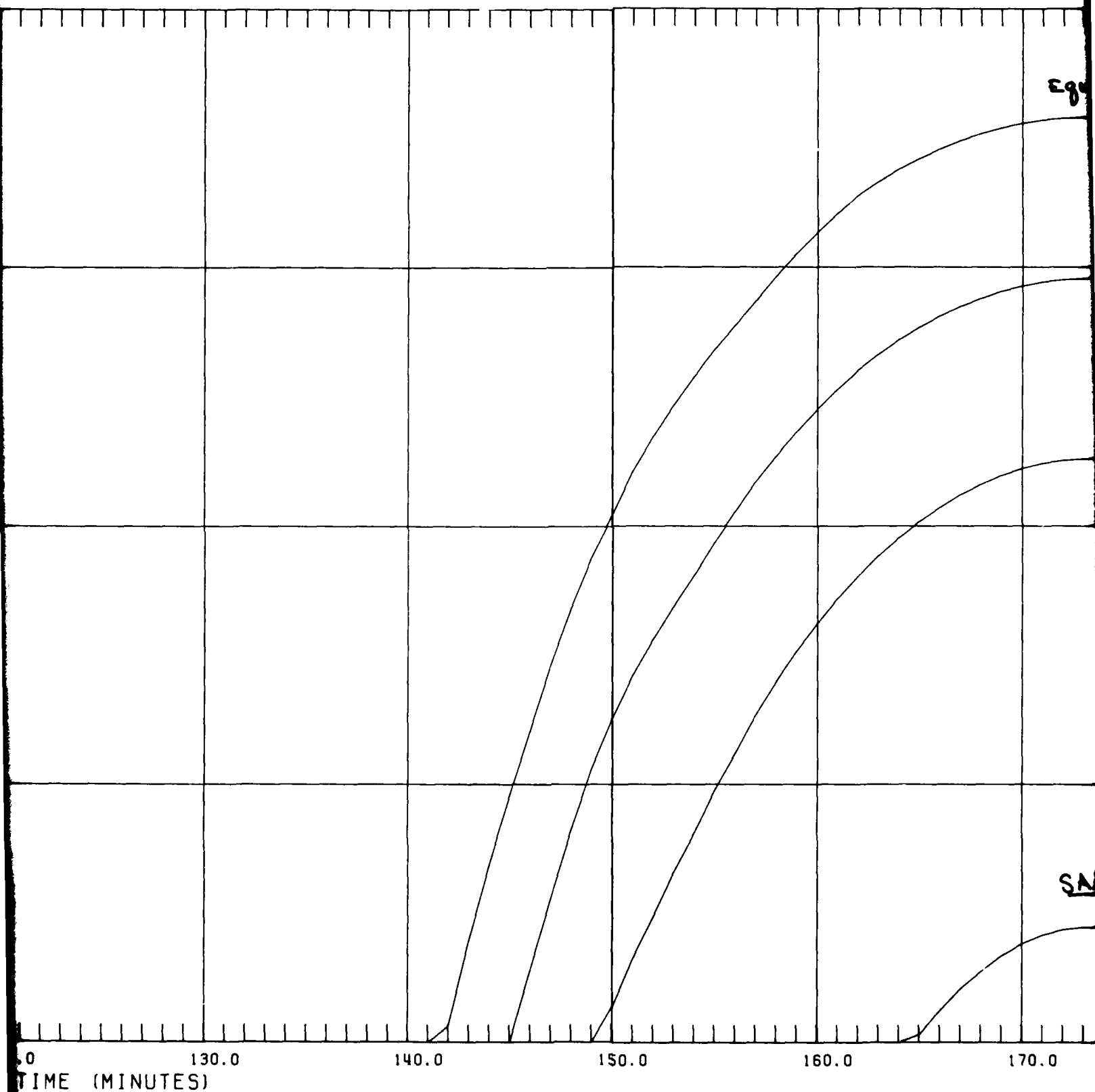
2¹



3'



4



5'

Equatorial Region

SAA

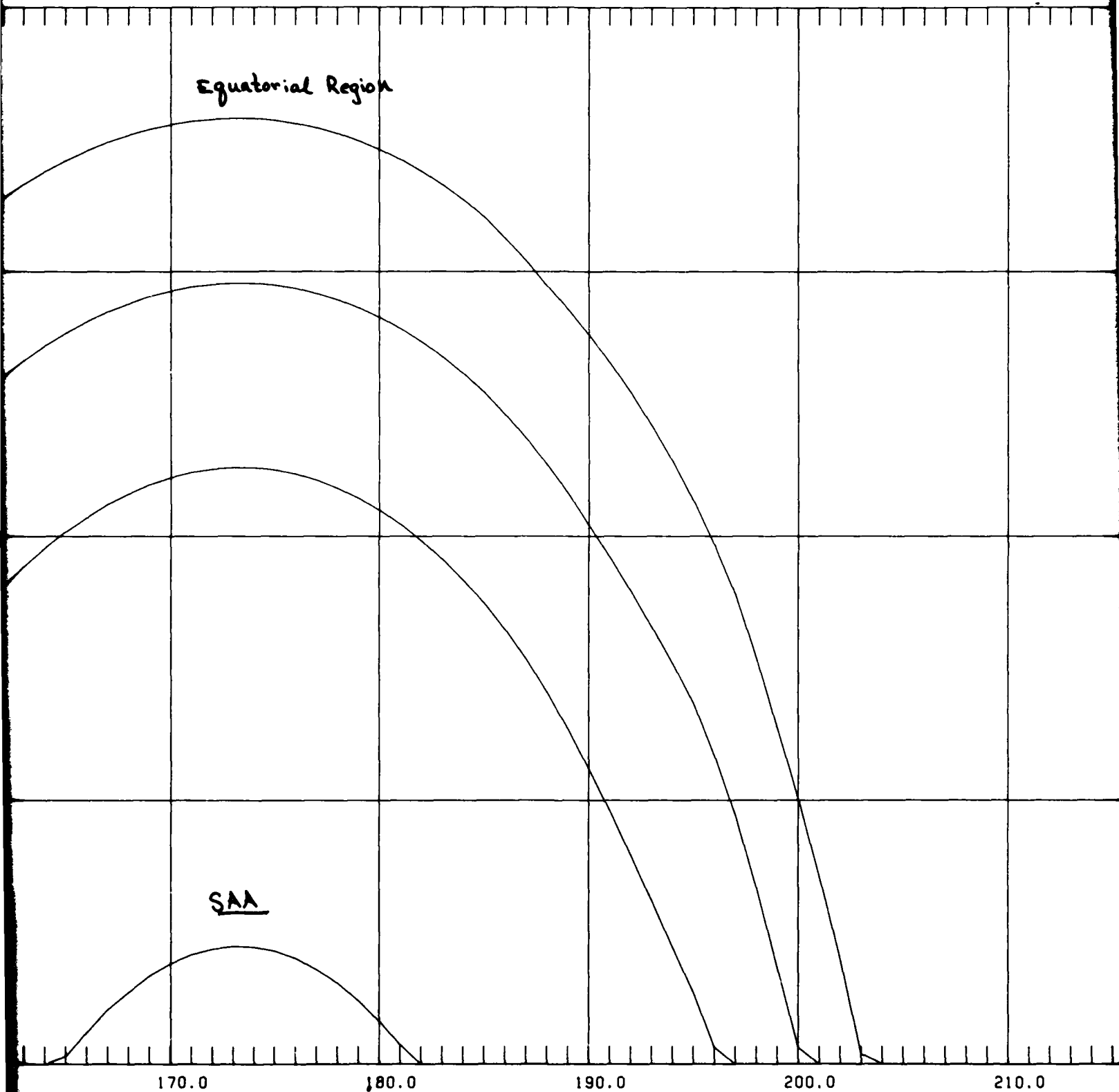
170.0

180.0

190.0

200.0

210.0



ORBIT: NAVELEX 5
60 DGR/6389-6889 KM

Figure III

EPOCH: 1989.5

MODELS:

FIELD: BARR/75

TRAPPED PROTONS: AP8

INNER ZN ELEC: AE6

OUTER ZN ELEC: AE17-L0

MISSION DURATION: 60.00 MO

EVALUATION PHASE: SOLAR MAX

UN FACTORS: NOT APPLIED

STOP TIME ON TAPE= 8.933318

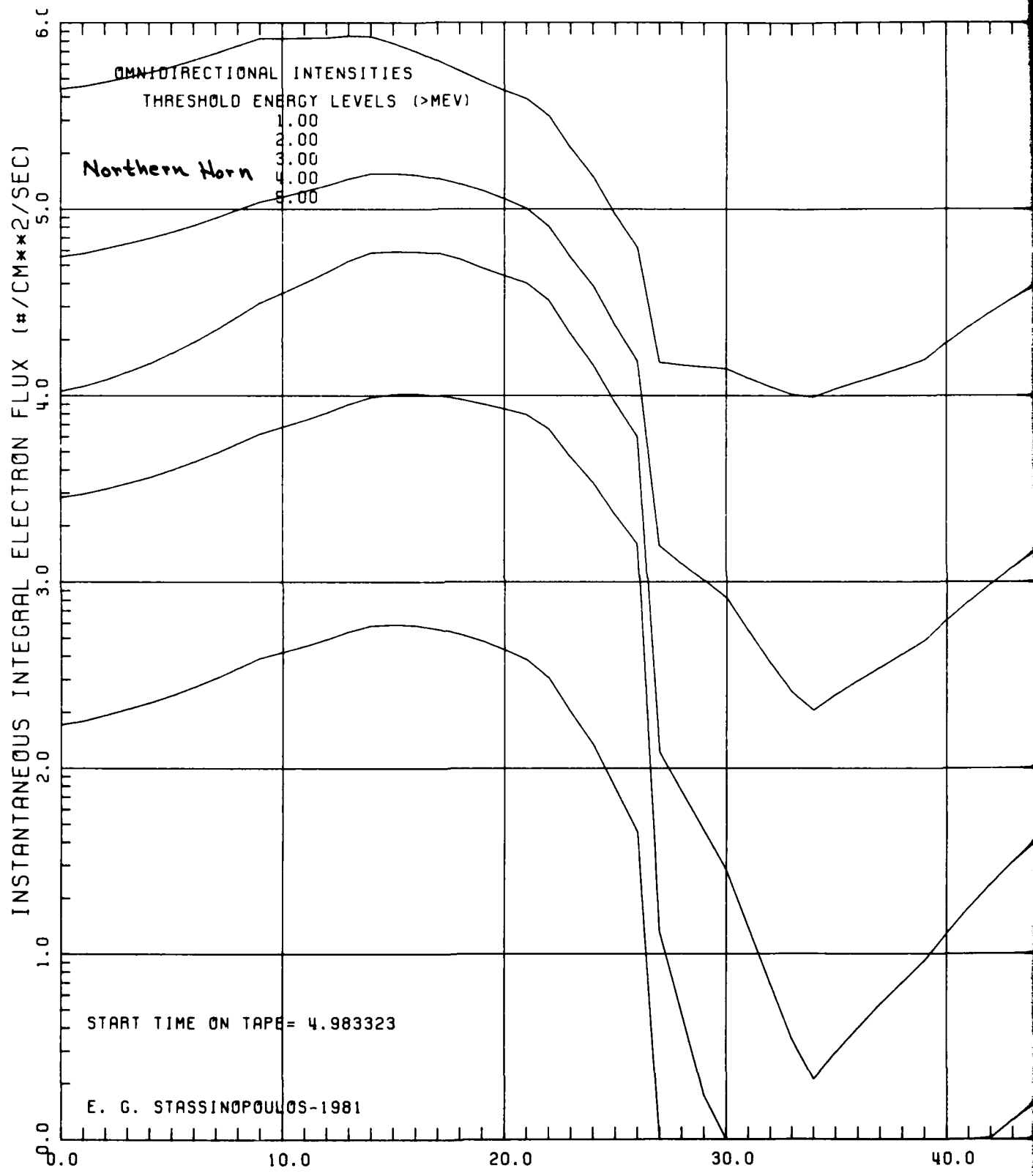
NASA-GSFC

210.0

220.0

230.0

240.0



2

Equatorial Region

SAA

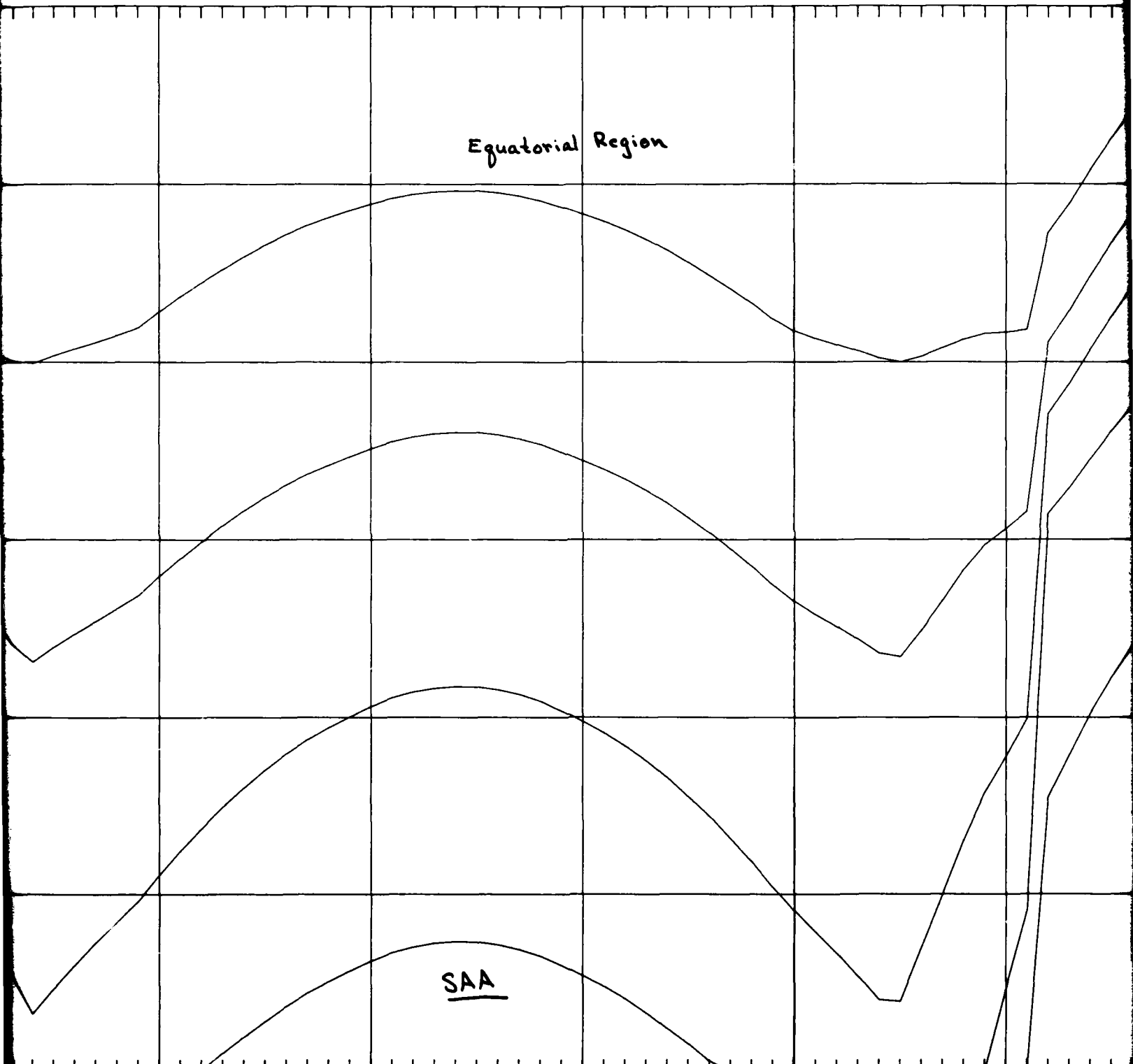
40.0

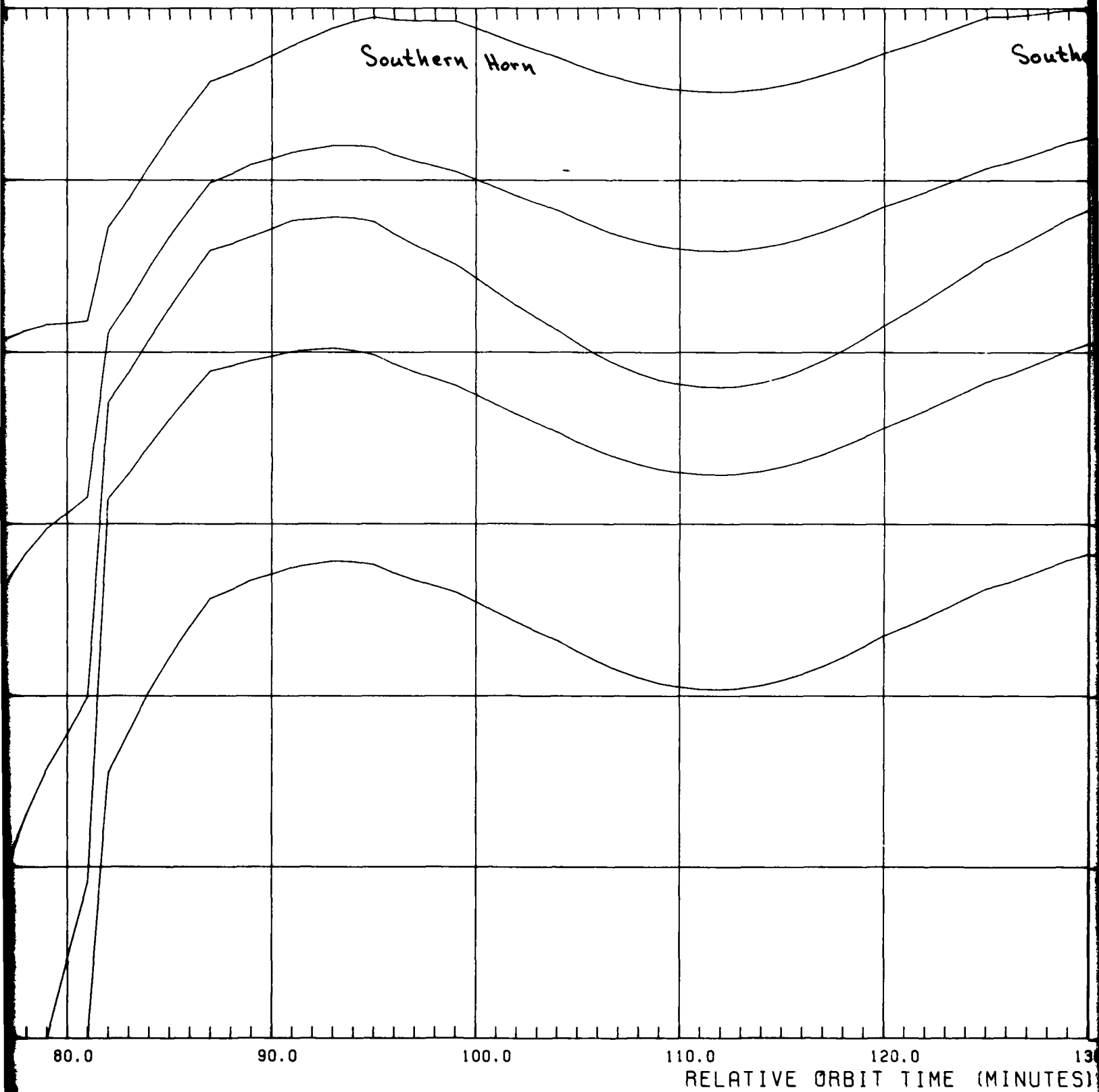
50.0

60.0

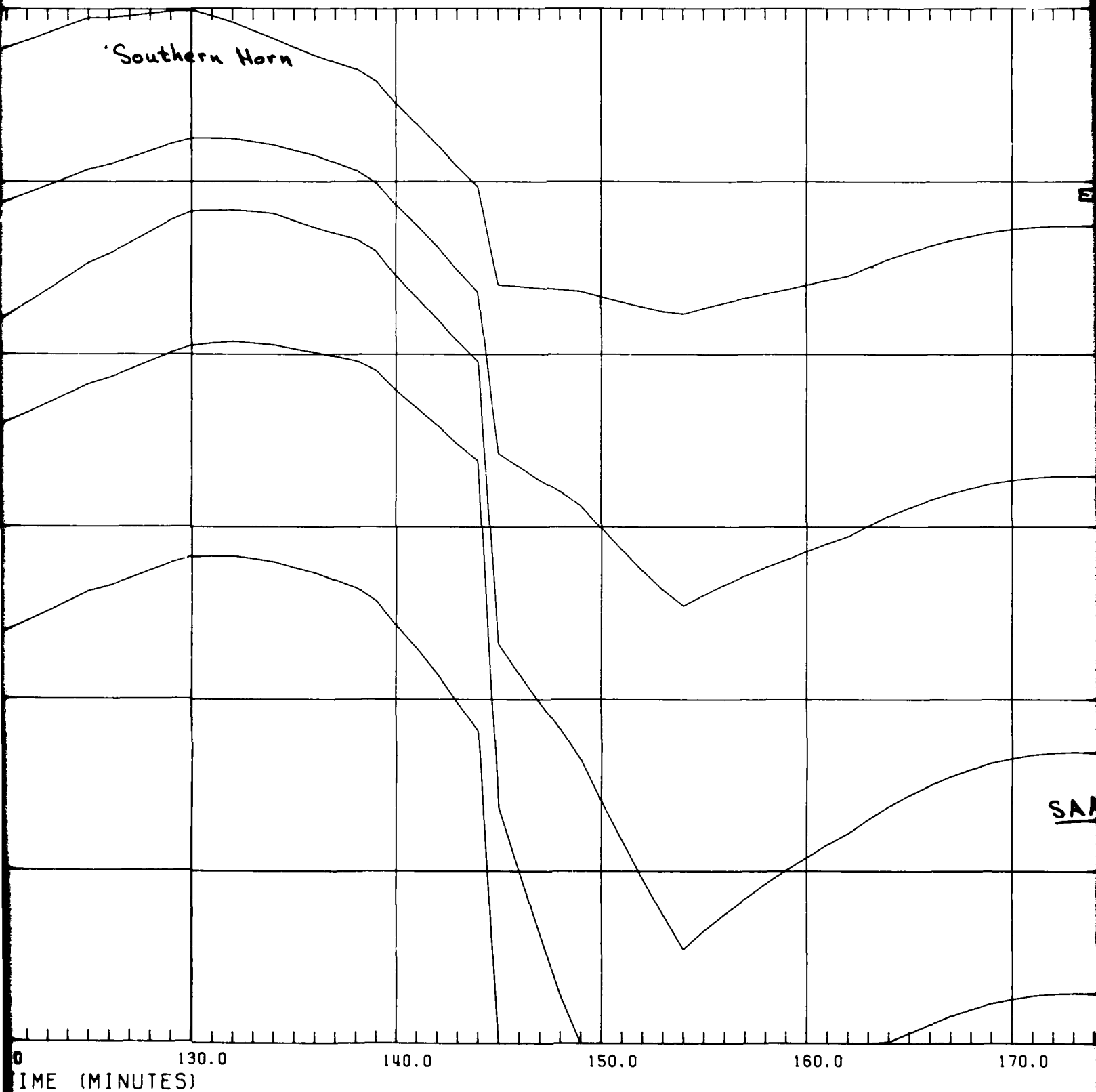
70.0

80.0

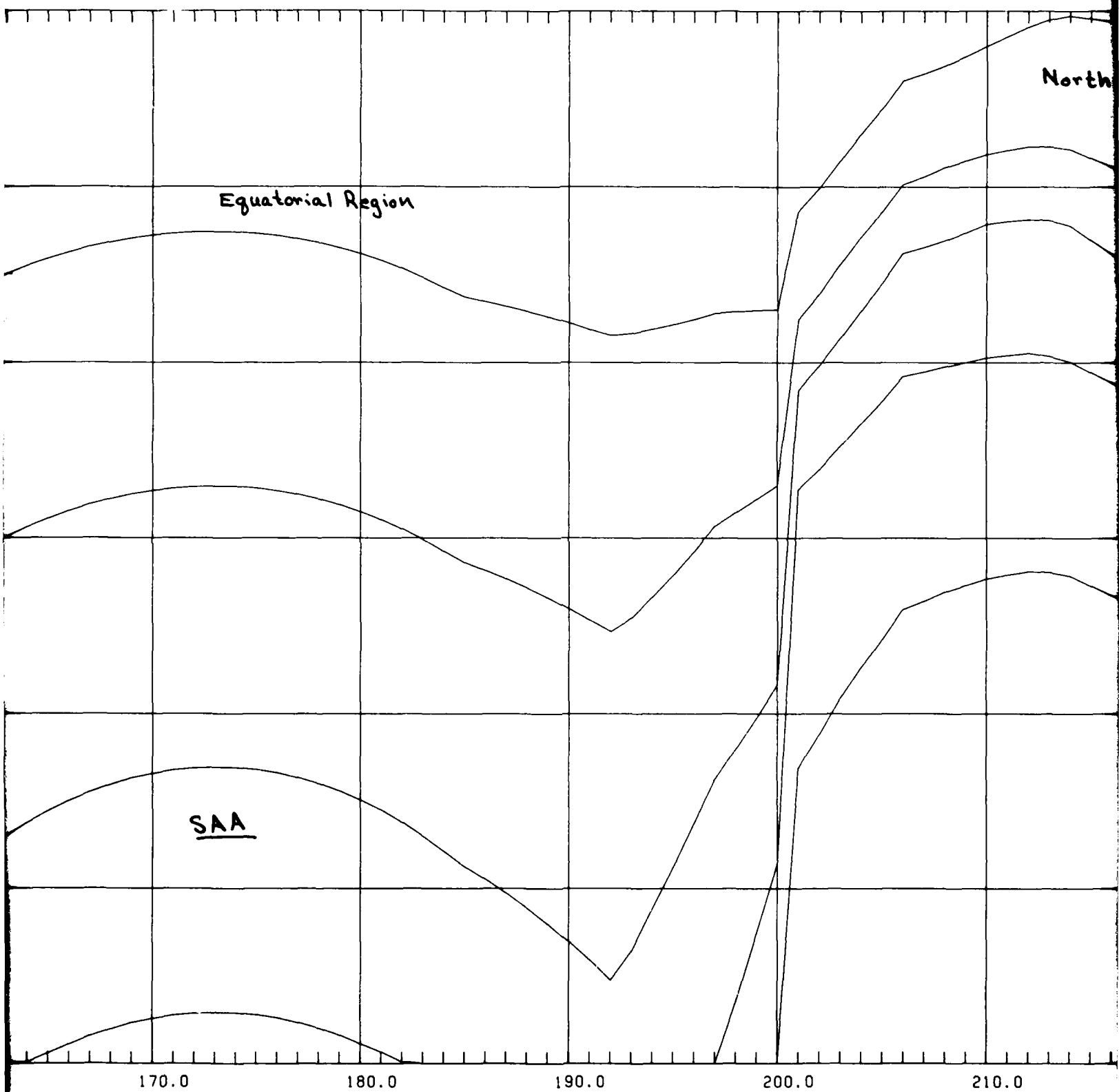


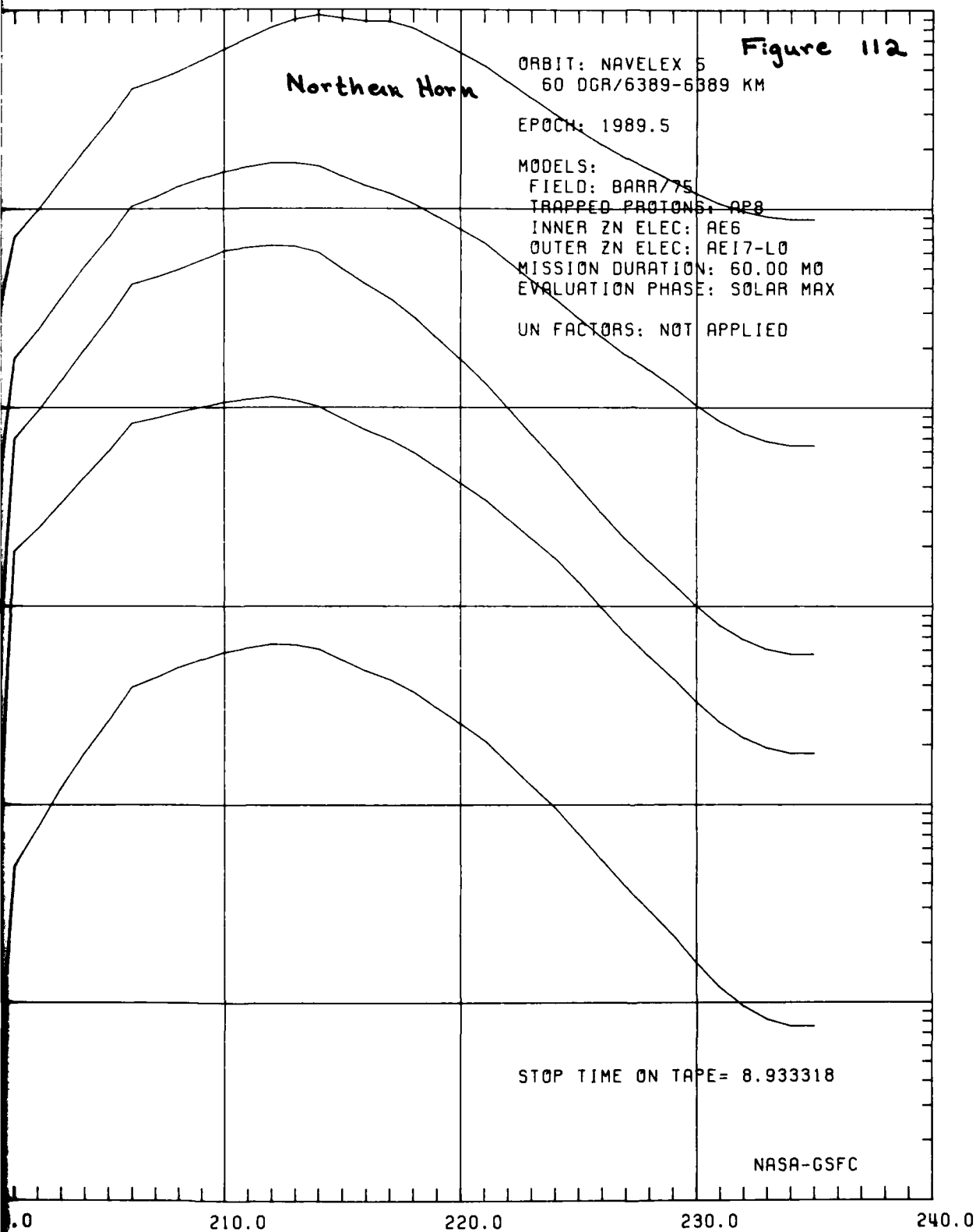


4



5





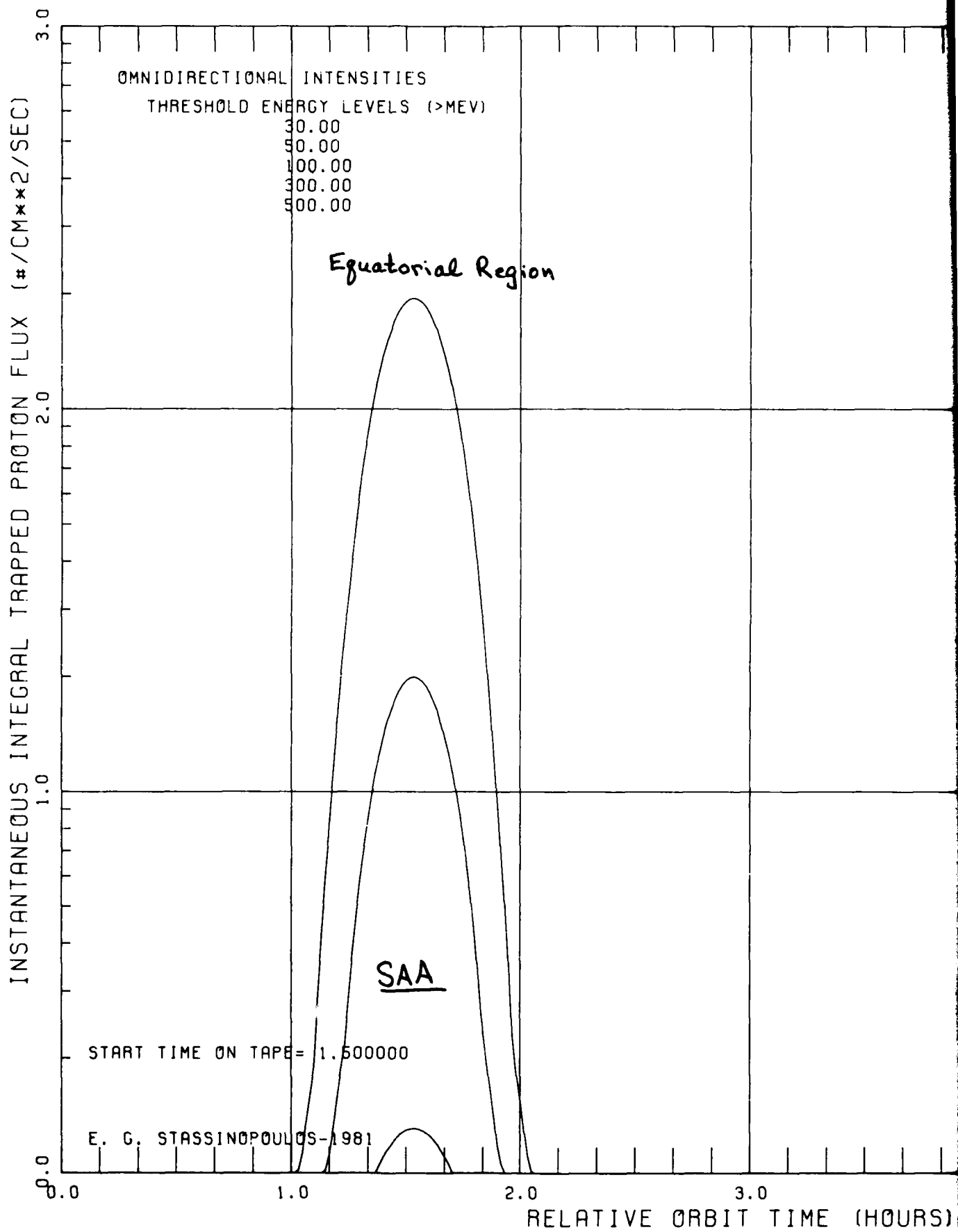


Figure 113

ORBIT: NAVELEX 6
60 DGR/10371-10371 KM

EPOCH: 1989.5

MODELS:

FIELD: BARR/75

TRAPPED PROTONS: AP8

INNER ZN ELEC: AE6

OUTER ZN ELEC: AE17-L0

MISSION DURATION: 60.00 MO

EVALUATION PHASE: SOLAR MAX

UN FACTORS: NOT APPLIED

Equatorial Region

SAA

STOP TIME ON TAPE= 7.483318

NASA-GSFC

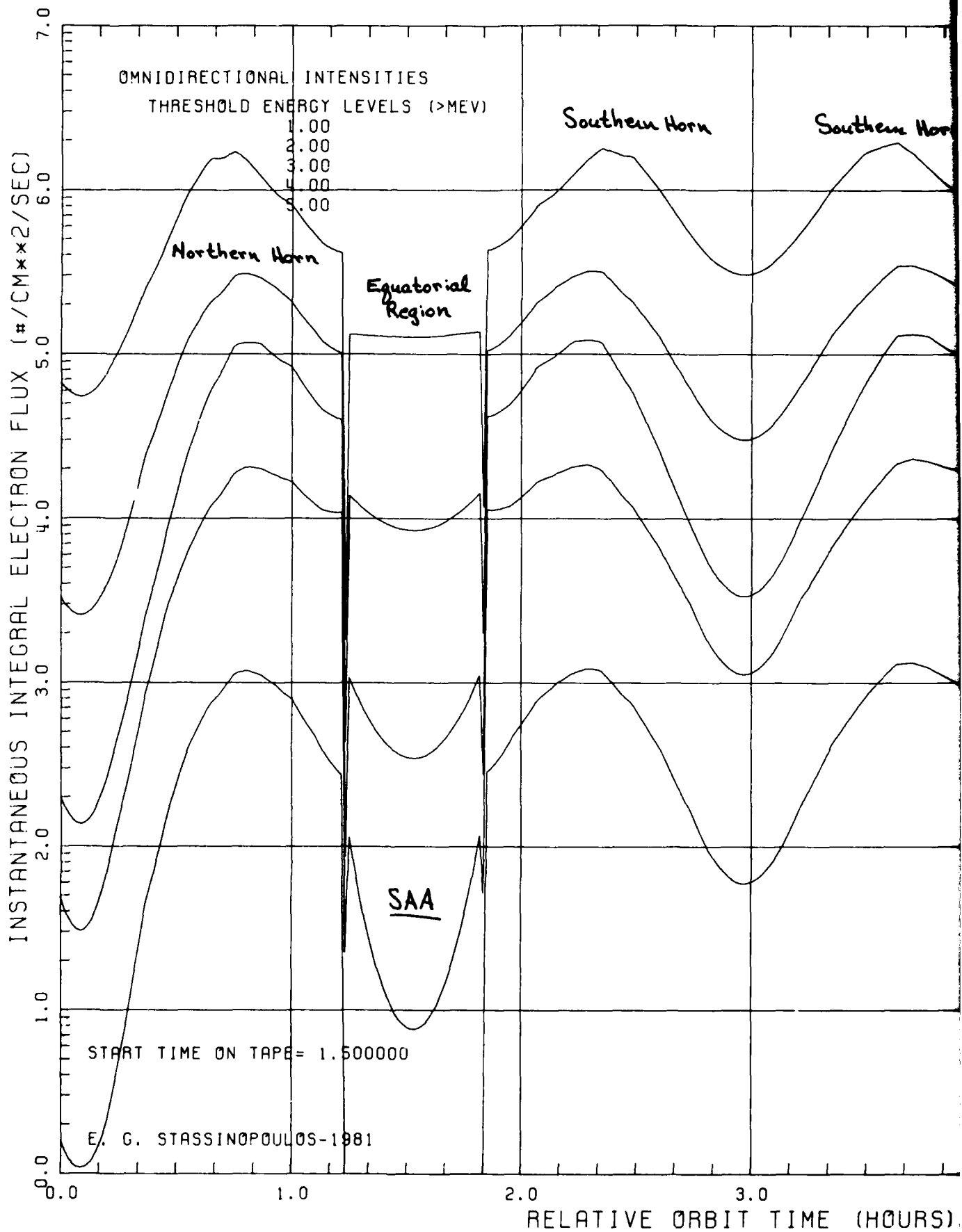
3.0

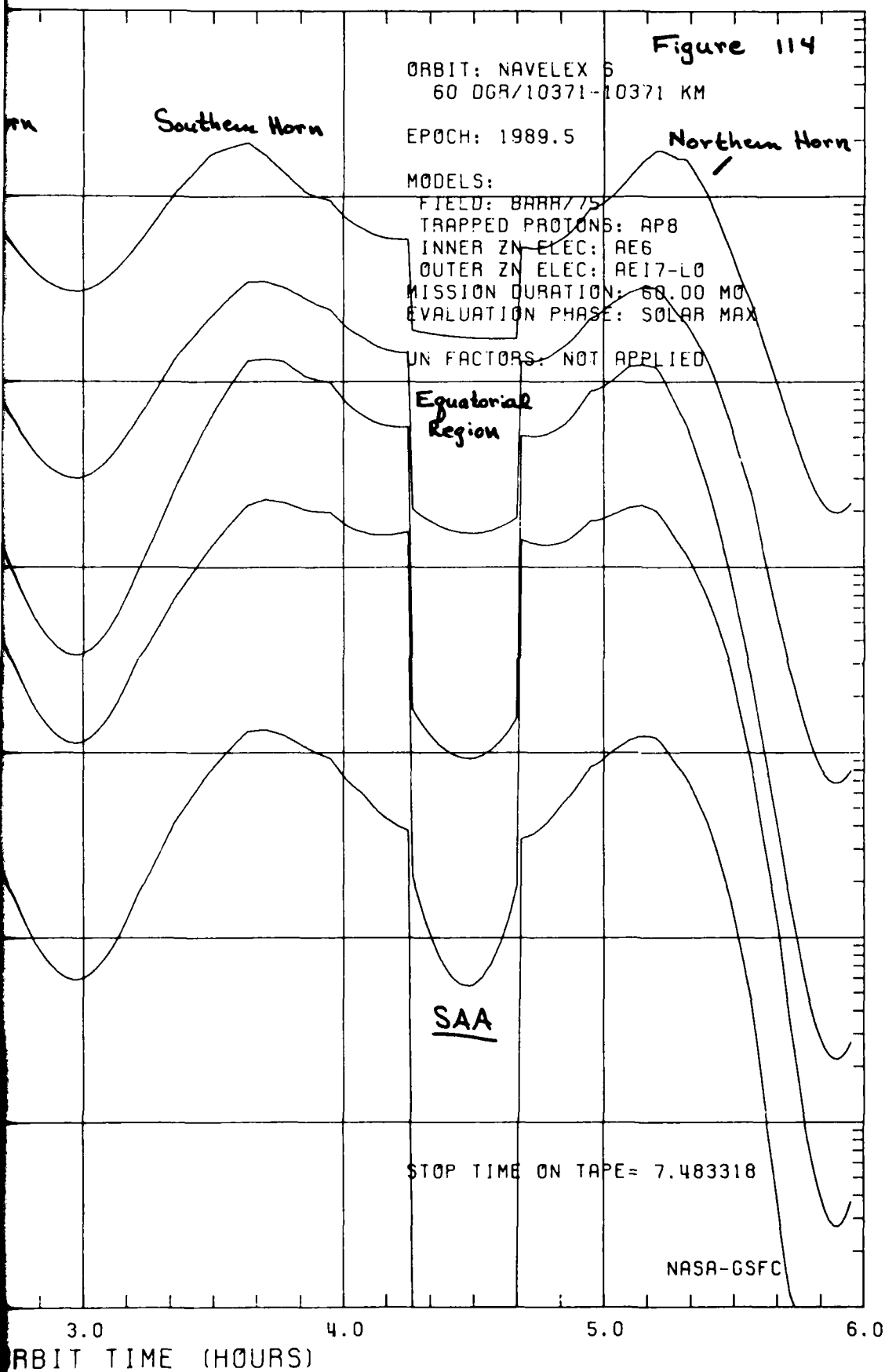
4.0

5.0

6.0

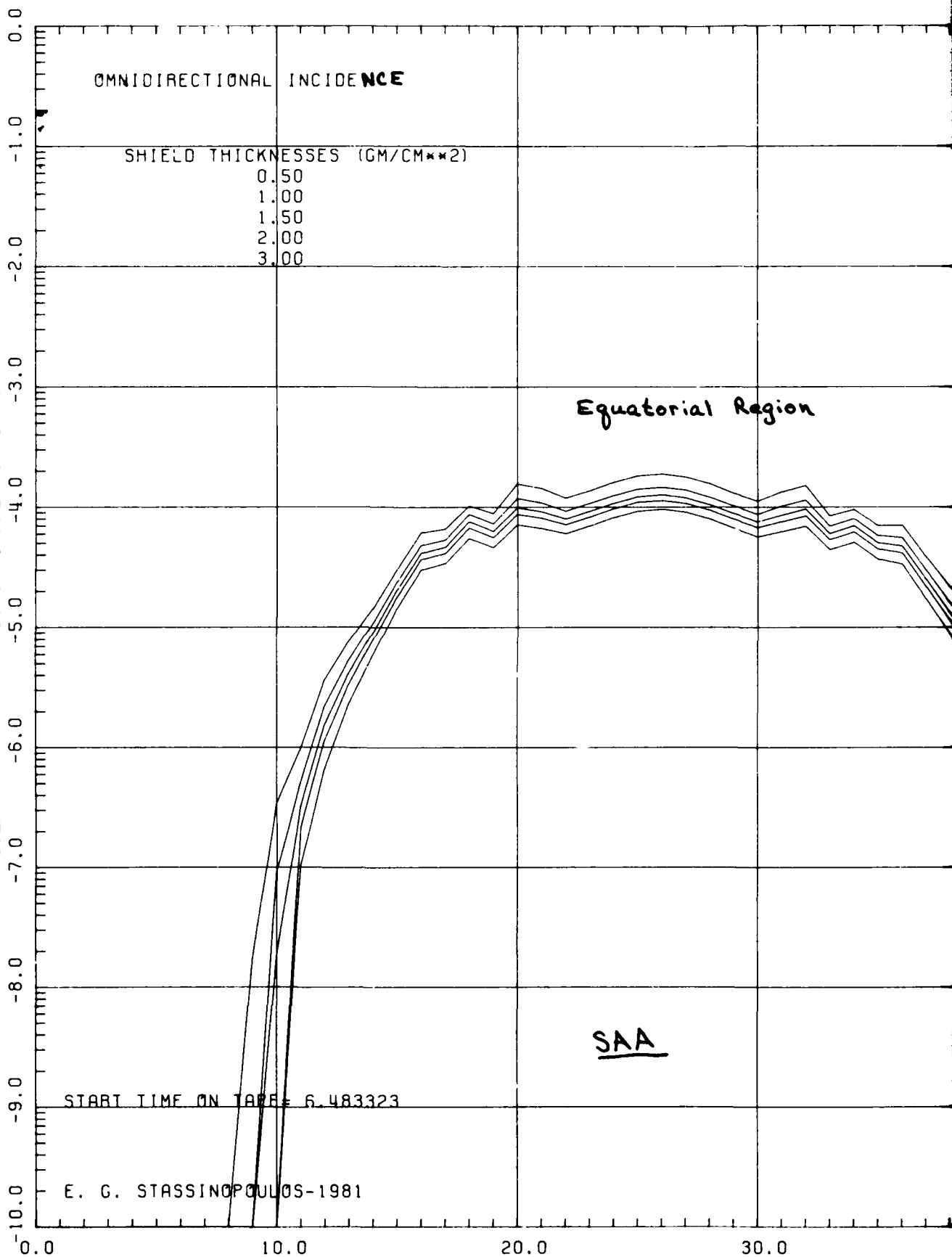
ORBIT TIME (HOURS)





INSTANTANEOUS ALUMINUM PROTON DOSE (RADS)

(SOLAR PROTON CONTRIBUTIONS ARE NOT INCLUDED)

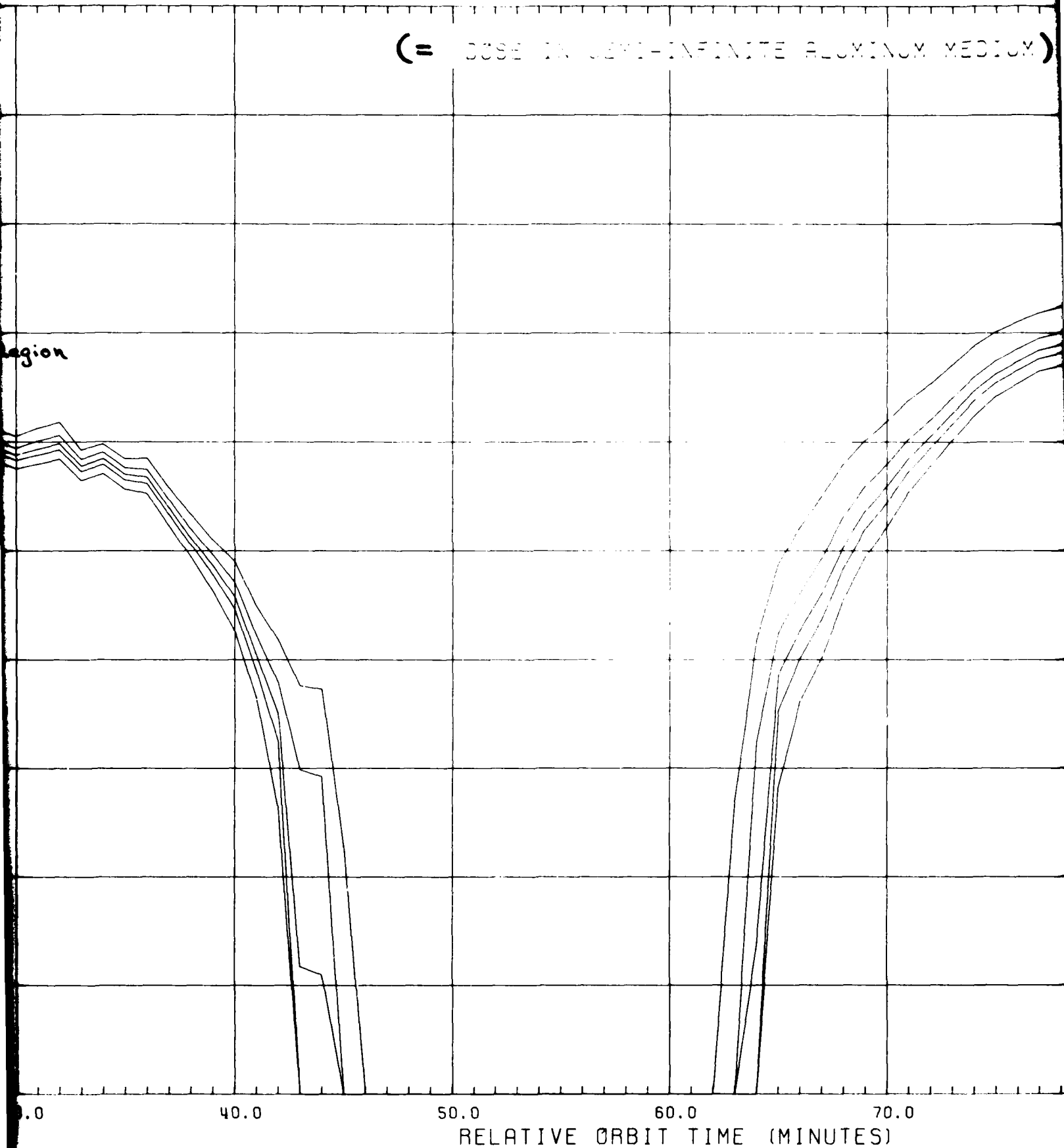


2

DOSE AT TRANSMISSION SURFACE OF FINITE ALUMINUM SLAB

(= DOSE IN SEMI-INFINITE ALUMINUM MEDIUM)

Region



RELATIVE ORBIT TIME (MINUTES)

OF FINITE ALUMINUM SLAB SHIELDS

FINITE ALUMINUM MEDIUM)

ORBIT: NAVELEX 1
60 DGR/1667-15

EPOCH: 1989.5

MODELS:
FIELD: BARR/75
TRAPPED PROTONS
INNER ZN ELEC:
OUTER ZN ELEC:

MISSION DURATION:
EVALUATION PHASE:

UN FACTORS: NOT A

Equatorial Region

SAA

STOP TIME ON TAPE

70.0 80.0 90.0 100.0 110.0
ME (MINUTES)

Figure 115

ORBIT: NAVELEX 1
60 DGR/1667-1667 KM

EPOCH: 1989.5

MODELS:

FIELD: BARR/75

TRAPPED PROTONS: AP8

INNER ZN ELEC: AE6

OUTER ZN ELEC: AE17 L0

MISSION DURATION: 60.00 MO

EVALUATION PHASE: SOLAR MAX

UN FACTORS: NOT APPLIED

STOP TIME ON TAPE = 8.449993

NASA-GSFC

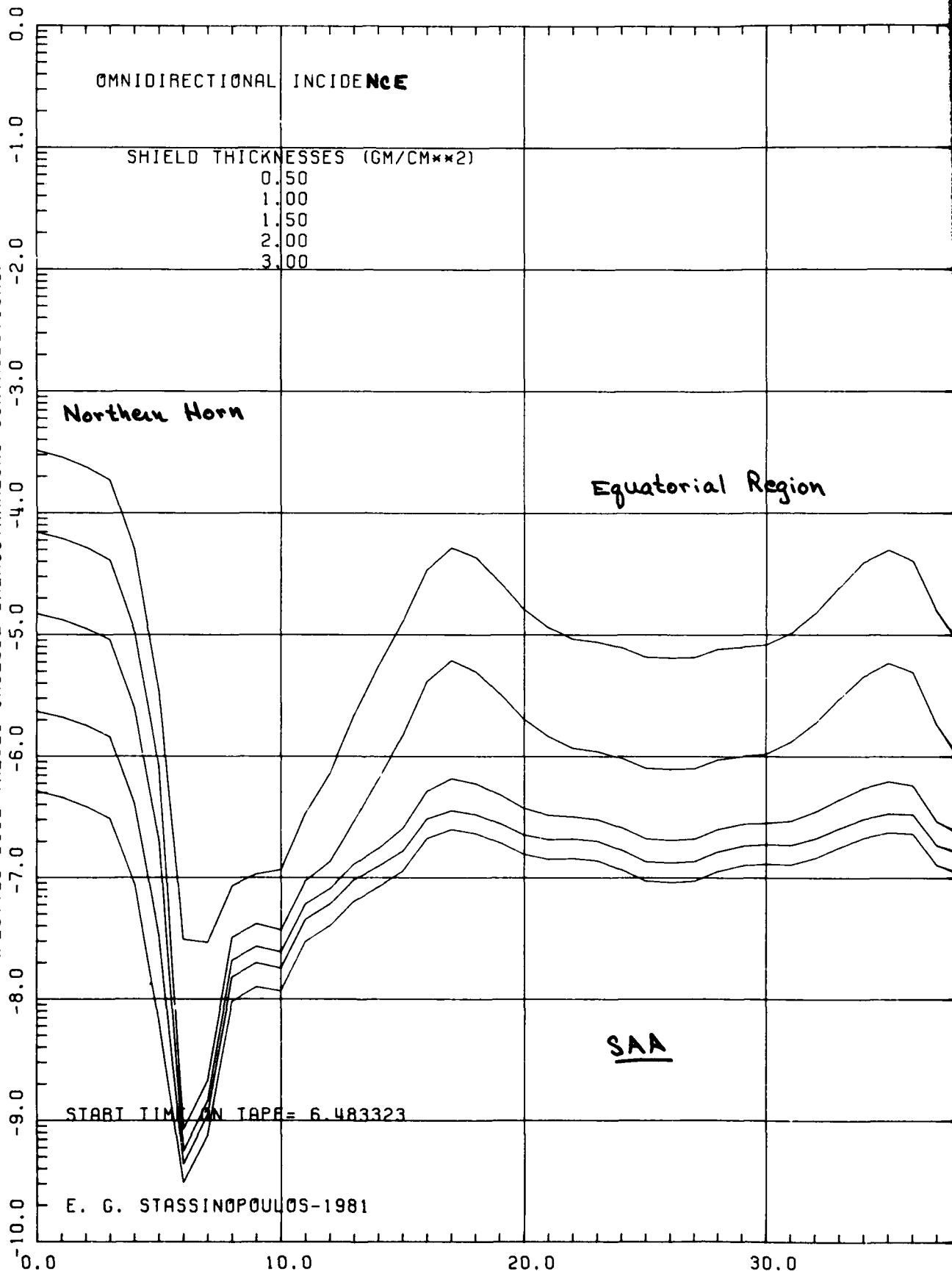
100.0

110.0

120.0

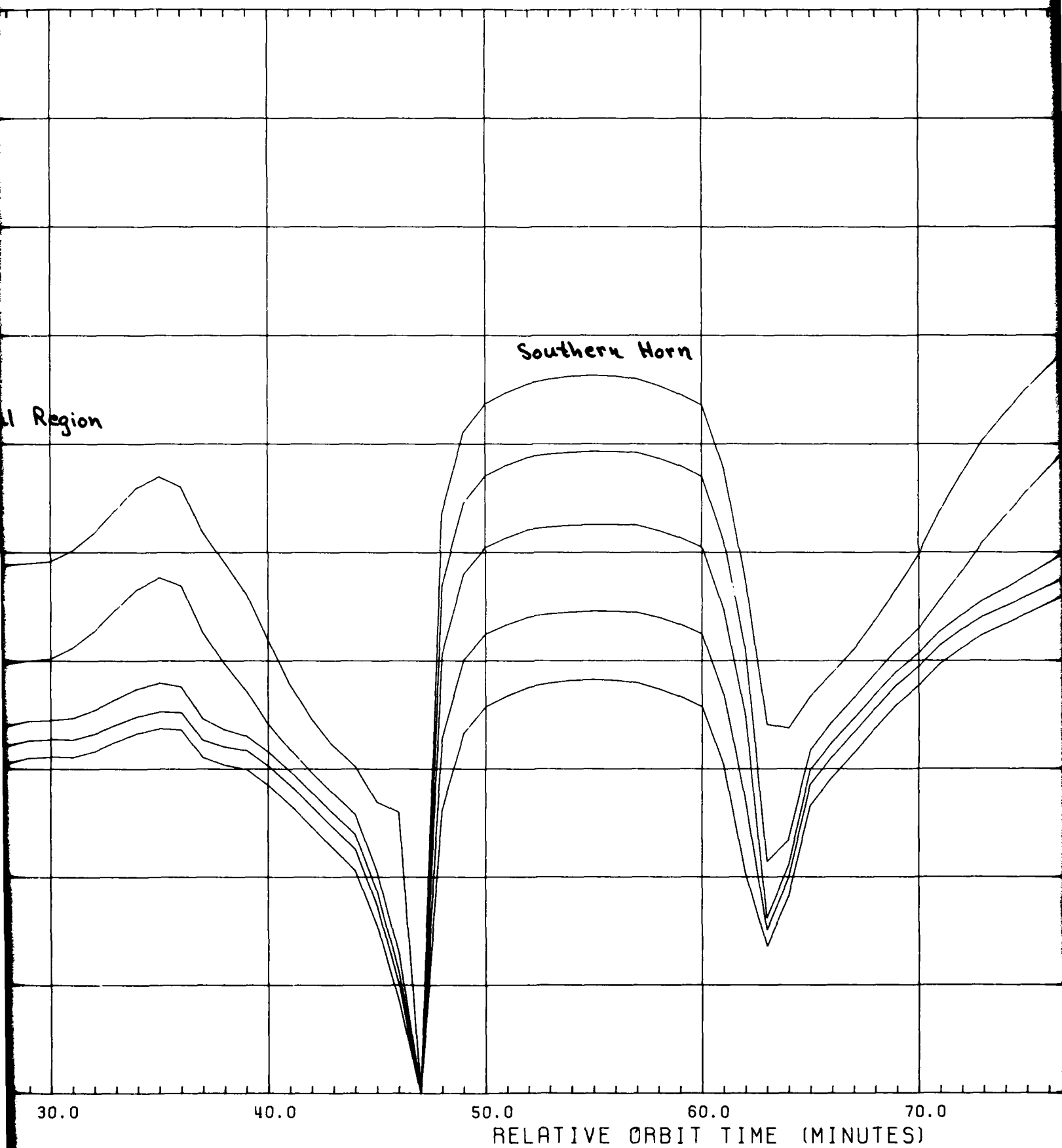
INSTANTANEOUS ALUMINUM ELECTRON DOSE (RADS)

(PLOTTED DOSE VALUES INCLUDE BREMSSTRAHLUNG CONTRIBUTIONS)



2

DOSE AT TRANSMISSION SURFACE OF FINITE ALUMINUM S



31

FINITE ALUMINUM SLAB SHIELDS

Figure

ORBIT: NAVELEX 1
60 DGR/1667-1667 KM
EPOCH: 1989.5
MODELS:
FIELD: BARR/75
TRAPPED PROTONS: AP8
INNER ZN ELEC: AE6
OUTER ZN ELEC: AE17 L
MISSION DURATION: 60.0
EVALUATION PHASE: SOLAR
UN FACTORS: NOT APPLIED

Equatorial Region

No

SAA

STOP TIME ON TAPE= 8.4

NA

70.0 80.0 90.0 100.0 110.0
MINUTES)

Figure 116

ORBIT: NAVELEX 1
60 DGR/1667-1667 KM

EPOCH: 1989.5

MODELS:
FIELD: BARR/75
TRAPPED PROTONS: AP8
INNER ZN ELEC: AE6
OUTER ZN ELEC: AE17 L0

MISSION DURATION: 60.00 MO
EVALUATION PHASE: SOLAR MAX

UN FACTORS: NOT APPLIED

Northern Horn

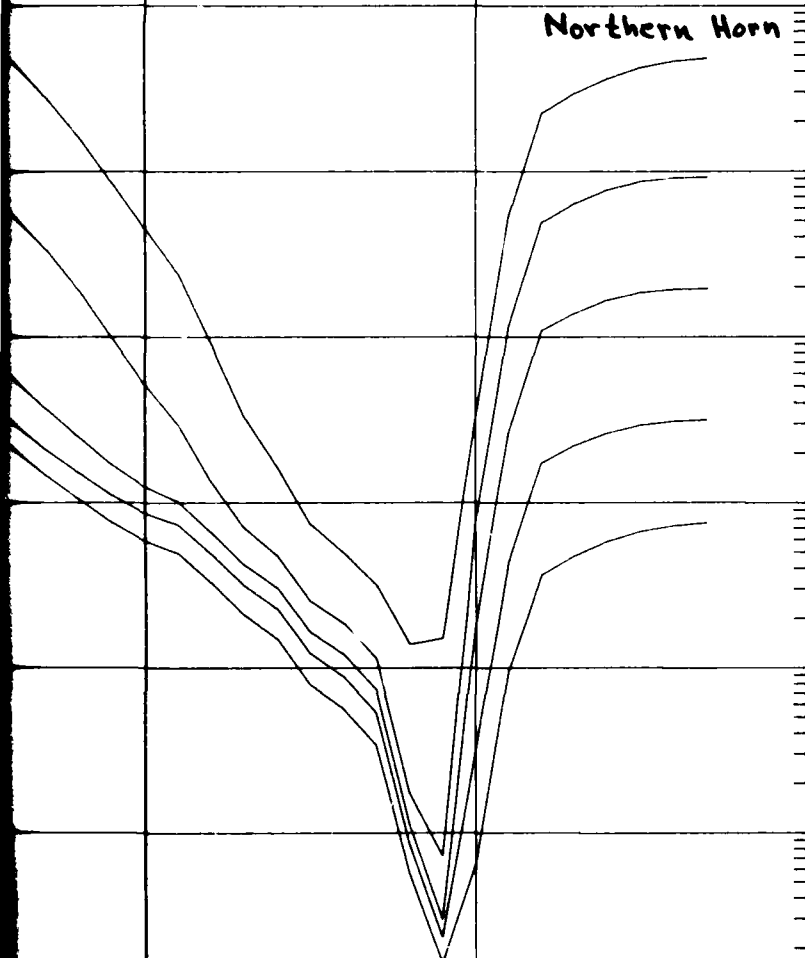
STOP TIME ON IAPF = 8.449993

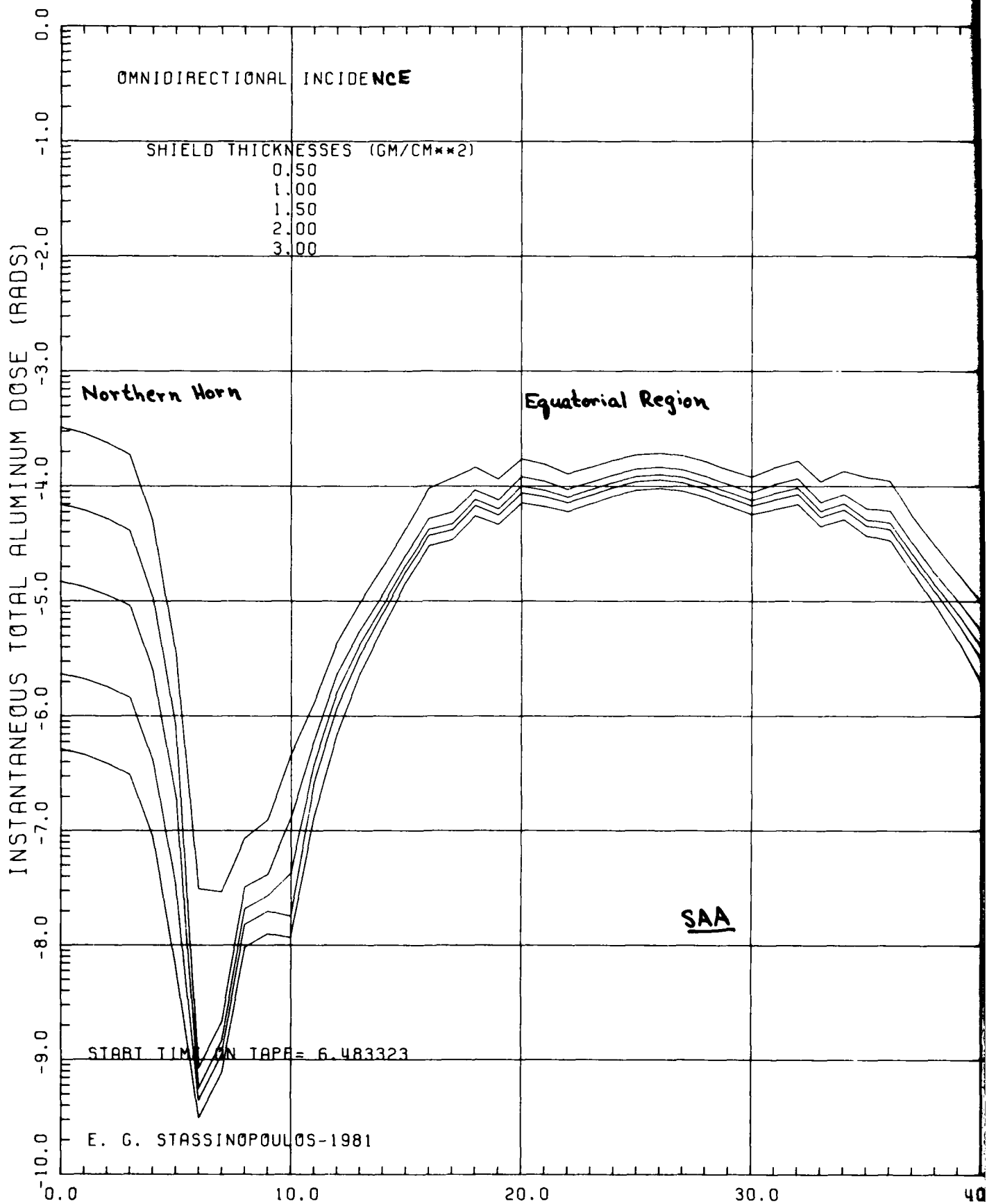
NASA-GSFC

100.0

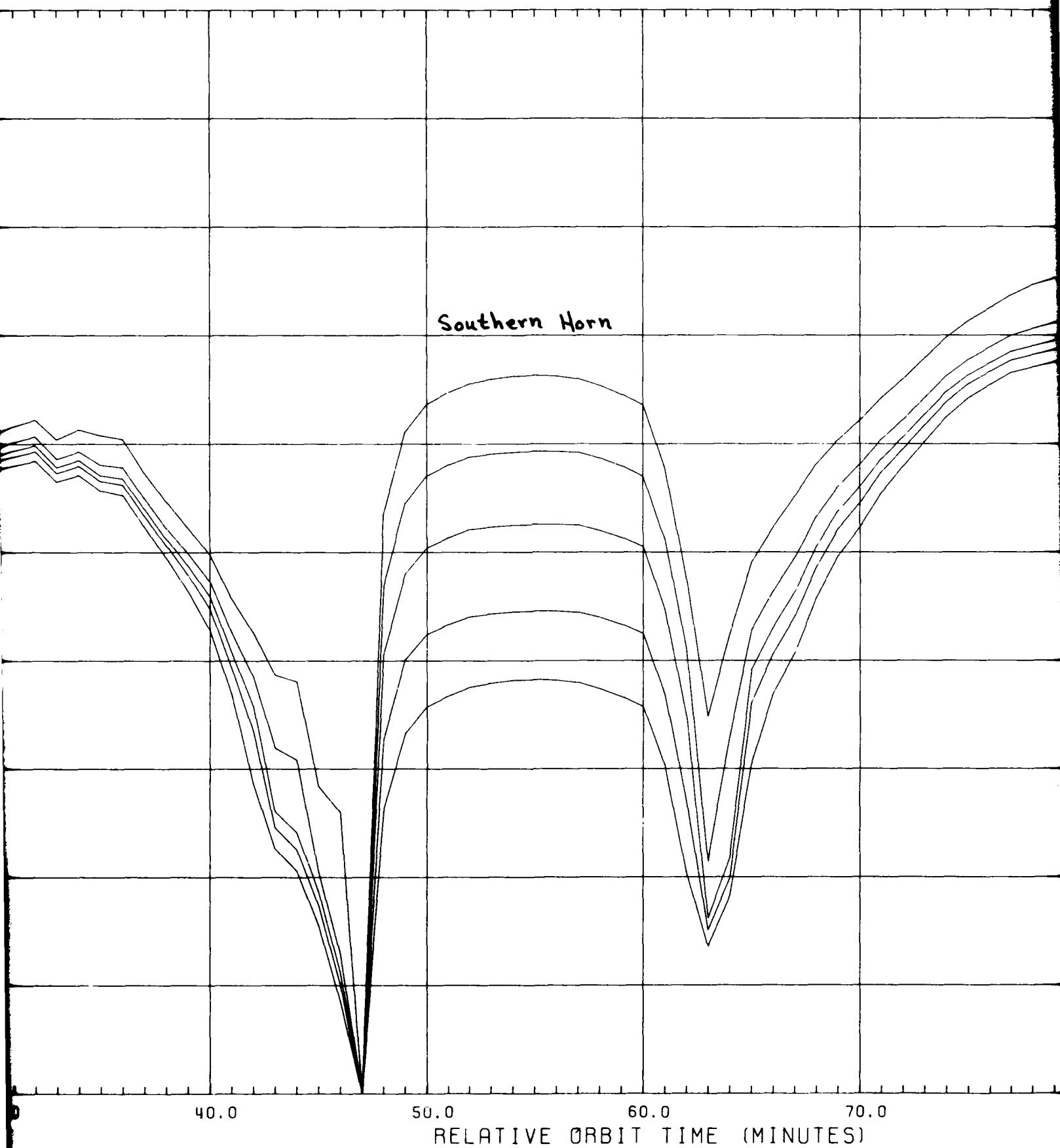
110.0

120.0





DOSE AT TRANSMISSION SURFACE OF FINITE ALUMINUM SLAB



ALUMINUM SLAB SHIELDS

Figure

ORBIT: NAVELEX 1
60 DGR/1667-1667 KM

EPOCH: 1989.5

MODELS:
FIELD: BARR/75
TRAPPED PROTONS: AP8
INNER ZN ELEC: AE6
OUTER ZN ELEC: AE17 L0
MISSION DURATION: 60.00 MO
EVALUATION PHASE: SOLAR MAX
UN FACTORS: NOT APPLIED

Equatorial Region

Northern

SAA

STOP TIME ON TAPE = 8.449993

NASA-GSF

70.0

80.0

90.0

100.0

110.0

TES)

Figure 117

ORBIT: NAVELEX 1
60 DGR/1667-1667 KM

EPOCH: 1989.5

MODELS:

FIELD: BARR/75

TRAPPED PROTONS: AP8

INNER ZN ELEC: AE6

OUTER ZN ELEC: AE17-L0

MISSION DURATION: 60.00 MO

EVALUATION PHASE: SOLAR MAX

UN FACTORS: NOT APPLIED

Northern Horn

STOP TIME ON TAPE = 8.449993

NASA-GSFC

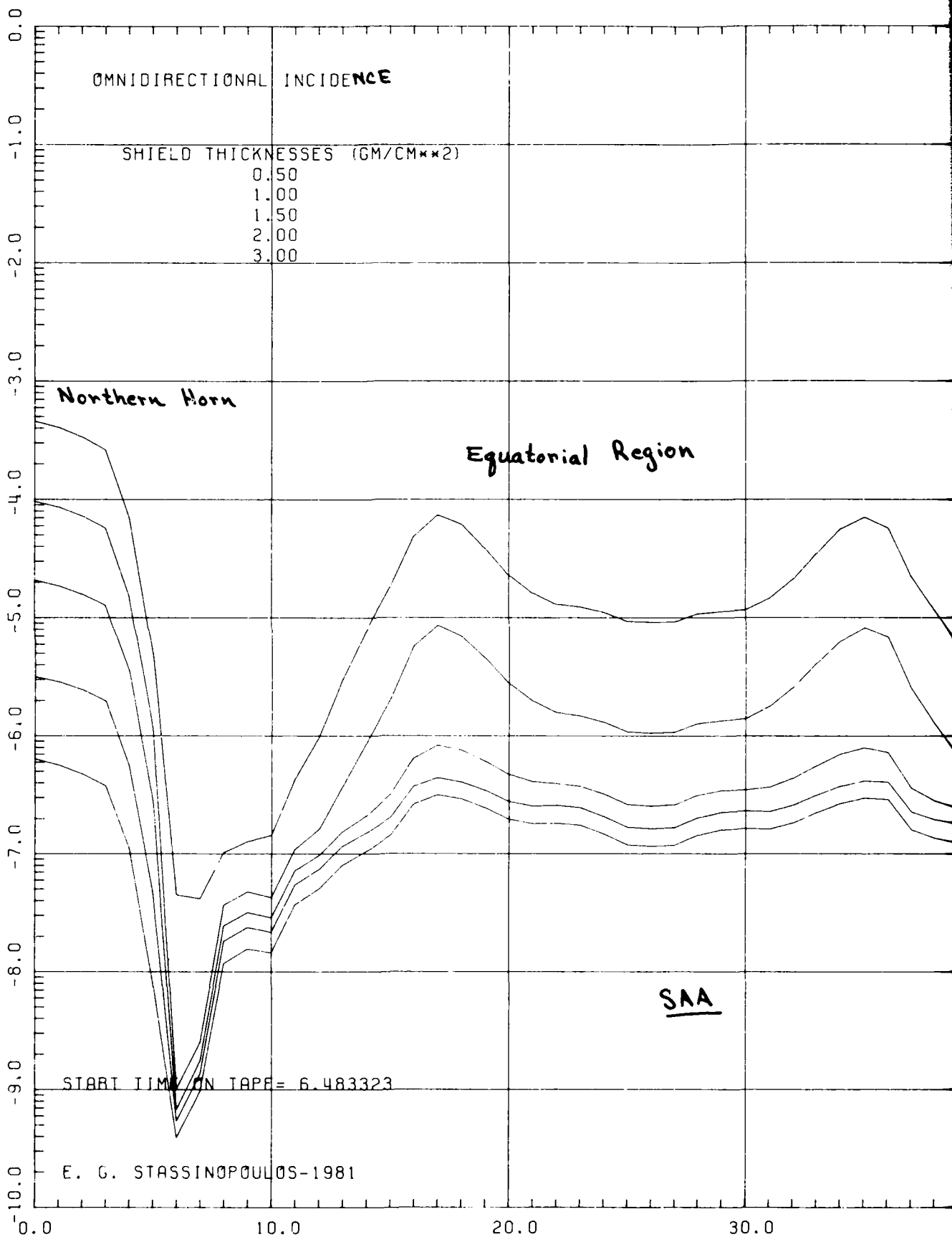
100.0

110.0

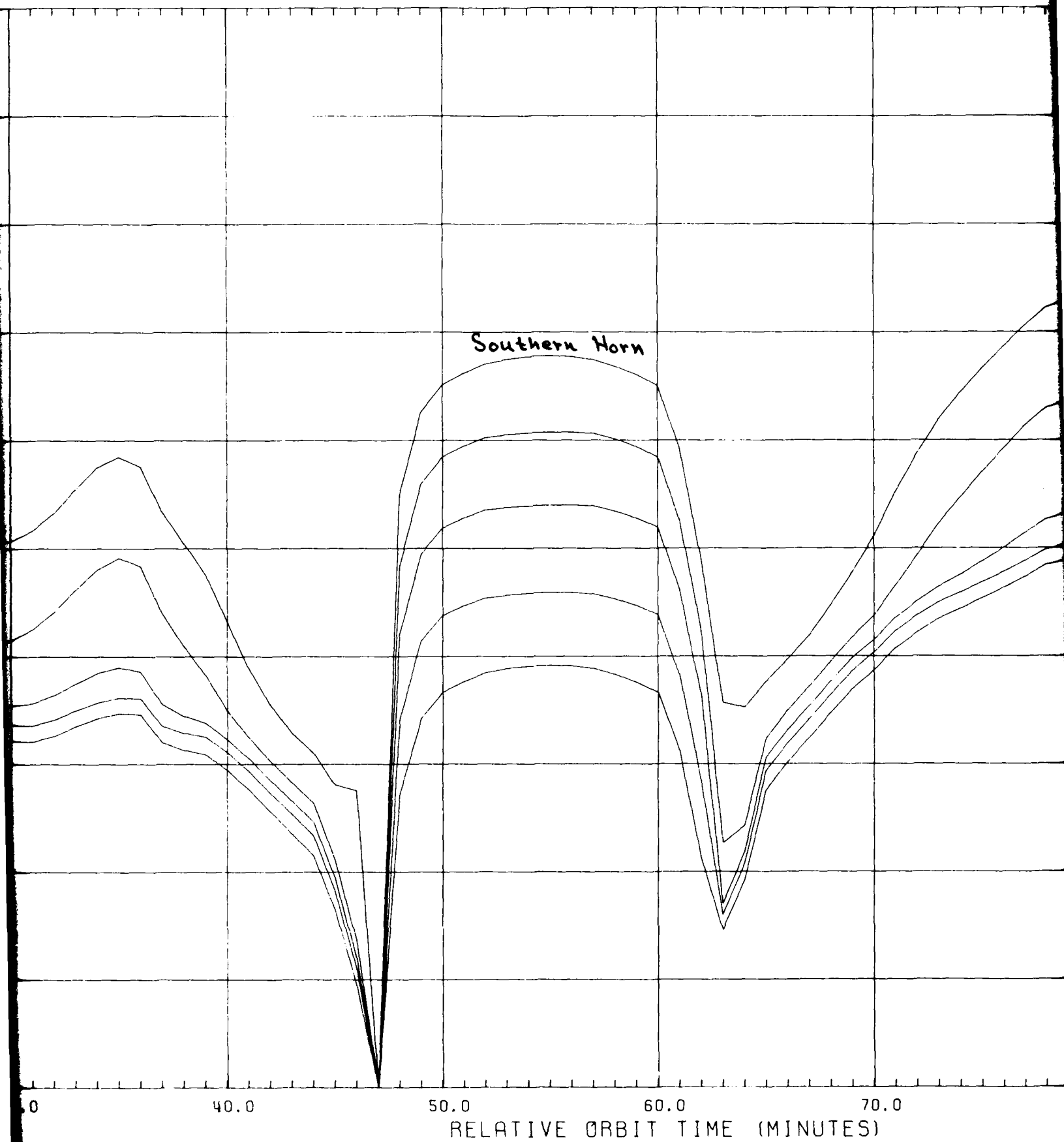
120.0

INSTANTANEOUS ALUMINUM ELECTRON DOSE (RADS)

(PLOTTED DOSE VALUES INCLUDE BREMSSTRAHLUNG CONTRIBUTIONS)



DOSE IN SEMI-INFINITE ALUMINUM MEDIUM



IUM

Figure 118

ORBIT: NAVELEX 1
60 DGR/1667-1667 KM

EPOCH: 1989.5

MODELS:
FIELD: BARR/75
TRAPPED PROTONS: AP8
INNER ZN ELEC: AE6
OUTER ZN ELEC: AE17-L0
MISSION DURATION: 60.00 MO
EVALUATION PHASE: SOLAR MAX

UN FACTORS: NOT APPLIED

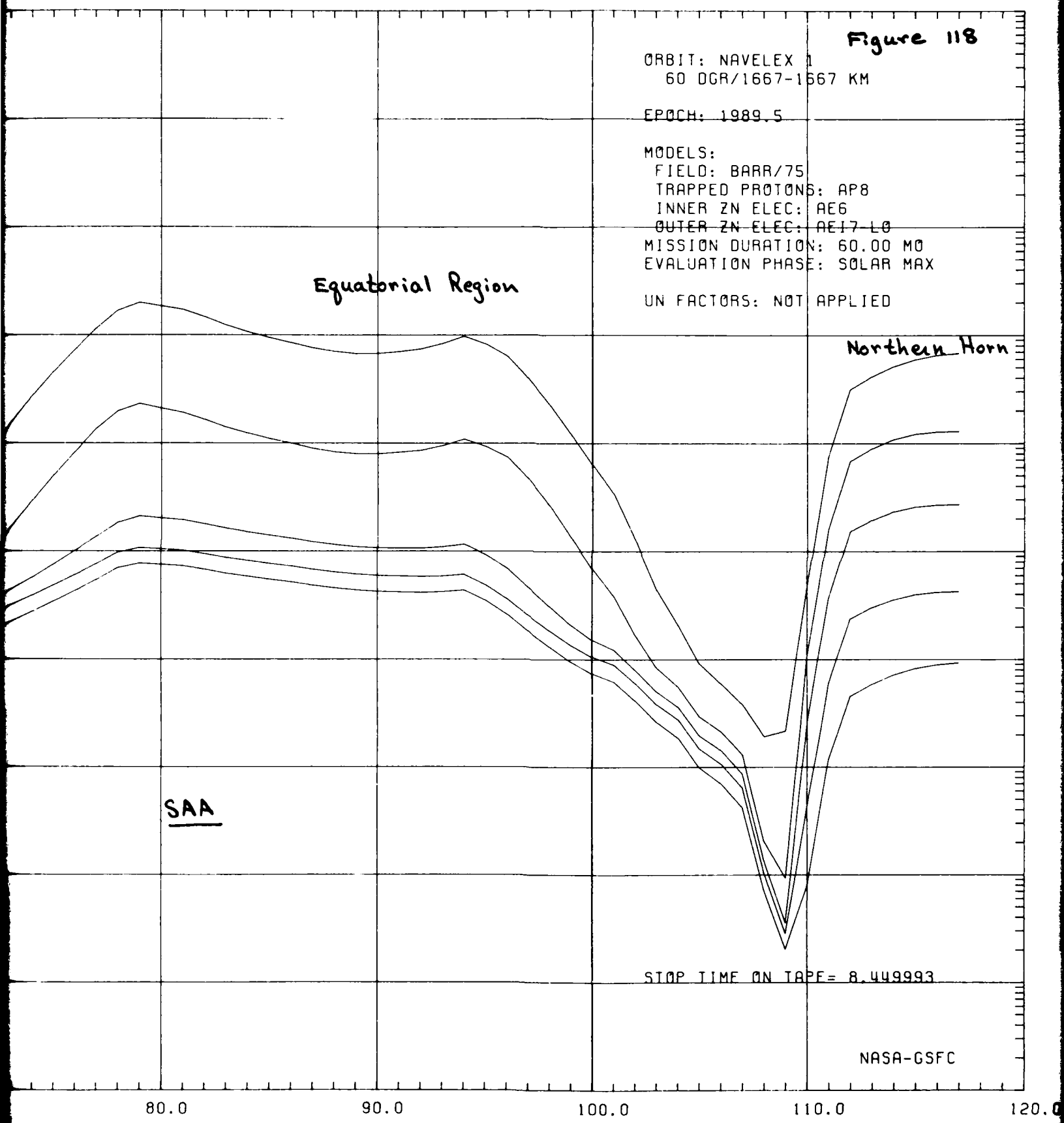
Equatorial Region

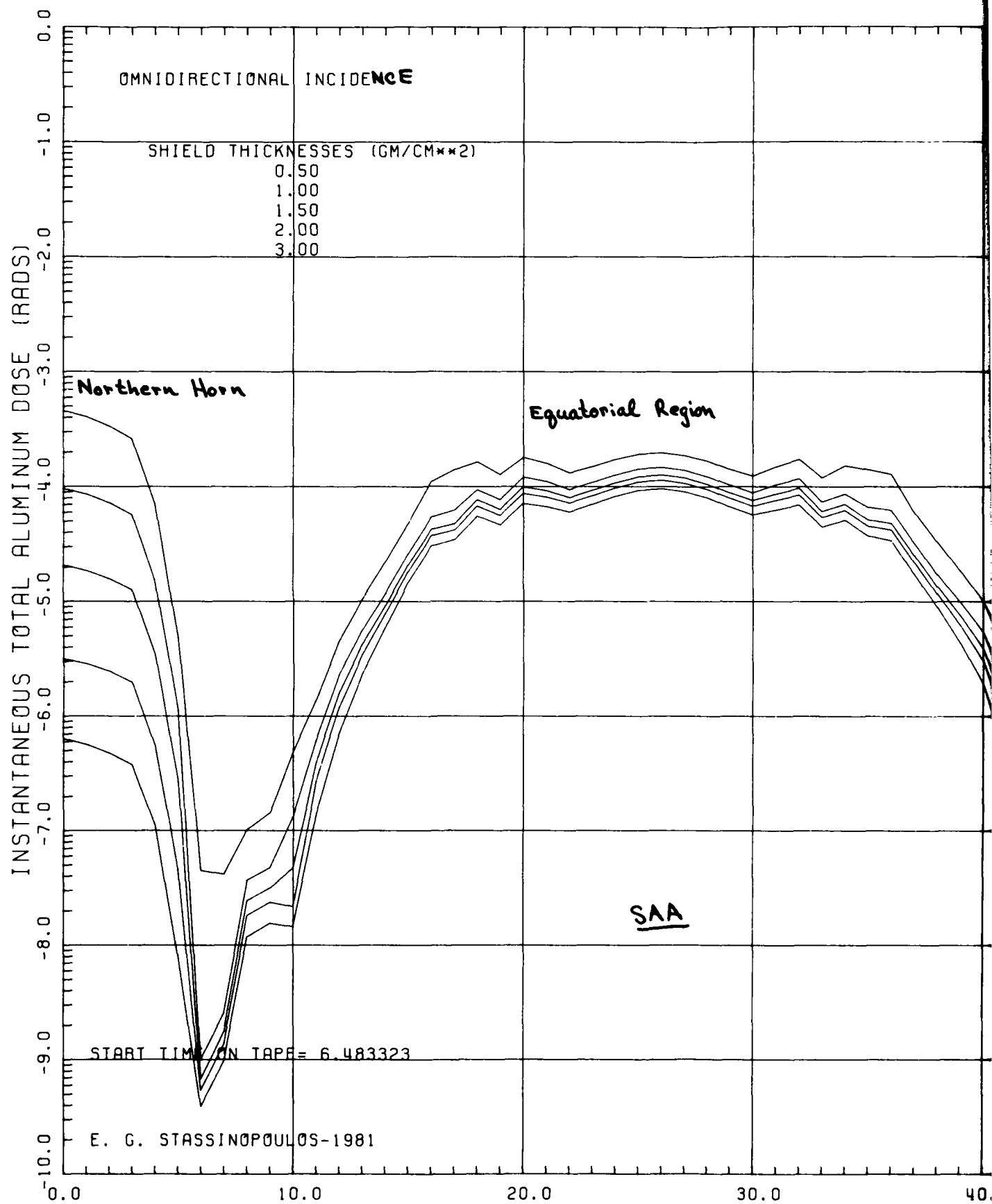
Northern Horn

SAA

STOP TIME ON TAPE= 8.449993

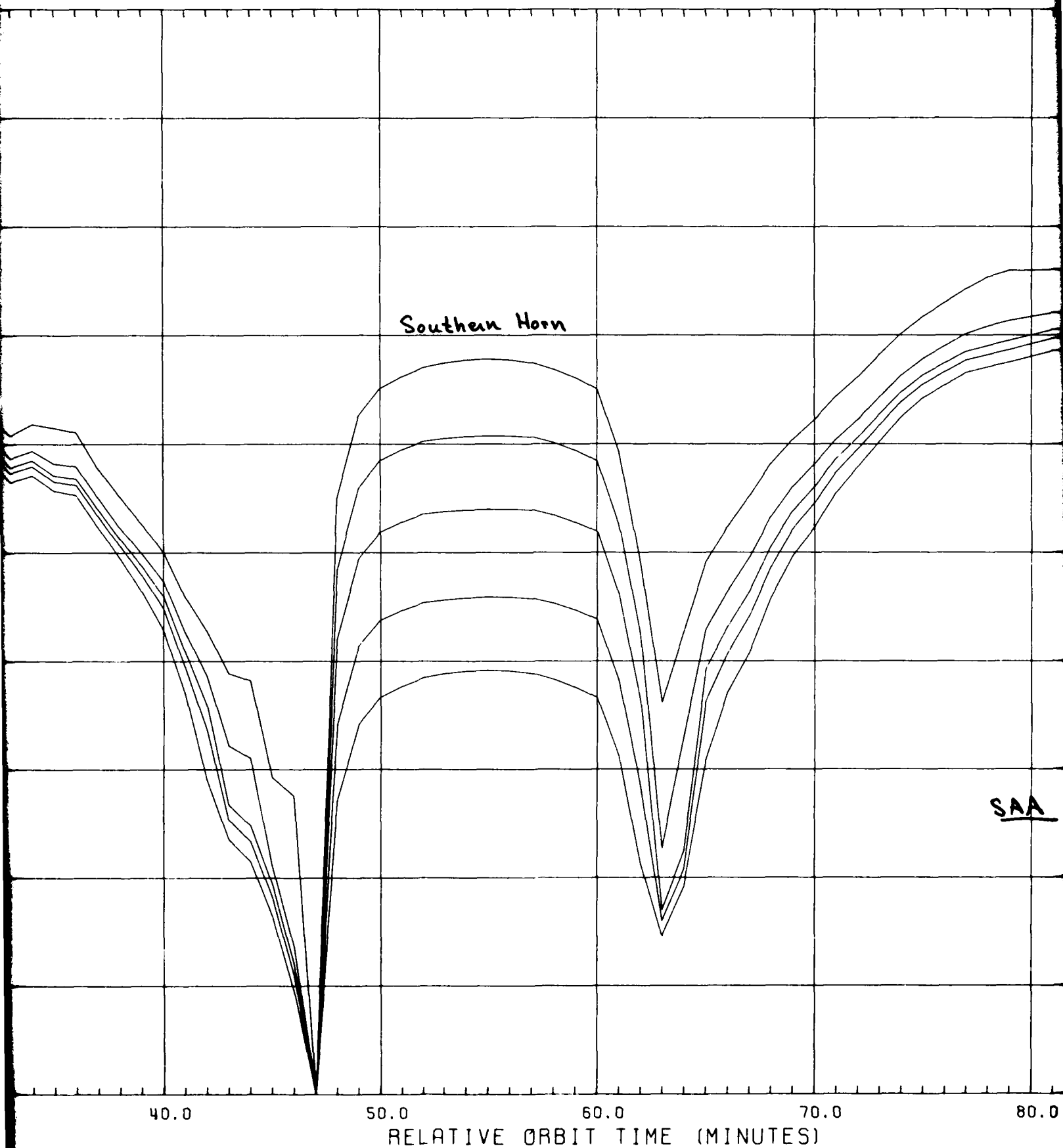
NASA-GSFC





2

DOSE IN SEMI-INFINITE ALUMINUM MEDIUM



UM

Figure 119

ORBIT: NAVELEX 1
60 DGR/1667-1567 KM

EPOCH: 1989.5

MODELS:
FIELD: BARR/75
TRAPPED PROTONS: AP8
INNER ZN ELEC: AE6
OUTER ZN ELEC: AE17 L0
MISSION DURATION: 60.00 MO
EVALUATION PHASE: SOLAR MAX

UN FACTORS: NOT APPLIED

Equatorial Region

Northern Horn

SAA

STOP TIME ON TAPF= 8.449993

NASA-GSFC

80.0

90.0

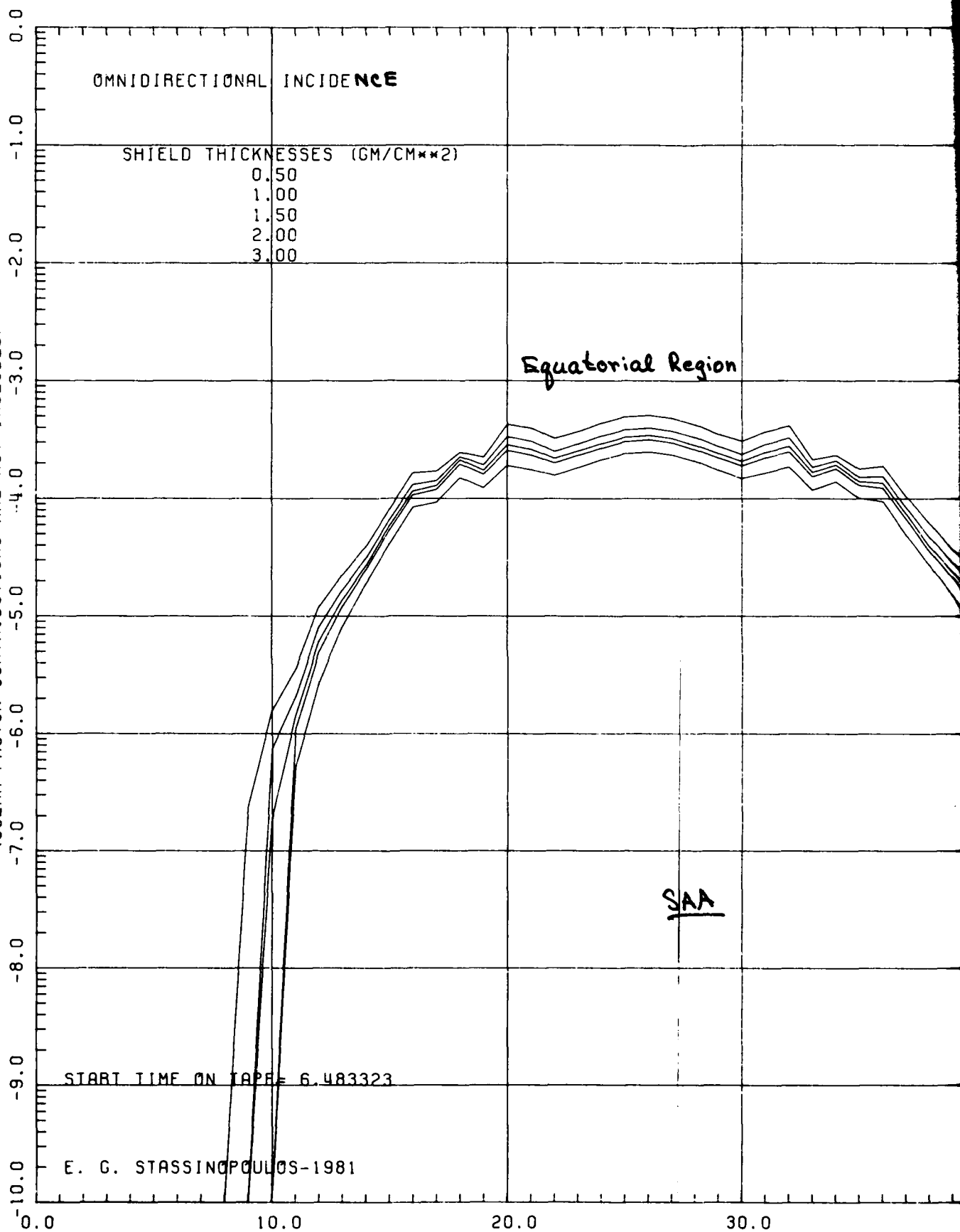
100.0

110.0

120.0

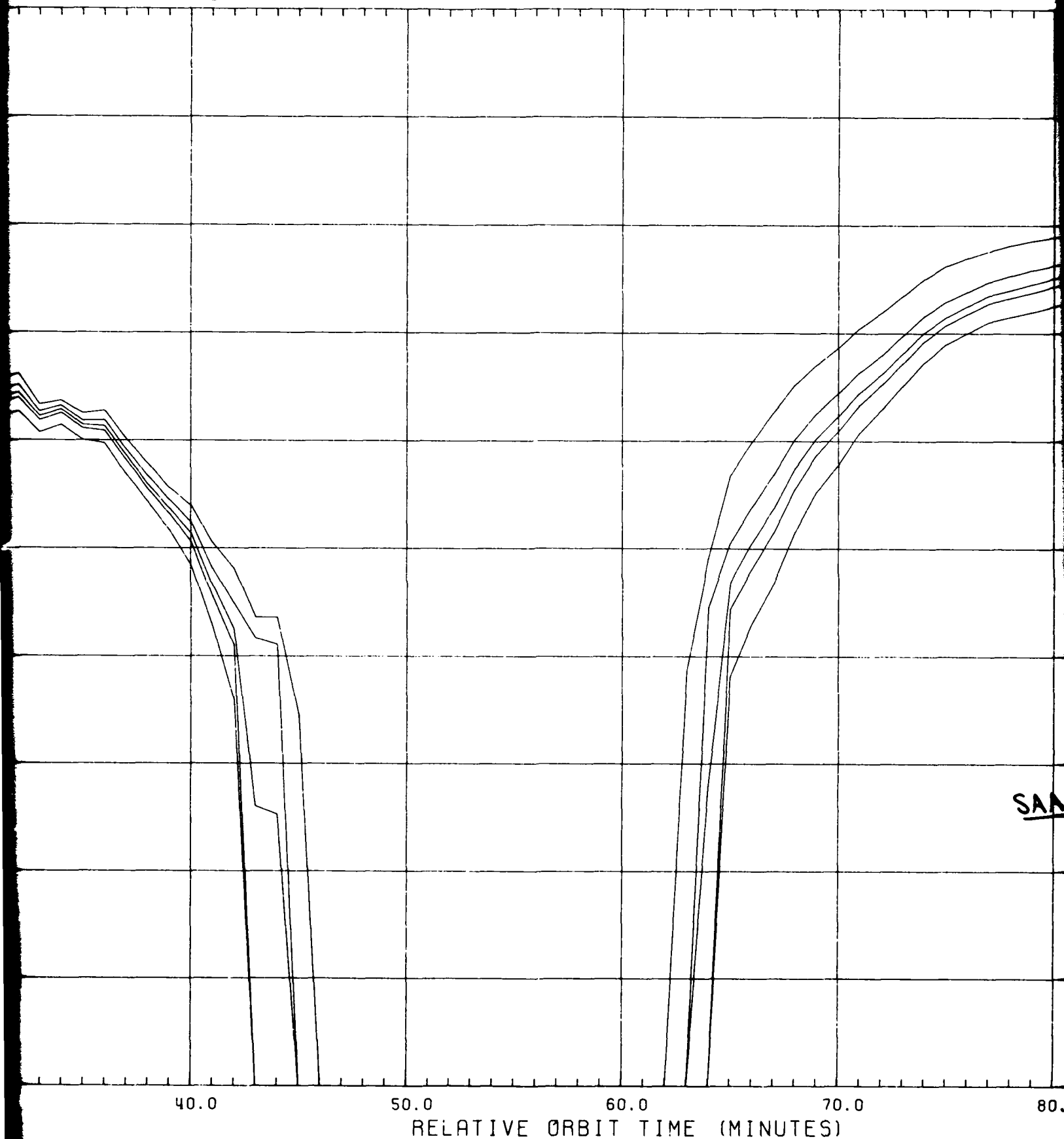
INSTANTANEOUS ALUMINUM PROTON DOSE (RADS)

(SOLAR PROTON CONTRIBUTIONS ARE NOT INCLUDED)



'2

DOSE AT CENTER OF ALUMINUM SPHERES



3

Figure 120

ORBIT: NAVELEX 1
60 DGR/1667-1567 KM

EPOCH: 1989.5

MODELS:
FIELD: BARR/75
TRAPPED PROTONS: AP8
INNER ZN ELEC: AE6
OUTER ZN ELEC: AE17 L0
MISSION DURATION: 60.00 MO
EVALUATION PHASE: SOLAR MAX

UN FACTORS: NOT APPLIED

Equatorial Region

SAA

STOP TIME ON TAPE = 8.449993

NASA-GSFC

80.0

90.0

100.0

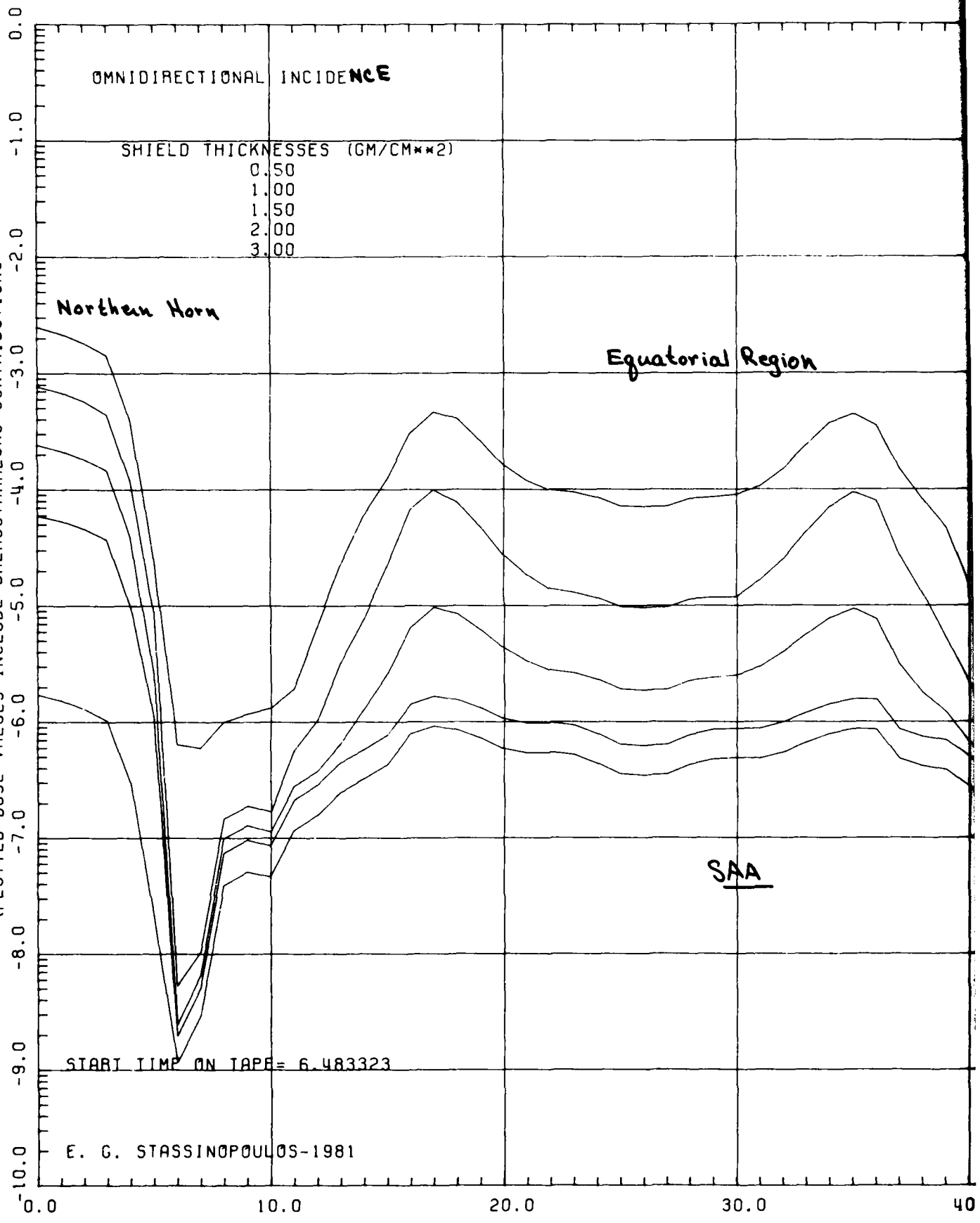
110.0

120.0

S

INSTANTANEOUS ALUMINUM ELECTRON DOSE (RADS)

(PLOTTED DOSE VALUES INCLUDE BREMSSTRAHLUNG CONTRIBUTIONS)



DOSE AT CENTER OF ALUMINUM SPHERES

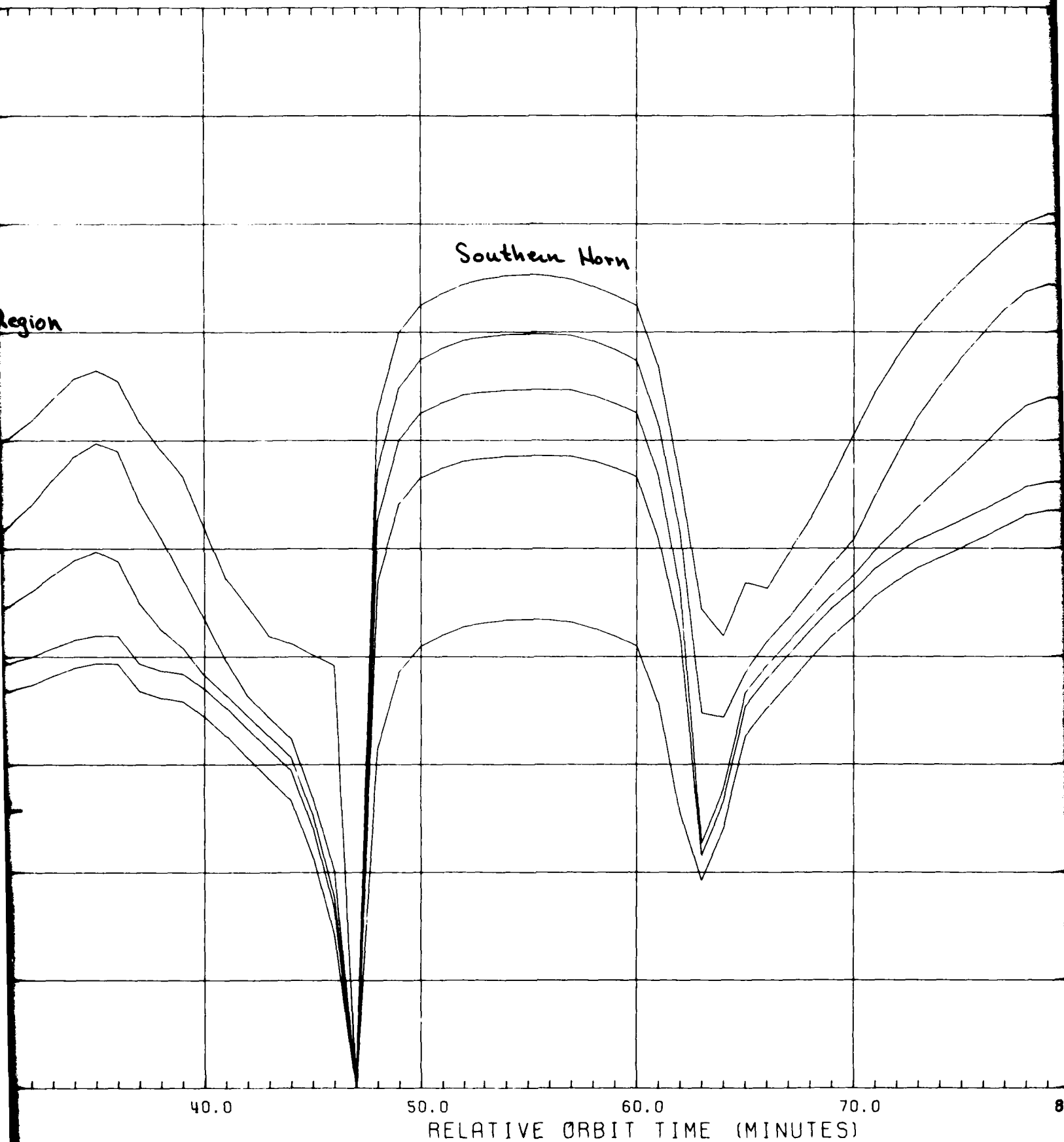


Figure 121

ORBIT: NAVELEX 1
60 DGR/1667-1567 KM

EPOCH: 1989.5

MODELS:
FIELD: BARR/75
TRAPPED PROTONS: AP8
INNER ZN ELEC: AE6
OUTER ZN ELEC: AE17-L0
MISSION DURATION: 60.00 MO
EVALUATION PHASE: SOLAR MAX

UN FACTORS: NOT APPLIED

Equatorial Region

Northern Horn

SAA

STOP TIME ON TAPE = 8.449993

NASA-GSFC

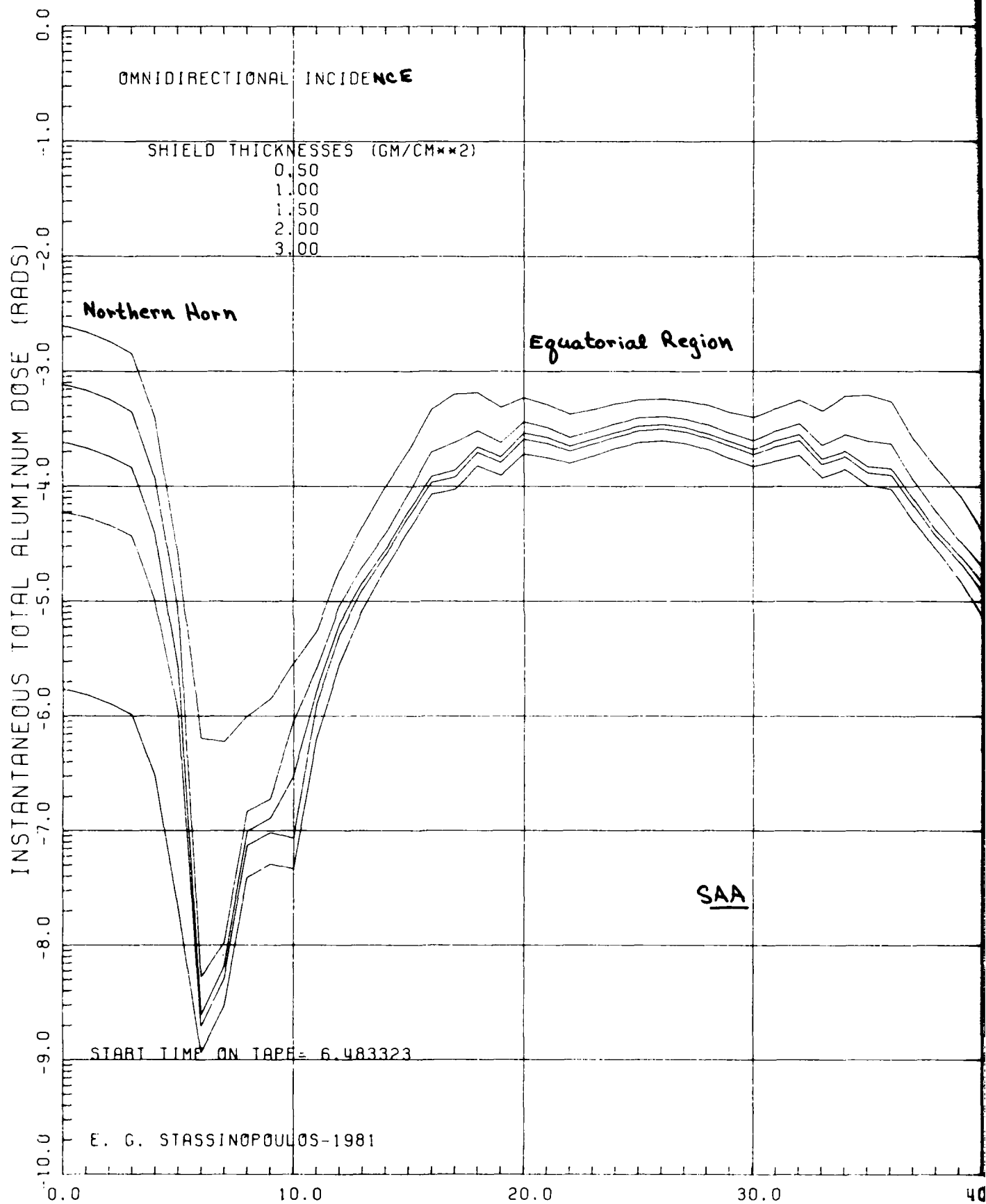
80.0

90.0

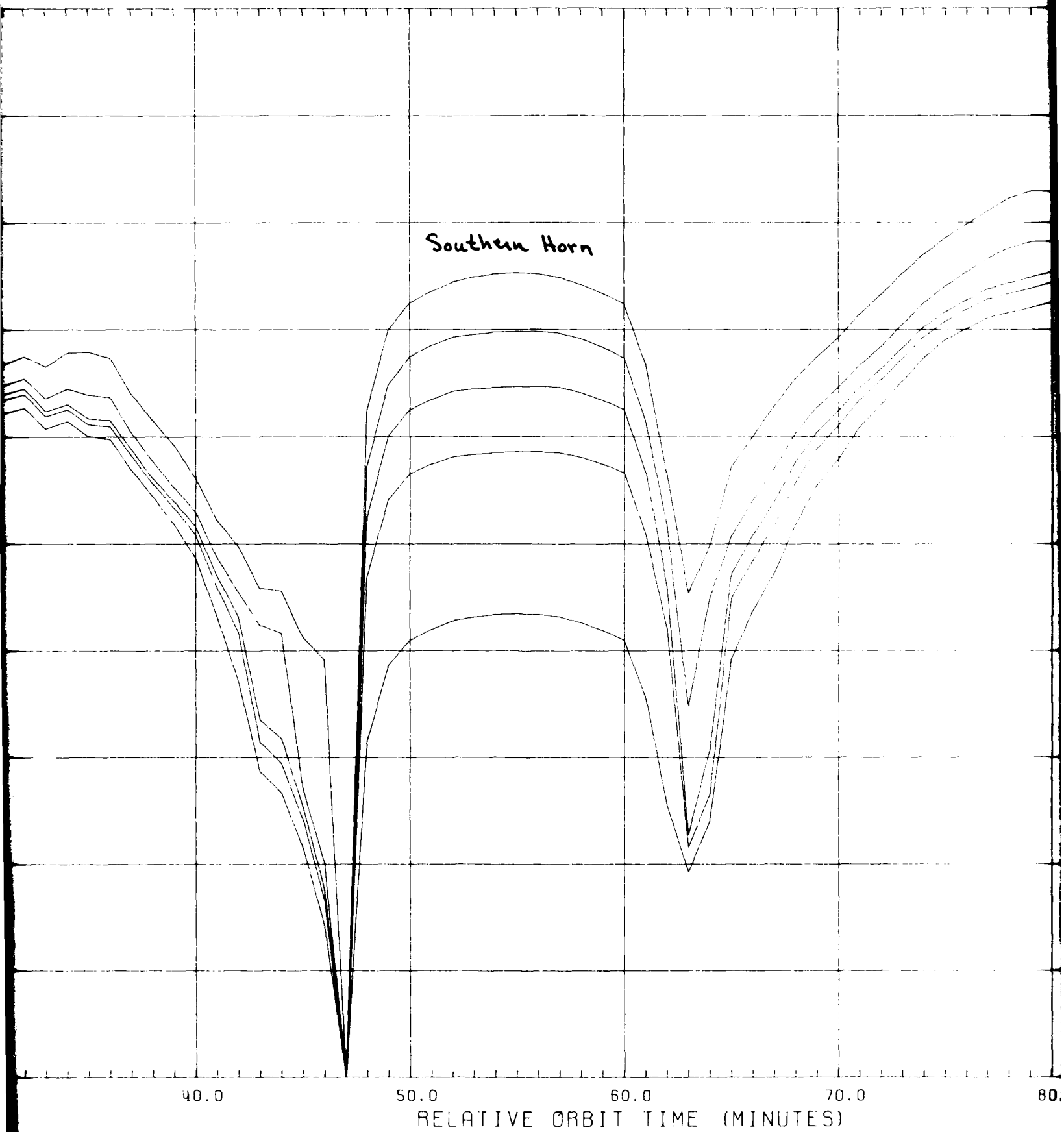
100.0

110.0

120.0



DOSE AT CENTER OF ALUMINUM SPHERES



S

Figure 122

ORBIT: NAVELEX 1
60 DGR/1667-1667 KM

EPOCH: 1989.5

MODELS:
FIELD: BARR/75
TRAPPED PROTONS: AP8
INNER ZN ELEC: AE6
OUTER ZN ELEC: AE17 L0
MISSION DURATION: 60.00 MO
EVALUATION PHASE: SOLAR MAX

UN FACTORS: NOT APPLIED

Equatorial Region

Northern Horn

SAA

STOP TIME ON TAPE= 8.449993

NASA-GSFC

80.0

90.0

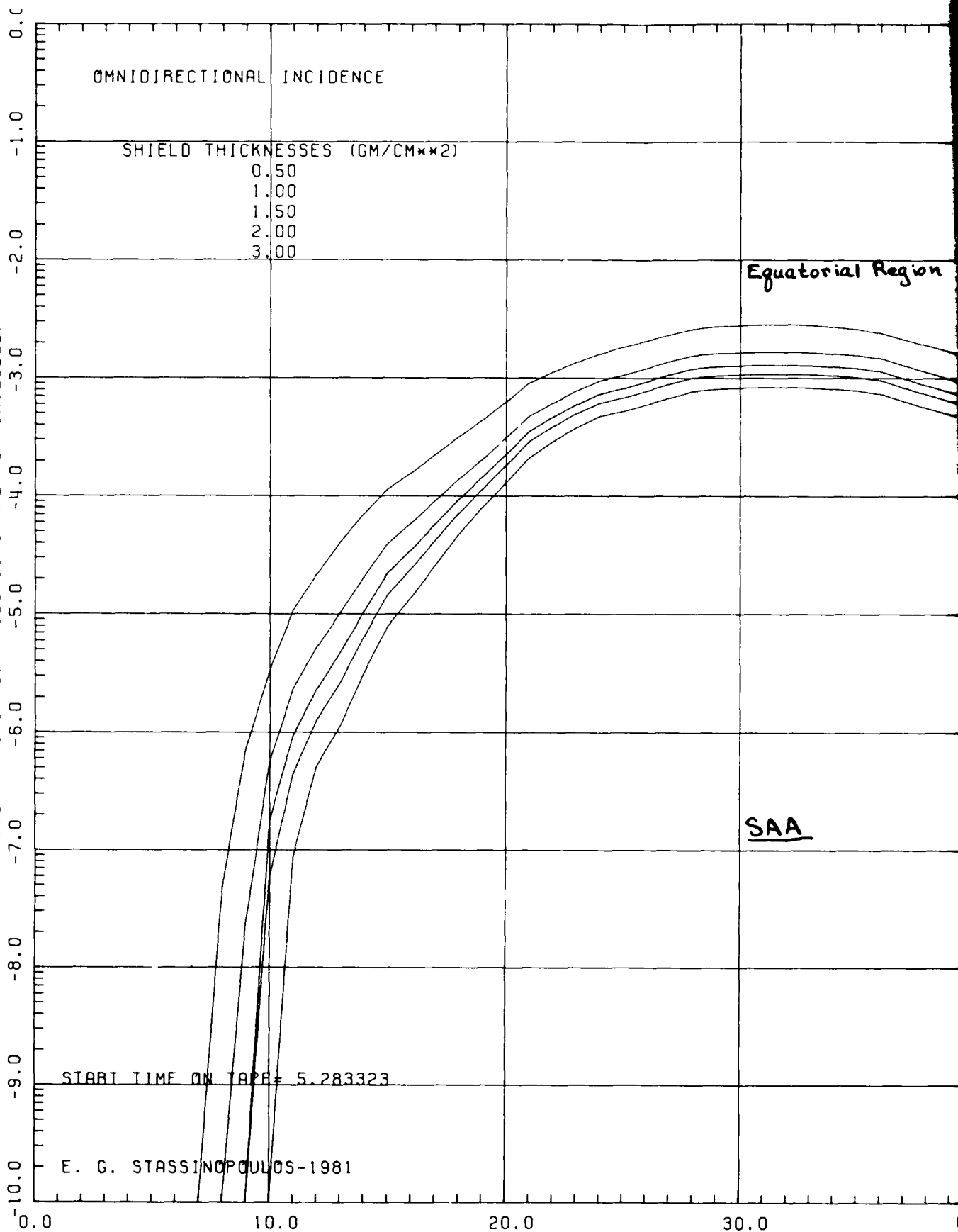
100.0

110.0

120.0

INSTANTANEOUS ALUMINUM PROTON DOSE (RADS)

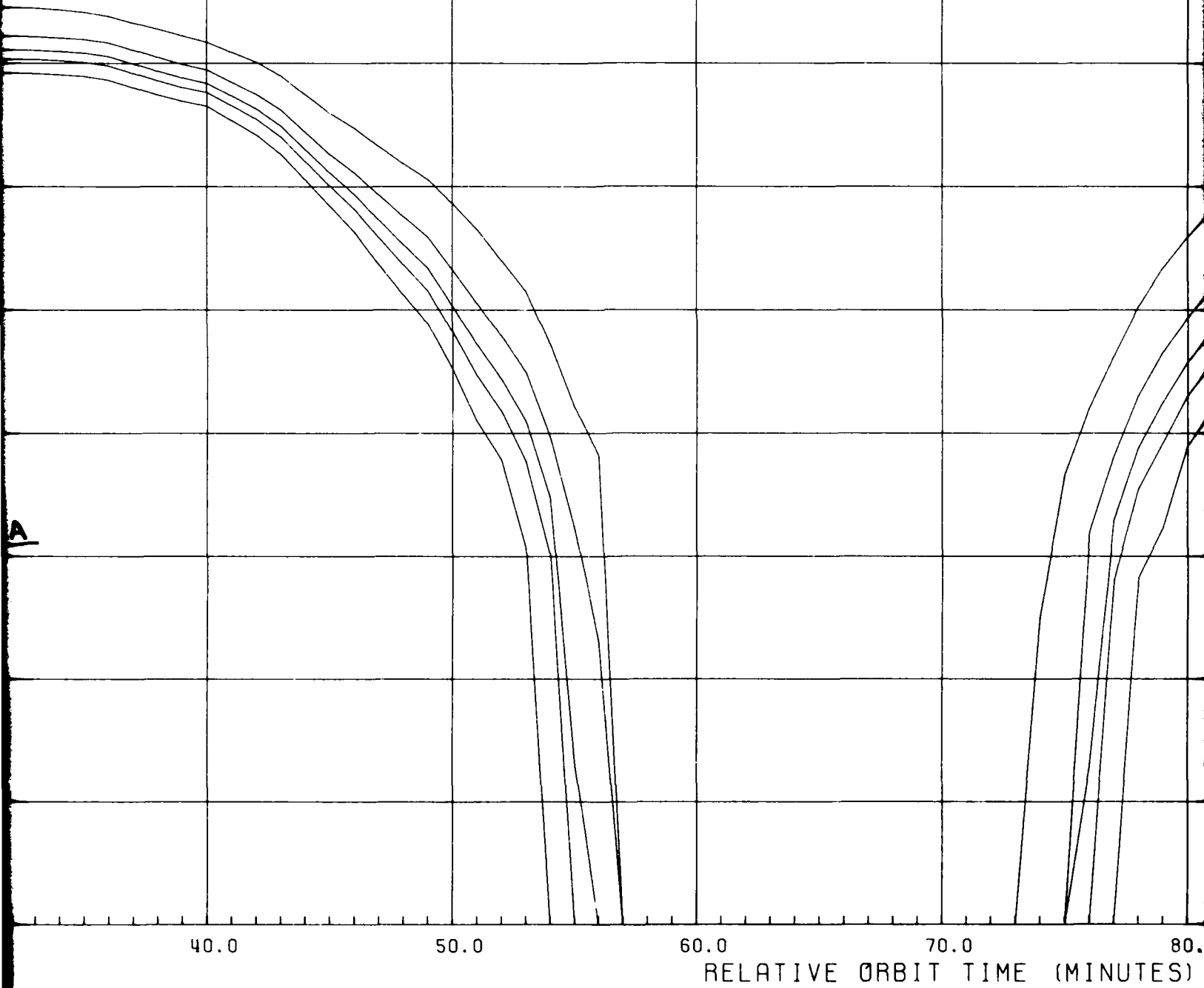
(SOLAR PROTON CONTRIBUTIONS ARE NOT INCLUDED)



DOSE AT TRANSMISSION SURFACE OF FINITE AL

(= DOSE IN SEMI-INFINITE ALUMINUM

atorial Region



3

OF FINITE ALUMINUM SLAB SHIELDS

FINITE ALUMINUM MEDIUM)

Equatorial Region

SAA

80.0

90.0

100.0

110.0

120.0

(MINUTES)

Figure 123

ORBIT: NAVELEX 2
60 DGR/2593-2593 KM

EPOCH: 1989.5

MODELS:
FIELD: BARR/75
TRAPPED PROTONS: AP8
INNER ZN ELEC: AE6
OUTER ZN ELEC: AE17-L0
MISSION DURATION: 60.00 MO
EVALUATION PHASE: SOLAR MAX

UN FACTORS: NOT APPLIED

orial Region

STOP TIME ON TAPE= 7.616656

NASA-GSFC

110.0

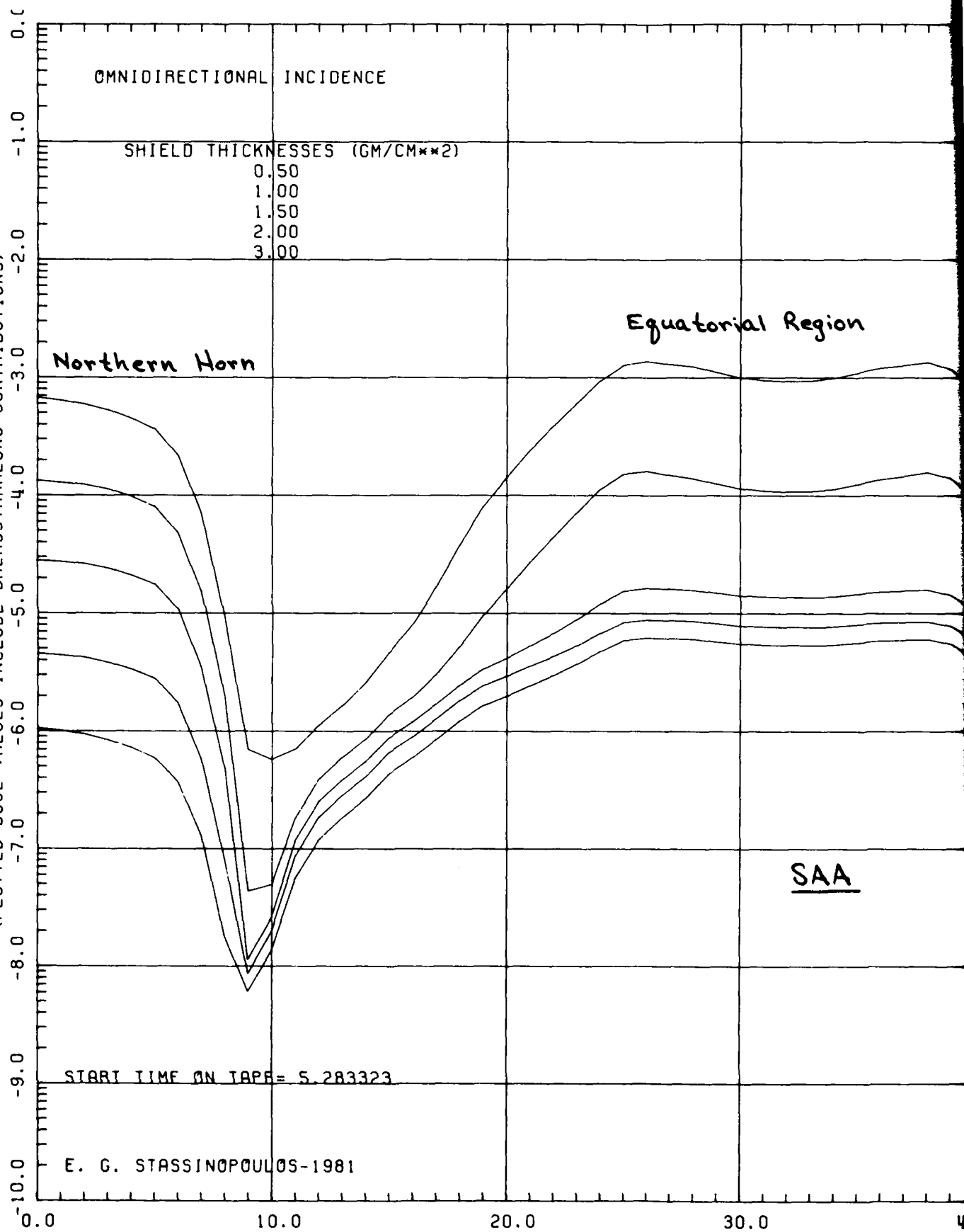
120.0

130.0

140.0

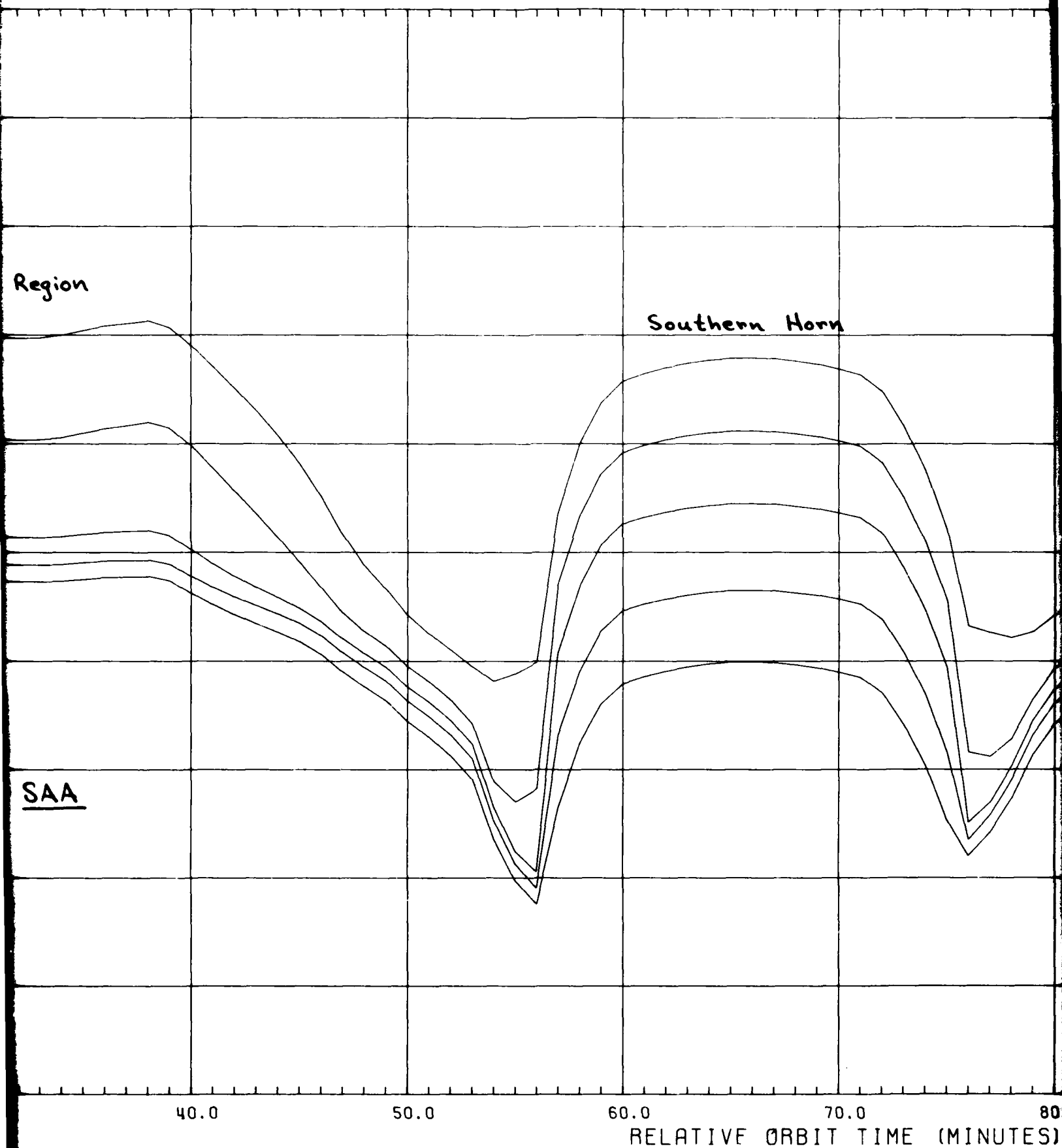
INSTANTANEOUS ALUMINUM ELECTRON DOSE (RADS)

(PLOTTED DOSE VALUES INCLUDE BREMSSTRAHLUNG CONTRIBUTIONS)



2

DOSE AT TRANSMISSION SURFACE OF FINITE R



3'

OF FINITE ALUMINUM SLAB SHIELDS

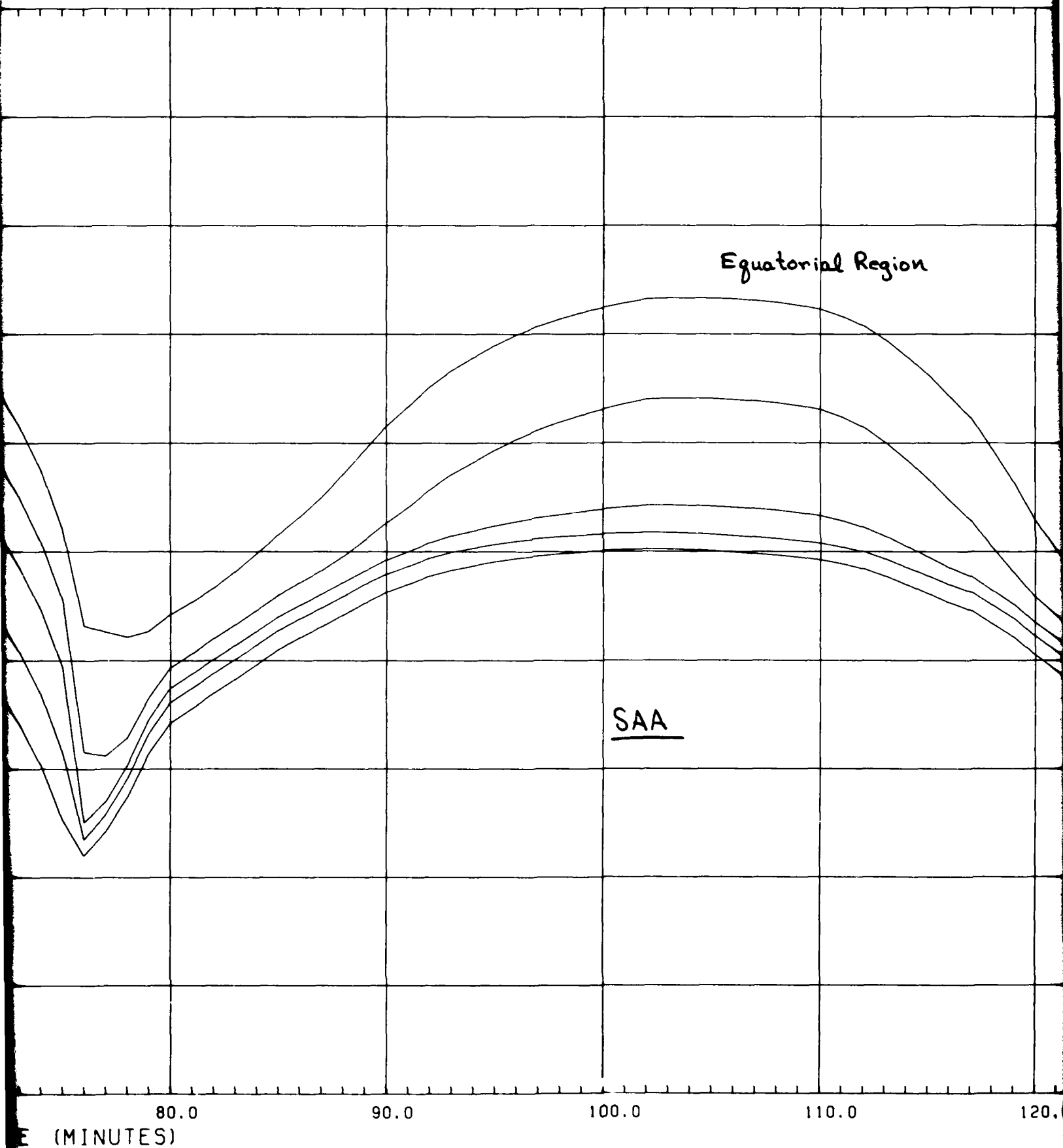


Figure 124

ORBIT: NAVELEX 2
60 DGR/2593-2593 KM

EPOCH: 1989.5

MODELS:

FIELD: BARR/75

TRAPPED PROTONS: AP8

INNER ZN ELEC: AE6

OUTER ZN ELEC: AE17 L0

MISSION DURATION: 60.00 MO

EVALUATION PHASE: SOLAR MAX

UN FACTORS: NOT APPLIED

Northern Horn

region

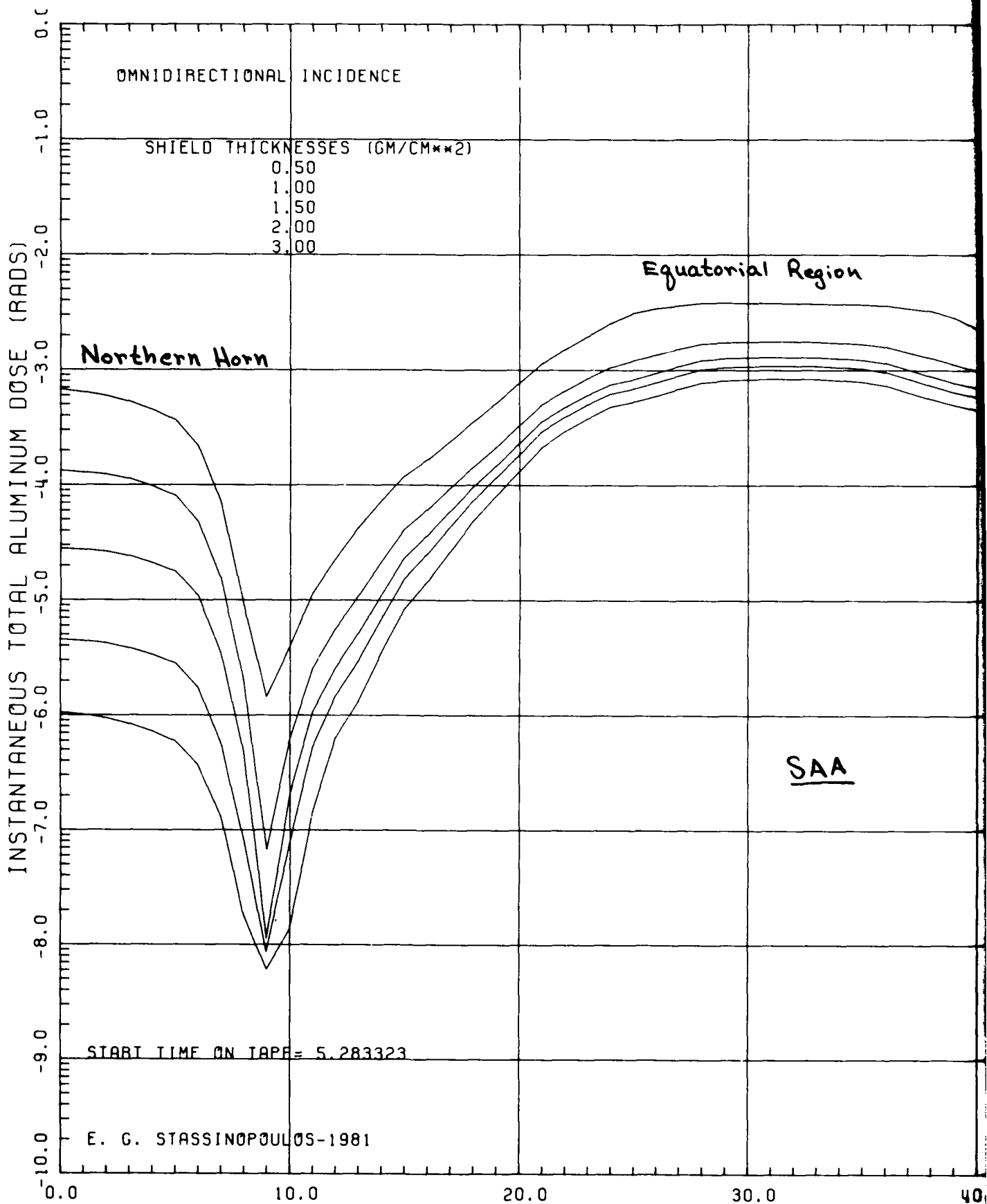
STOP TIME ON TAPE = 7.616656

NASA-GSFC

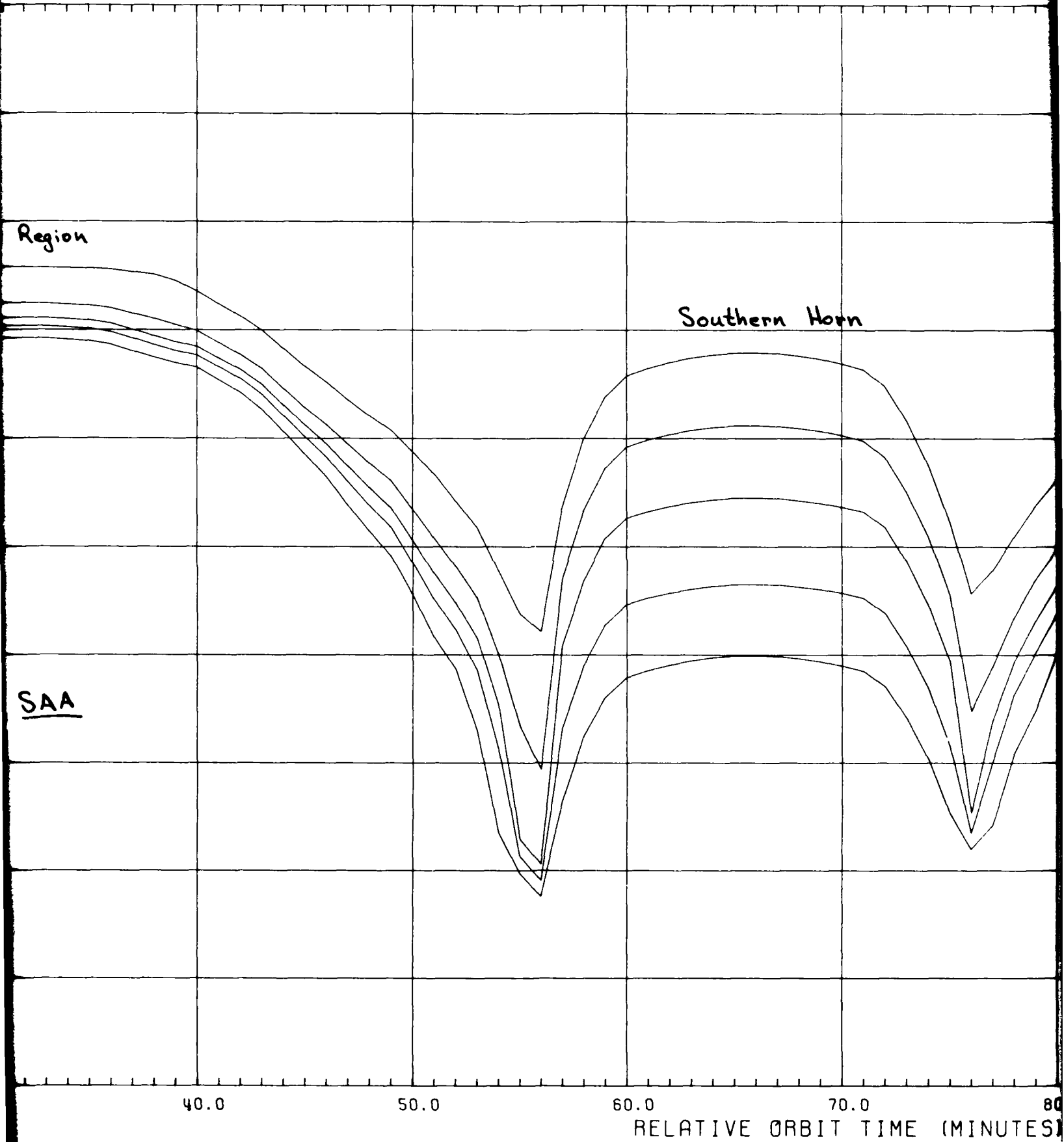
120.0

130.0

140.0



DOSE AT TRANSMISSION SURFACE OF FINITE P



E OF FINITE ALUMINUM SLAB SHIELDS

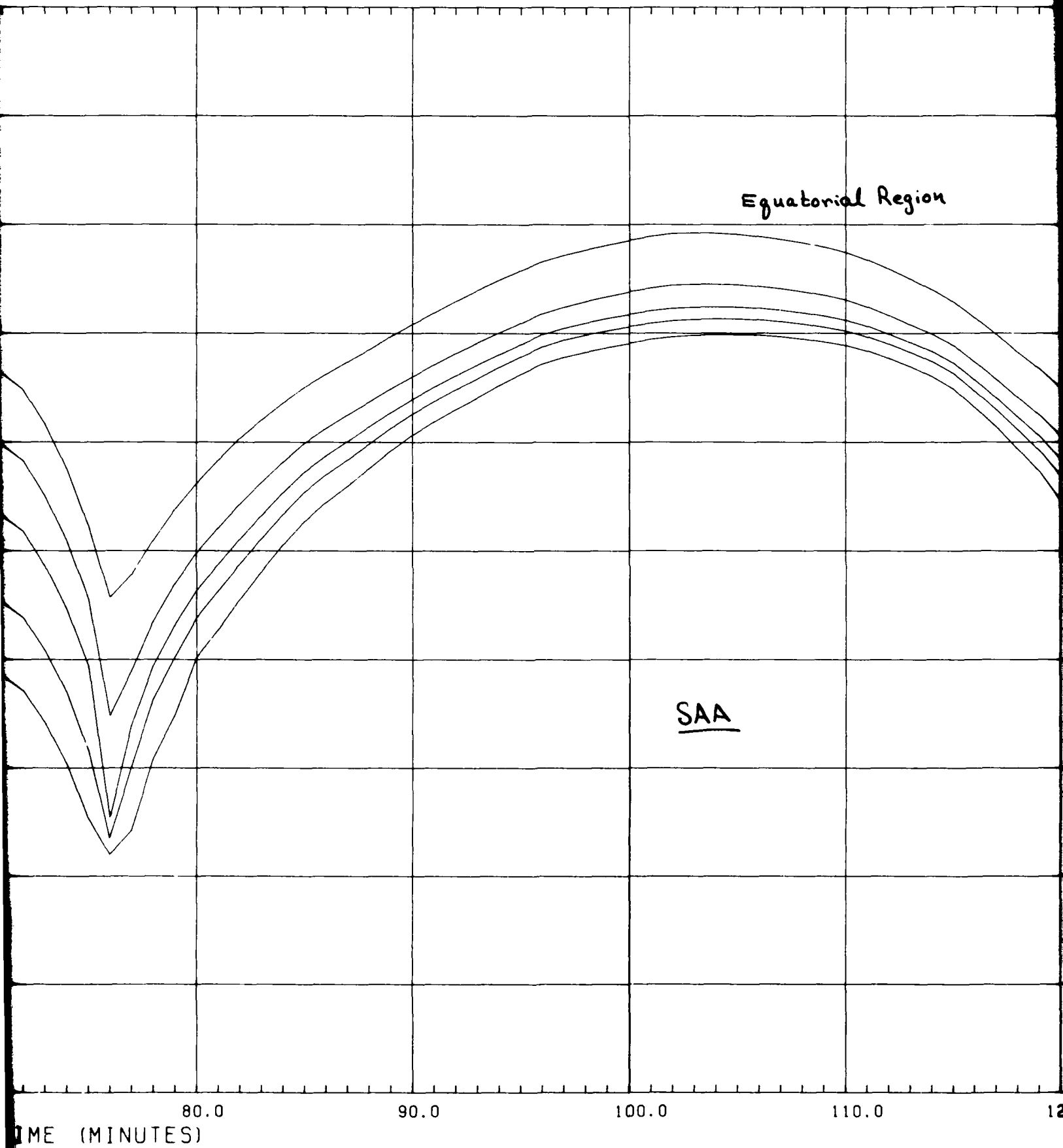


Figure 125

ORBIT: NAVELEX 2
60 DGR/2593-2593 KM

EPOCH: 1989.5

MODELS:

FIELD: BARR/75

TRAPPED PROTONS: AP8

INNER ZN ELEC: AE6

OUTER ZN ELEC: AE17 L0

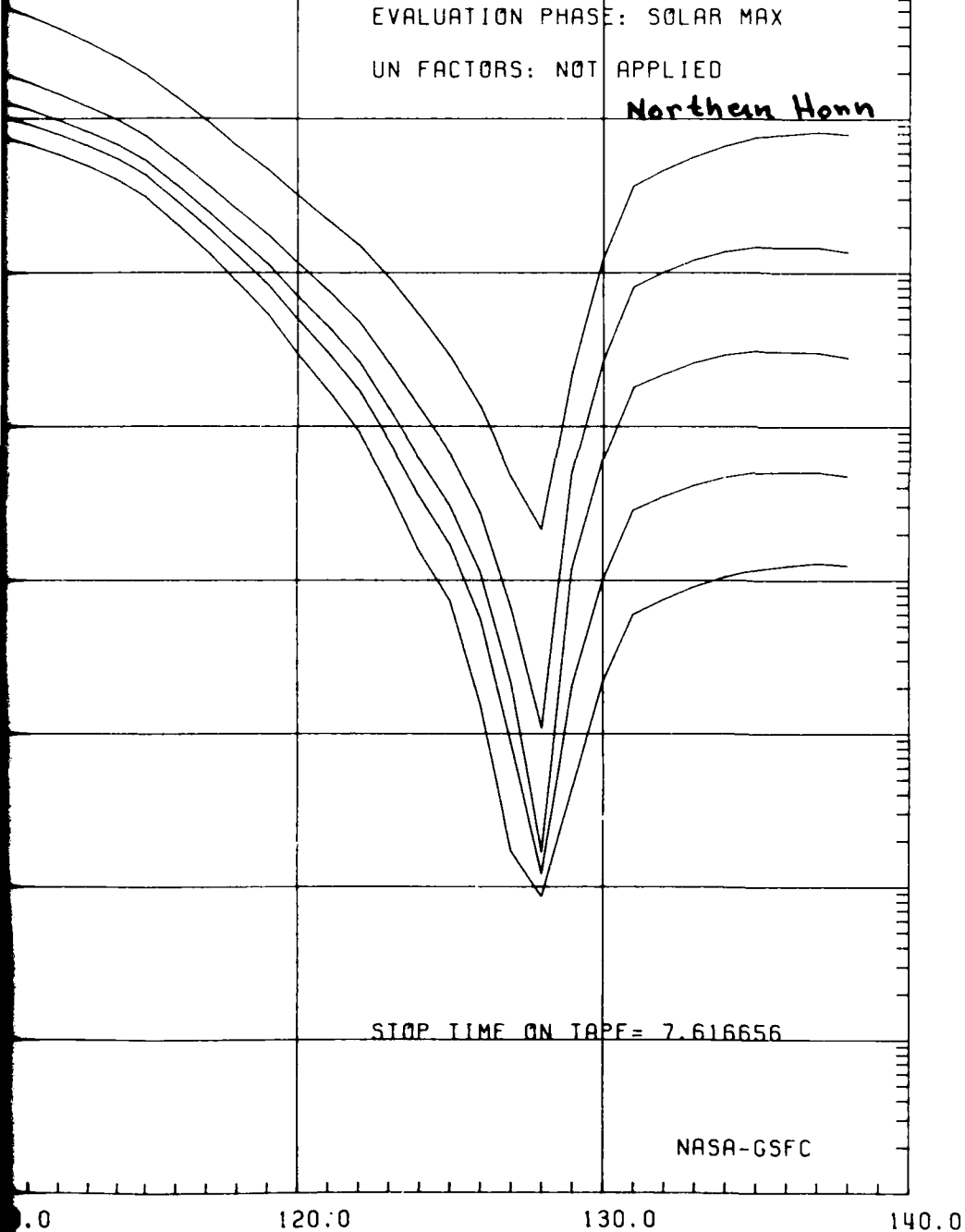
MISSION DURATION: 60.00 MO

EVALUATION PHASE: SOLAR MAX

UN FACTORS: NOT APPLIED

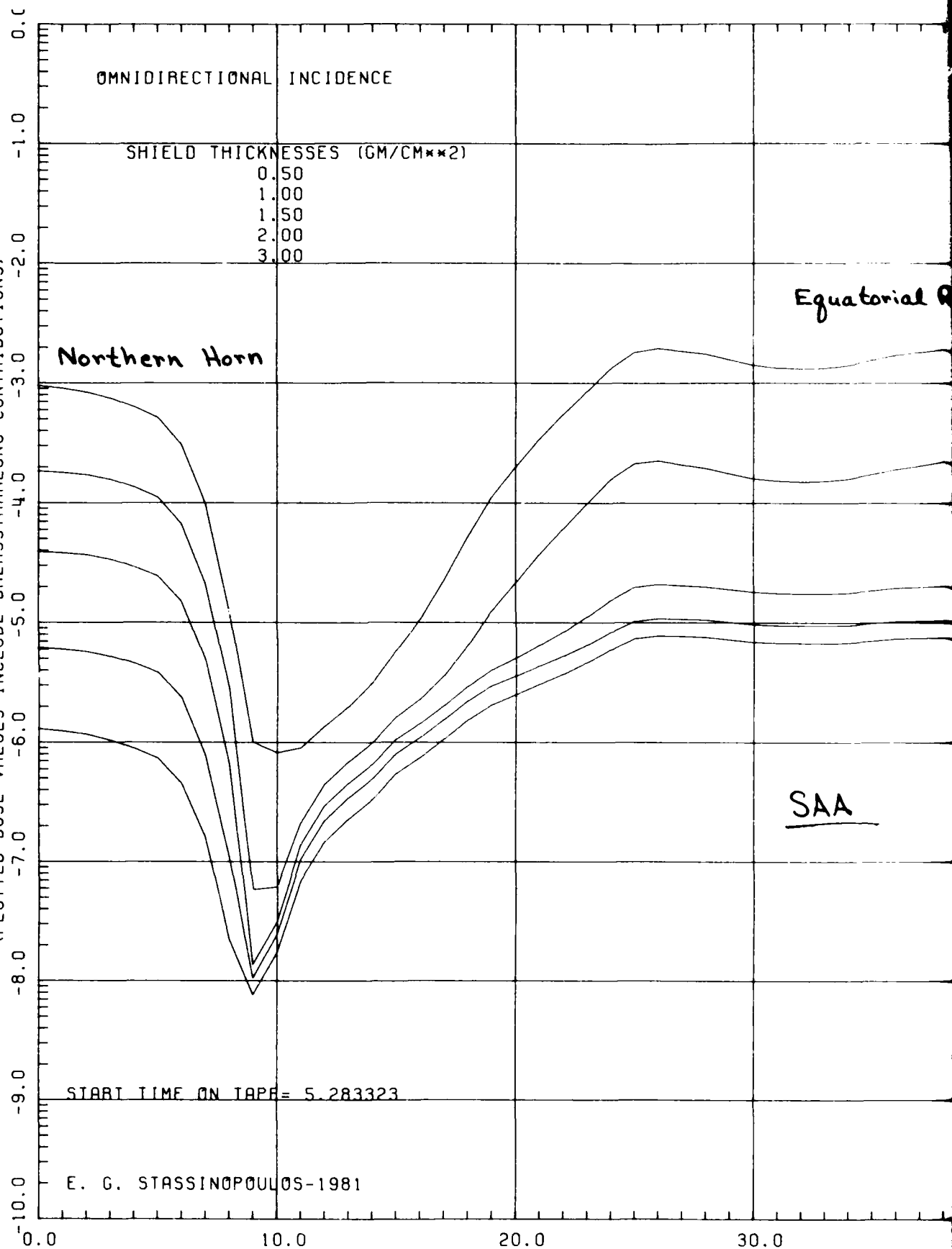
Northern Hemisphere

Region

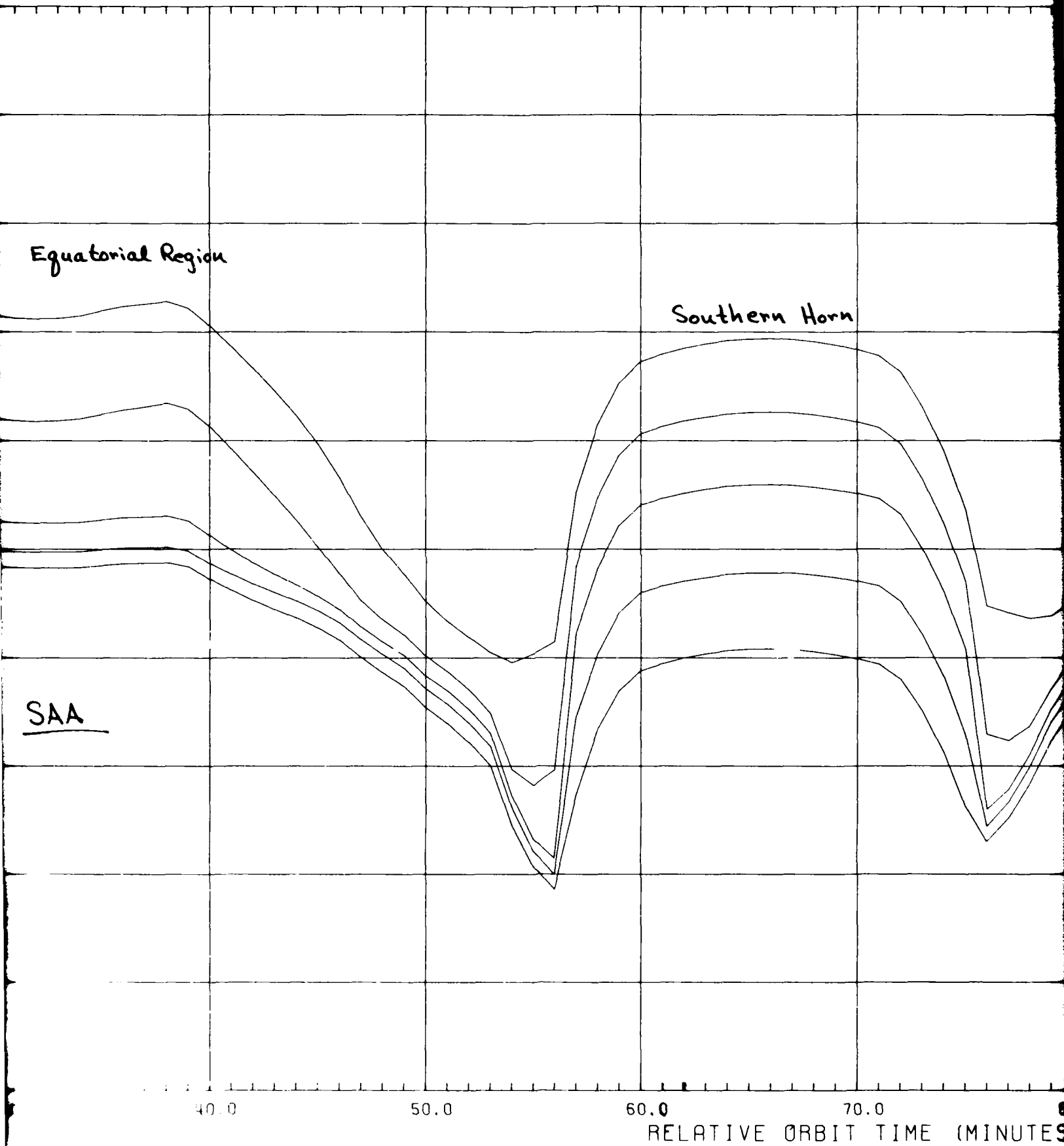


INSTANTANEOUS ALUMINUM ELECTRON DOSE (RADS)

(PLOTTED DOSE VALUES INCLUDE BREMSSTRAHLUNG CONTRIBUTIONS)



DOSE IN SEMI-INFINITE ALUMINUM



RELATIVE ORBIT TIME (MINUTES)

3
ITE ALUMINUM MEDIUM

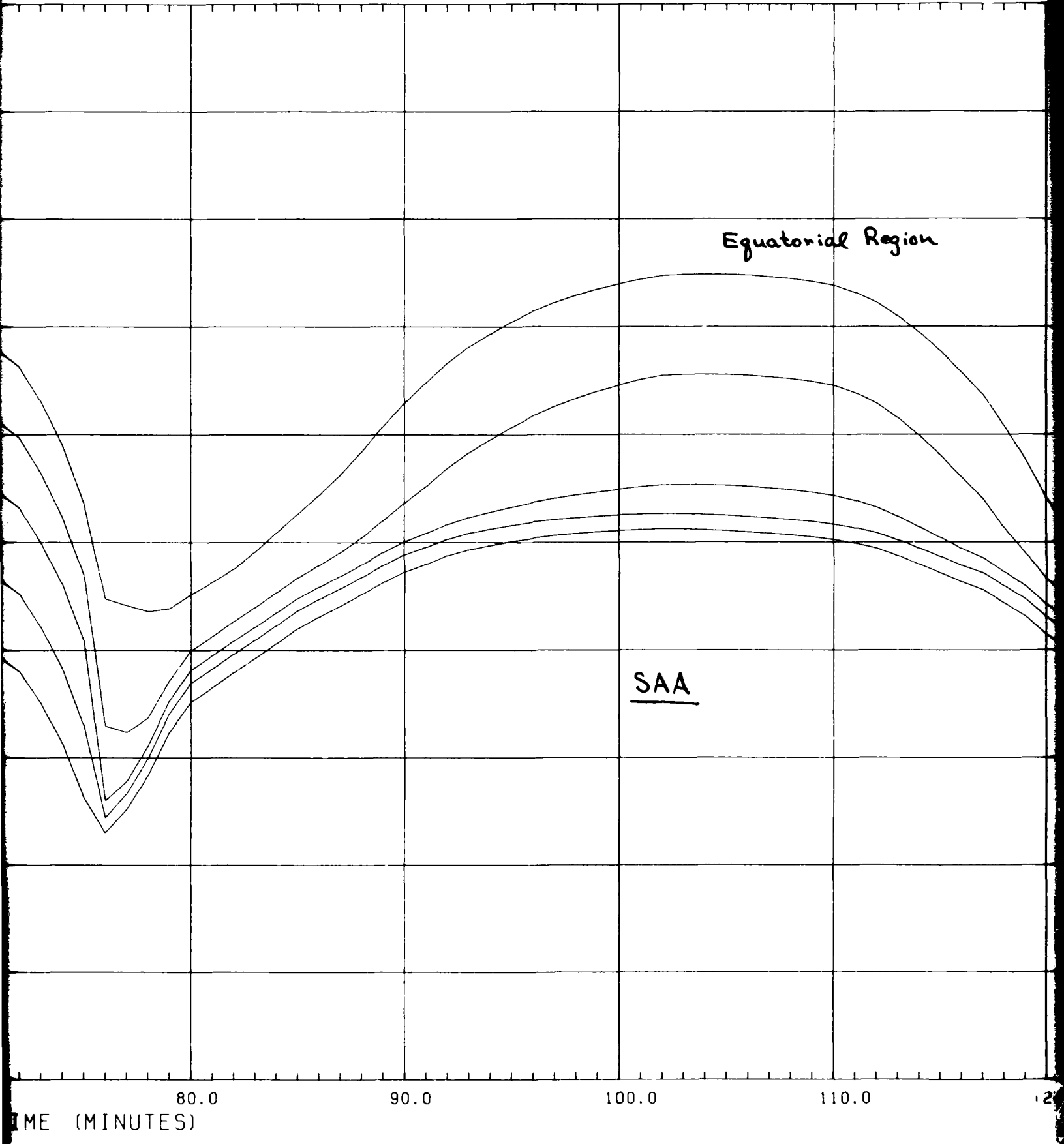


Figure 126

ORBIT: NAVELEX 2
60 DGR/2593-2593 KM

EPOCH: 1989.5

MODELS:

FIELD: BARR/75

TRAPPED PROTONS: AP8

INNER ZN ELEC: AE6

OUTER ZN ELEC: AE17-L0

MISSION DURATION: 60.00 MO

EVALUATION PHASE: SOLAR MAX

UN FACTORS: NOT APPLIED

atorial Region

Northern
Horn

STOP TIME ON TAPE = 7.616656

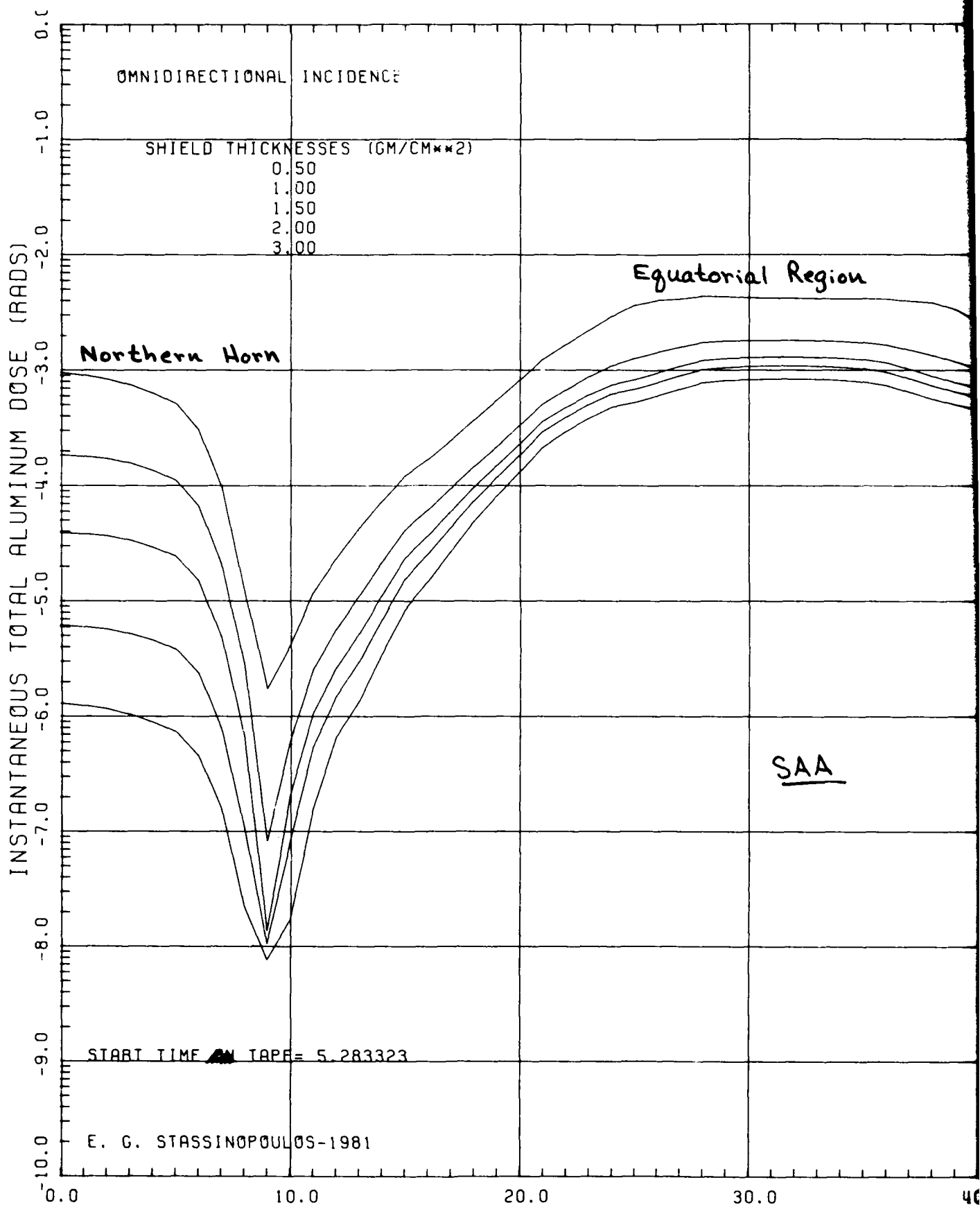
NASA-GSFC

110.0

120.0

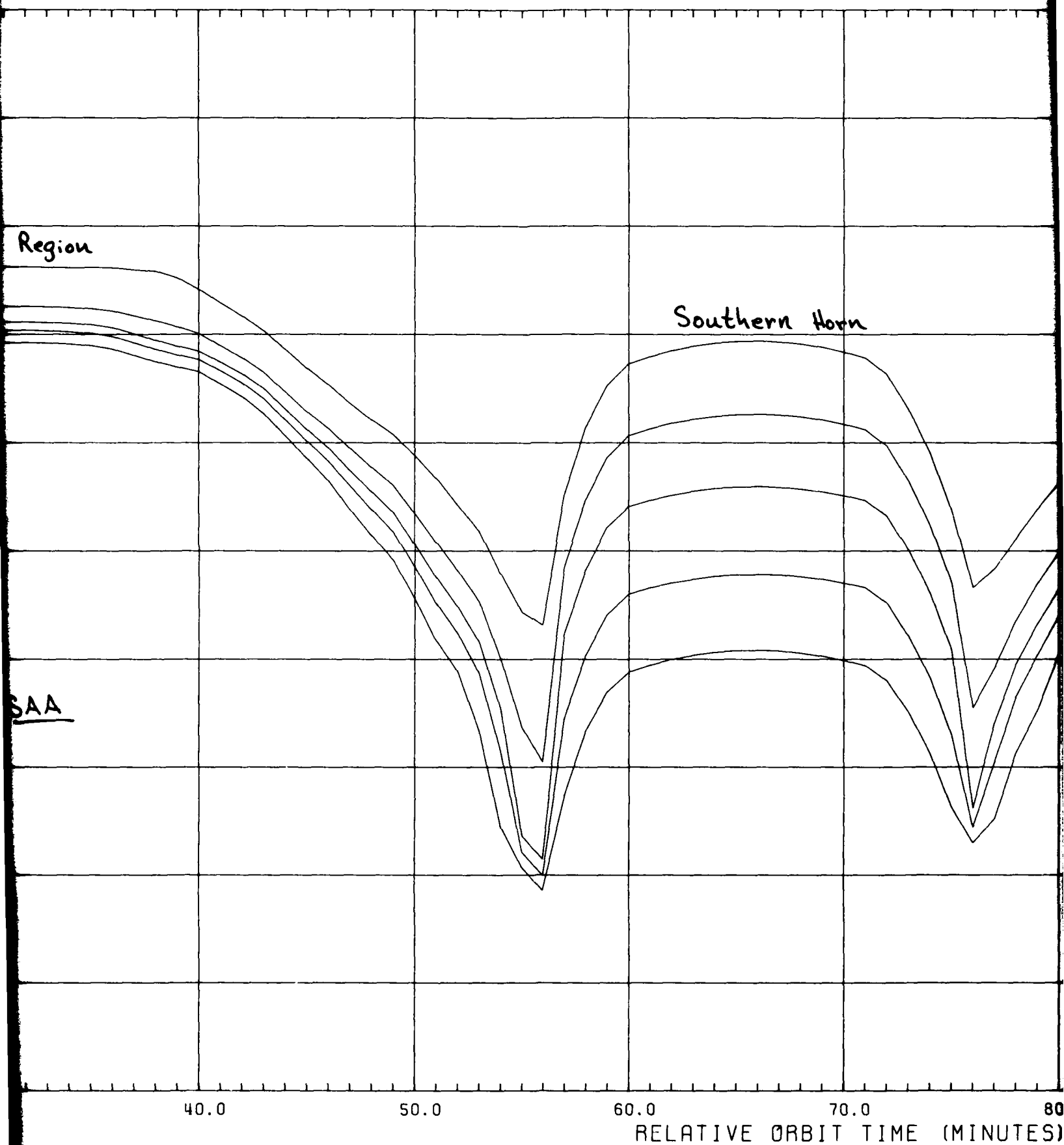
130.0

140.0



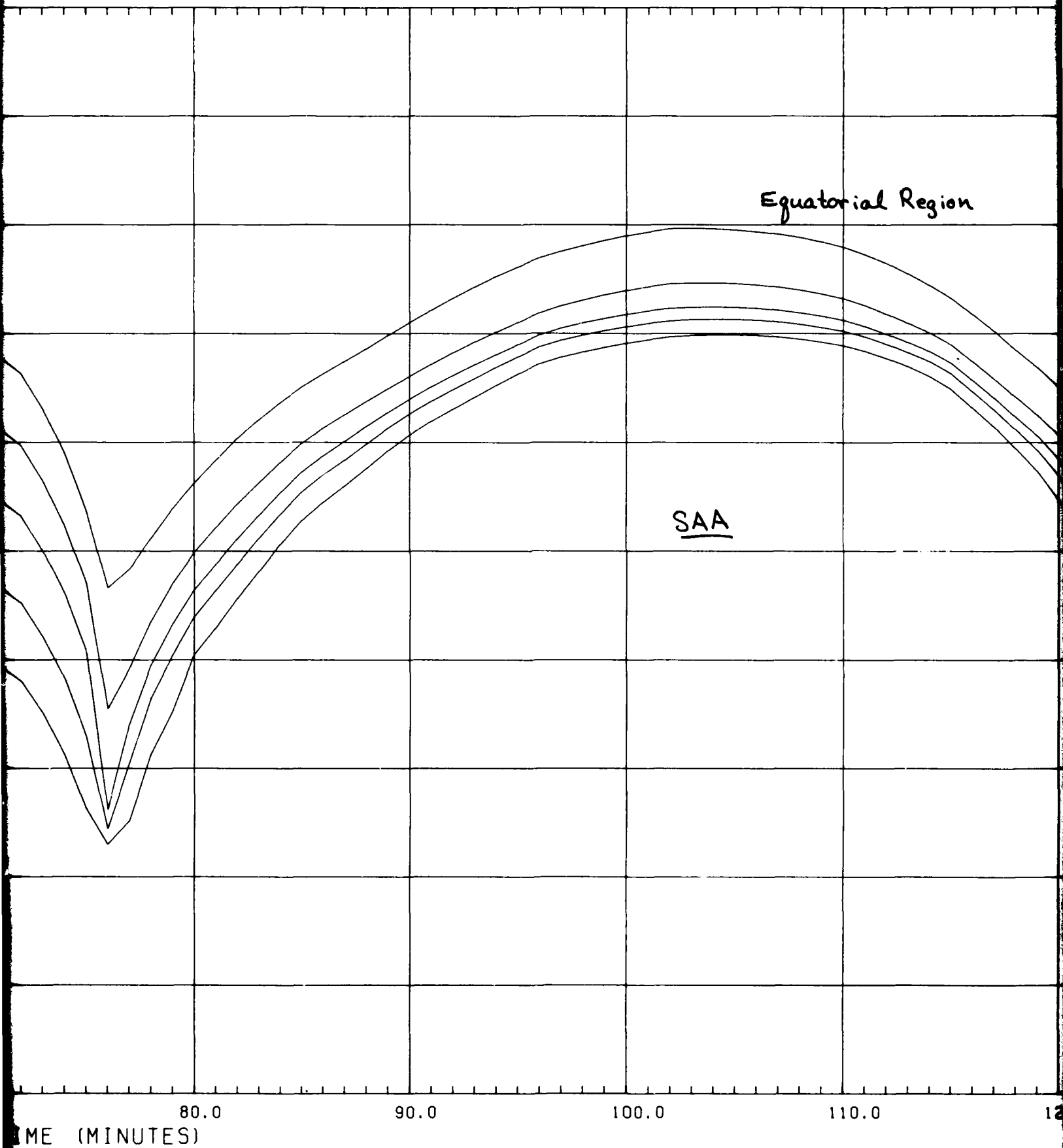
2'

DOSE IN SEMI-INFINITE ALUMINUM



13

ITE ALUMINUM MEDIUM



Equatorial Region

SAA

TIME (MINUTES)

Figure 127

ORBIT: NAVELEX 2
60 DGR/2593-2593 KM

EPOCH: 1989.5

MODELS:

FIELD: BARR/75

TRAPPED PROTONS: AP8

INNER ZN ELEC: AE6

OUTER ZN ELEC: AE17-L0

MISSION DURATION: 60.00 MO

EVALUATION PHASE: SOLAR MAX

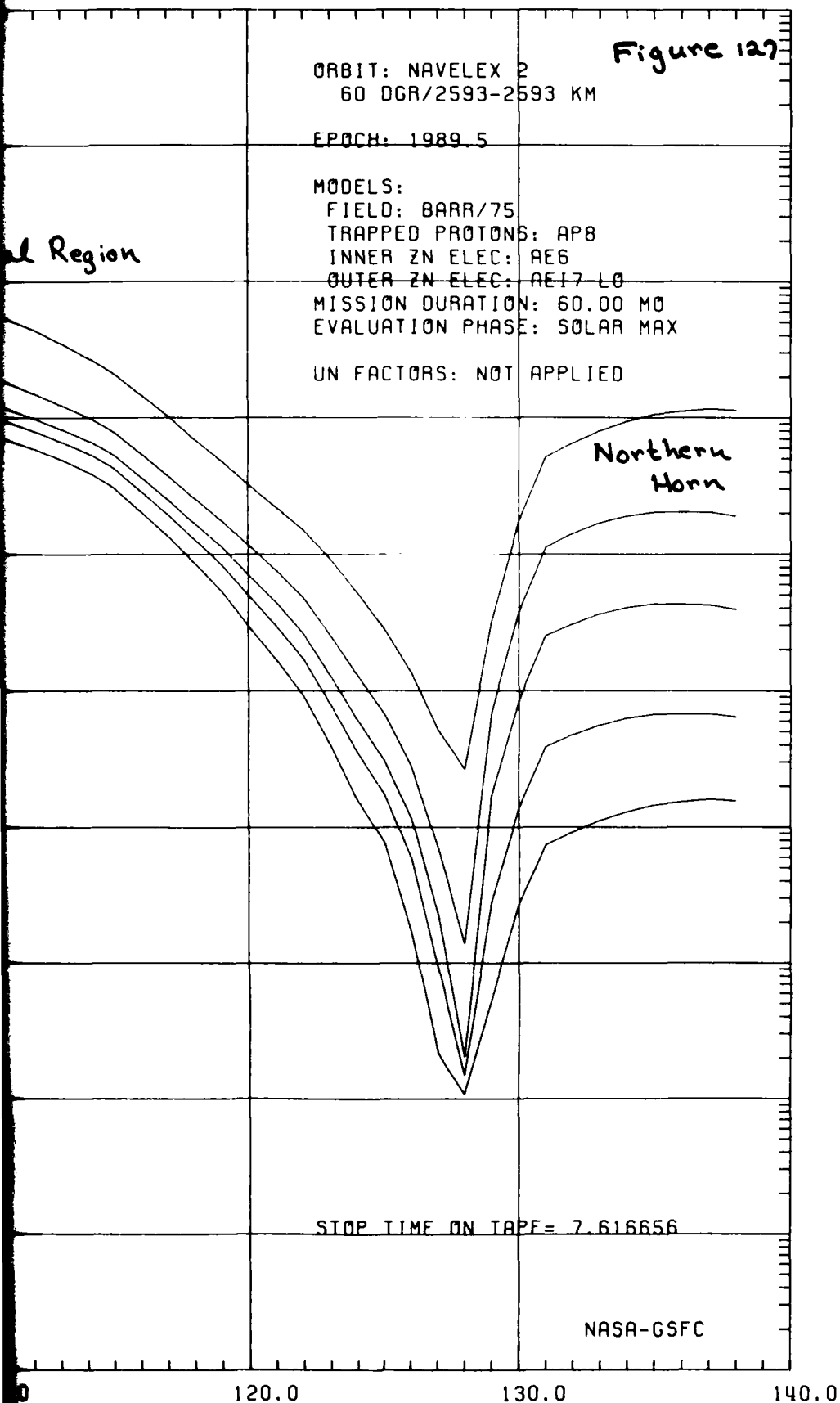
UN FACTORS: NOT APPLIED

al Region

Northern
Horn

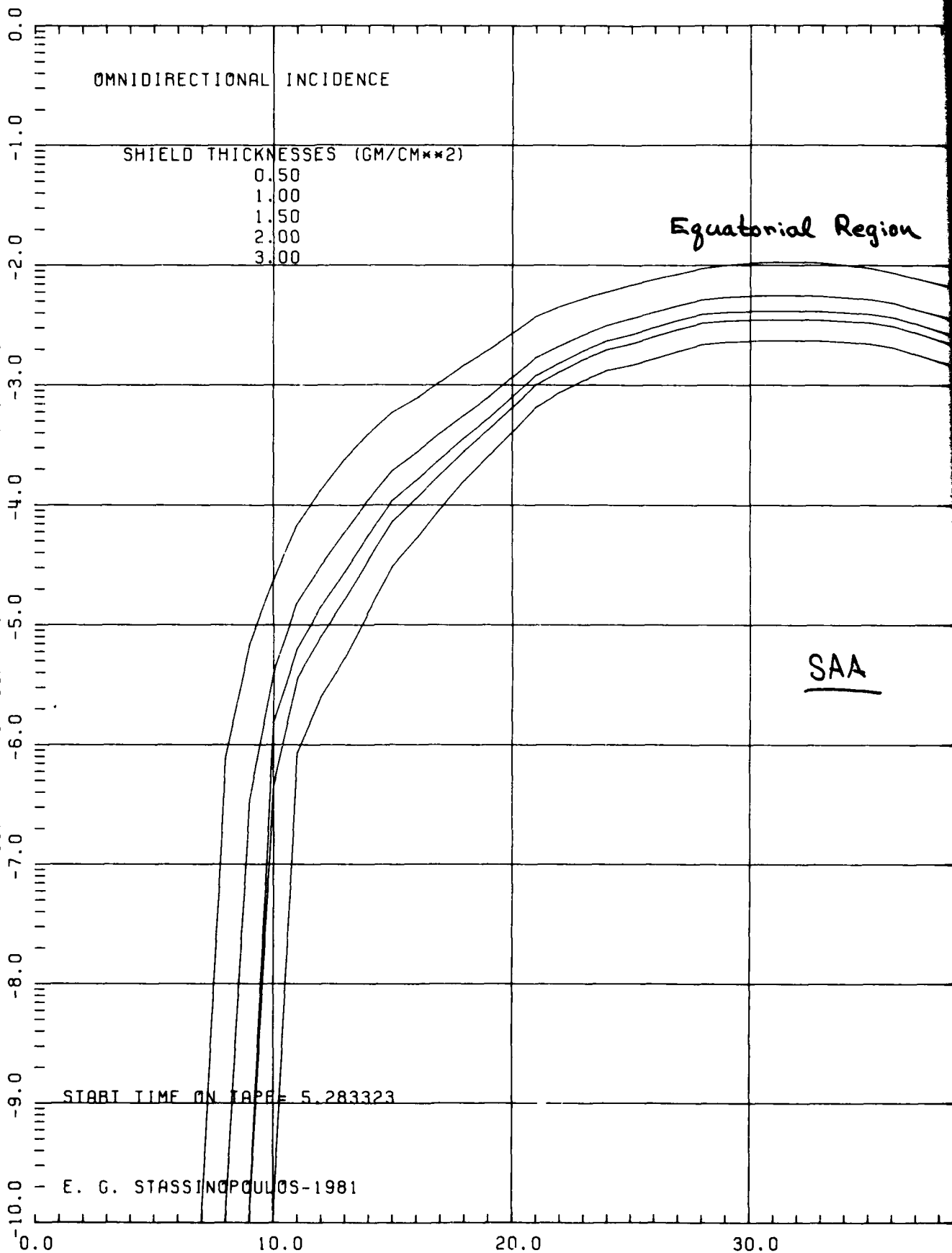
STOP TIME ON TAPE = 7.616656

NASA-GSFC



INSTANTANEOUS ALUMINUM PROTON DOSE (RADS)

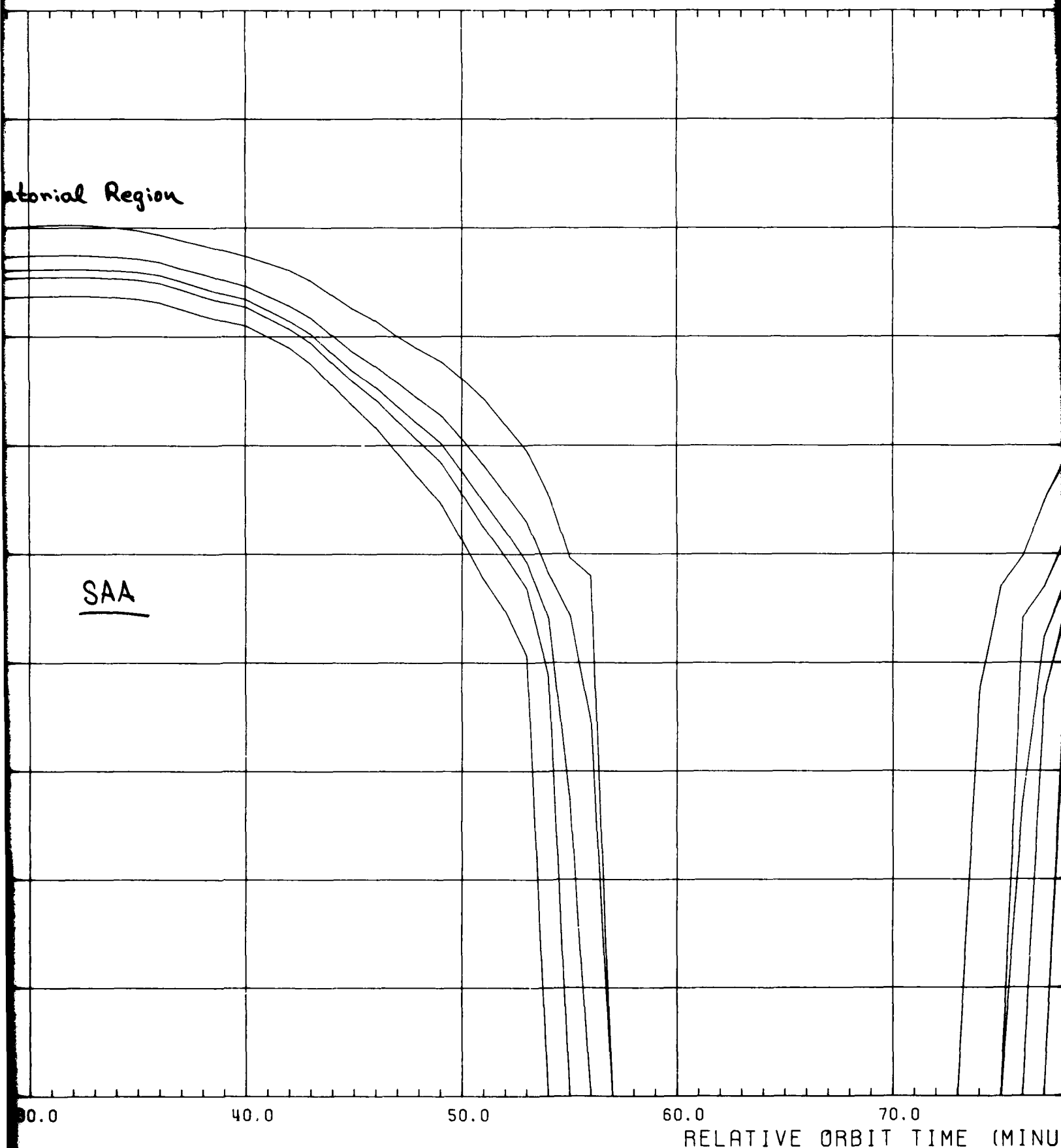
(SOLAR PROTON CONTRIBUTIONS ARE NOT INCLUDED)



DOSE AT CENTER OF ALUMINUM

atorial Region

SAA



RELATIVE ORBIT TIME (MINU)

OF ALUMINUM SPHERES

Equatorial Region

SAA

TIME (MINUTES)

80.0

90.0

100.0

110.0

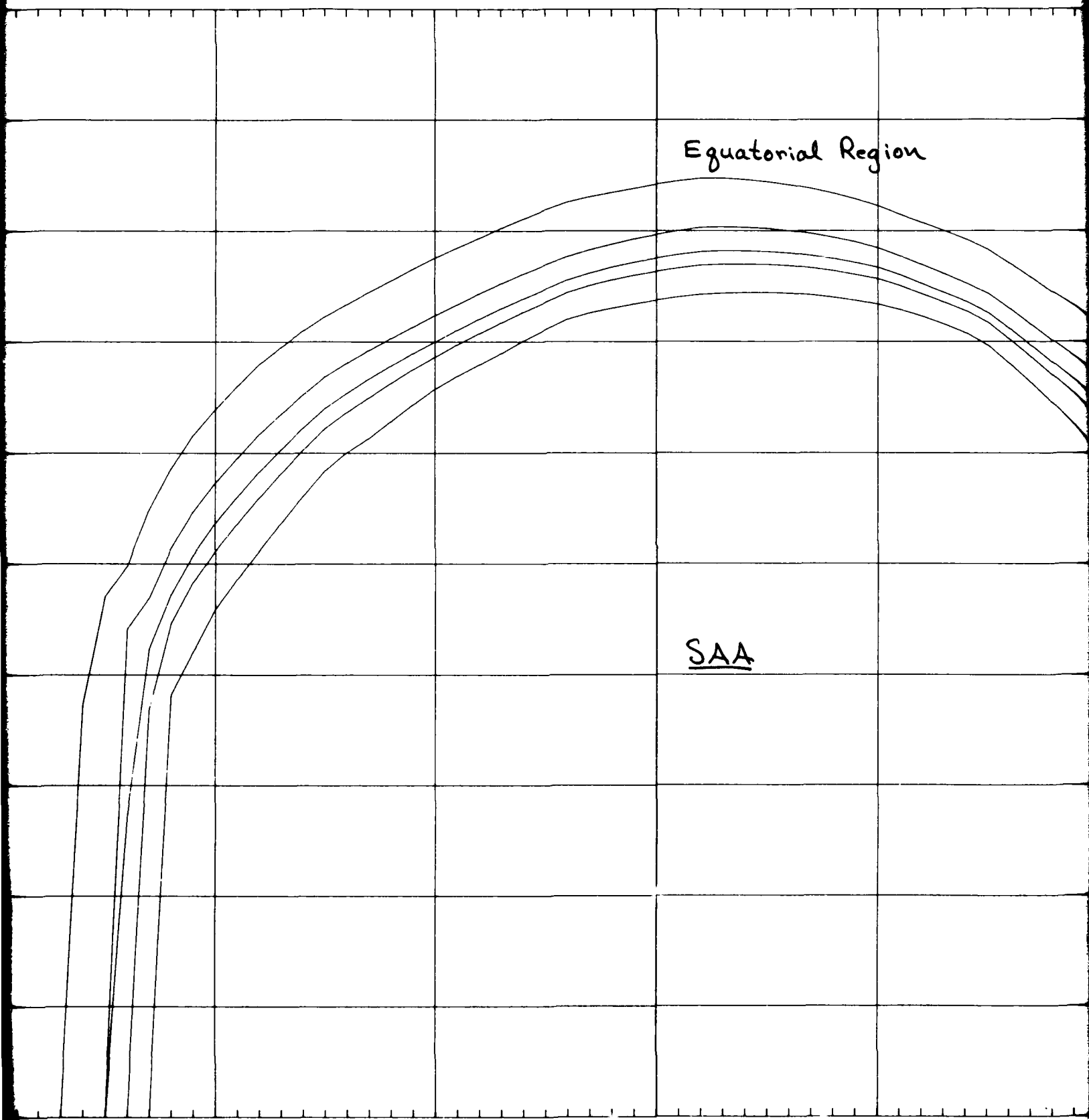


Figure 128

ORBIT: NAVELEX 2
60 DGR/2593-2593 KM

EPOCH: 1989.5

MODELS:
FIELD: BARR/75
TRAPPED PROTONS: AP8
INNER ZN ELEC: AE6
OUTER ZN ELEC: AE17-L0
MISSION DURATION: 60.00 MO
EVALUATION PHASE: SOLAR MAX

UN FACTORS: NOT APPLIED

Region

STOP TIME ON TAPE = 7.616656

NASA-GSFC

110.0

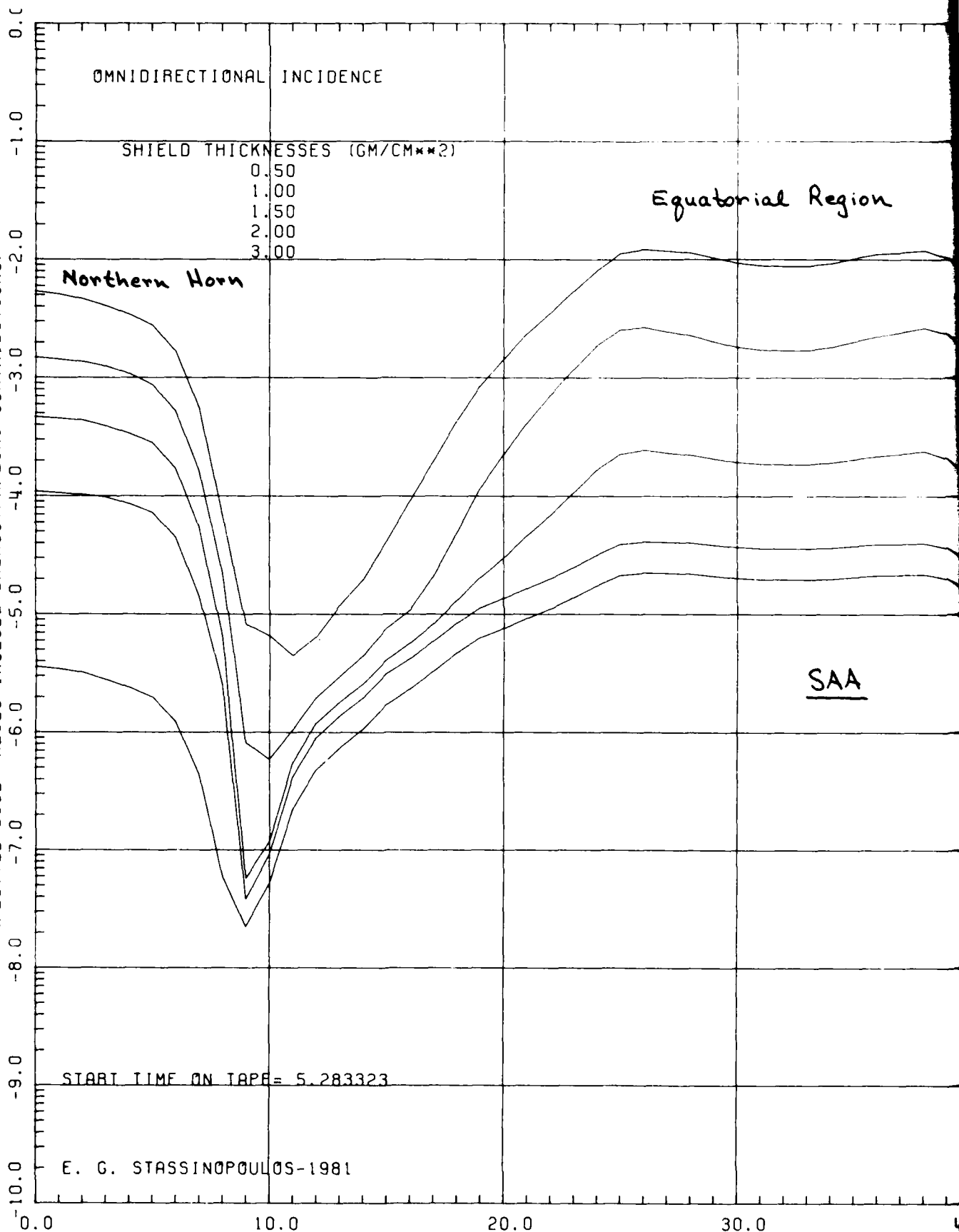
120.0

130.0

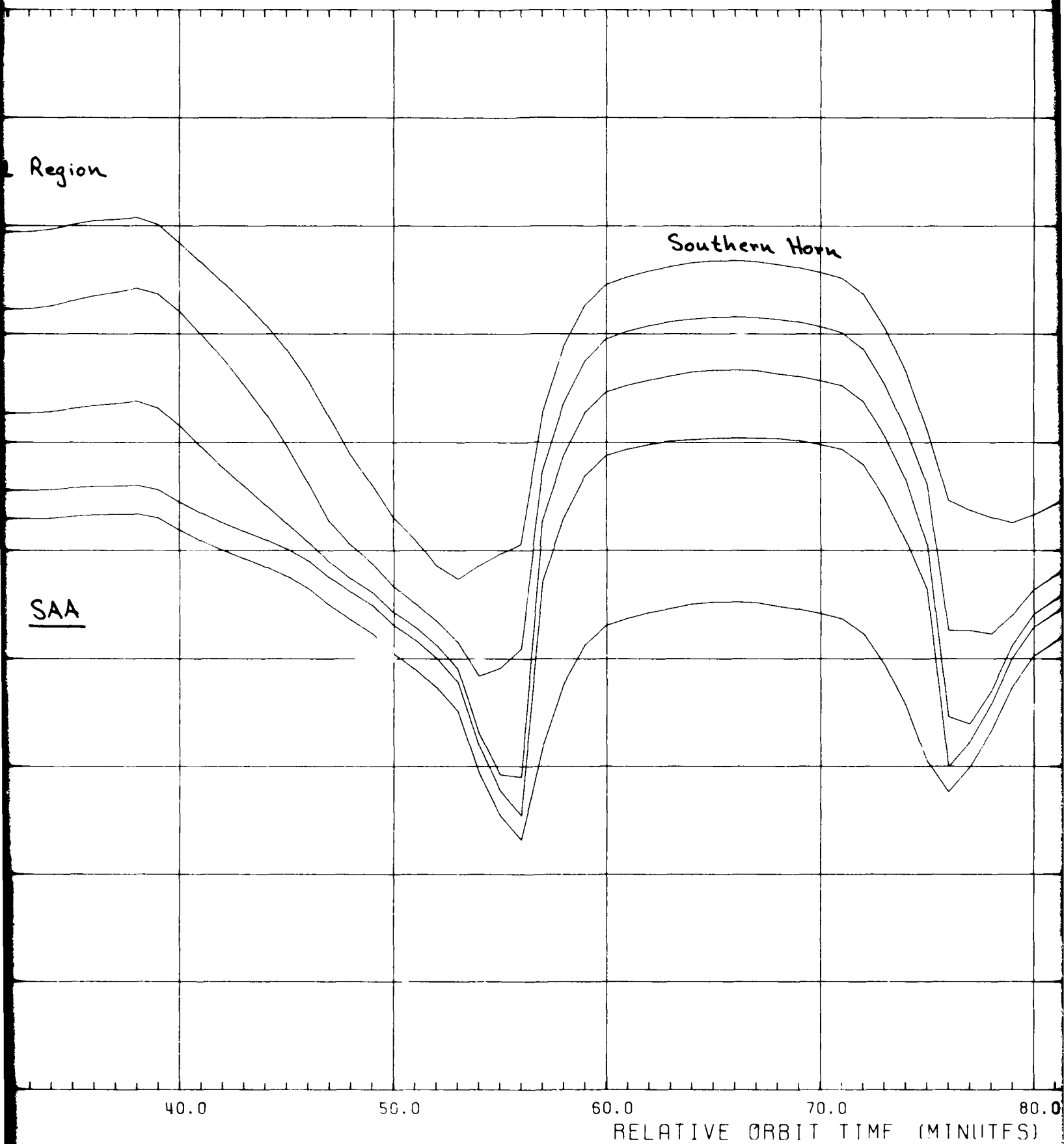
140.0

INSTANTANEOUS ALUMINUM ELECTRON DOSE (RADS)

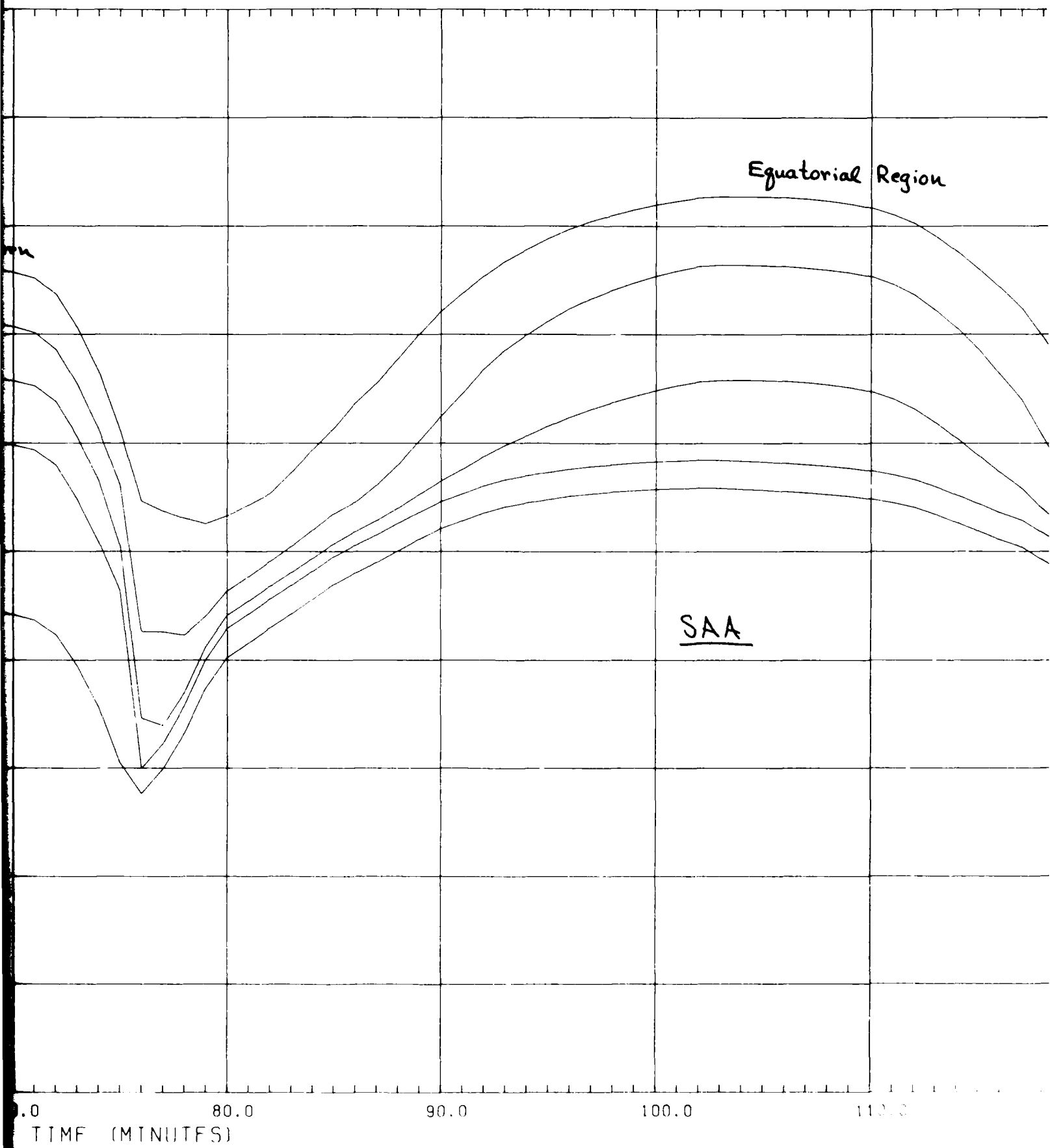
(PLOTTED DOSE VALUES INCLUDE BREMSSTRAHLUNG CONTRIBUTIONS)



DOSE AT CENTER OF ALUMINUM SPH



OF ALUMINUM SPHERES



AD-A141 849

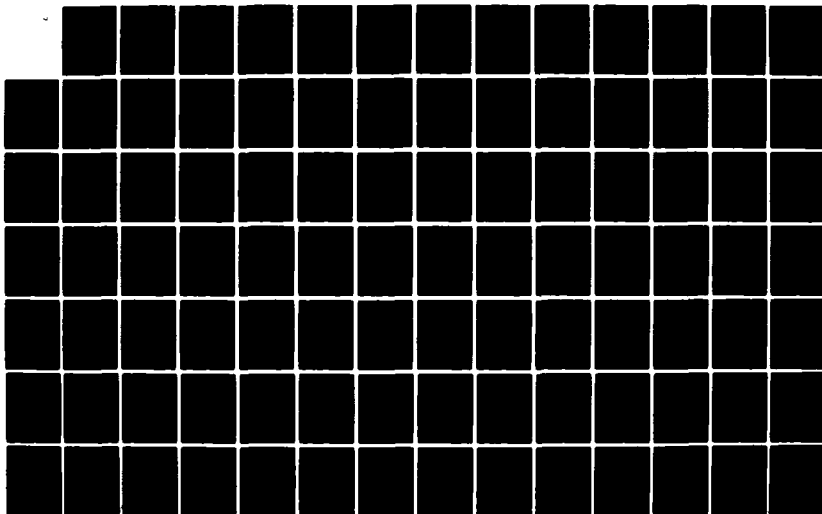
ORBITAL RADIATION STUDY FOR INCLINED CIRCULAR
TRAJECTORIES(U) NATIONAL AERONAUTICS AND SPACE
ADMINISTRATION GREENBELT MD GO.. E G STASSINOPOULOS
NOV 81 NASA-GSFC-X-601-81-28

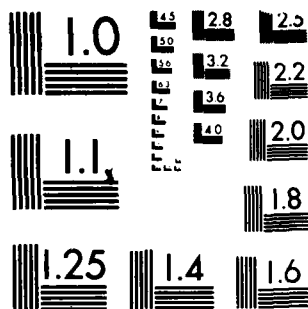
UNCLASSIFIED

F/G 22/3

NL

4/ 5





MICROCOPY RESOLUTION TEST CHART
NATIONAL BUREAU OF STANDARDS 1963-A

4

Figure 129

ORBIT: NAVELEX 2
60 DGR/2593-2593 KM

EPOCH: 1989.5

MODELS:

FIELD: BARR/75

TRAPPED PROTONS: AP8

INNER ZN ELEC: AE6

OUTER ZN ELEC: AE17 L0

MISSION DURATION: 60.00 MO

EVALUATION PHASE: SOLAR MAX

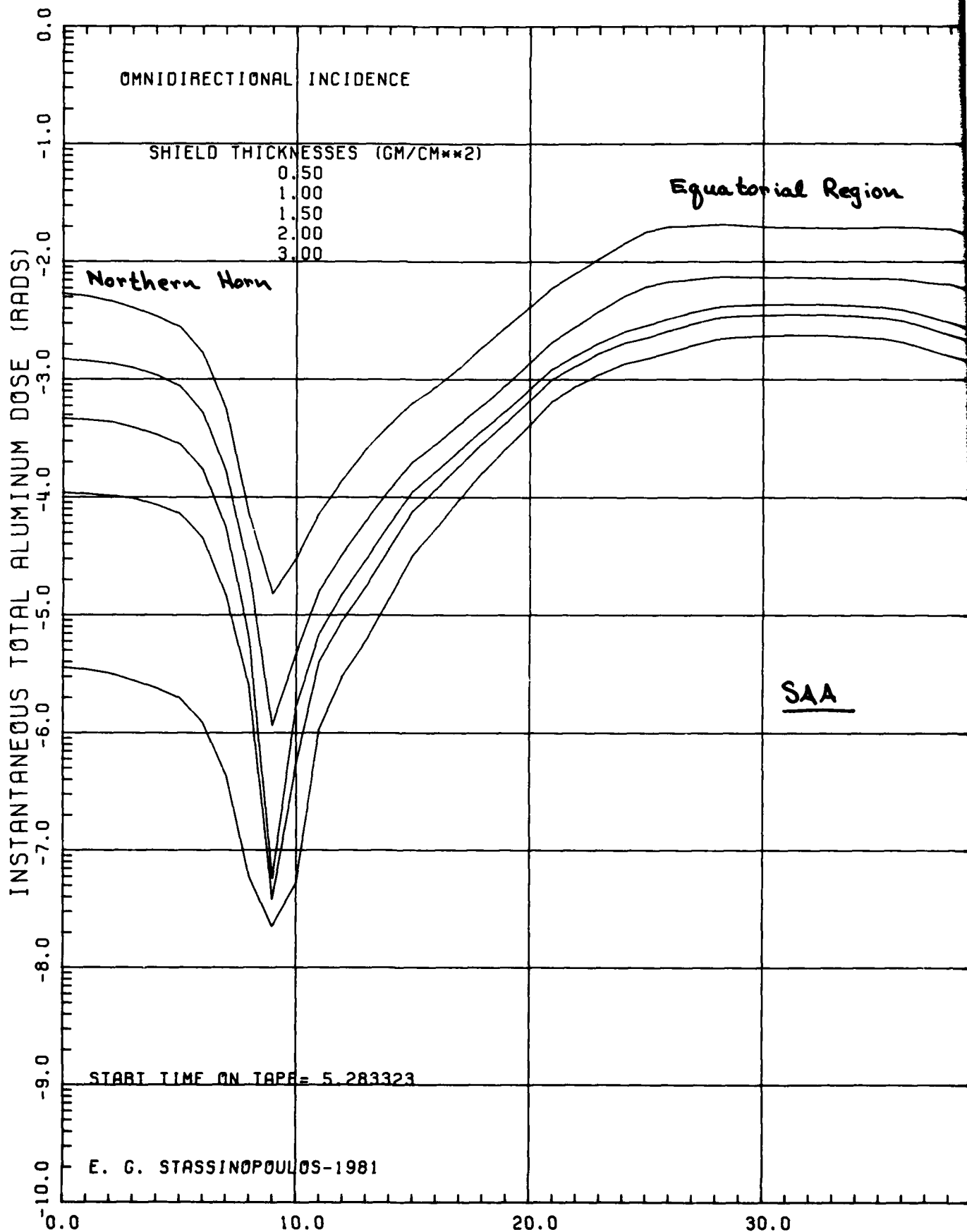
UN FACTORS: NOT APPLIED

Northern Horn

STOP TIME ON TAPE = 7.616656

NASA-GSFC

0.0 120.0 130.0 140.0



2

DOSE AT CENTER OF ALUMINUM S

Region

Southern Horn

AA

40.0

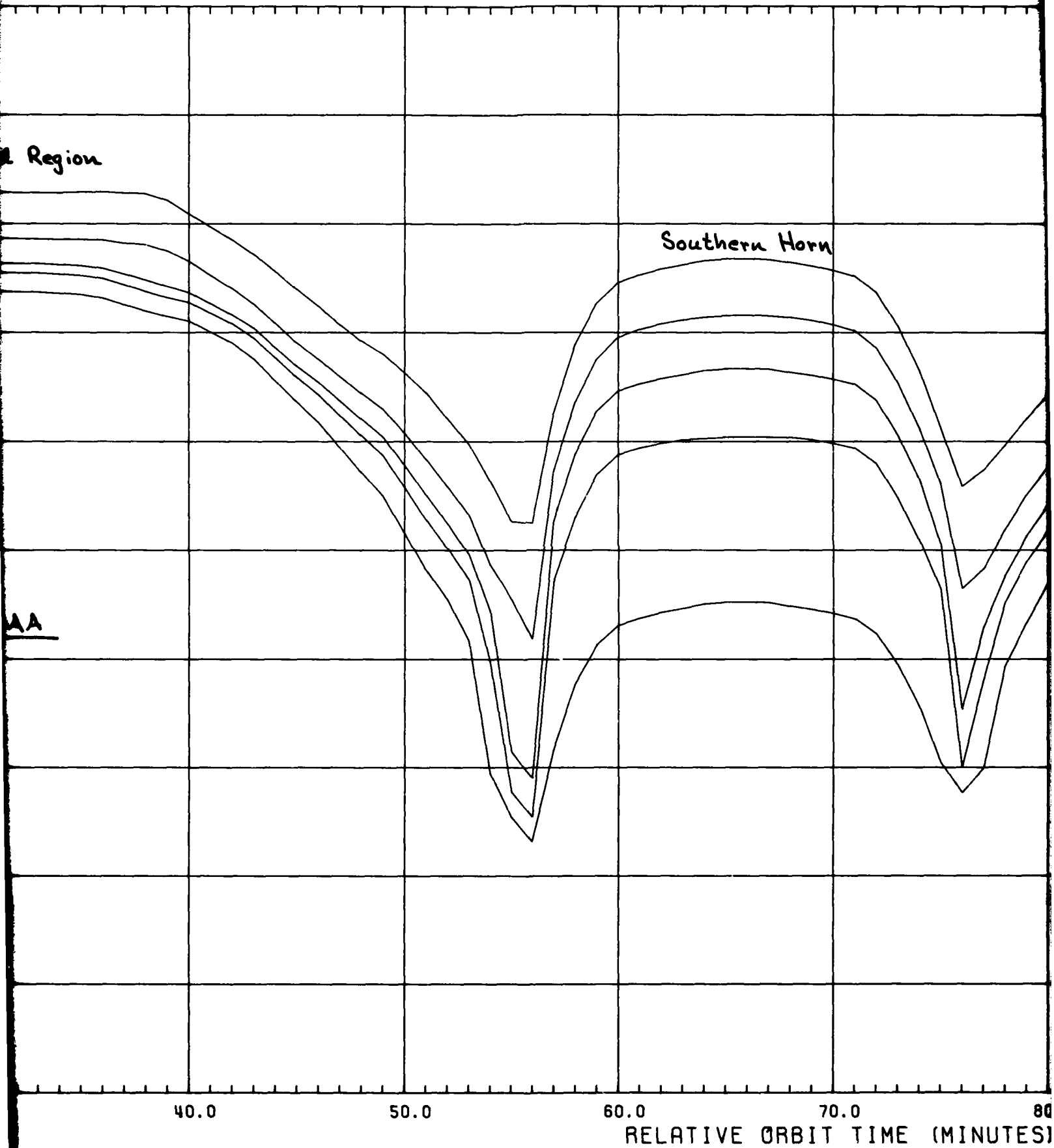
50.0

60.0

70.0

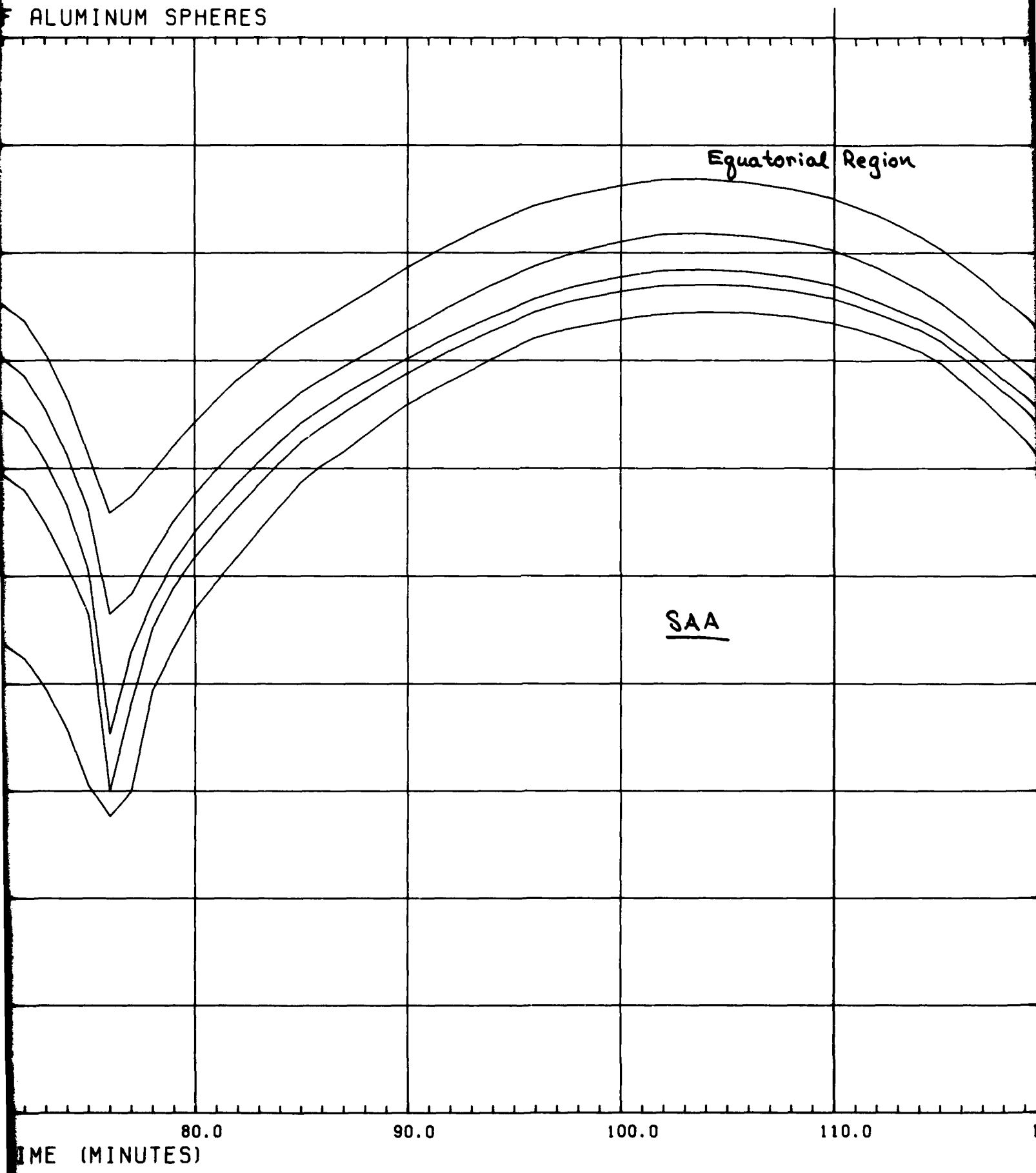
80

RELATIVE ORBIT TIME (MINUTES)



13

ALUMINUM SPHERES



4

Figure 130

ORBIT: NAVELEX 2
60 DGR/2593-2593 KM

EPOCH: 1989.5

MODELS:
FIELD: BARR/75
TRAPPED PROTONS: AP8
INNER ZN ELEC: AE6
OUTER ZN ELEC: AE17 L8
MISSION DURATION: 60.00 MO
EVALUATION PHASE: SOLAR MAX

UN FACTORS: NOT APPLIED

Orbital Region

Northern
Horn

STOP TIME ON TAPE = 7.616656

NASA-GSFC

110.0

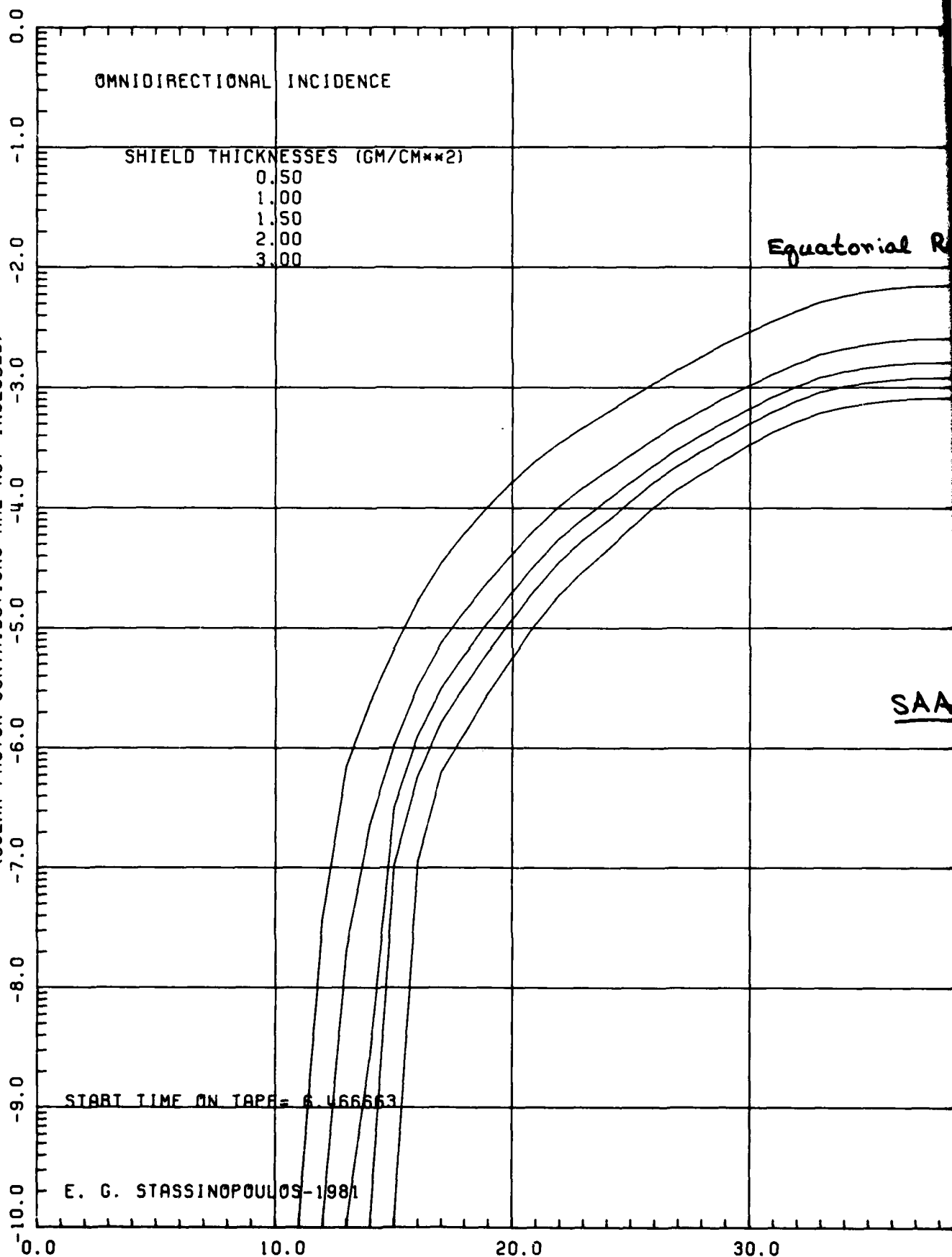
120.0

130.0

140.0

INSTANTANEOUS ALUMINUM PROTON DOSE (RADS)

(SOLAR PROTON CONTRIBUTIONS ARE NOT INCLUDED)



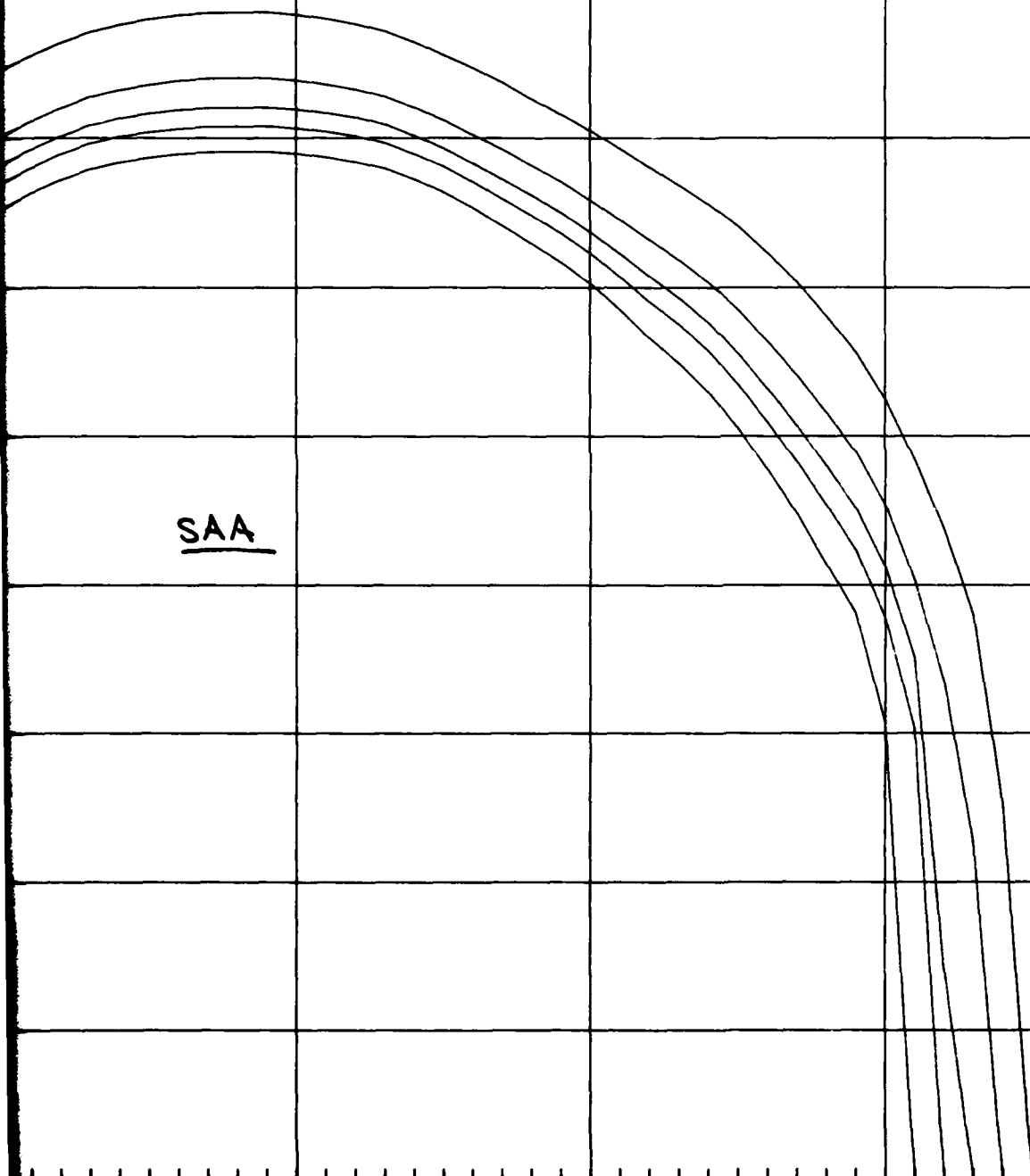
2

DOSE AT T

(= 0

Equatorial Region

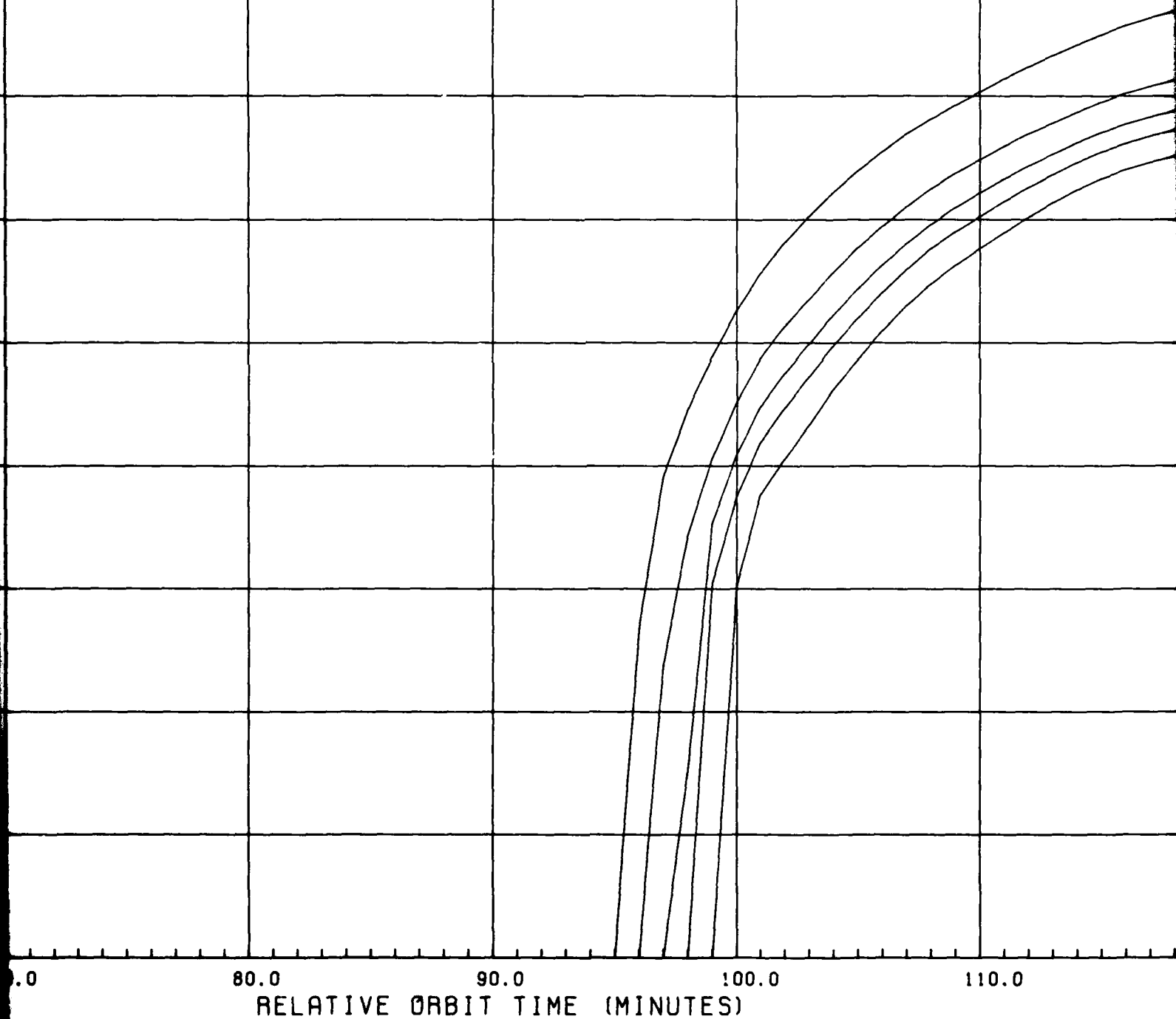
SAA



12

DOSE AT TRANSMISSION SURFACE OF FINITE ALUMINUM SLAB SHIELDS

(= DOSE IN SEMI-INFINITE ALUMINUM MEDIUM)



HIELDS

14

Equatorial Region

SAA

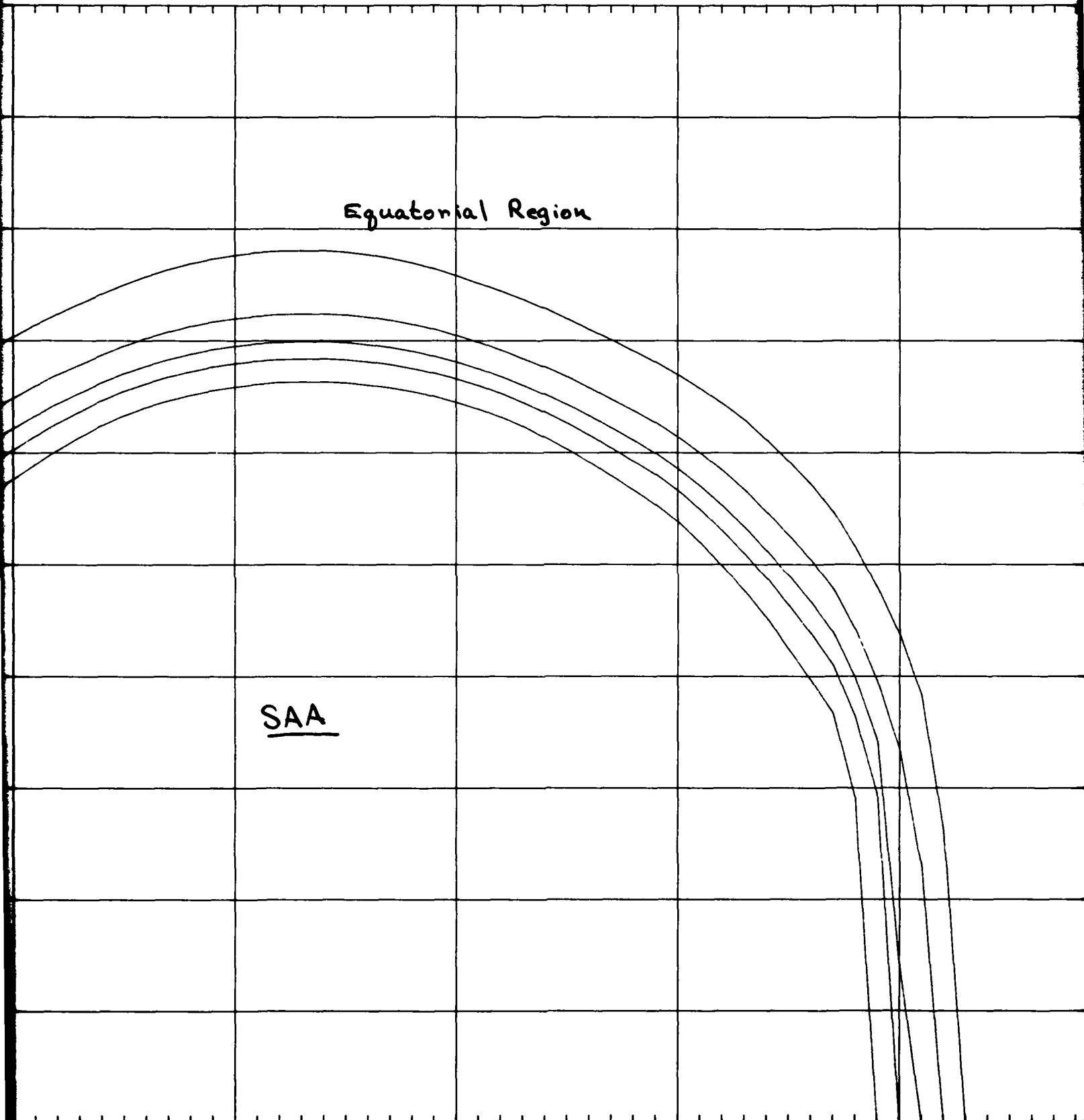
0.0

120.0

130.0

140.0

150.0



5

Figure 131

ORBIT: NAVELEX B
60 DGR/3889-3889 KM

EPOCH: 1989.5

MODELS:
FIELD: BARR/75
TRAPPED PROTONS: AP8
INNER ZN ELEC: AE6
OUTER ZN ELEC: AE17 L8
MISSION DURATION: 60.00 MO
EVALUATION PHASE: SOLAR MAX

UN FACTORS: NOT APPLIED

STOP TIME ON TAPE = 9.316662

NASA-GSFC

150.0

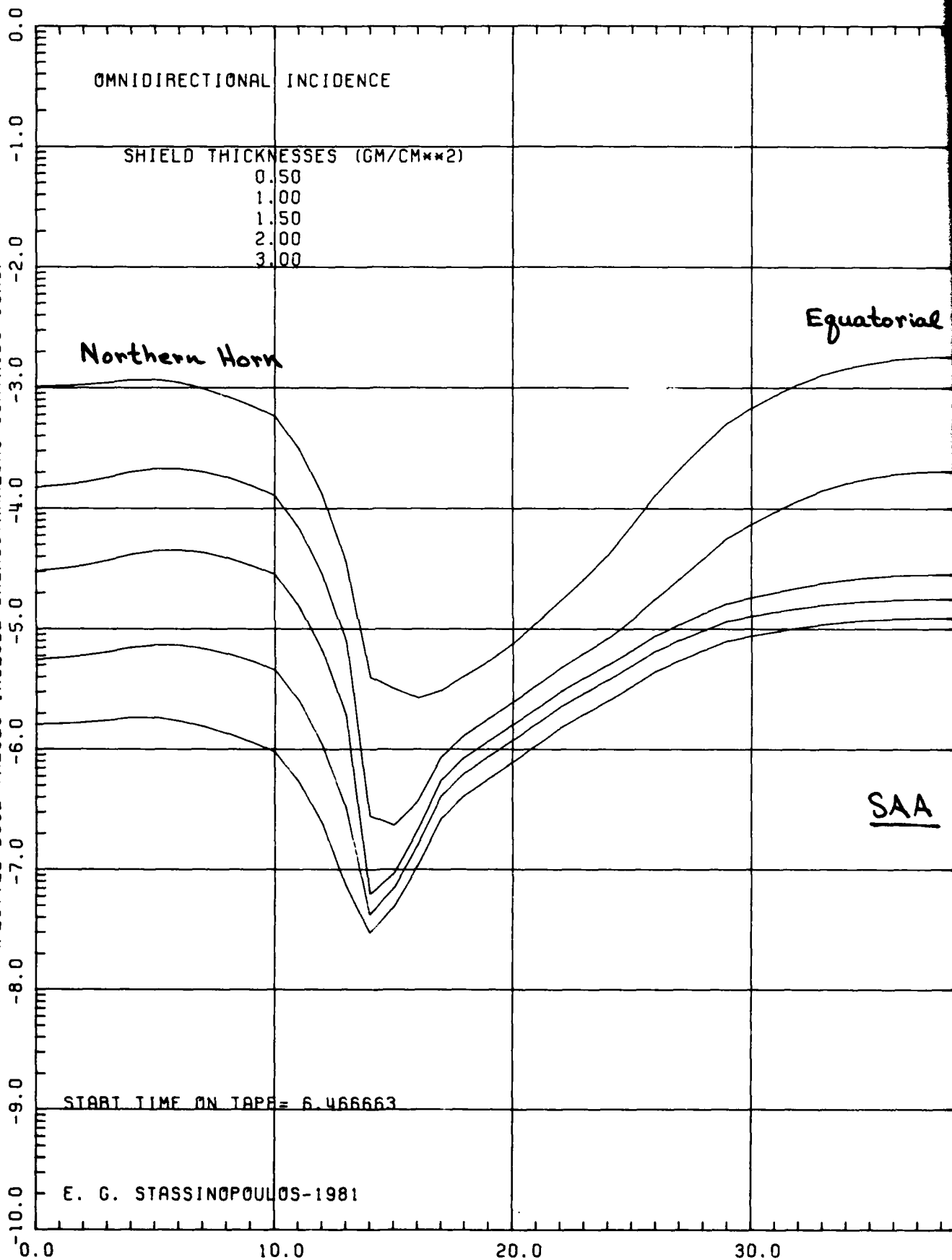
160.0

170.0

180.0

INSTANTANEOUS ALUMINUM ELECTRON DOSE (RADS)

(PLOTTED DOSE VALUES INCLUDE BREMSSTRAHLUNG CONTRIBUTIONS)



1 2

DOSE AT TRA

Equatorial Region

Southern Horn

SAA

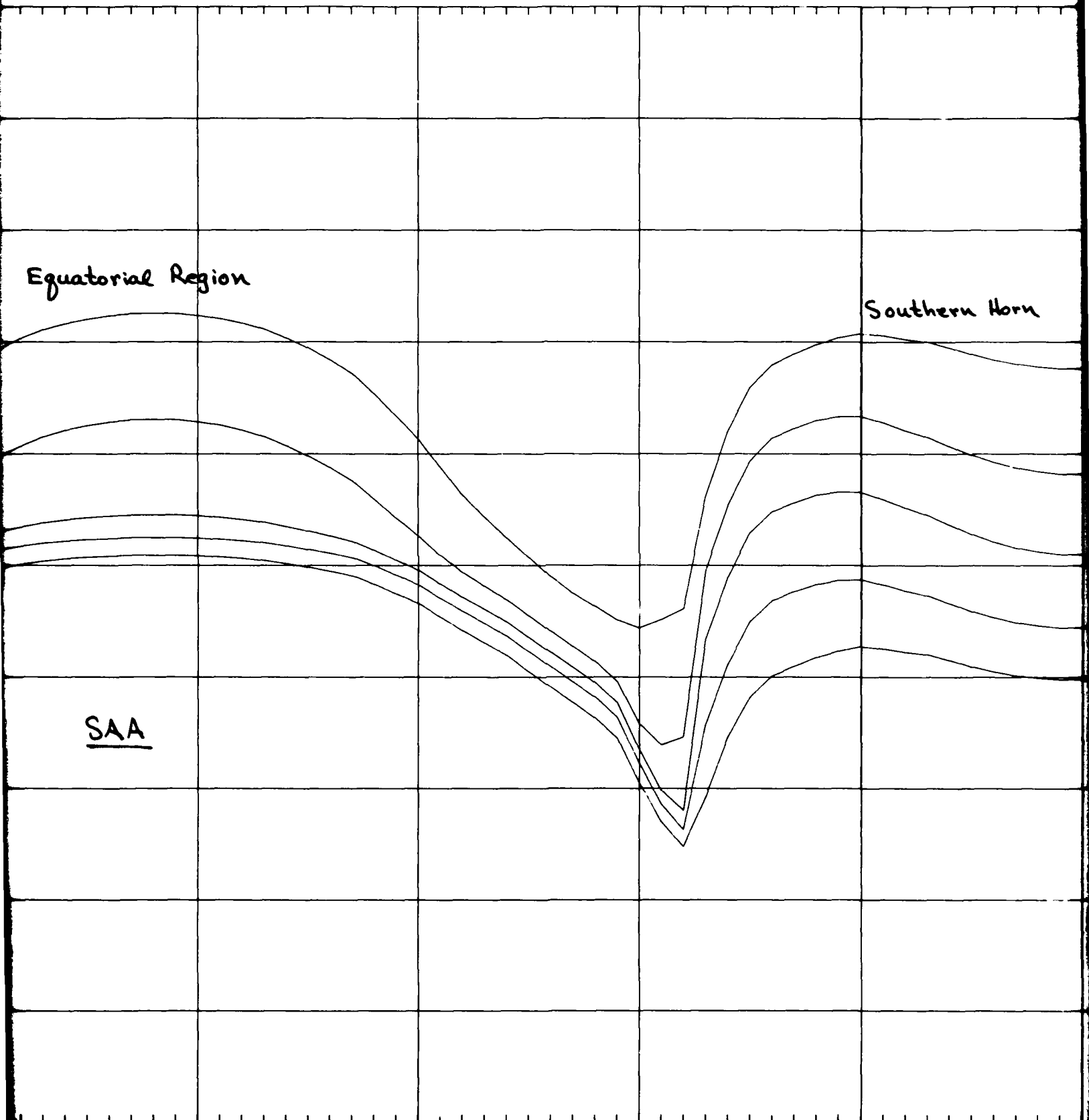
40.0

50.0

60.0

70.0

80.0



13

DOSE AT TRANSMISSION SURFACE OF FINITE ALUMINUM SLAB SHIELDS

Southern Horn

Southern Horn

80.0

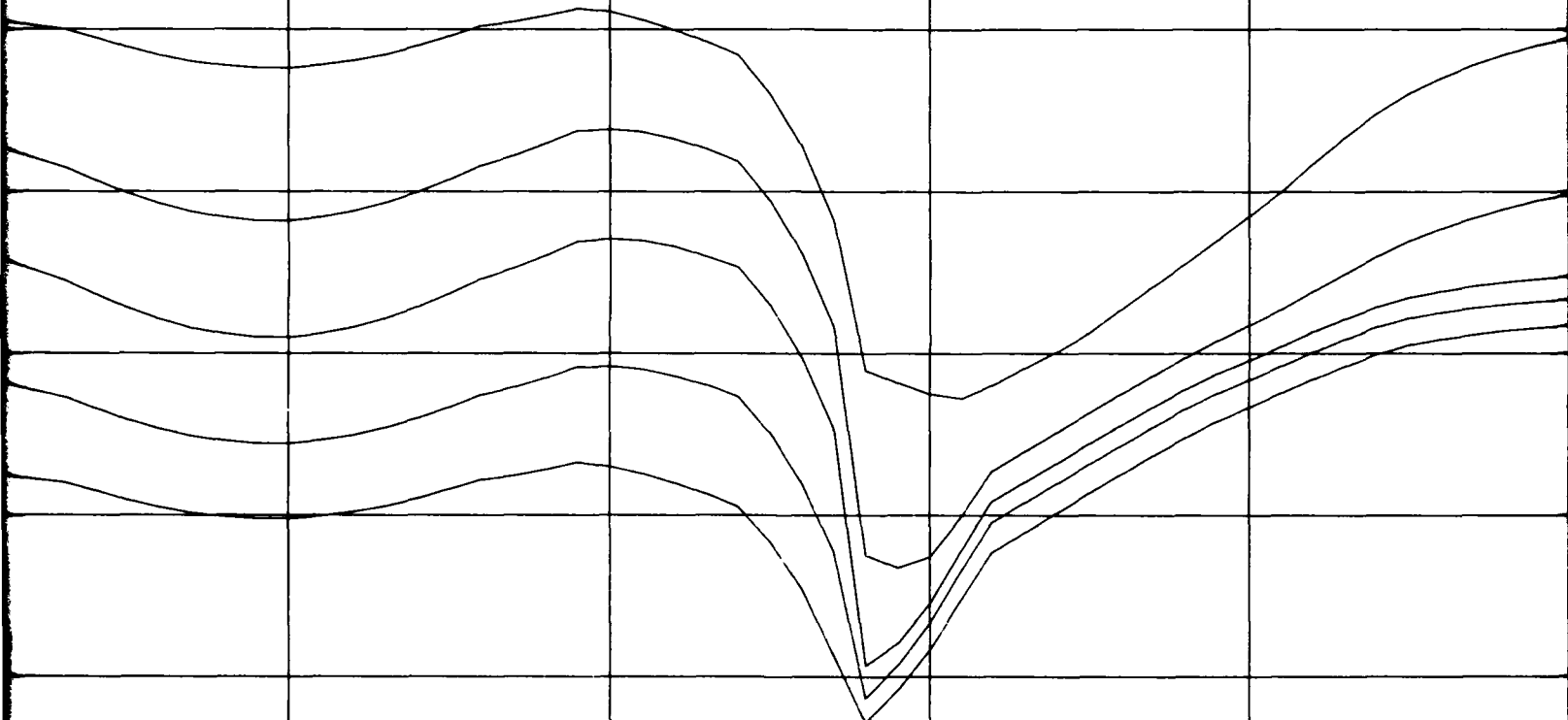
90.0

100.0

110.0

120.0

RELATIVE ORBIT TIME (MINUTES)



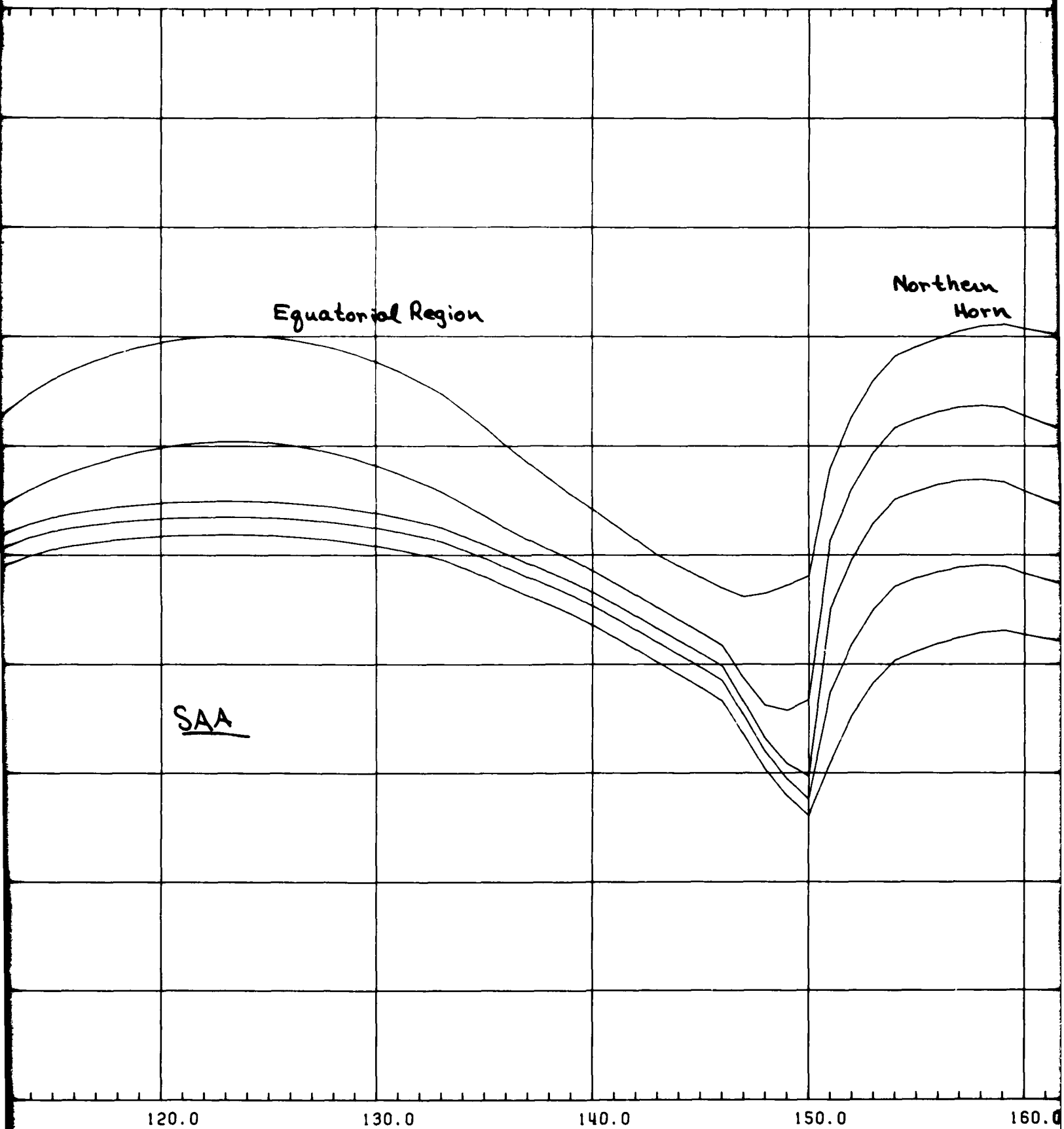


Figure 132

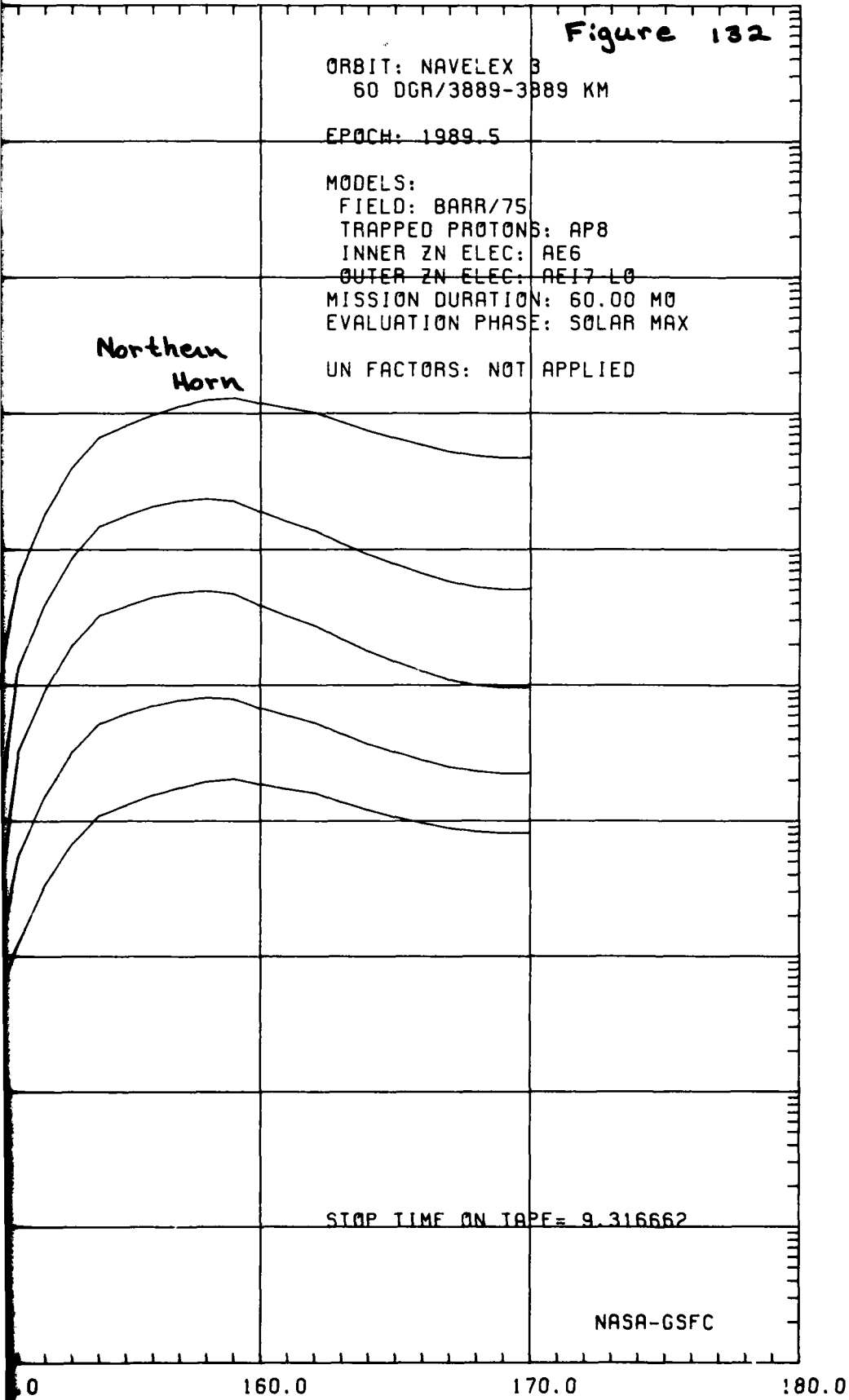
ORBIT: NAVELEX B
60 DGR/3889-3889 KM

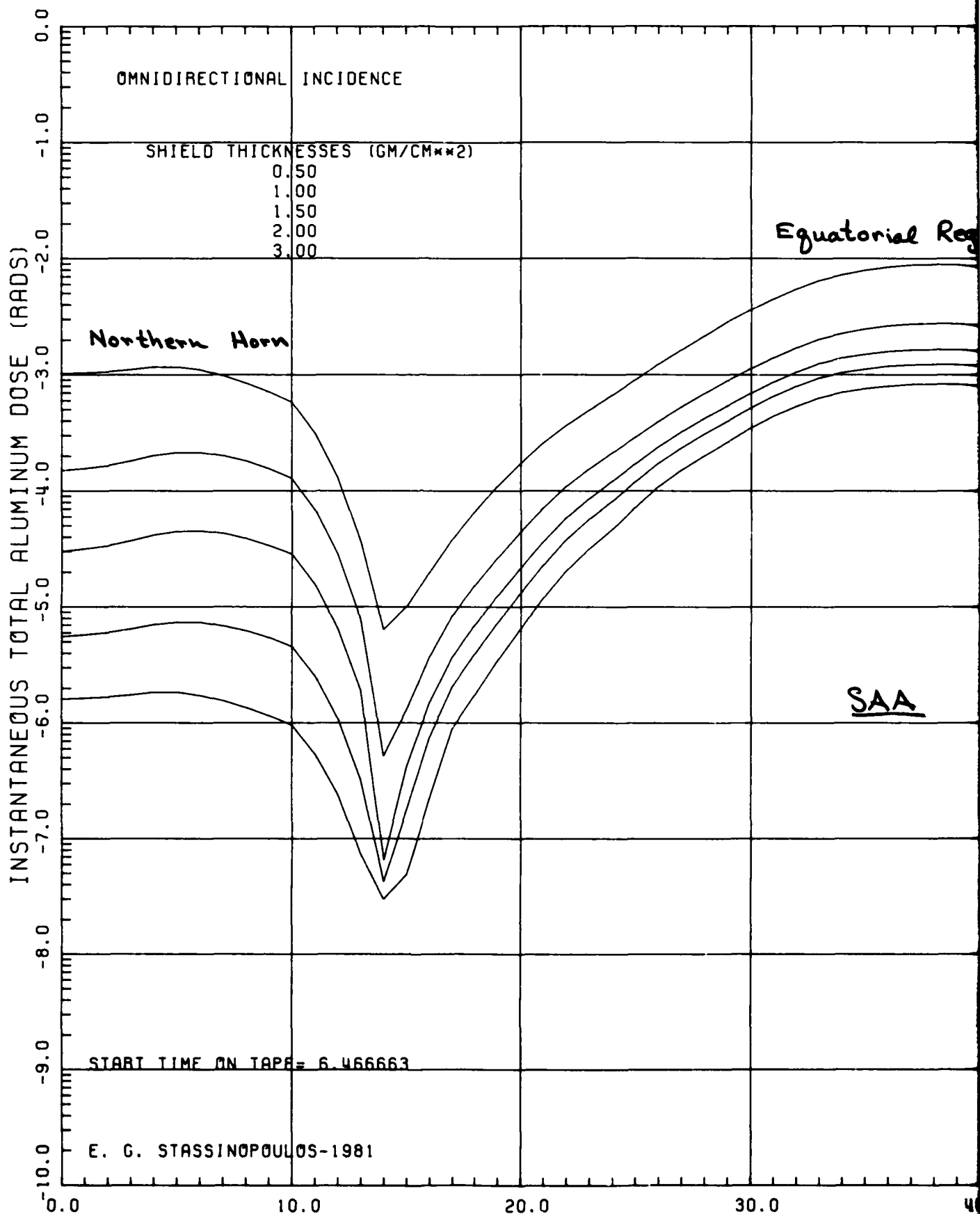
EPOCH: 1989.5

MODELS:
FIELD: BARR/75
TRAPPED PROTONS: AP8
INNER ZN ELEC: AE6
OUTER ZN ELEC: AE17 L0
MISSION DURATION: 60.00 MO
EVALUATION PHASE: SOLAR MAX

UN FACTORS: NOT APPLIED

Northern
Horn





2

DOSE AT TR

Equatorial Region

Southern Horn

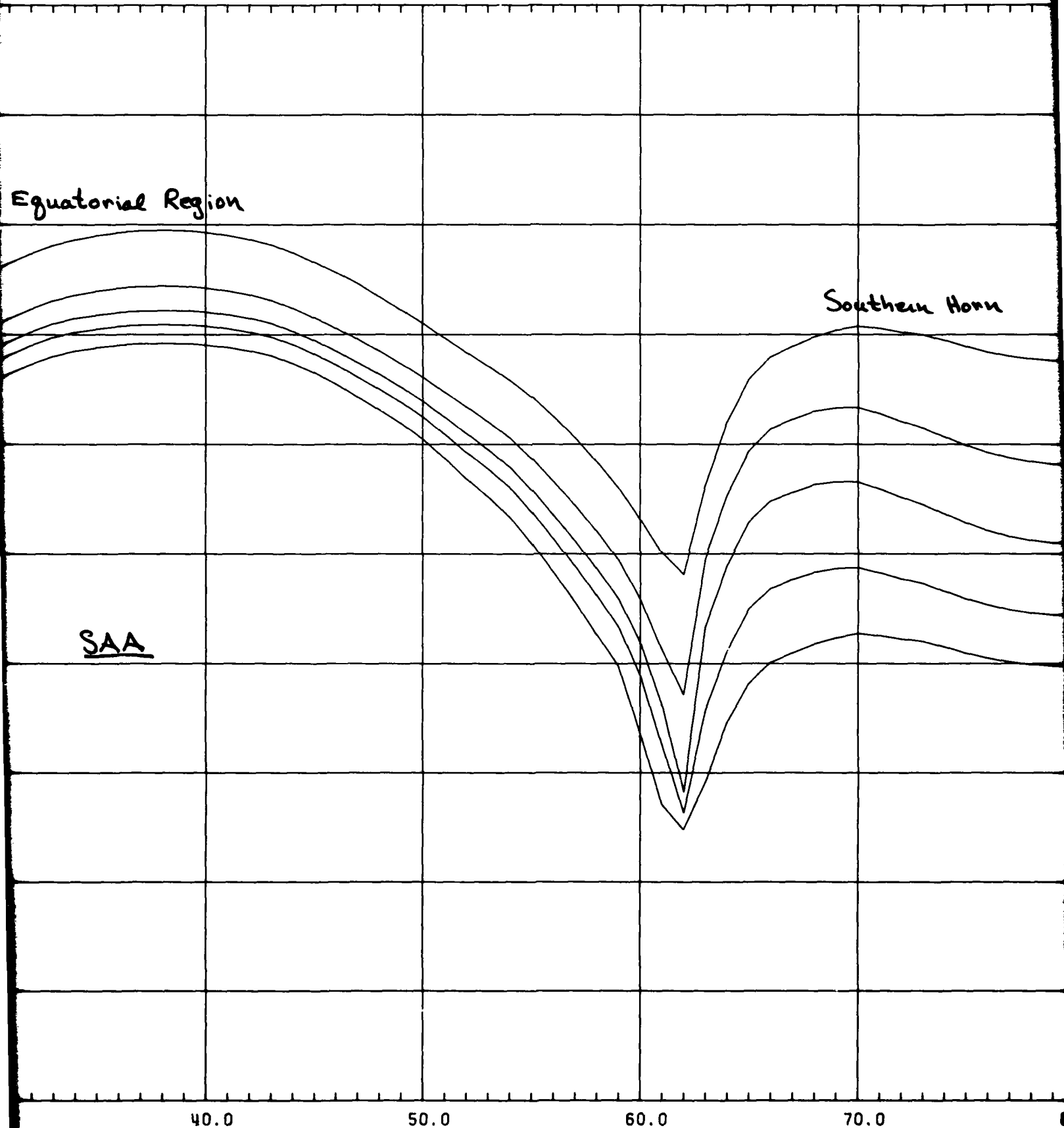
SAA

40.0

50.0

60.0

70.0

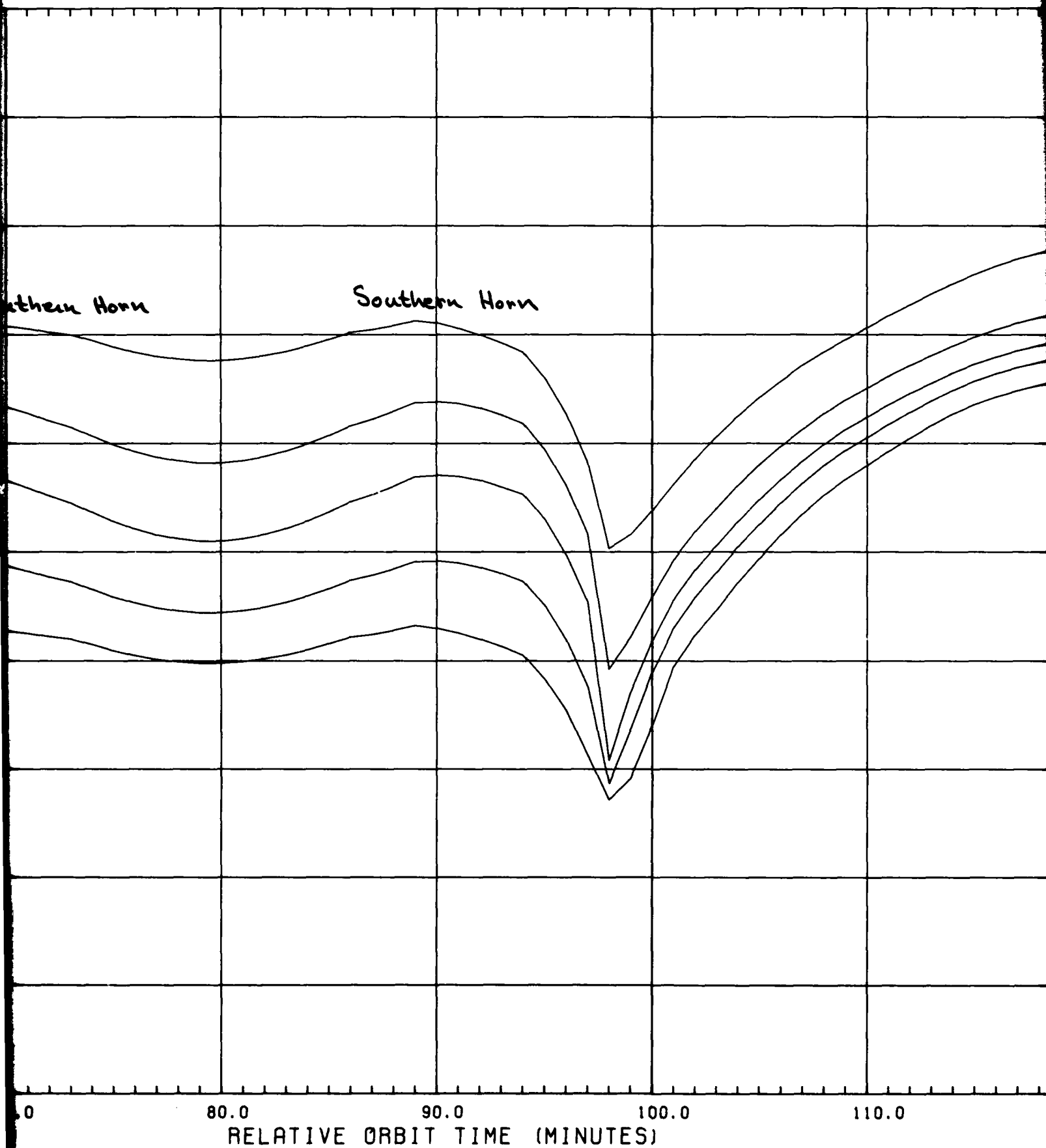


13

DOSE AT TRANSMISSION SURFACE OF FINITE ALUMINUM SLAB SHIELDS

Northern Horn

Southern Horn



IELDS

4

Equatorial Region

Northern
Horn

SAA

0.0

120.0

130.0

140.0

150.0

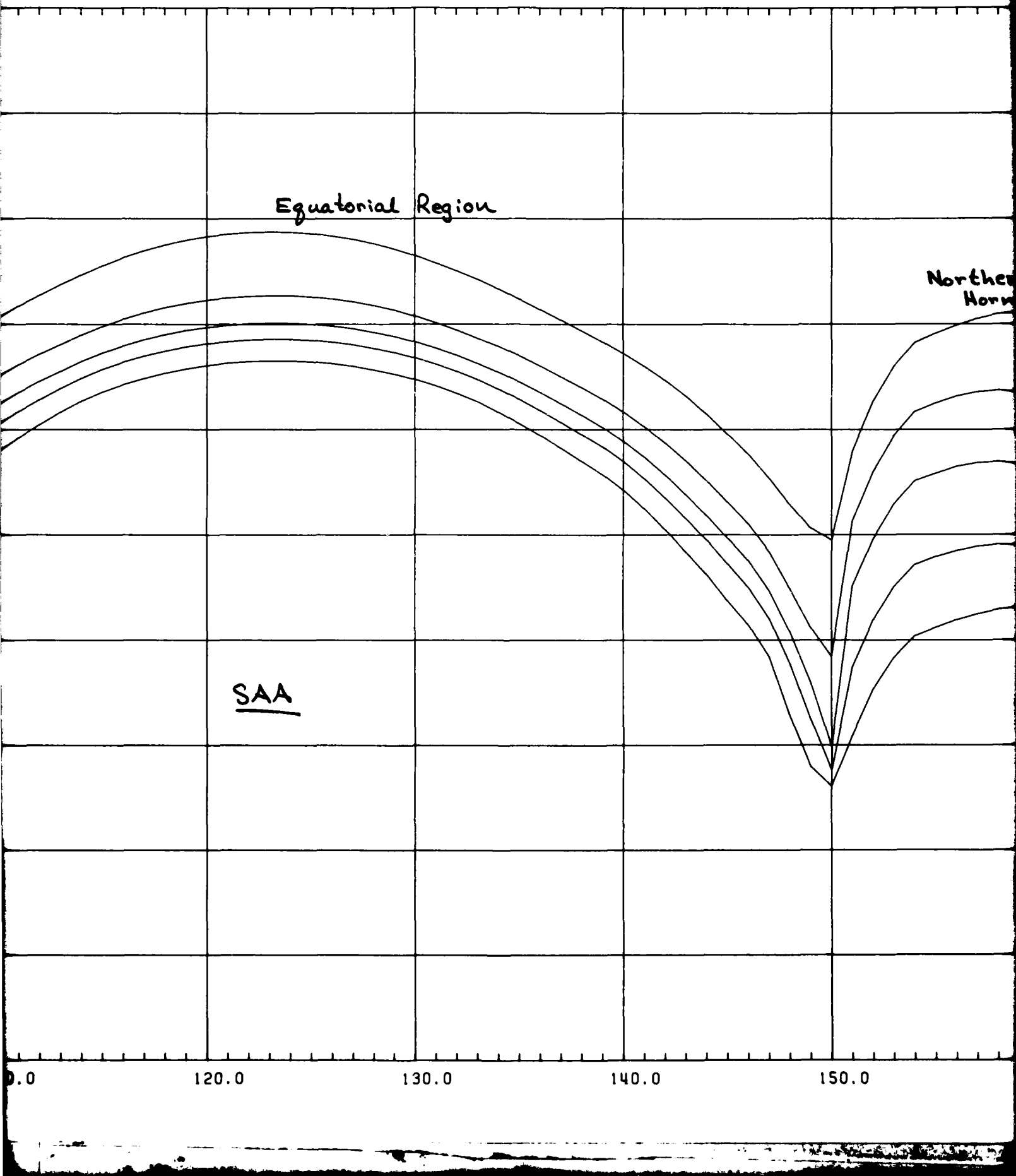


Figure 133

ORBIT: NAVELEX 3
60 DGR/3889-3889 KM

EPOCH: 1989.5

MODELS:
FIELD: BARR/75
TRAPPED PROTONS: AP8
INNER ZN ELEC: AE6
OUTER ZN ELEC: AE17-L0
MISSION DURATION: 60.00 MO
EVALUATION PHASE: SOLAR MAX

UN FACTORS: NOT APPLIED

Northern
Horn

STOP TIME ON TAPE = 9.316662

NASA-GSFC

0.0

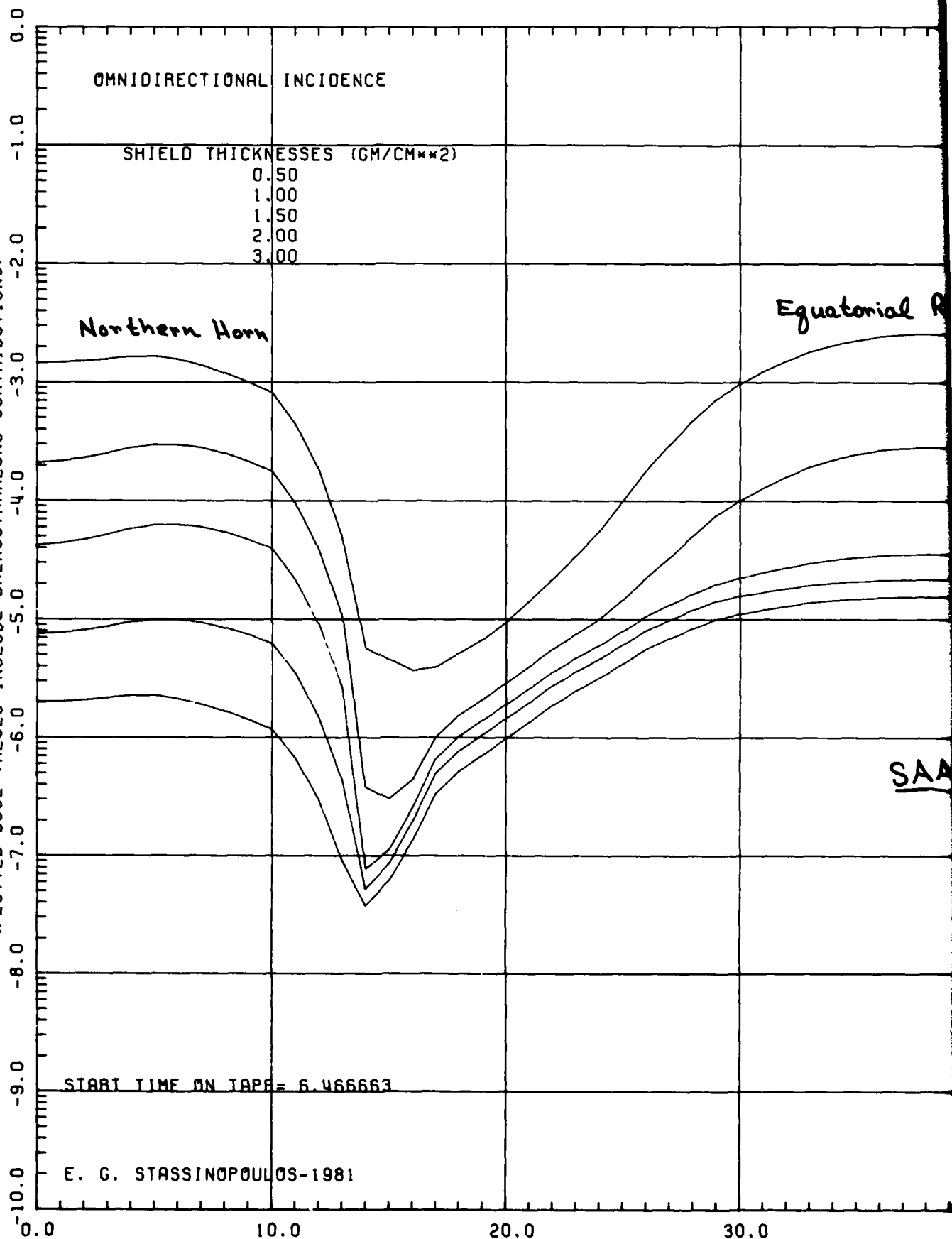
160.0

170.0

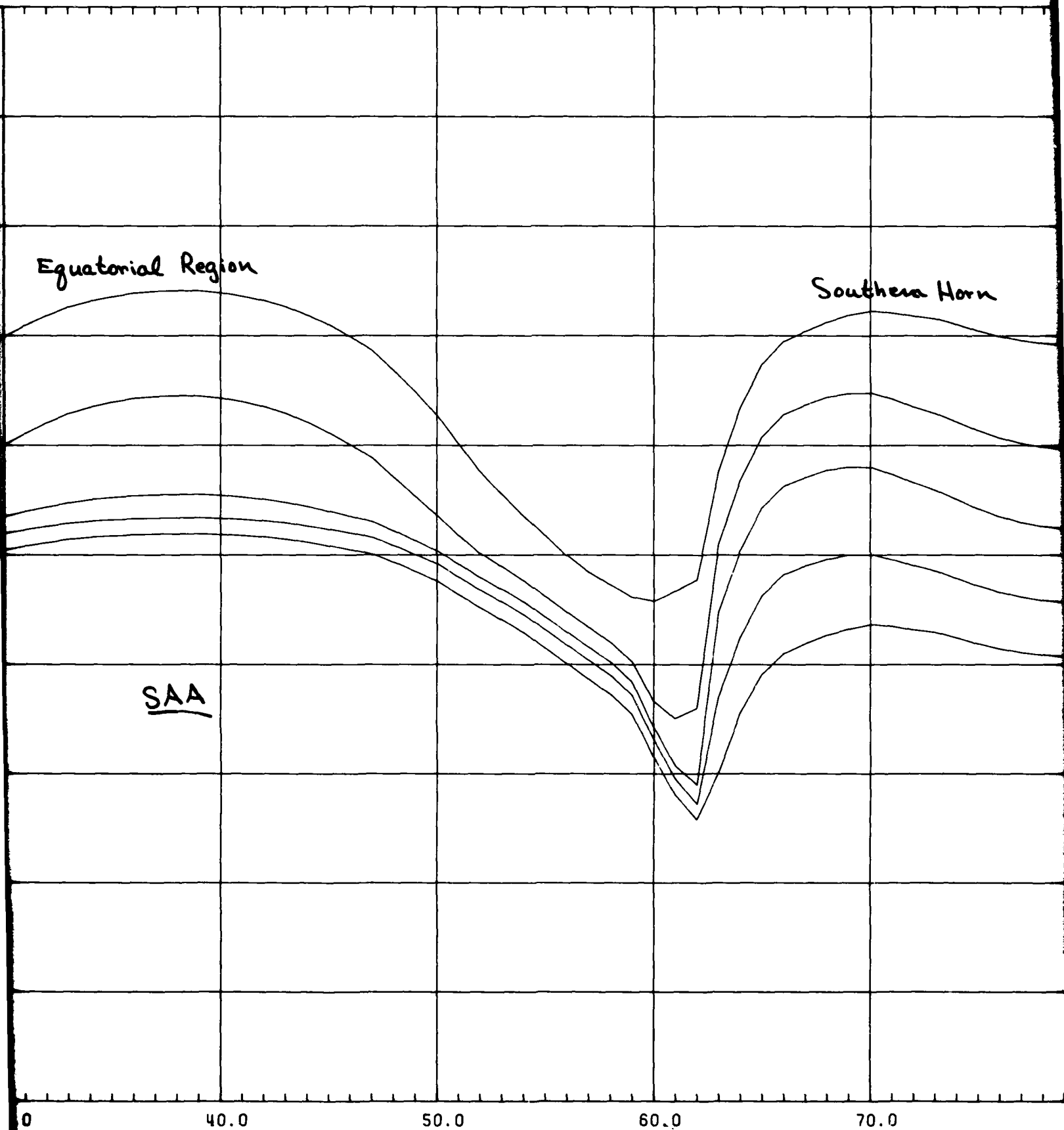
180.0

INSTANTANEOUS ALUMINUM ELECTRON DOSE (RADS)

(PLOTTED DOSE VALUES INCLUDE BREMSSTRAHLUNG CONTRIBUTIONS)

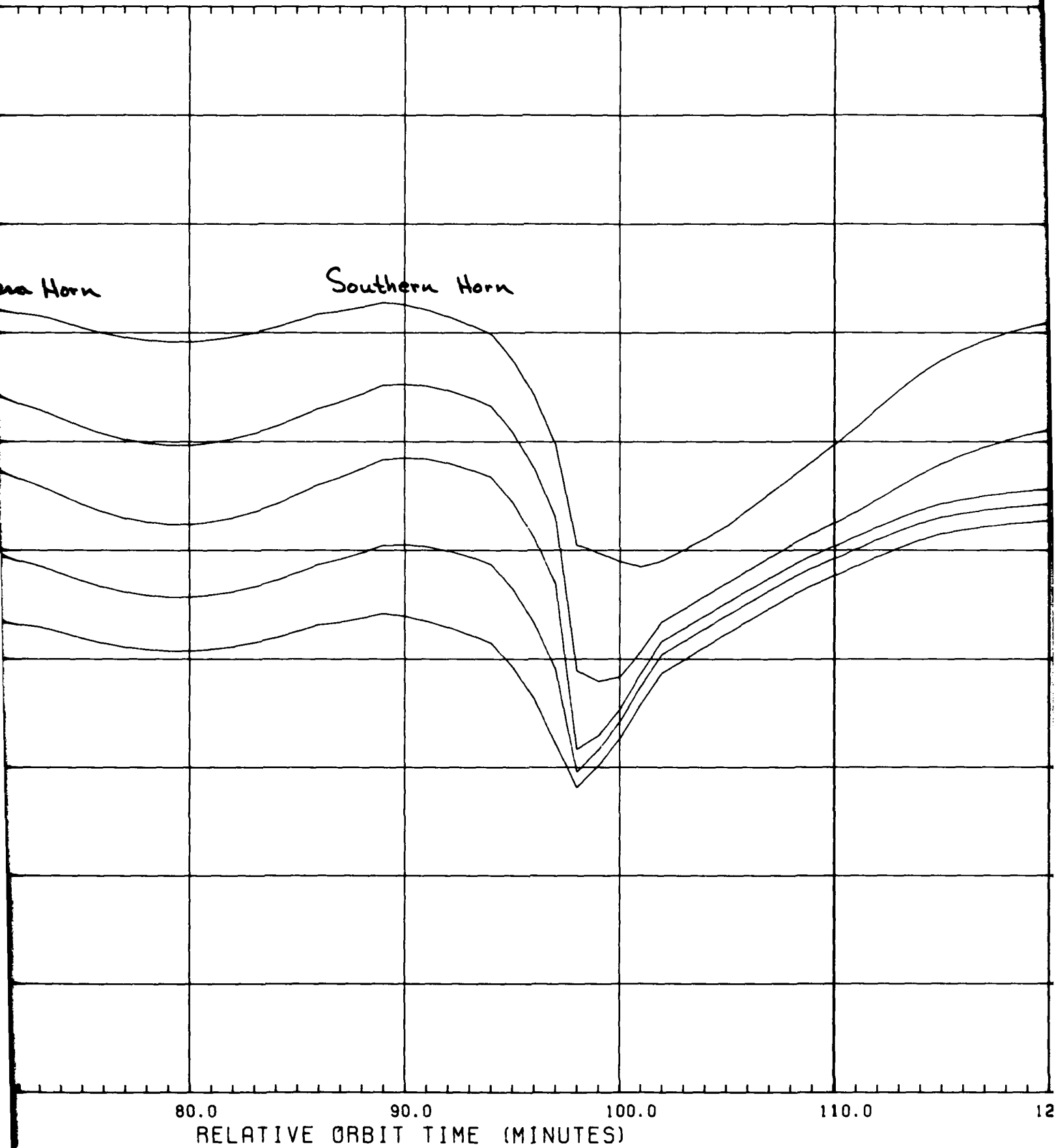


2



3

DOSE IN SEMI-INFINITE ALUMINUM MEDIUM



14

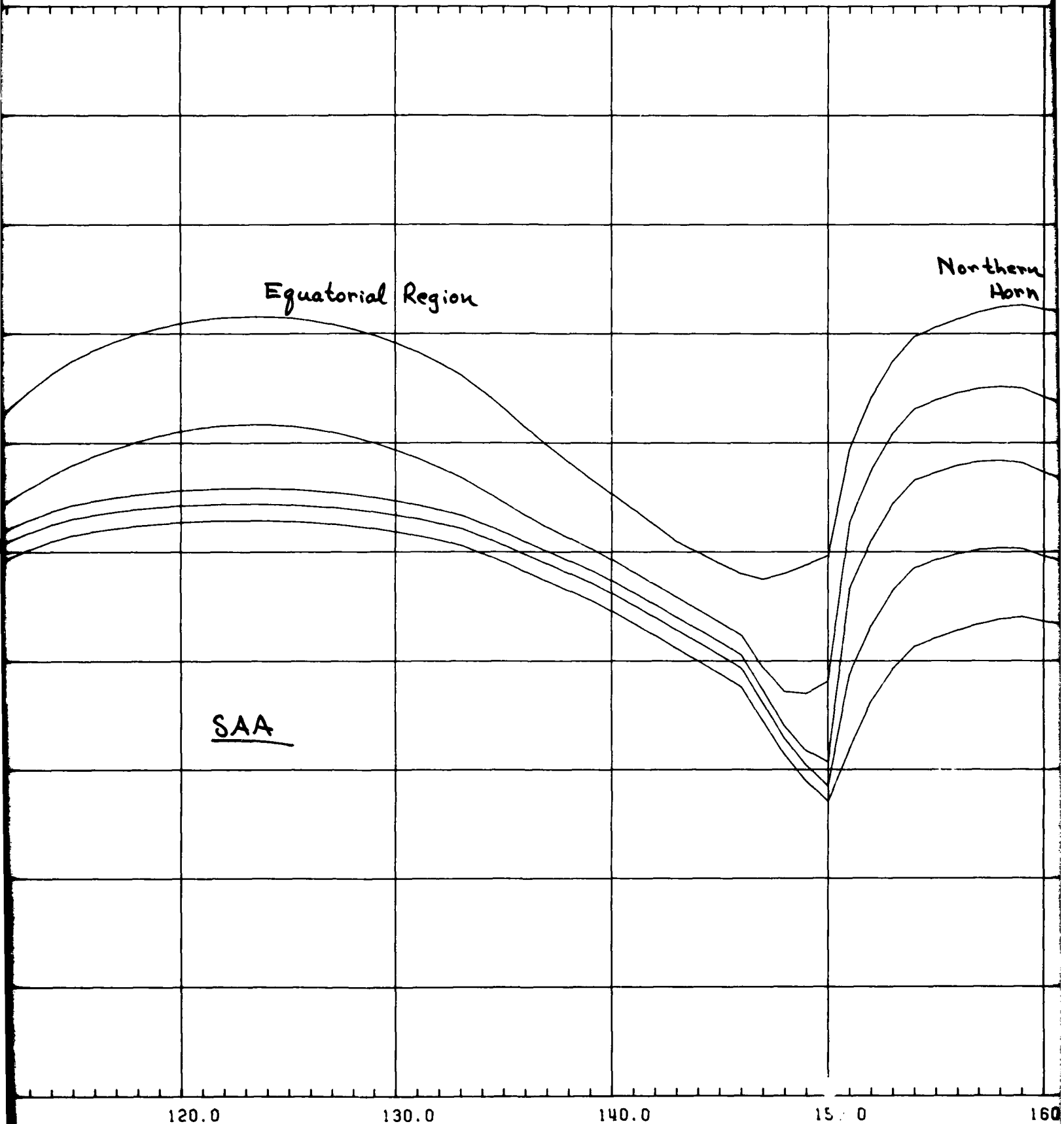


Figure 134

ORBIT: NAVELEX 3
60 DGR/3889-3889 KM

EPOCH: 1989.5

MODELS:

FIELD: BARR/75

TRAPPED PROTONS: AP8

INNER ZN ELEC: AE6

OUTER ZN ELEC: AE17 L0

MISSION DURATION: 60.00 MO

EVALUATION PHASE: SOLAR MAX

UN FACTORS: NOT APPLIED

Northern
Horn

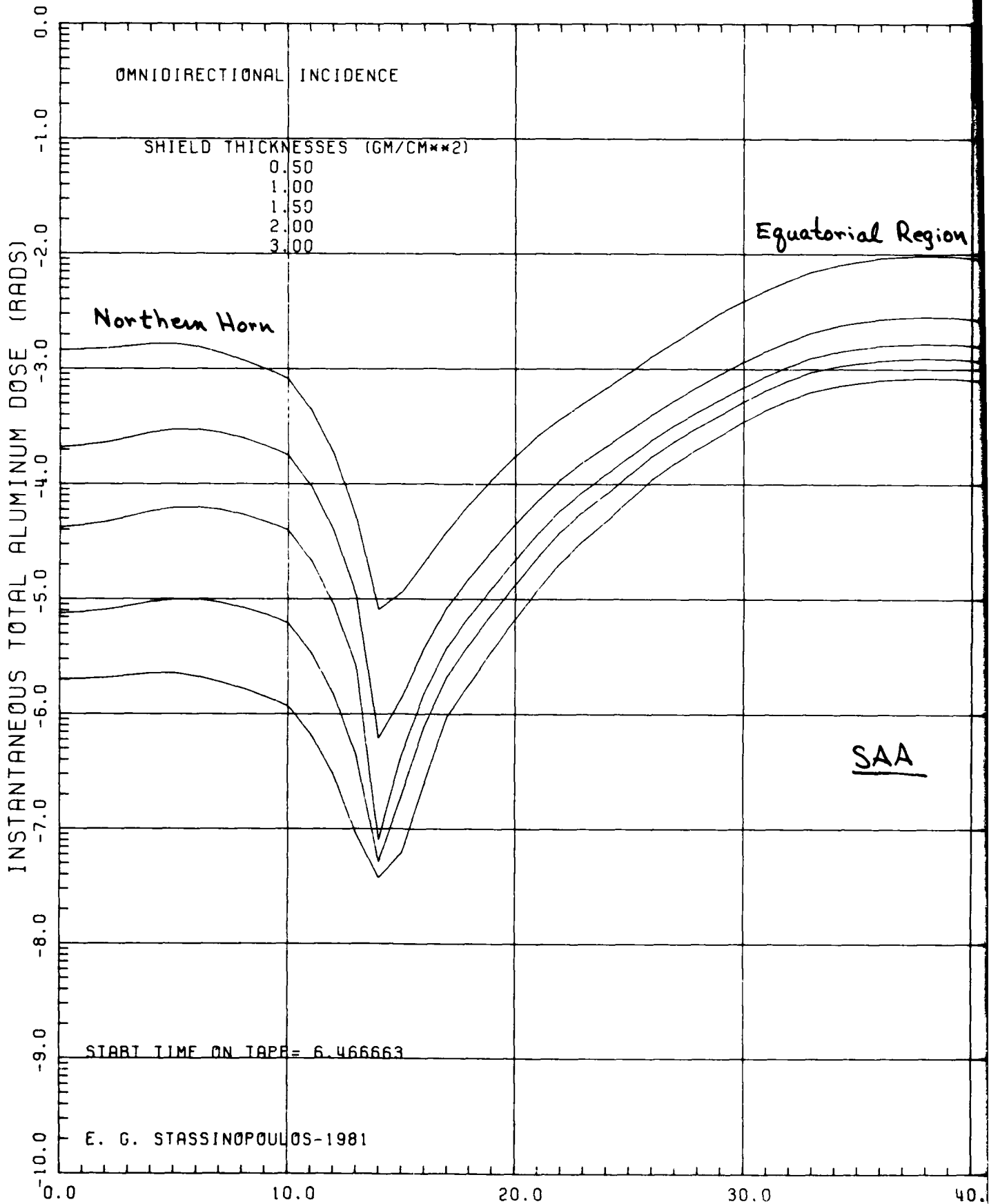
STOP TIME ON TAPE = 9.316662

NASA-GSFC

160.0

170.0

180.0



Equatorial Region

Southern Horn

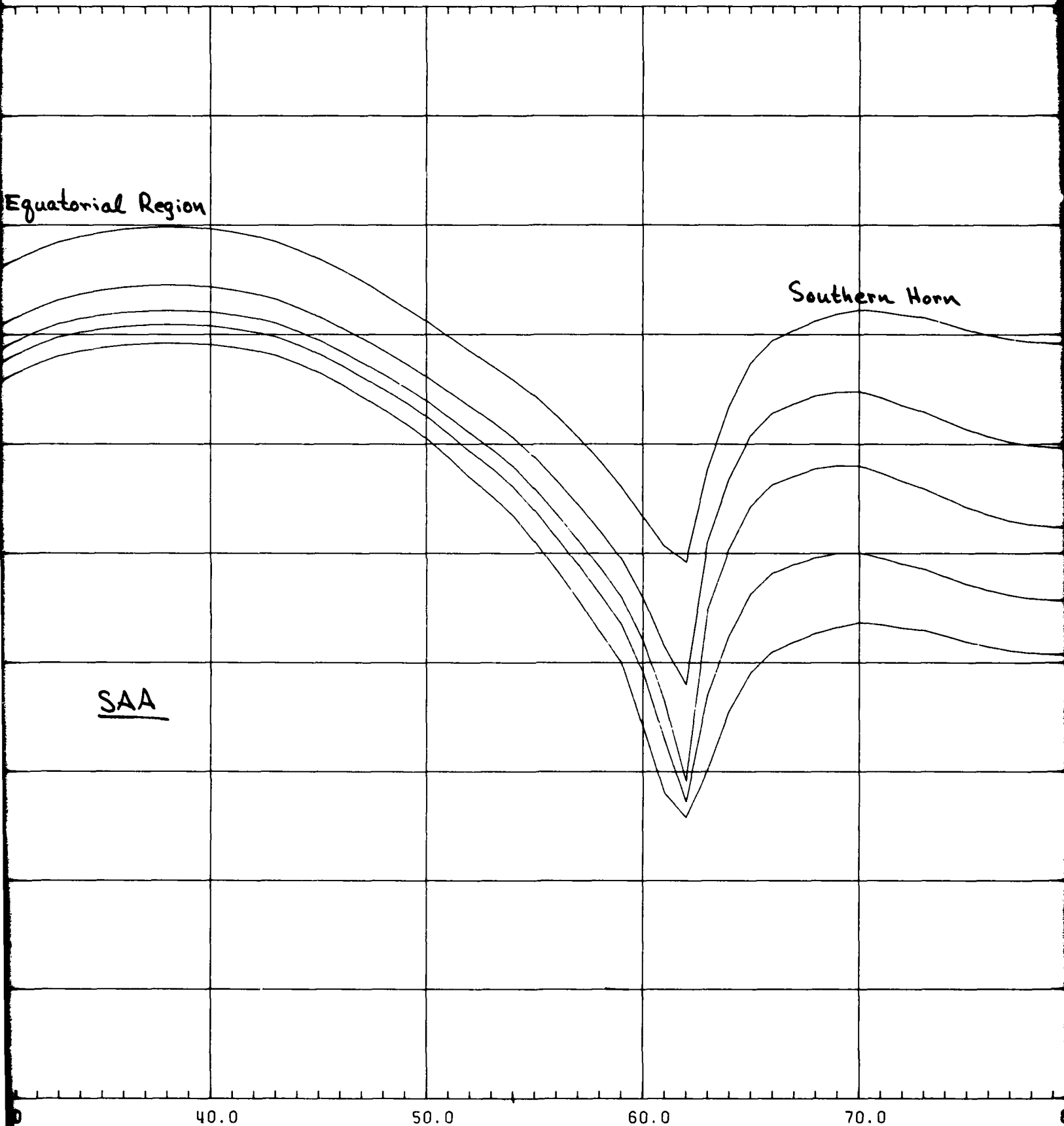
SAA

40.0

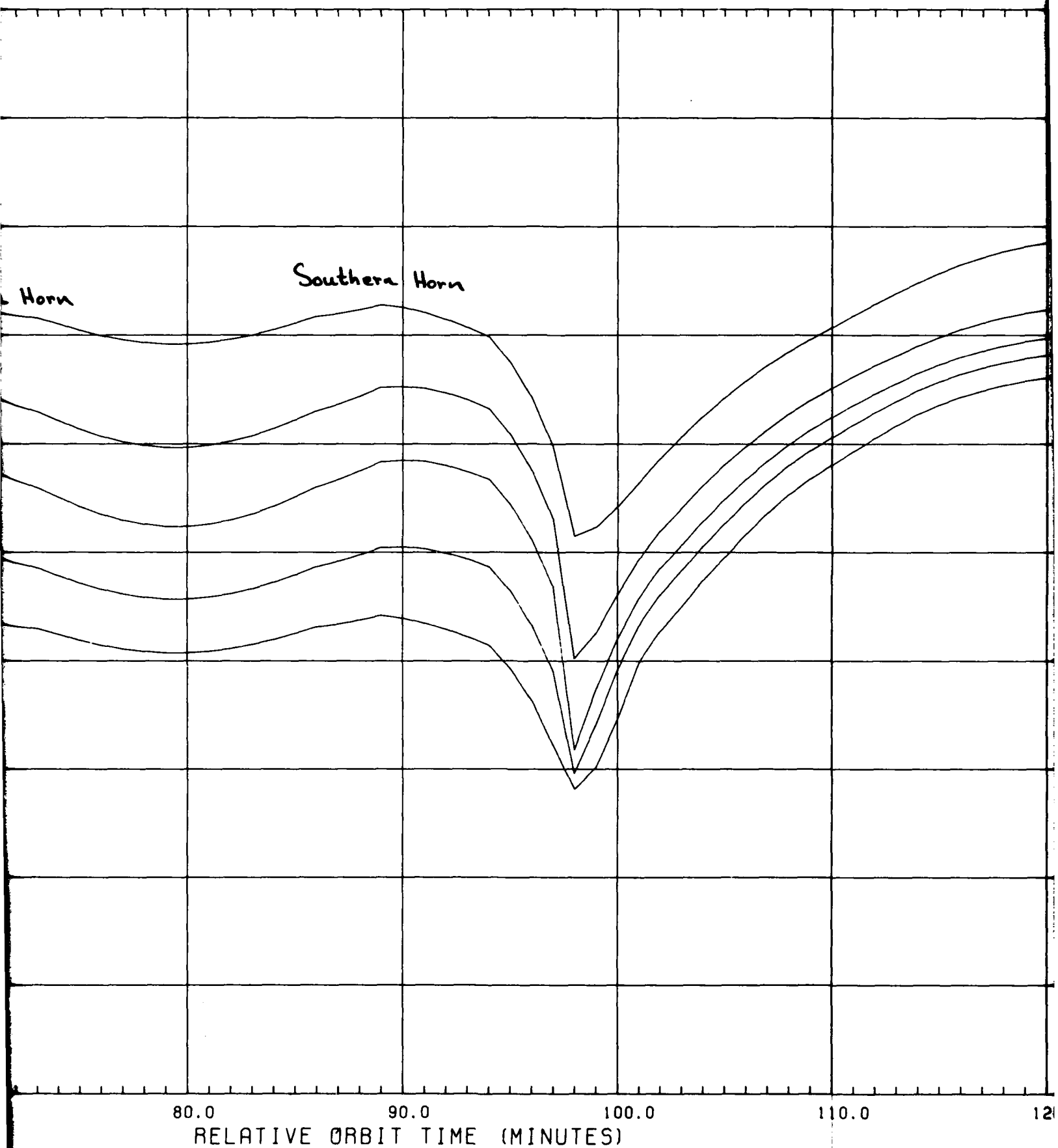
50.0

60.0

70.0



13
DOSE IN SEMI-INFINITE ALUMINUM MEDIUM



4

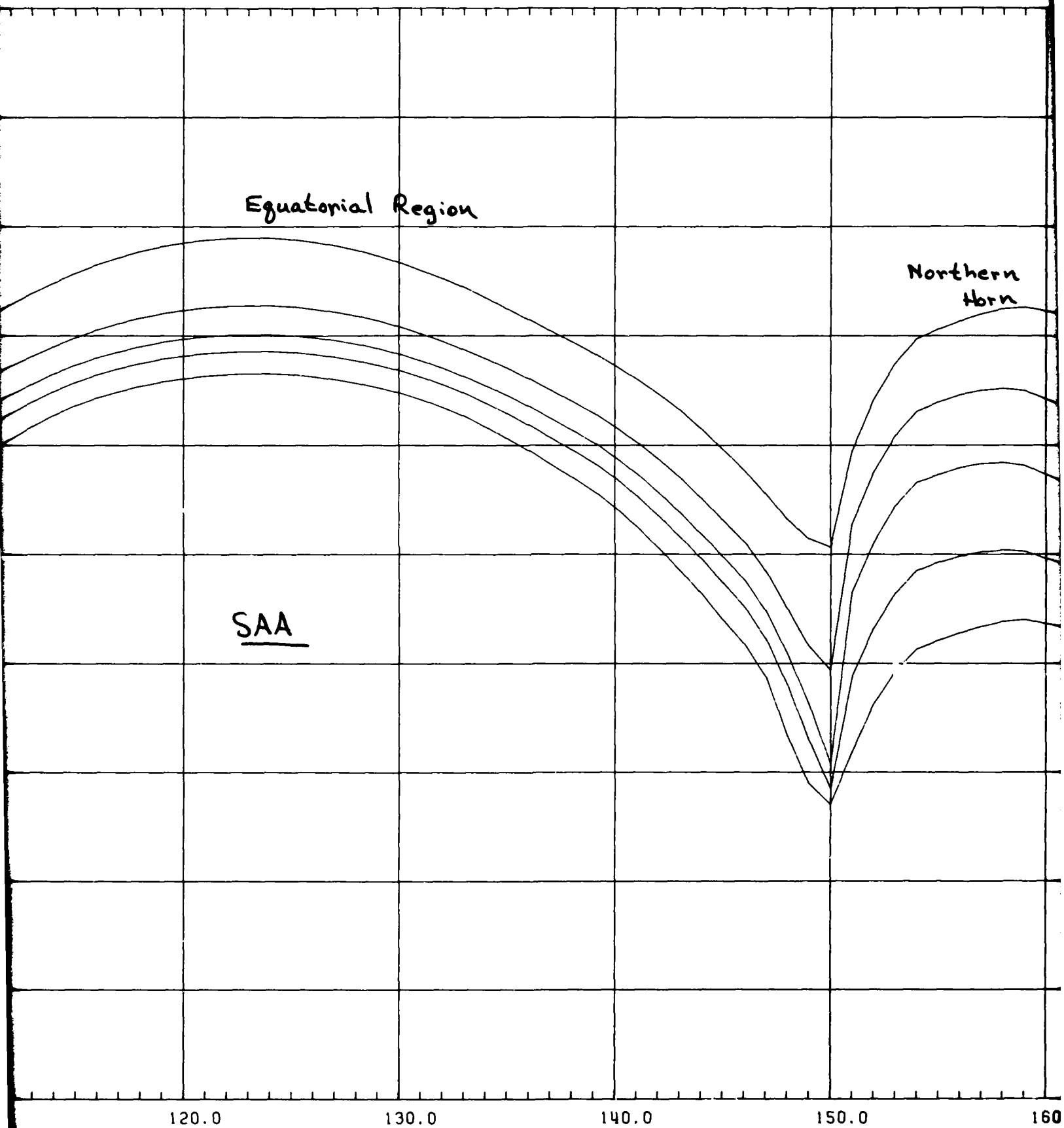


Figure 135

ORBIT: NAVELEX 8
60 DGR/3889-3889 KM

EPOCH: 1989.5

MODELS:

FIELD: BARR/75

TRAPPED PROTONS: AP8

INNER ZN ELEC: AE6

OUTER ZN ELEC: AE17 L0

MISSION DURATION: 60.00 MO

EVALUATION PHASE: SOLAR MAX

UN FACTORS: NOT APPLIED

Northern
Horn

STOP TIME ON TAPE= 9.316662

NASA-GSFC

150.0

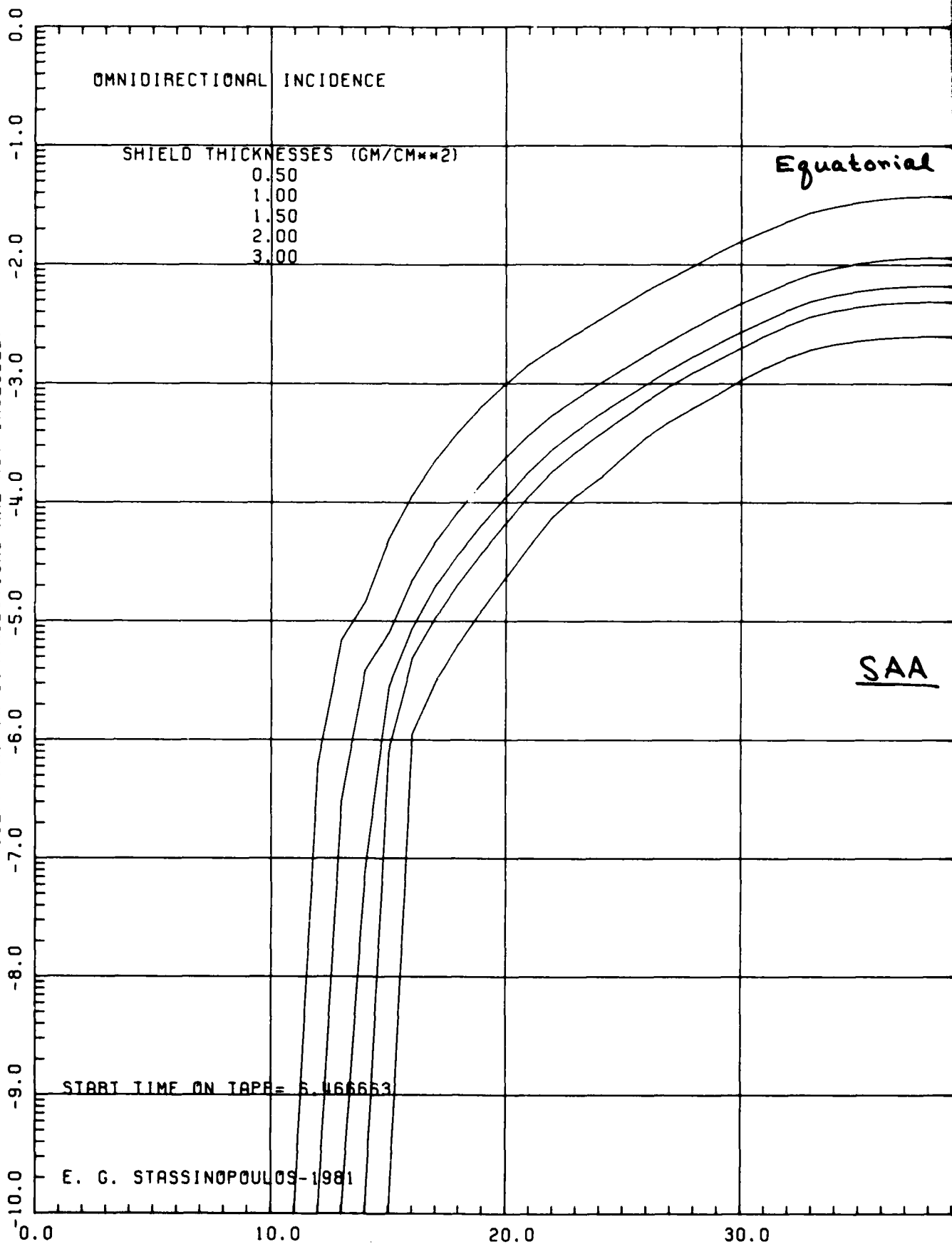
160.0

170.0

180.0

INSTANTANEOUS ALUMINUM PROTON DOSE (RADS)

(SOLAR PROTON CONTRIBUTIONS ARE NOT INCLUDED)



20

Equatorial Region

SAA

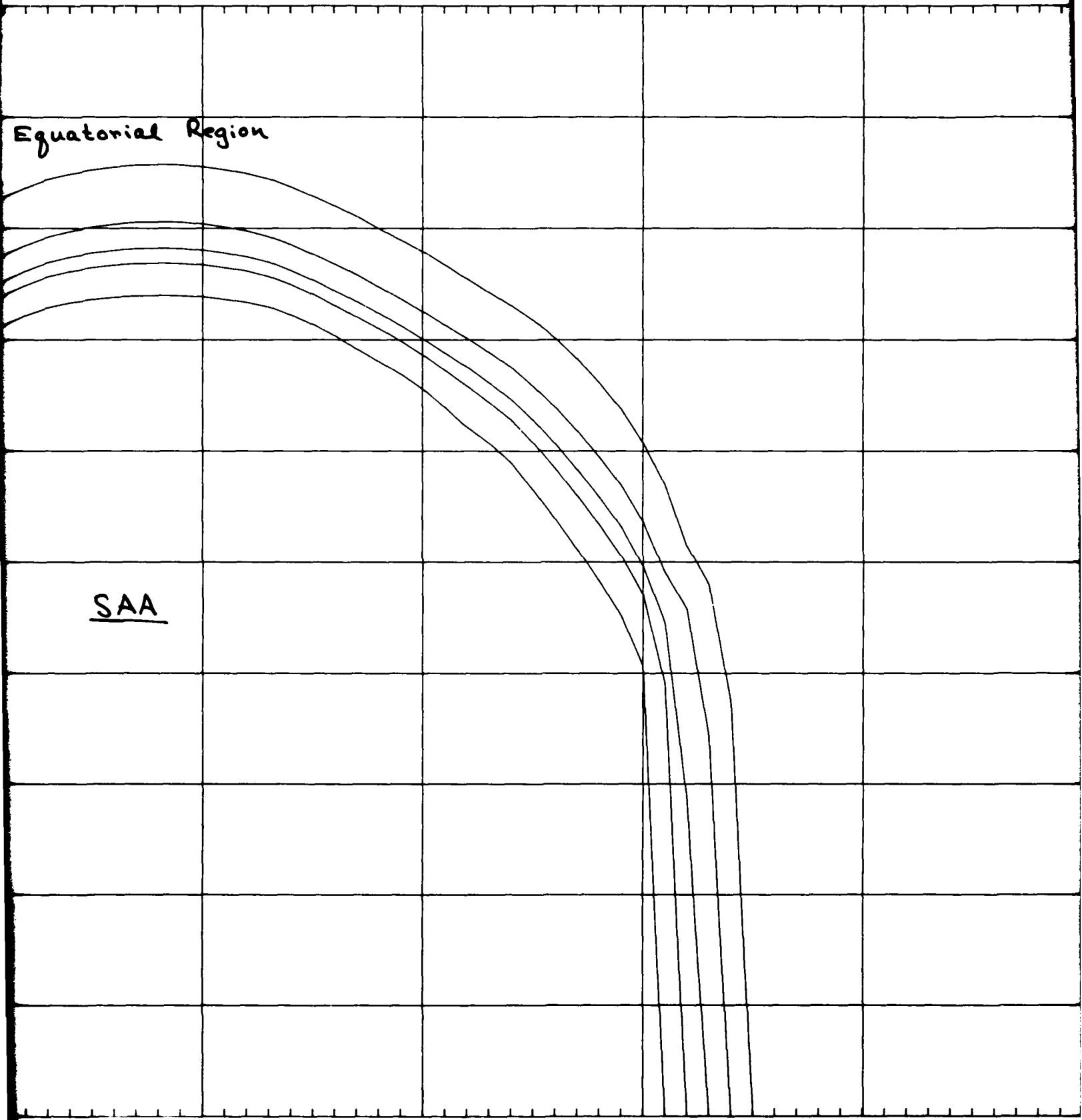
40.0

50.0

60.0

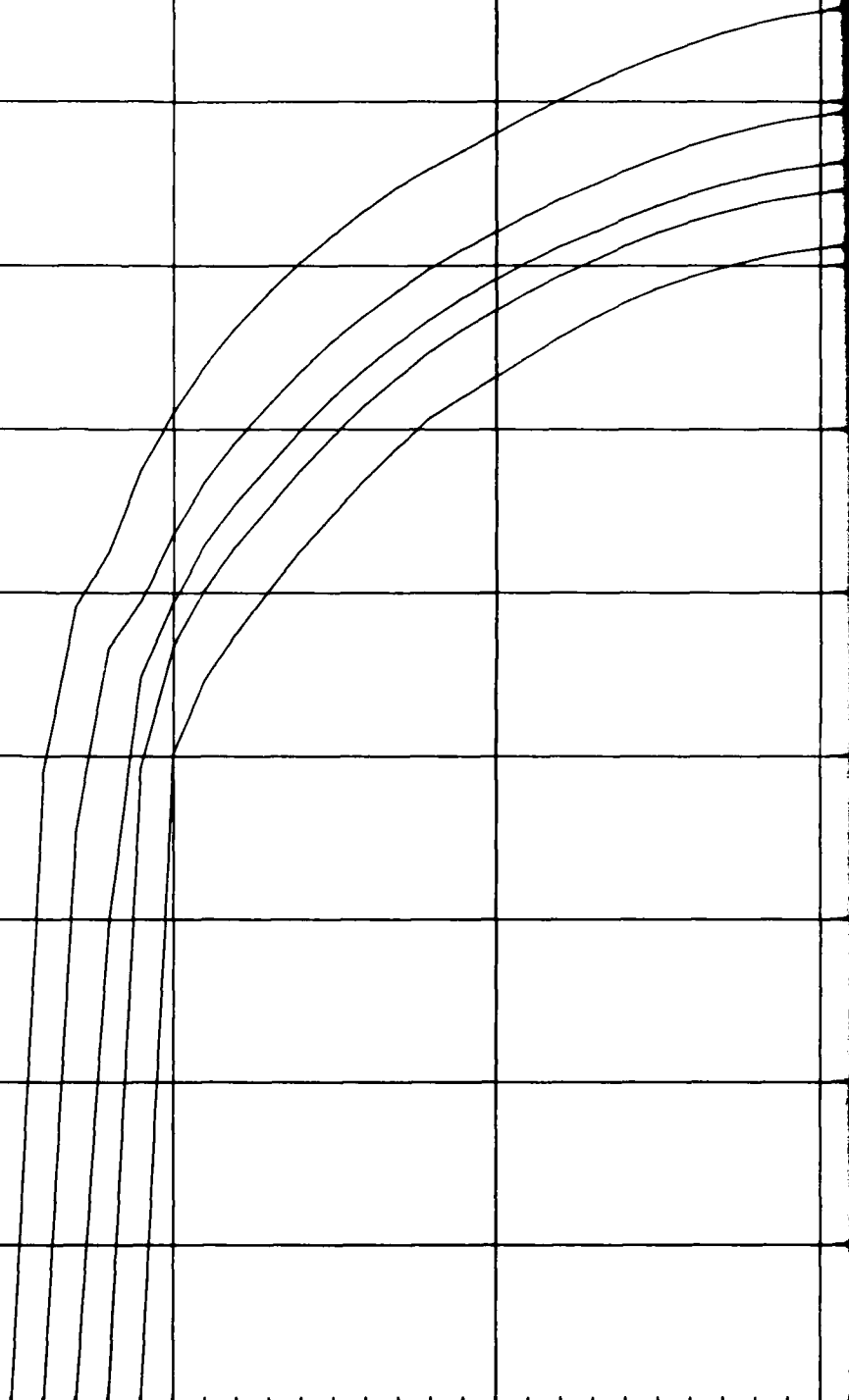
70.0

80



3
DOSE AT CENTER OF ALUMINUM SPHERES

80.0 90.0 100.0 110.0 120
RELATIVE ORBIT TIME (MINUTES)



41

Equatorial Region

SAA

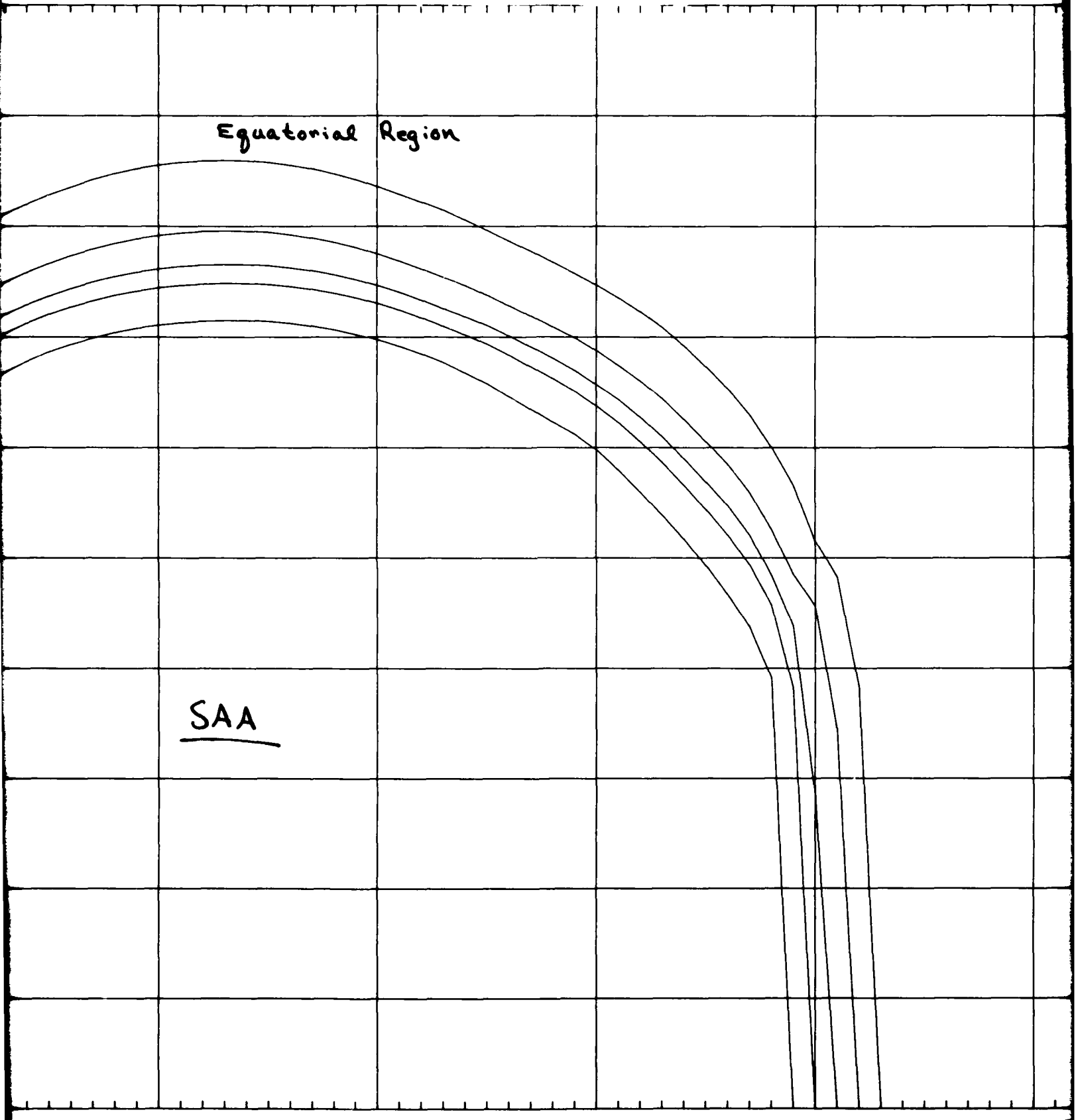
120.0

130.0

140.0

150.0

160.0



5

Figure 136

ORBIT: NAVELEX 3
60 DGR/3889-3889 KM

EPOCH: 1989.5

MODELS:

FIELD: BARR/75

TRAPPED PROTONS: AP8

INNER ZN ELEC: AE6

OUTER ZN ELEC: AE17-L0

MISSION DURATION: 60.00 MO

EVALUATION PHASE: SOLAR MAX

UN FACTORS: NOT APPLIED

STOP TIME ON TAPE= 9.316662

NASA-GSFC

0.0

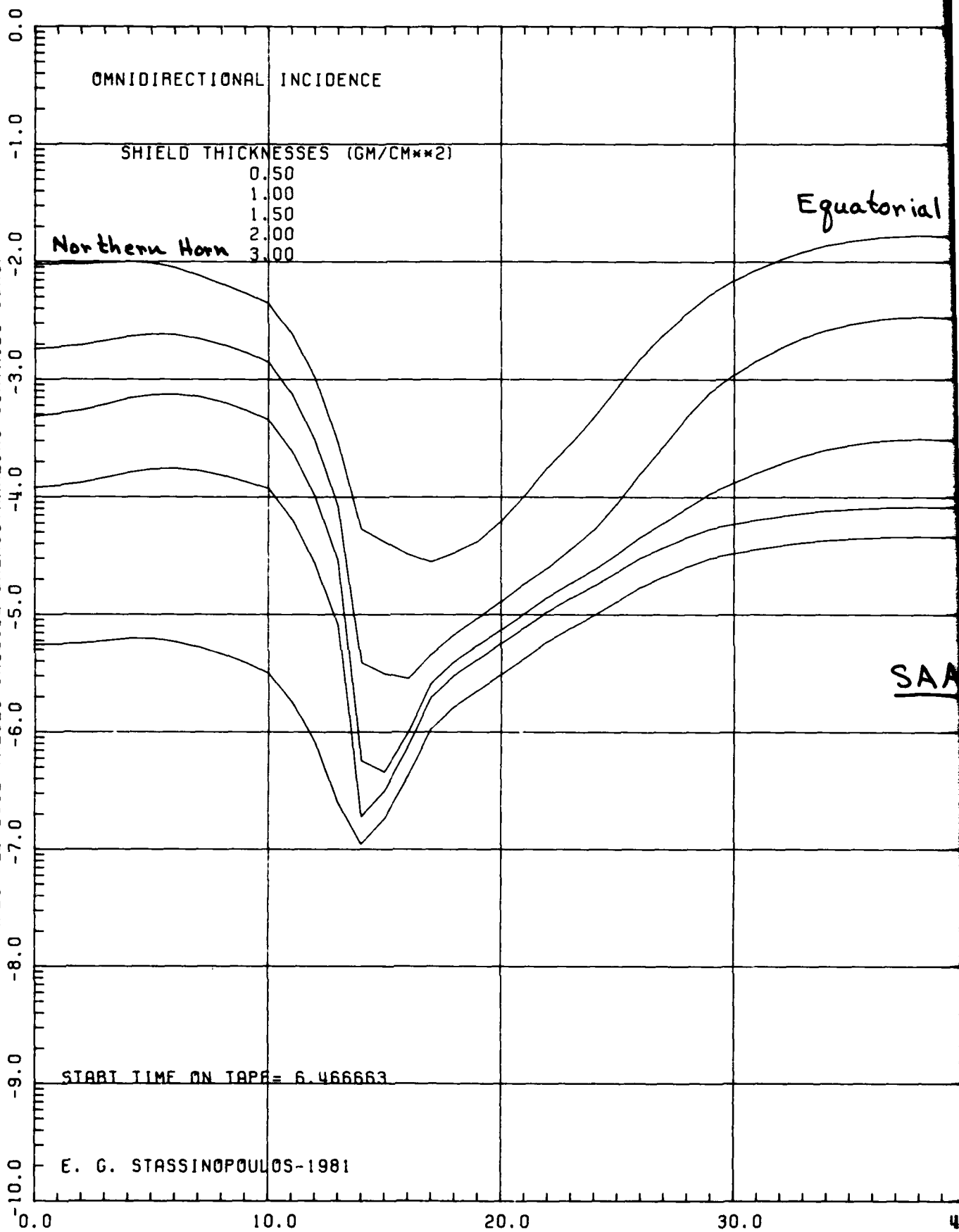
160.0

170.0

180.0

INSTANTANEOUS ALUMINUM ELECTRON DOSE (RADS)

(PLOTTED DOSE VALUES INCLUDE BREMSSTRAHLUNG CONTRIBUTIONS)



Equatorial Region

Southern Horn

SAA

40.0

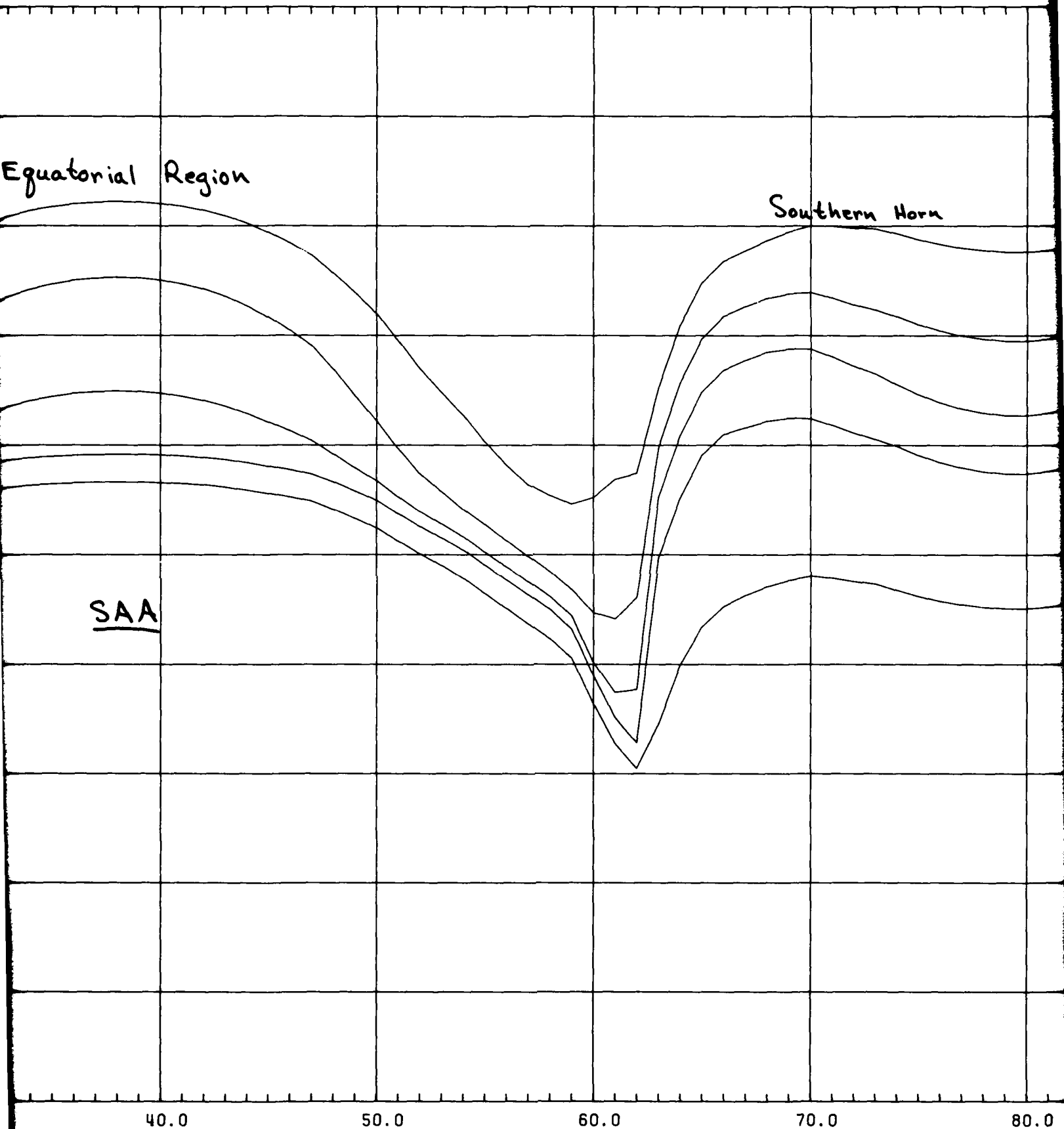
50.0

60.0

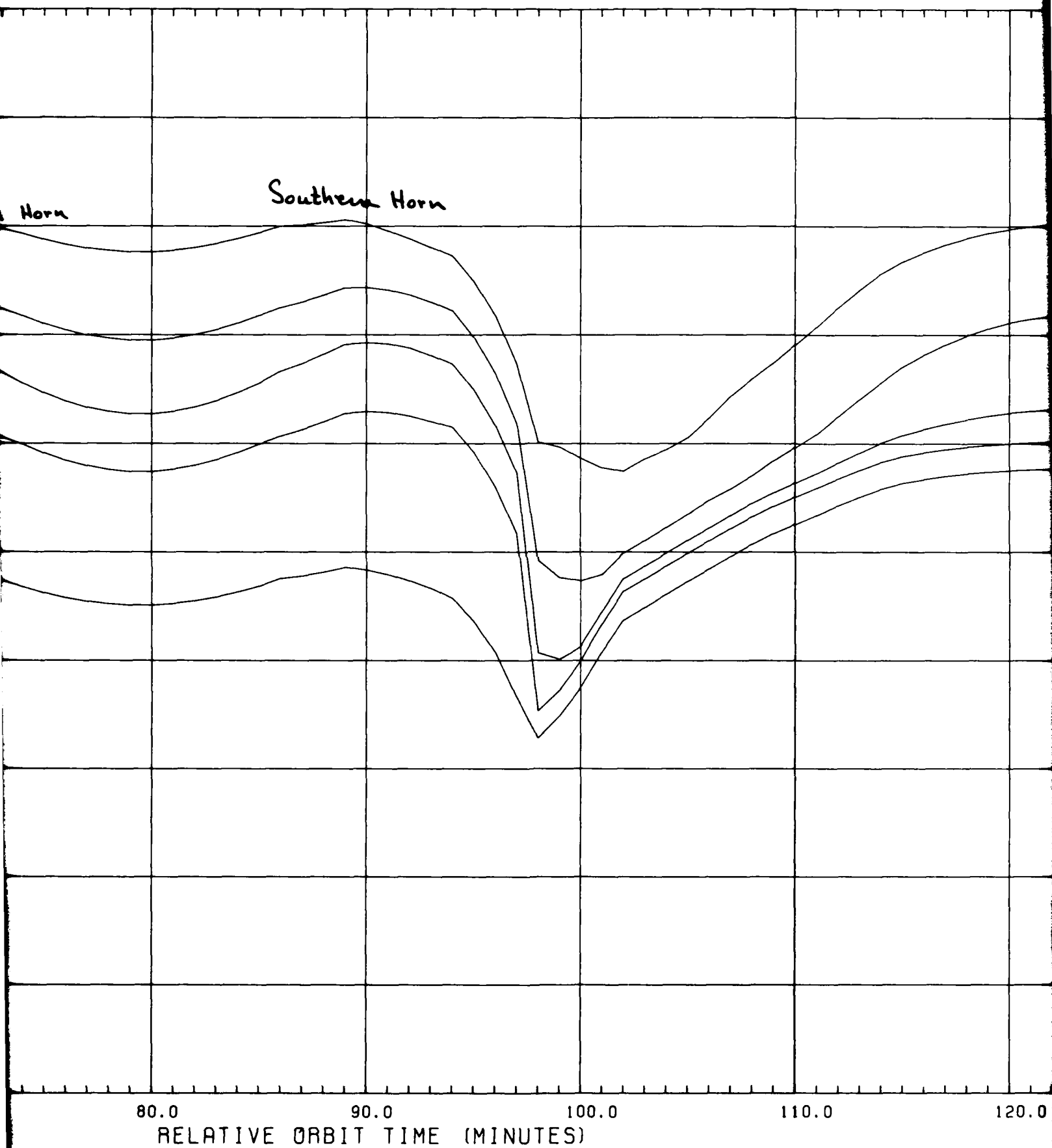
70.0

80.0

REI



3
DOSE AT CENTER OF ALUMINUM SPHERES



Equatorial Region

Northern
Horn

SAA

120.0

130.0

140.0

150.0

160.0

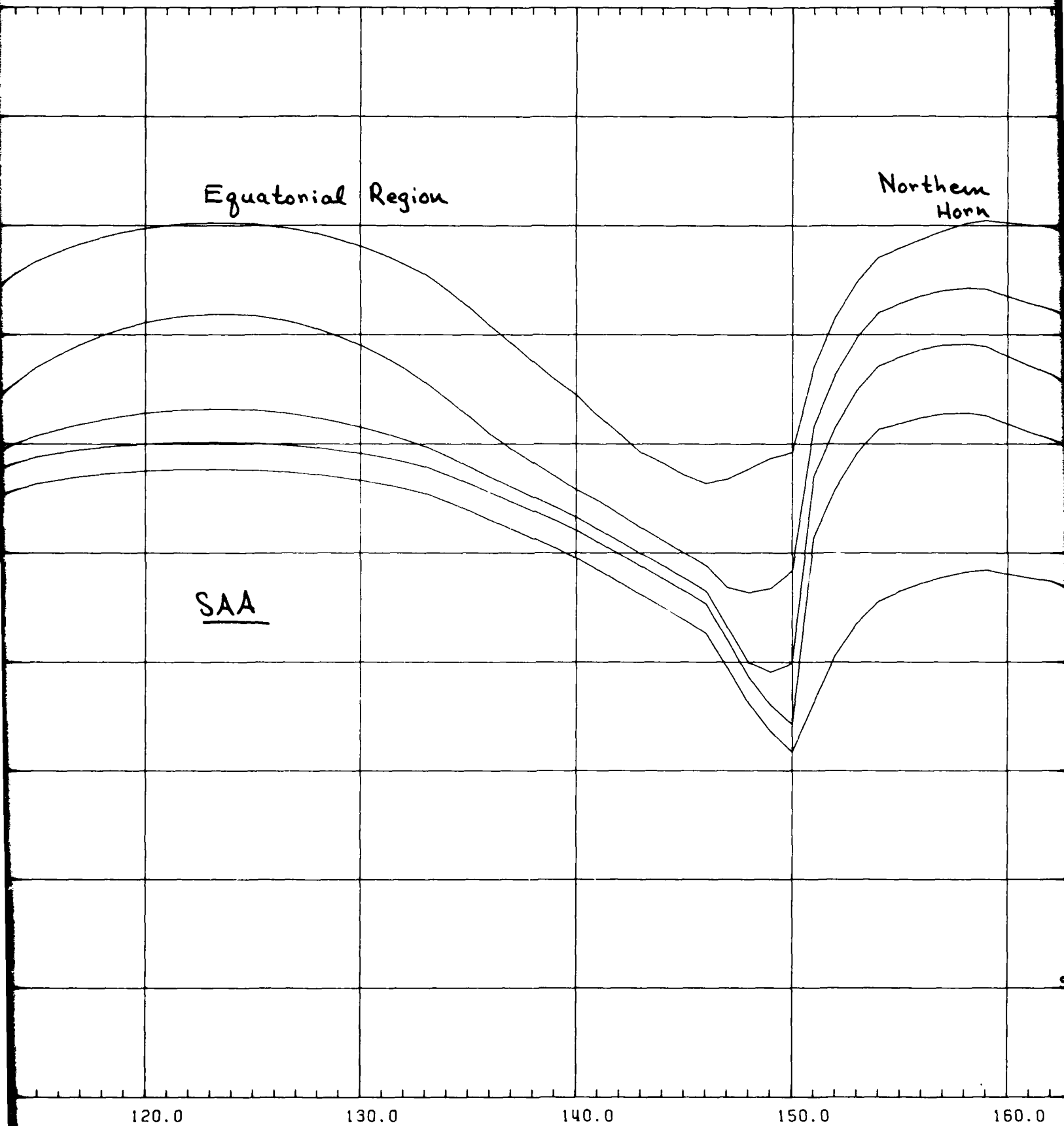


Figure 137

ORBIT: NAVELEX 3
60 DGR/3889-3889 KM

EPOCH: 1989.5

MODELS:
FIELD: BARR/75
TRAPPED PROTONS: AP8
INNER ZN ELEC: AE6
OUTER ZN ELEC: AE17 L0
MISSION DURATION: 60.00 MO
EVALUATION PHASE: SOLAR MAX

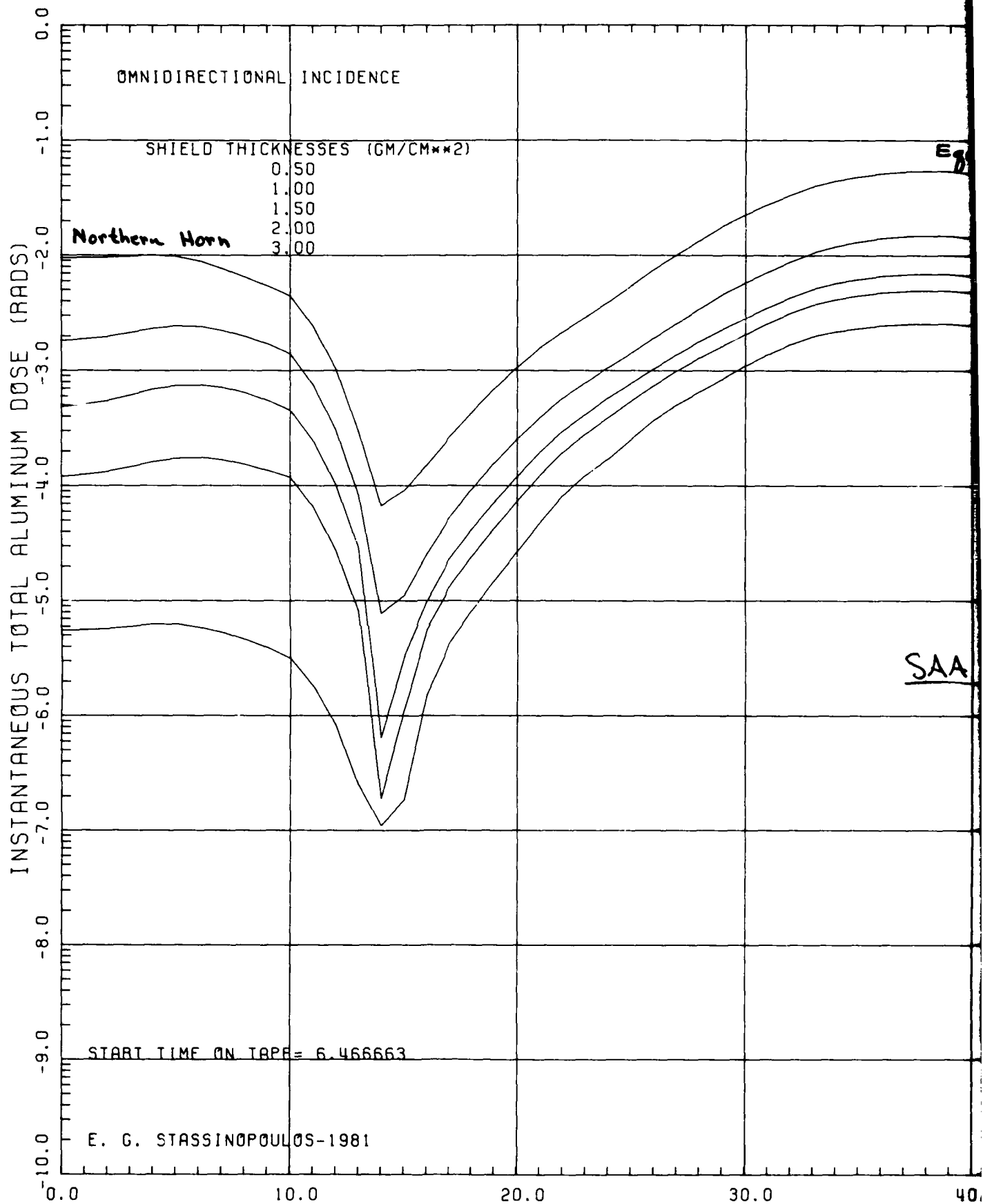
UN FACTORS: NOT APPLIED

Northern
Horn

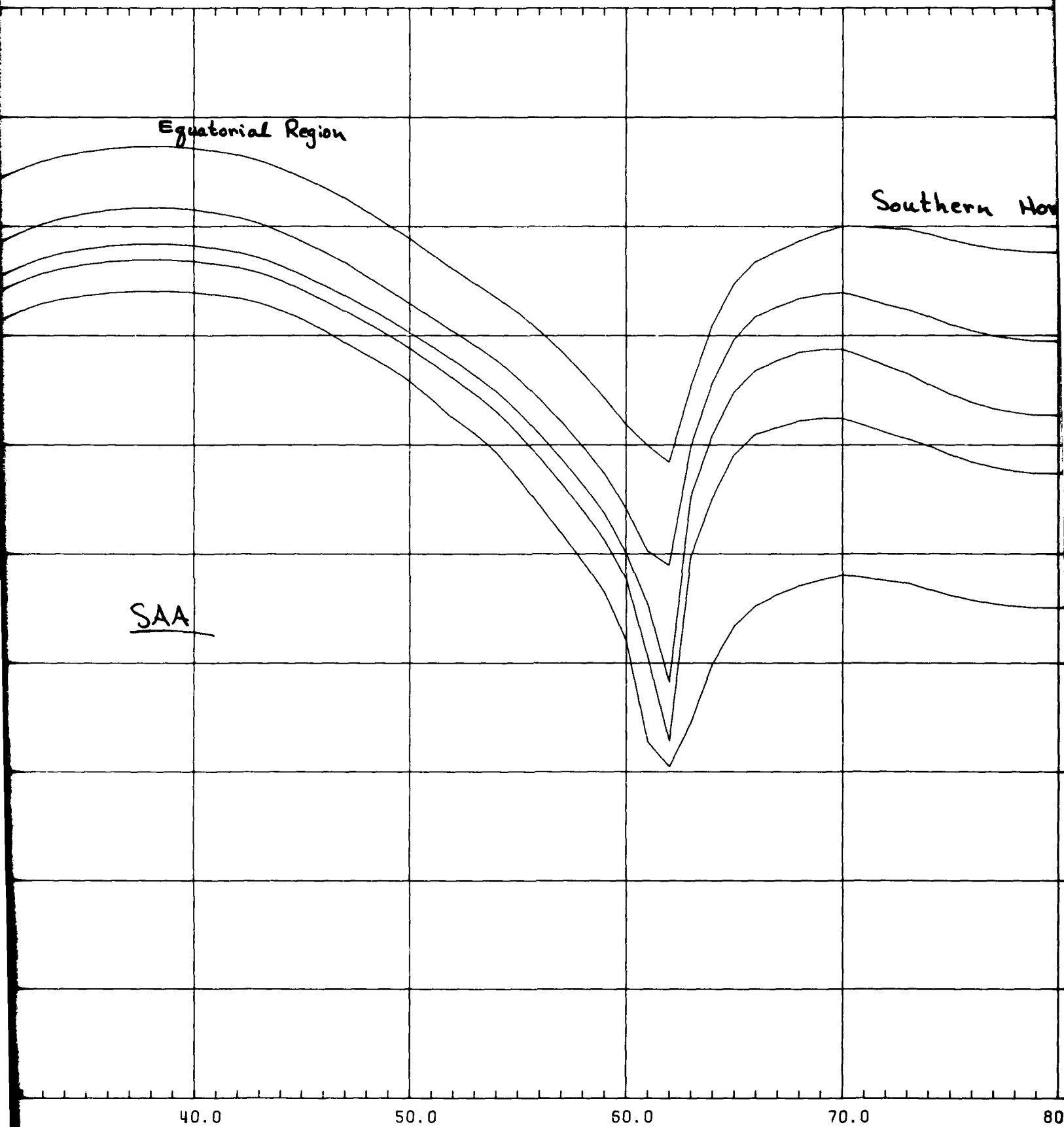
STOP TIME ON TAPE = 9.316662

NASA-GSFC

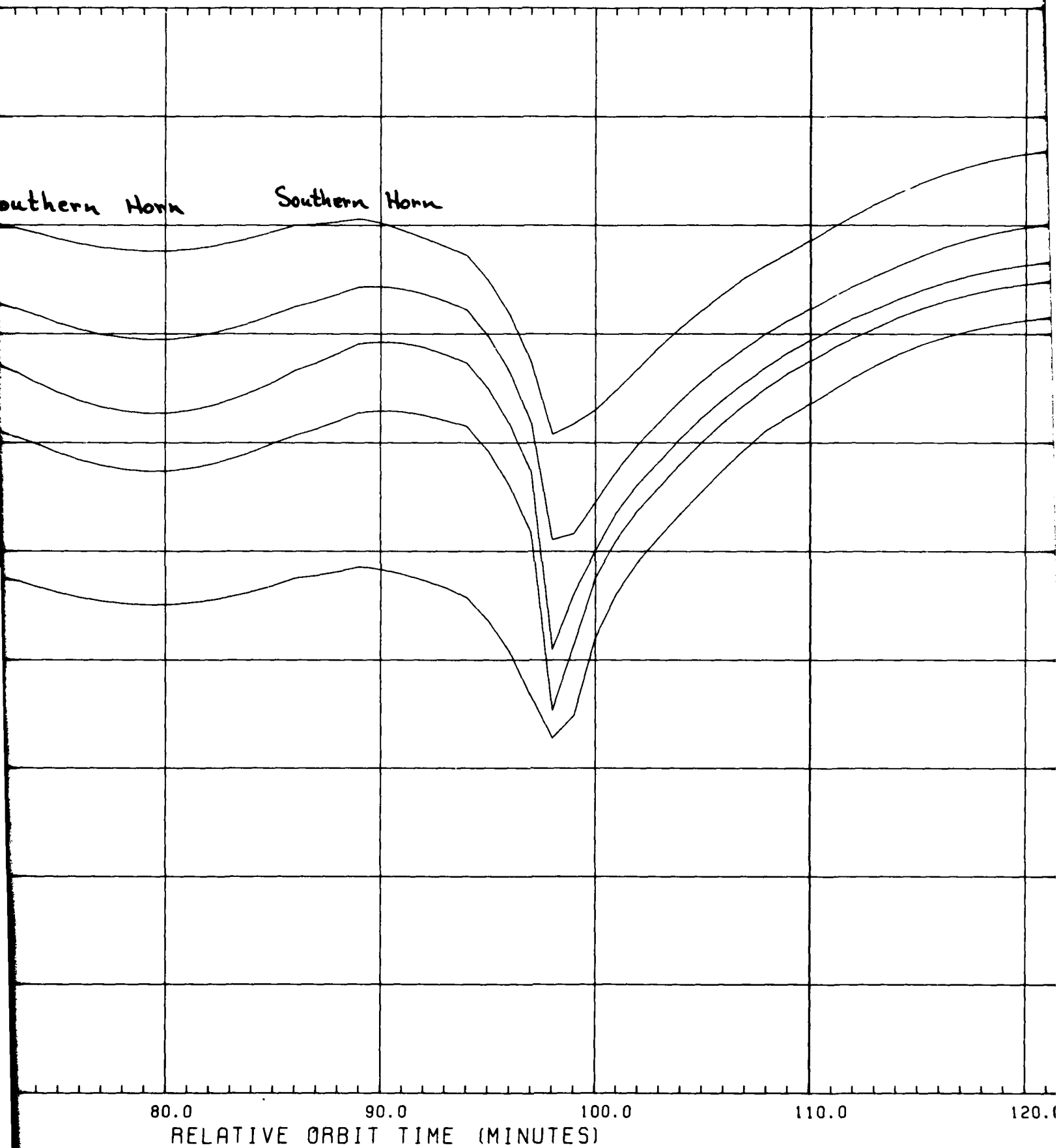
0 160.0 170.0 180.0



21



12
DOSE AT CENTER OF ALUMINUM SPHERES



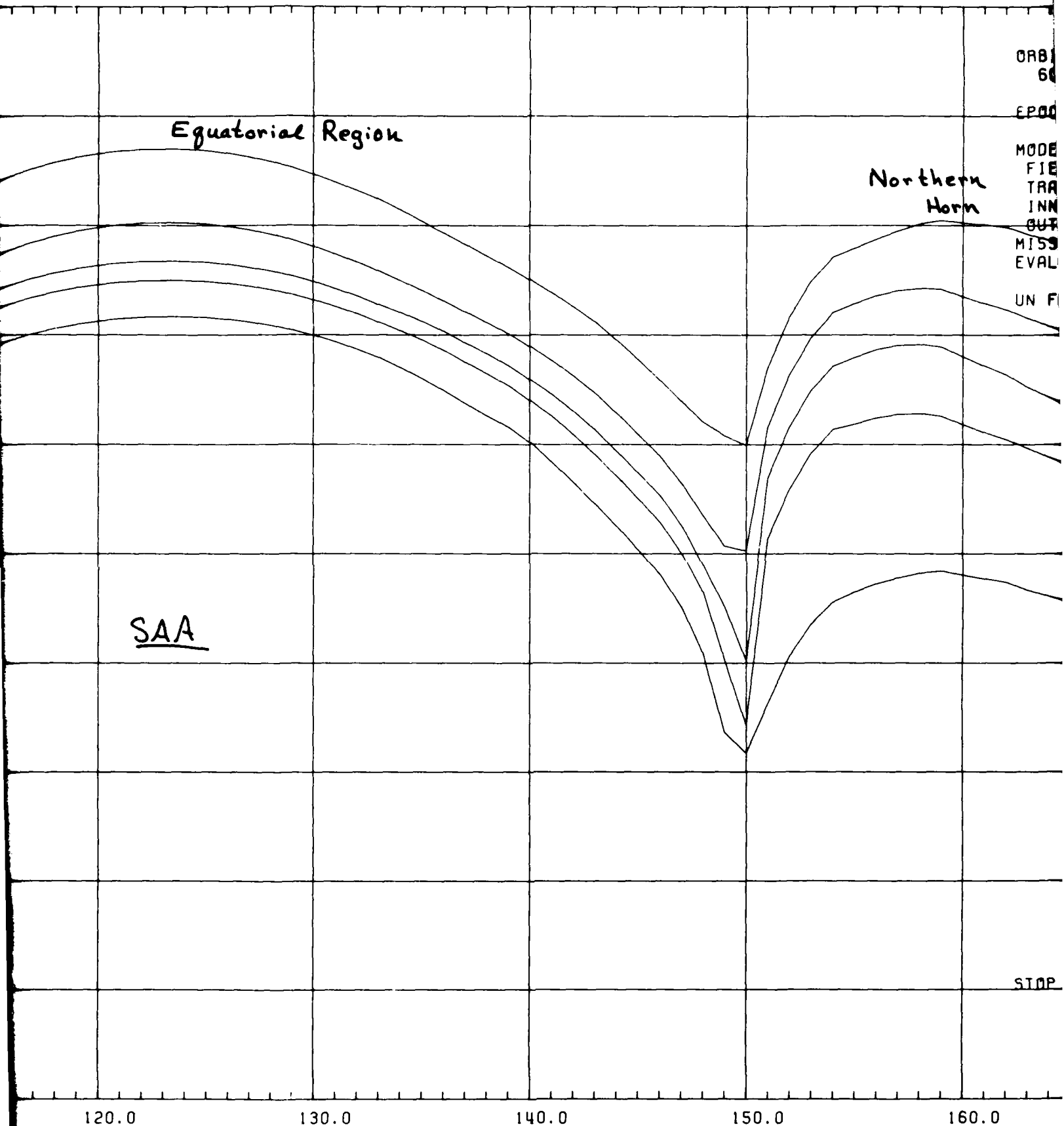


Figure 138

ORBIT: NAVELEX 3
60 DGR/3889-3889 KM

EPOCH: 1989.5

MODELS:
FIELD: BARR/75
TRAPPED PROTONS: AP8
INNER ZN ELEC: AE6
OUTER ZN ELEC: AE17-L0
MISSION DURATION: 60.00 MO
EVALUATION PHASE: SOLAR MAX

UN FACTORS: NOT APPLIED

Northern
Horn

STOP TIME ON TAPE= 9.316662

NASA-GSFC

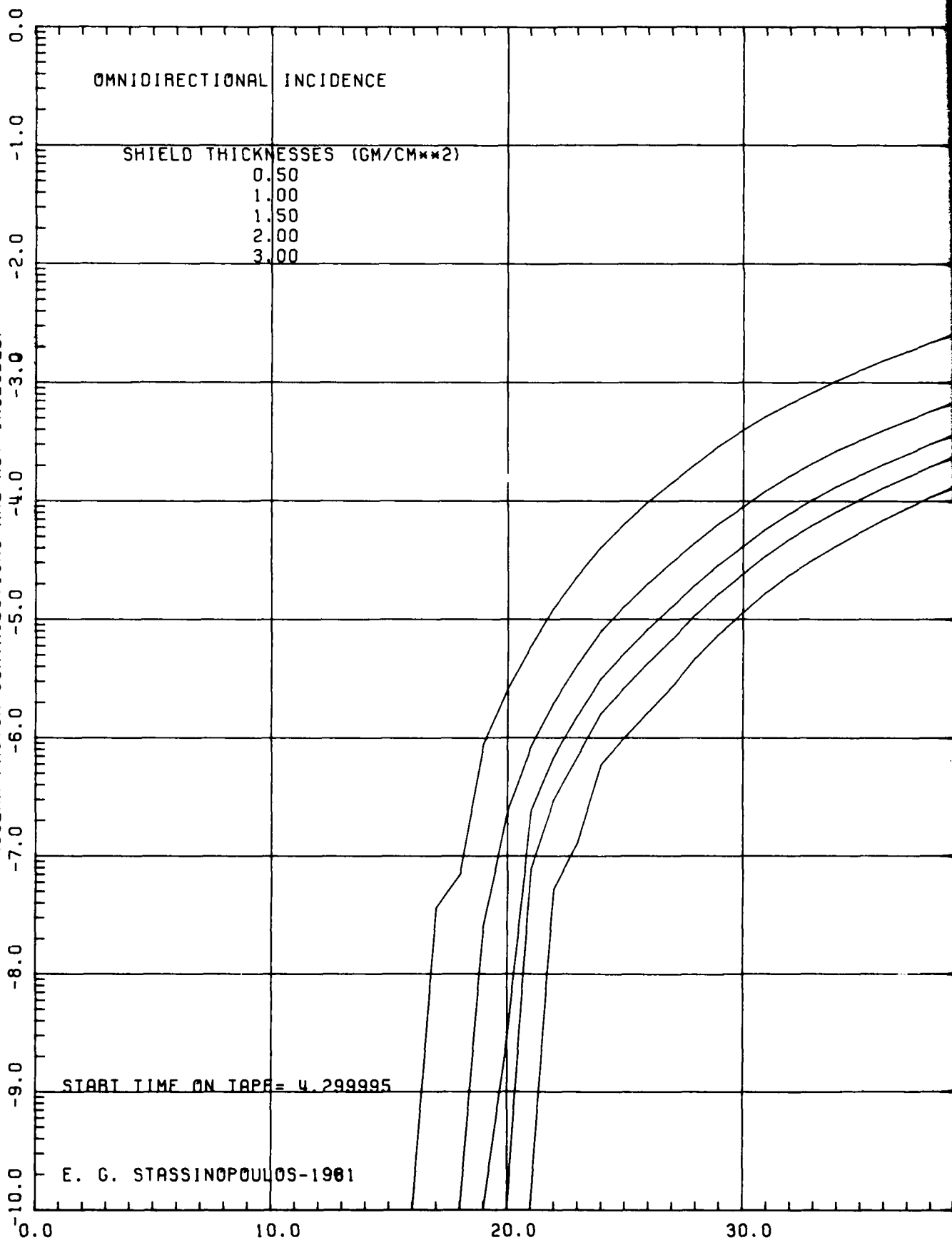
160.0

170.0

180.0

INSTANTANEOUS ALUMINUM PROTON DOSE (RADS)

(SOLAR PROTON CONTRIBUTIONS ARE NOT INCLUDED)



2

Equatorial Region

SAA

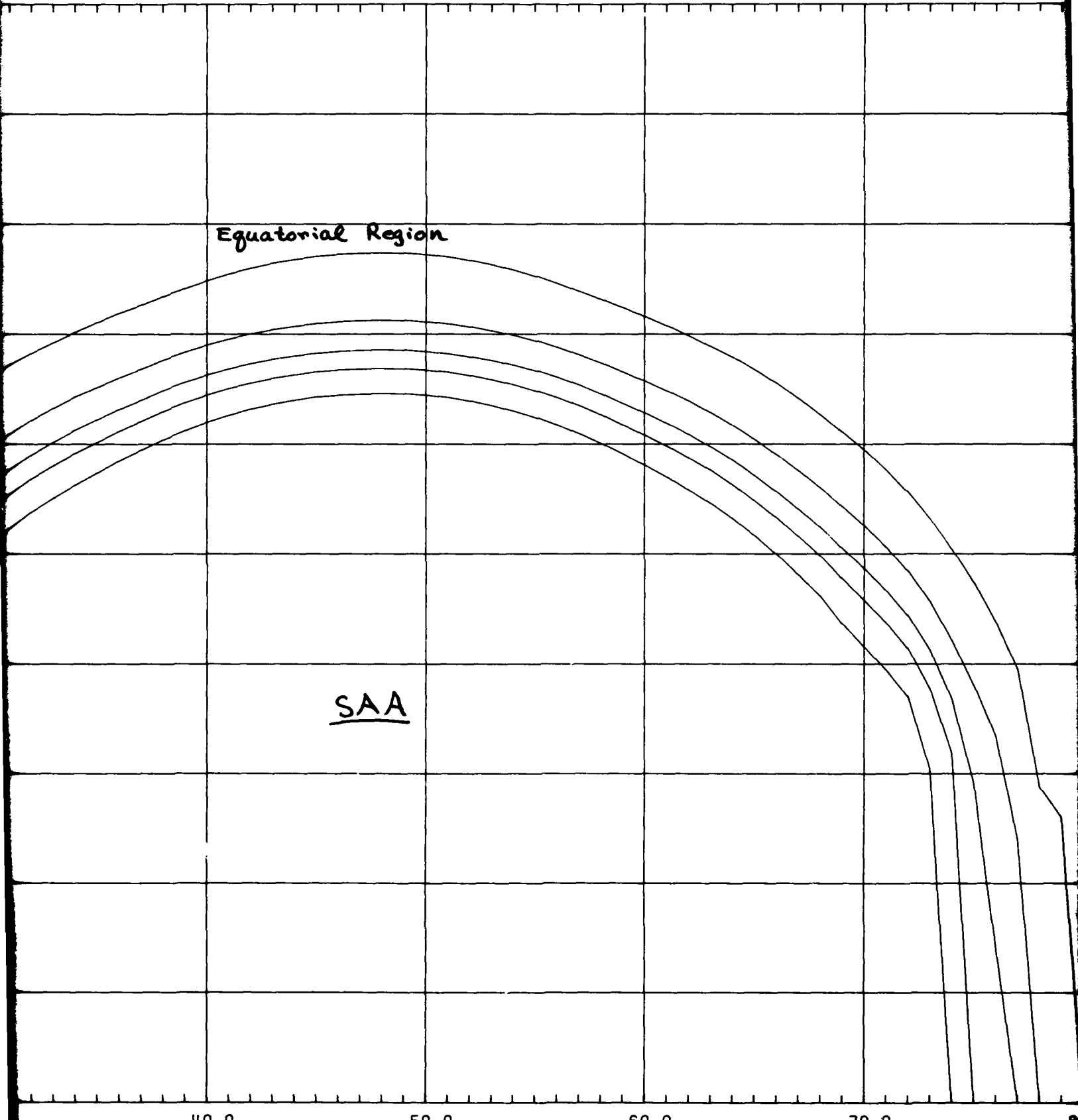
40.0

50.0

60.0

70.0

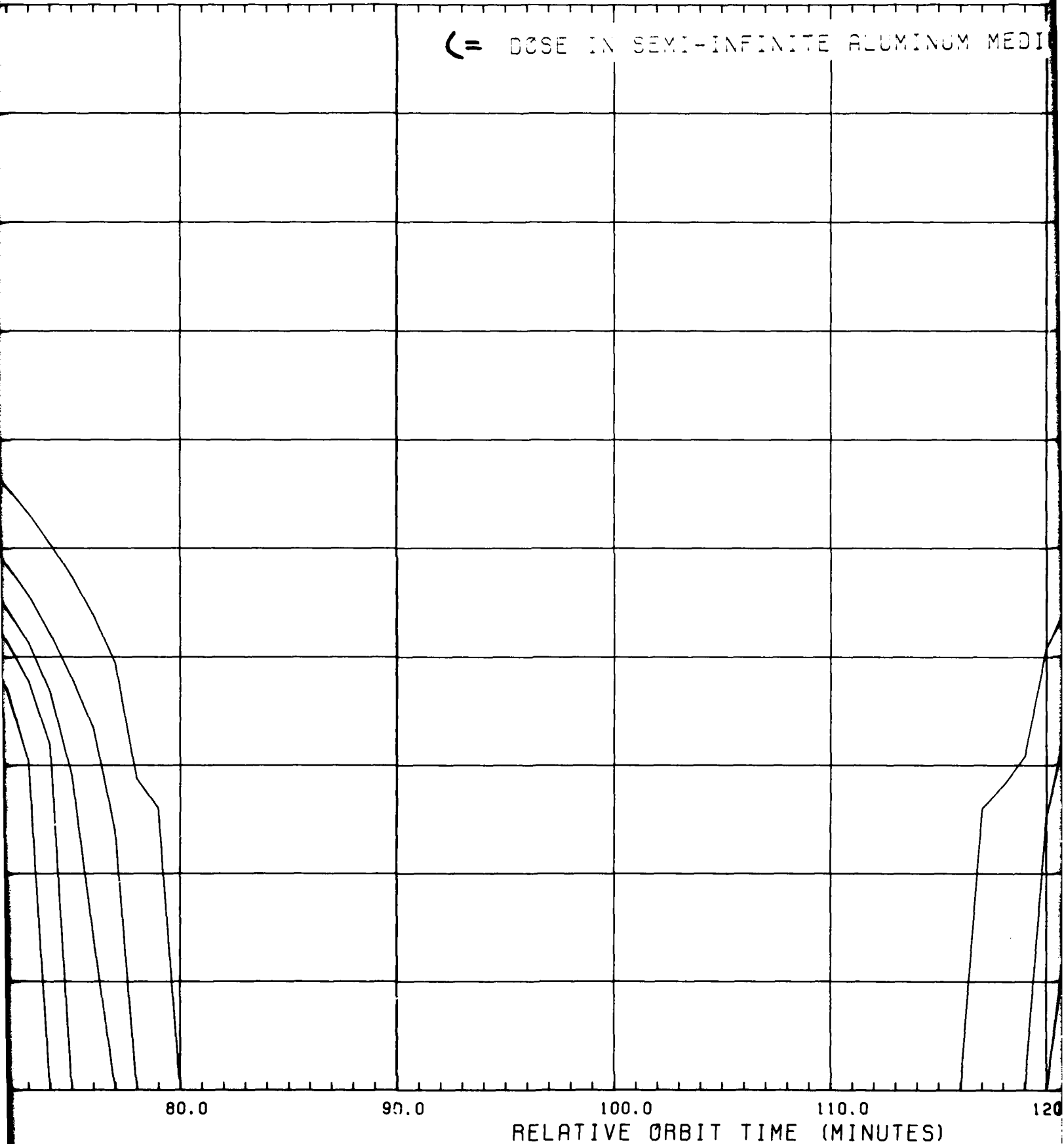
8



3

DOSE AT TRANSMISSION SURFACE OF FINITE ALUMINUM

(= DOSE IN SEMI-INFINITE ALUMINUM MEDIUM)



4

TE ALUMINUM SLAB SHIELDS

UMINUM MEDIUM)

Equatorial Re

SAA

120.0

130.0

140.0

150.0

160.0

TES)

5

ORBIT: NAVELE
60 DGR/5186

EPOCH: 1989.5

MODELS:
FIELD: BARR/7
TRAPPED PROTONS
INNER ZN ELE
OUTER ZN ELE
MISSION DURATION
EVALUATION PHASE

UN FACTORS: NONE

Equatorial Region

SAA

STOP TIME ON T

0 160.0 170.0 180.0 190.0

Figure 139

ORBIT: NAVELEX 4
60 DGR/5186-5186 KM

EPOCH: 1989.5

MODELS:

FIELD: BARR/75

TRAPPED PROTONS: AP8

INNER ZN ELEC: AE6

OUTER ZN ELEC: AE17 L0

MISSION DURATION: 60.00 MO

EVALUATION PHASE: SOLAR MAX

UN FACTORS: NOT APPLIED

STOP TIME ON TAPE= 7.716662

NASA-GSFC

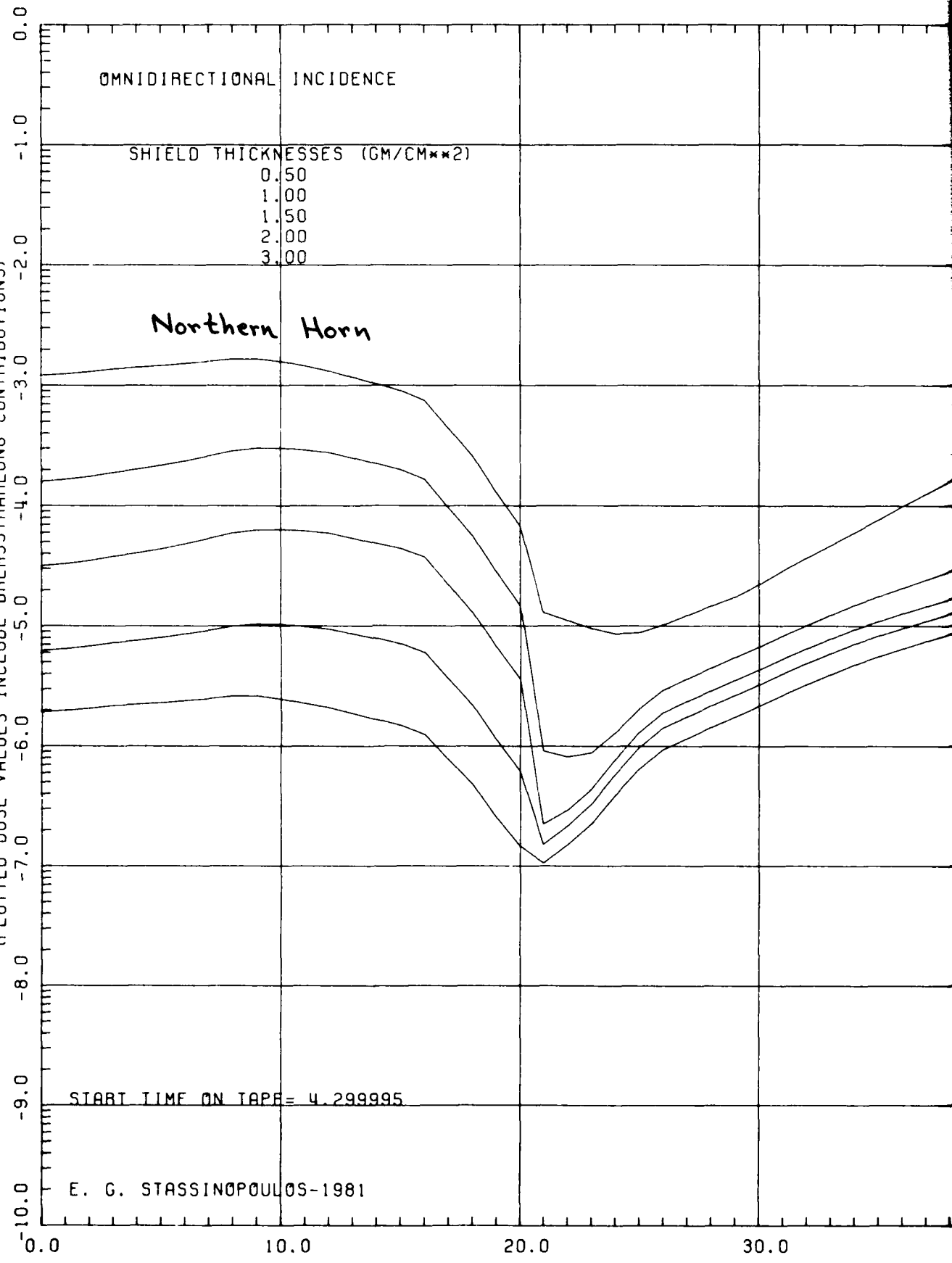
90.0

200.0

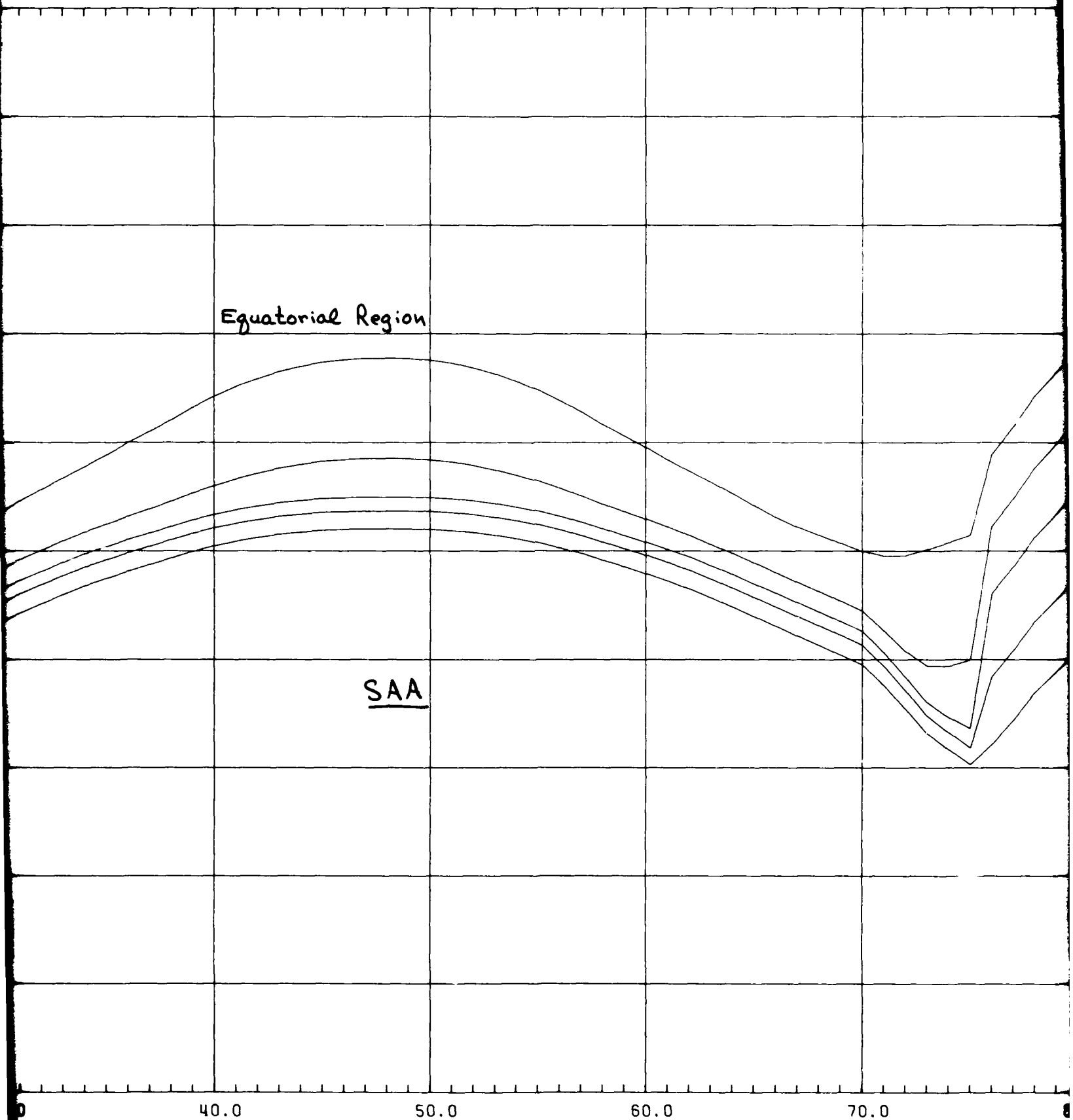
210.0

INSTANTANEOUS ALUMINUM ELECTRON DOSE (RADS)

(PLOTTED DOSE VALUES INCLUDE BREMSSTRAHLUNG CONTRIBUTIONS)



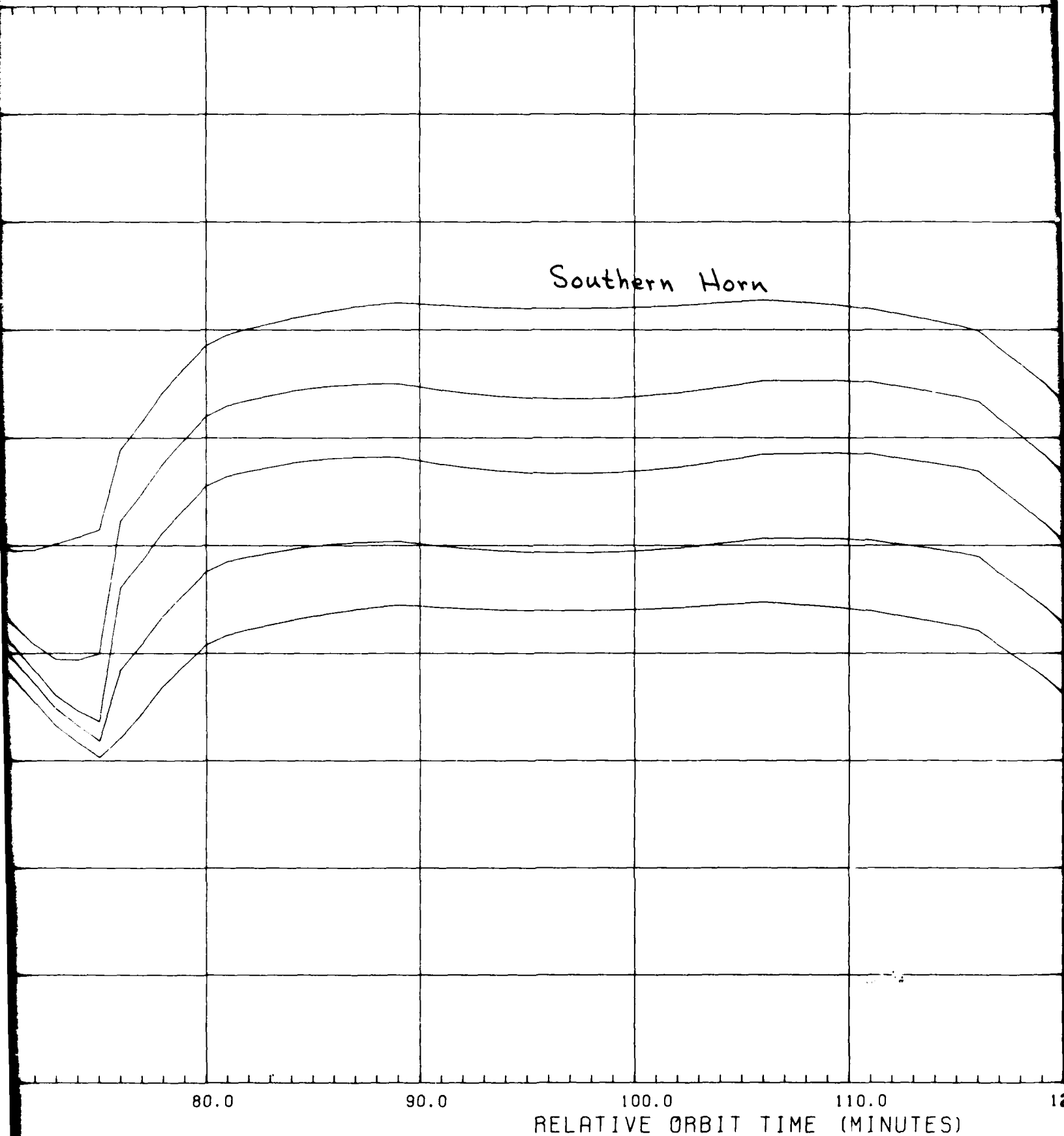
5



1
2

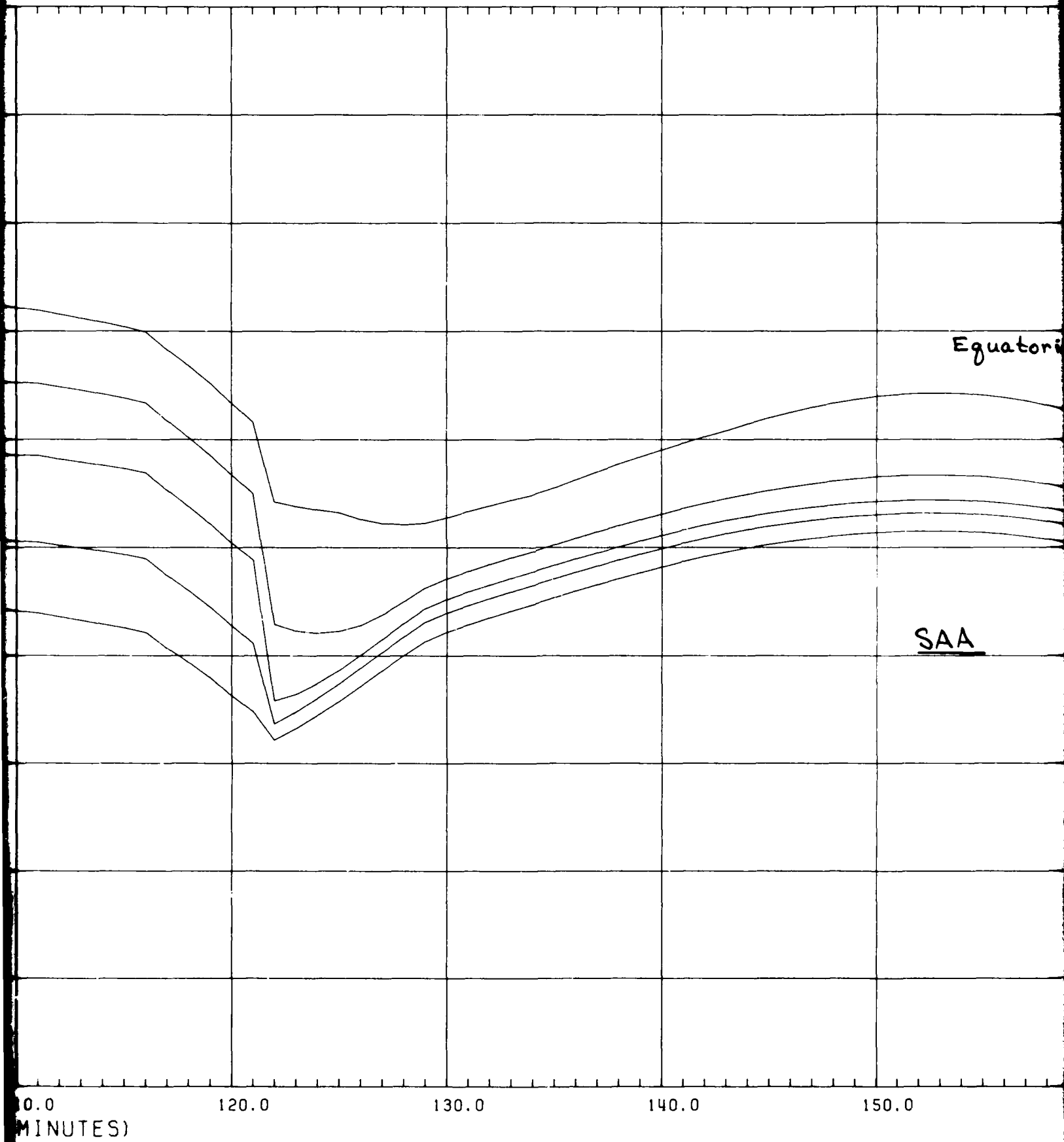
DOSE AT TRANSMISSION SURFACE OF FINITE ALUMINUM

Southern Horn



14

FINITE ALUMINUM SLAB SHIELDS



5

ORBIT: NA
60 DGR/

EPOCH: 19

MODELS:
FIELD: B
TRAPPED
INNER ZN
OUTER ZN
MISSION D
EVALUATION

UN FACTORS

Equatorial Region

Northern
Horn

SAA

STOP TIME

150.0

160.0

170.0

180.0

190.0

Figure 140

ORBIT: NAVELEX 4
60 DGR/5186-5186 KM

EPOCH: 1989.5

MODELS:

FIELD: BARR/75

TRAPPED PROTONS: AP8

INNER ZN ELEC: AE6

OUTER ZN ELEC: AE17 L0

MISSION DURATION: 60.00 MO

EVALUATION PHASE: SOLAR MAX

UN FACTORS: NOT APPLIED

Northern
Horn

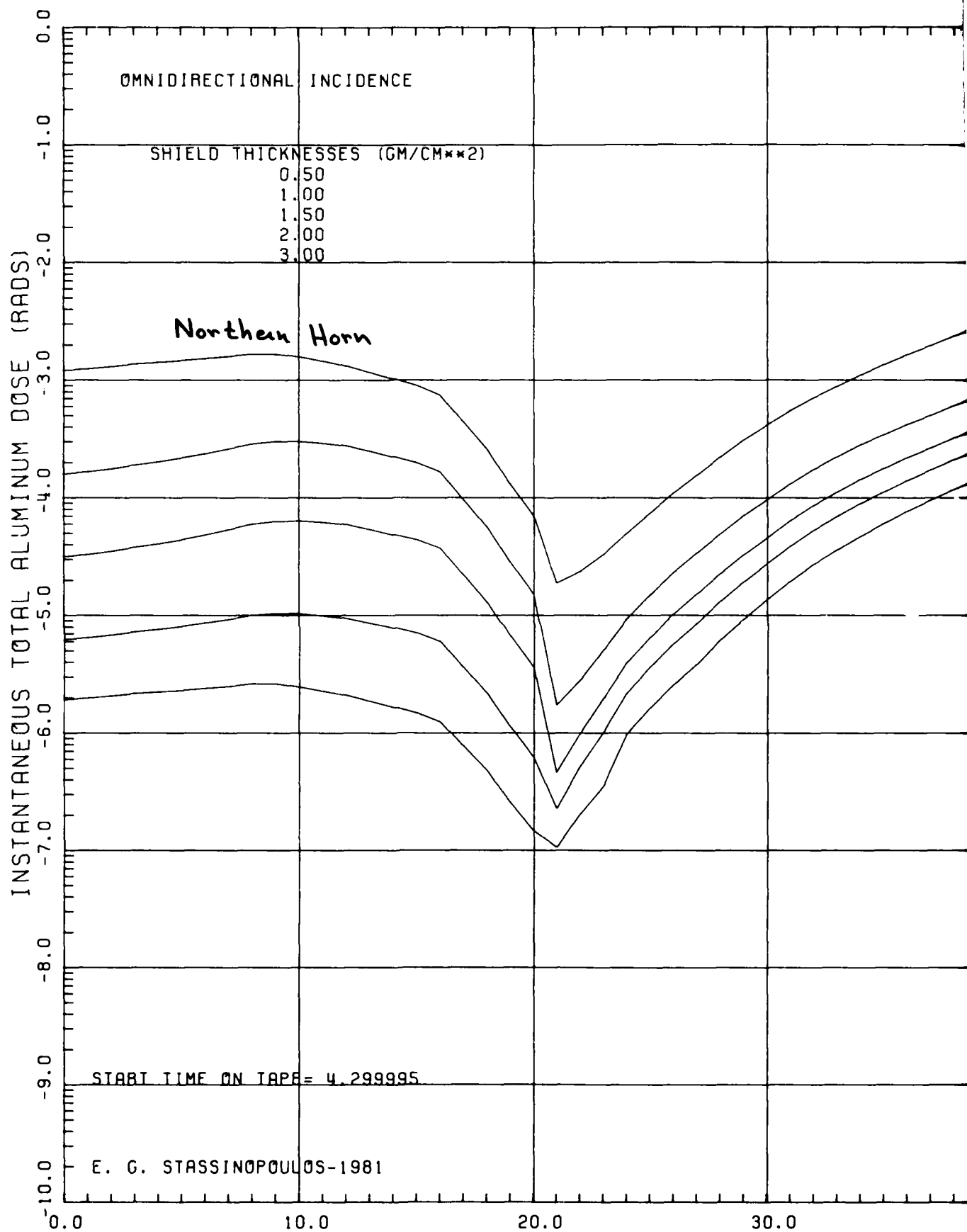
STOP TIME ON TAPE= 7.716662

NASA-GSFC

190.0

200.0

210.0



Equatorial Region

SAA

40.0

50.0

60.0

70.0

3'

DOSE AT TRANSMISSION SURFACE OF FINITE ALUMINUM

Southern Horn

80.0

90.0

100.0

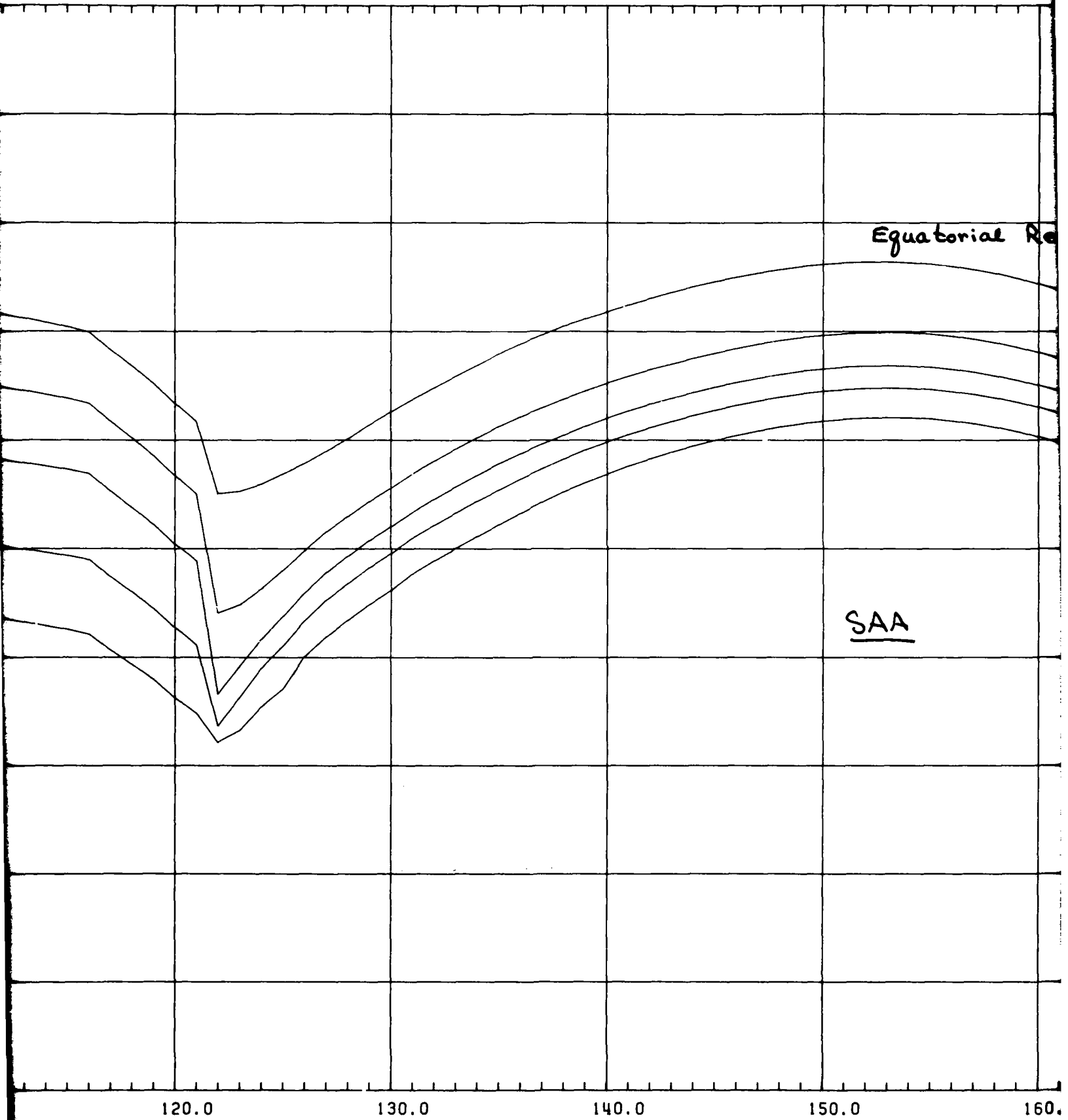
110.0

1

RELATIVE ORBIT TIME (MINUTES)

4

TE ALUMINUM SLAB SHIELDS



Equatorial Re

SAA

TES)

5

ORBIT: NAVELEX
60 DGR/5186-

EPOCH: 1989.5

MODELS:
FIELD: BARR/7
TRAPPED PROTON
INNER ZN ELEC
OUTER ZN ELEC
MISSION DURATION
EVALUATION PHASE

UN FACTORS: NO

Equatorial Region

Northern
Horn

SAA

STOP TIME ON T

0 160.0 170.0 180.0 190.0

Figure 141

ORBIT: NAVELEX 4
60 DGR/5186-5186 KM

EPOCH: 1989.5

MODELS:

FIELD: BARR/75

TRAPPED PROTONS: AP8

INNER ZN ELEC: AE6

OUTER ZN ELEC: AE17 L8

MISSION DURATION: 60.00 MO

EVALUATION PHASE: SOLAR MAX

UN FACTORS: NOT APPLIED

STOP TIME ON TAPF= 7.716662

NASA-GSFC

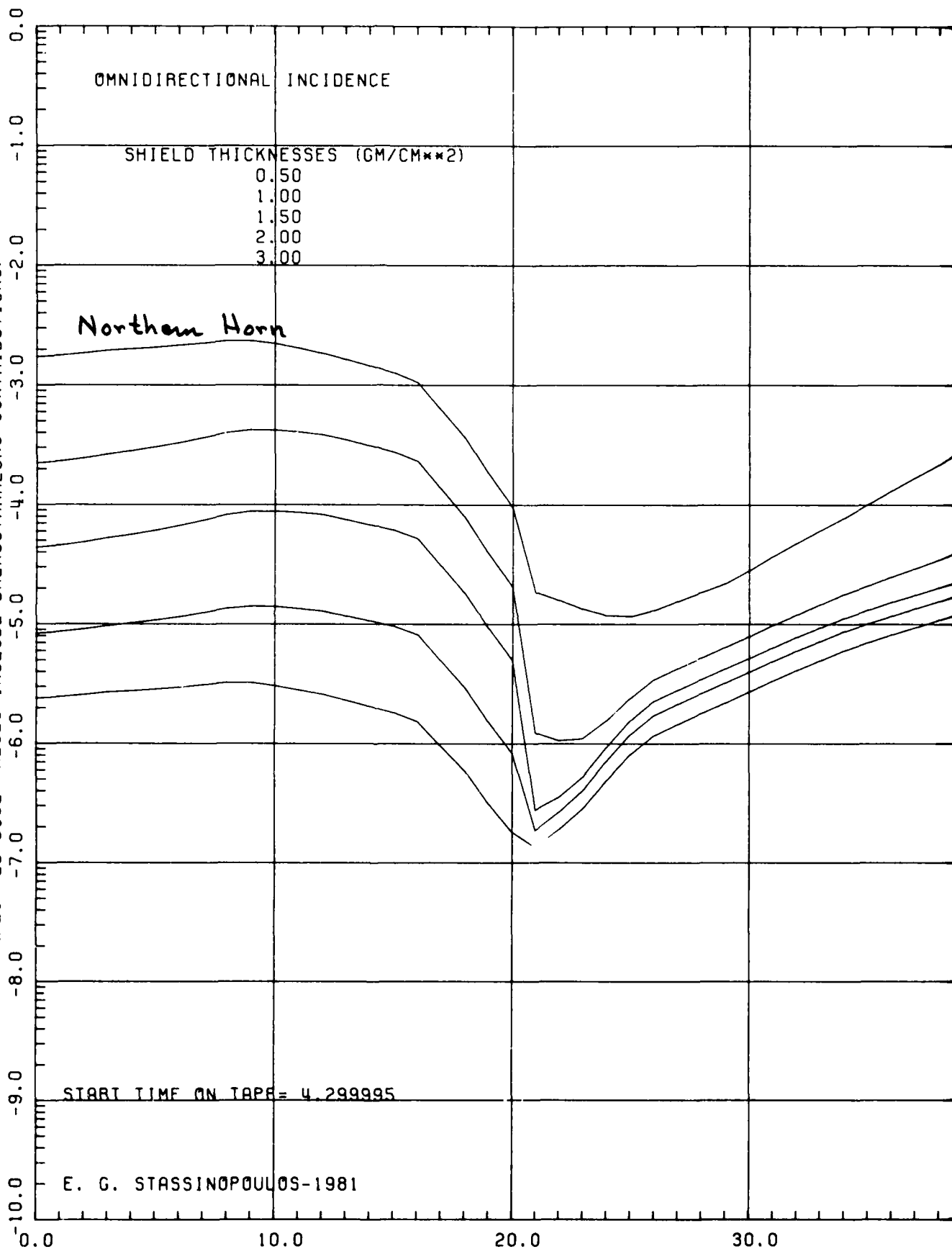
190.0

200.0

210.0

INSTANTANEOUS ALUMINUM ELECTRON DOSE (RADS)

(PLOTTED DOSE VALUES INCLUDE BREMSSTRAHLUNG CONTRIBUTIONS)



1
2

Equatorial Region

SAA

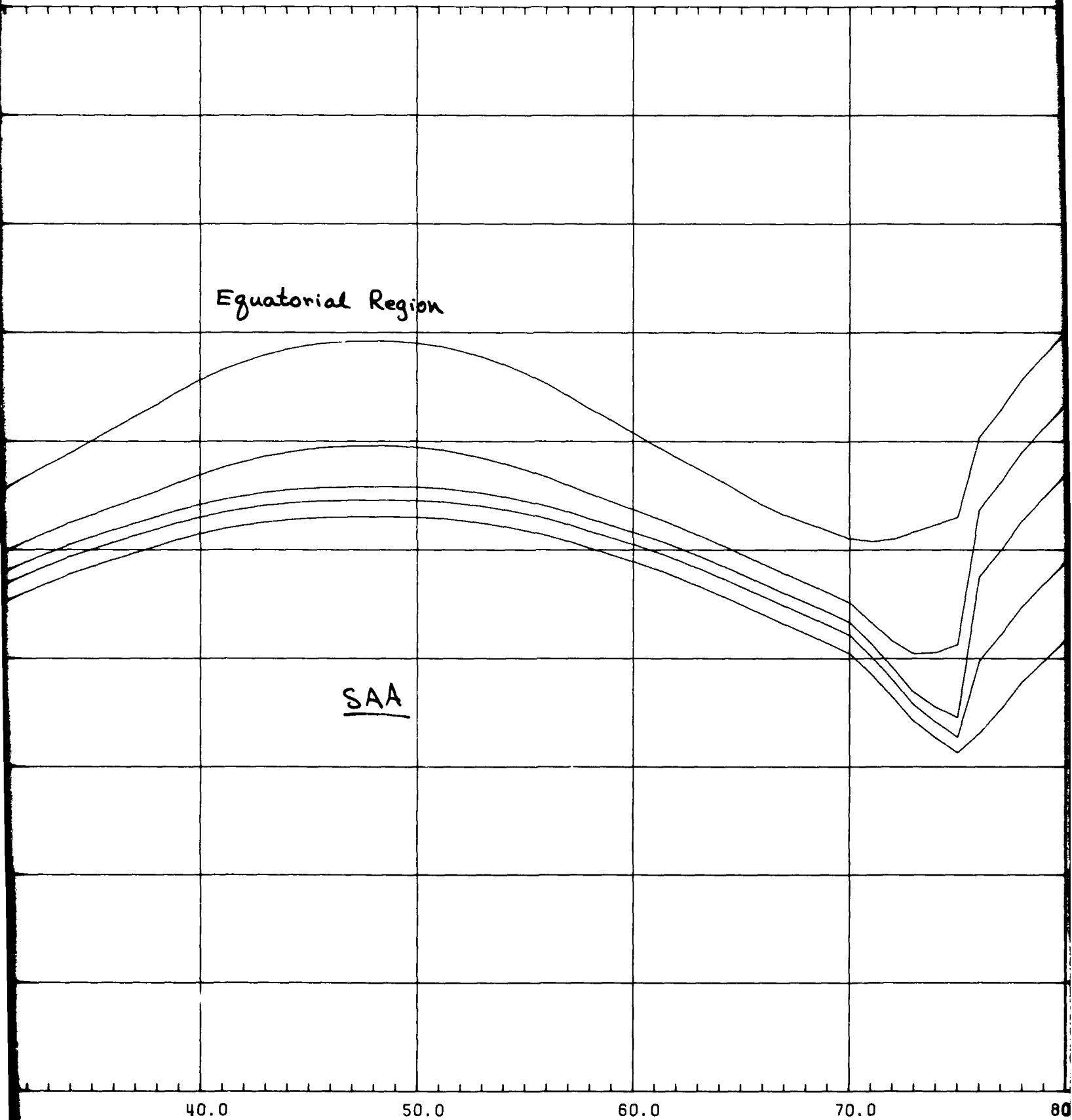
40.0

50.0

60.0

70.0

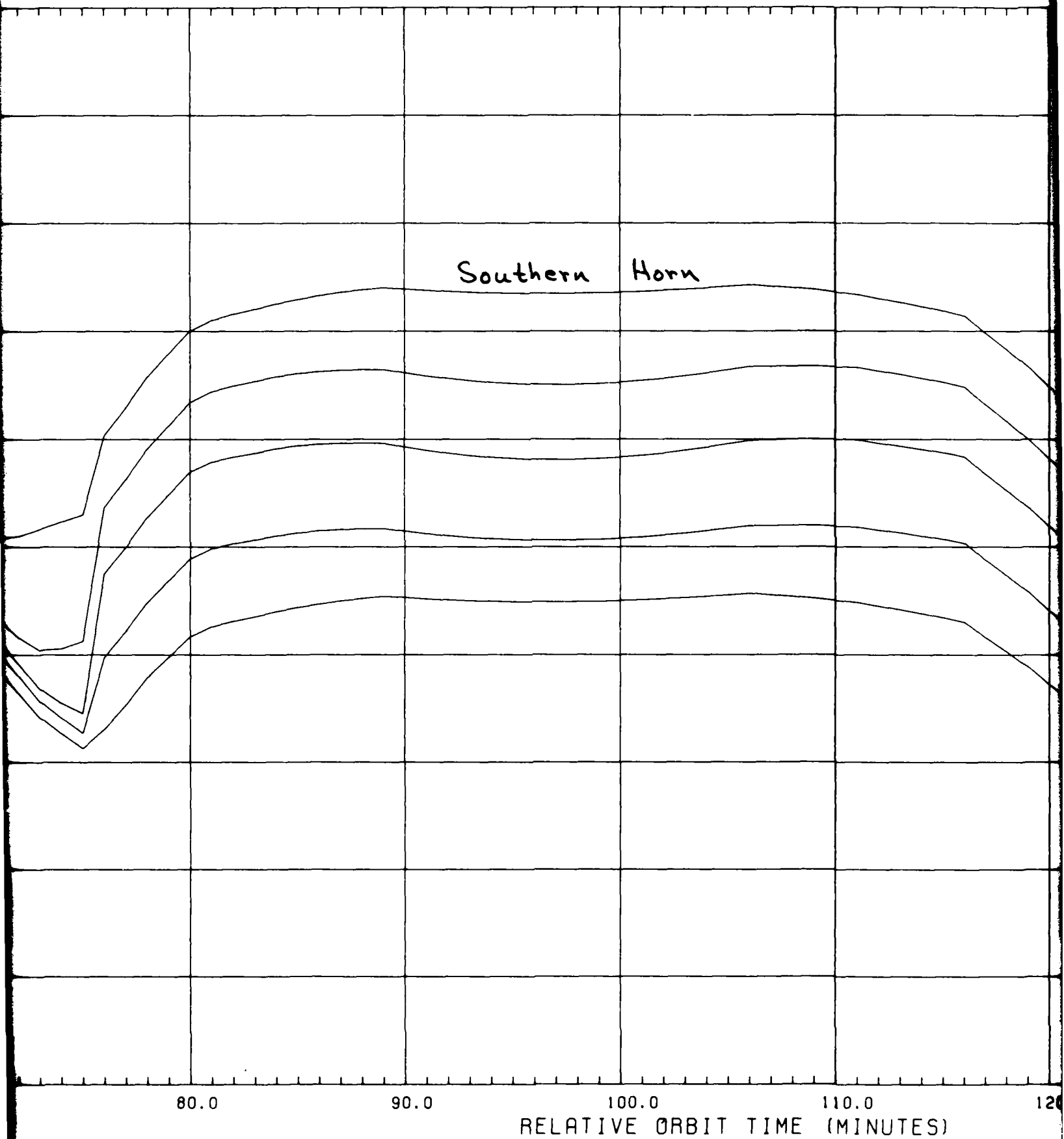
80.0



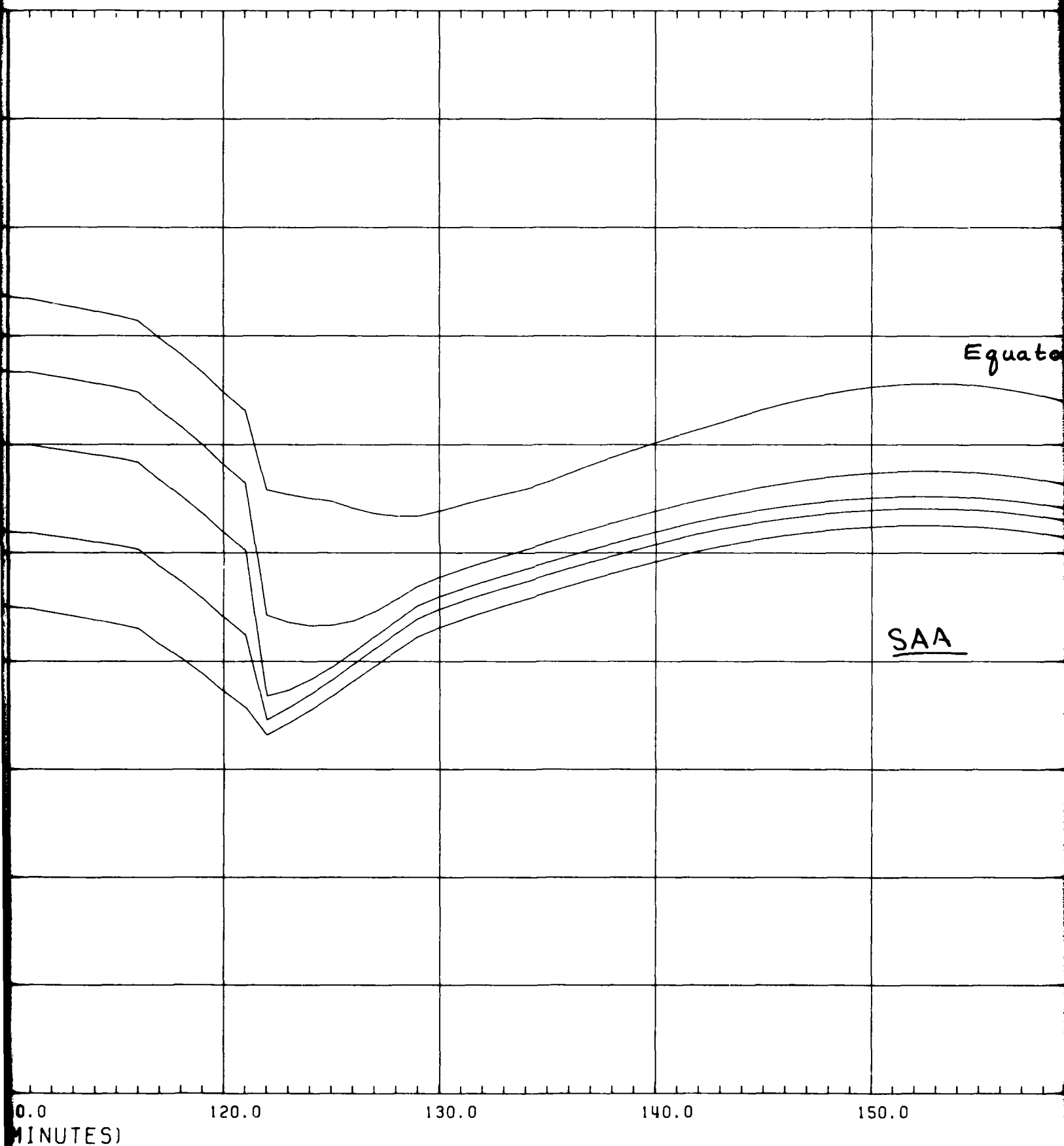
3

DOSE IN SEMI-INFINITE ALUMINUM MEDIUM

Southern Horn



LUMINUM MEDIUM



ORBIT: NAVELEX
60 DGR/5186

EPOCH: 1989.5

MODELS:
FIELD: BARR/7
TRAPPED PROTONS
INNER ZN ELECTRON
OUTER ZN ELECTRON
MISSION DURATION
EVALUATION PHASE

UN FACTORS: NO

Northern
Horn

Equatorial Region

SAA

STOP TIME ON T

150.0

160.0

170.0

180.0

190.0

Figure 142

ORBIT: NAVELEX 4
60 DGR/5186-5186 KM

EPOCH: 1989.5

MODELS:

FIELD: BARR/75

TRAPPED PROTONS: AP8

INNER ZN ELEC: AE6

OUTER ZN ELEC: AE17-L0

MISSION DURATION: 60.00 MO

EVALUATION PHASE: SOLAR MAX

UN FACTORS: NOT APPLIED

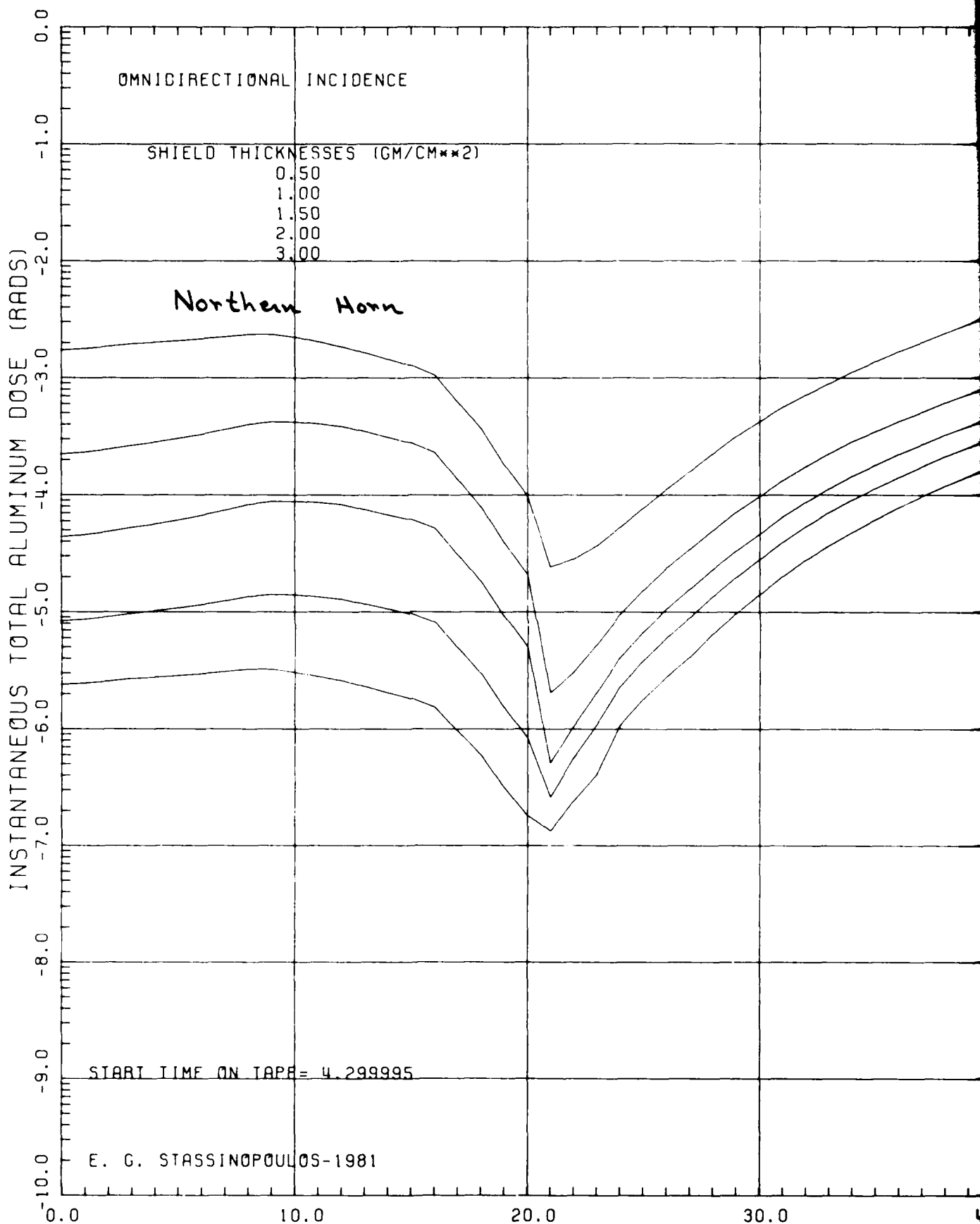
STOP TIME ON TAPE= 7.716662

NASA-GSFC

190.0

200.0

210.0



Equatorial Region

SAA

40.0

50.0

60.0

70.0

80

DOSE IN SEMI-INFINITE ALUMINUM MEDIUM

Southern Horn

80.0

90.0

100.0

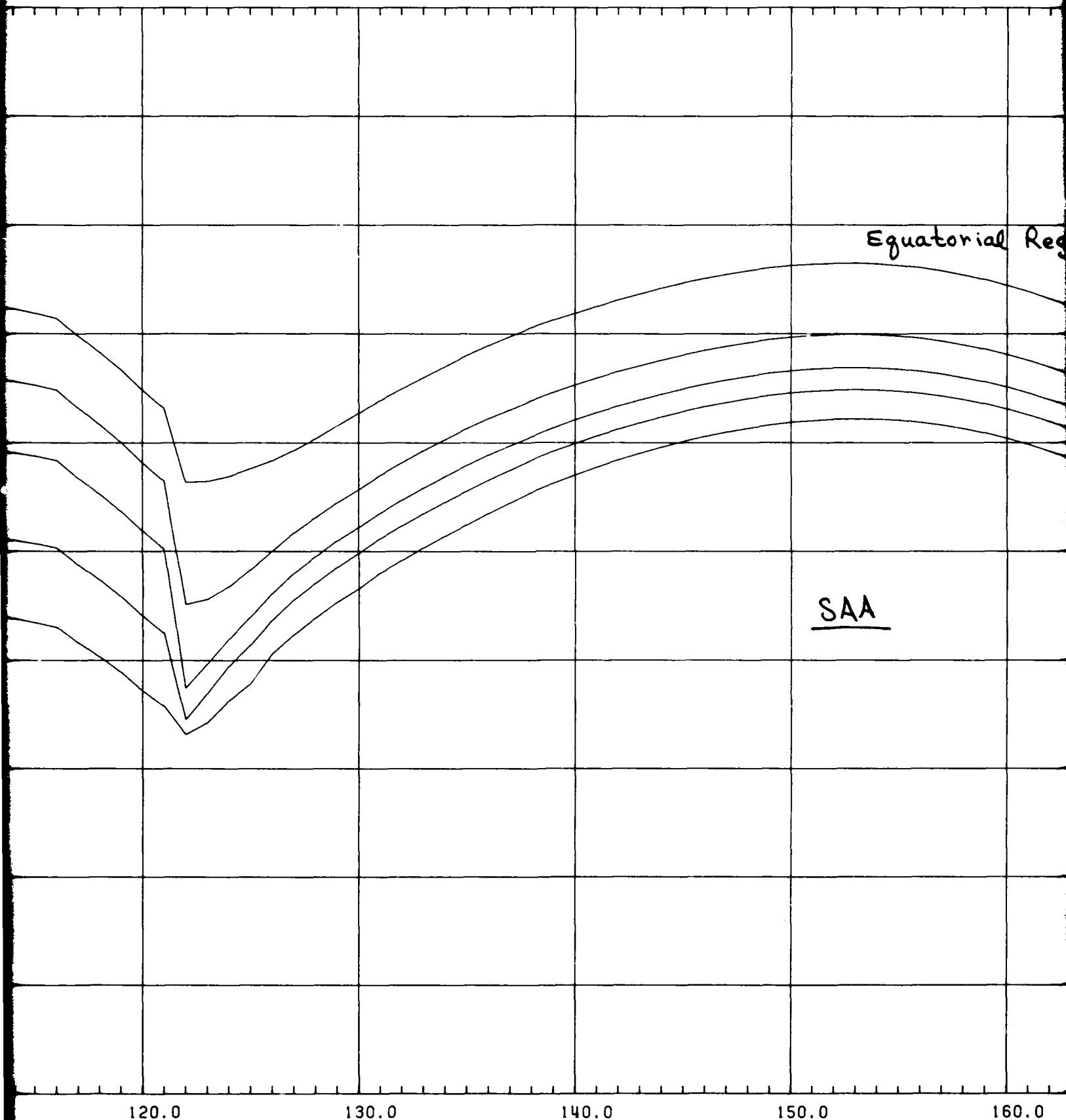
110.0

120.0

RELATIVE ORBIT TIME (MINUTES)

UM MEDIUM

4



Equatorial Reg

SAA

120.0

130.0

140.0

150.0

160.0

S)

5.

ORBIT: NAVELEX 4
60 DGR/5186-518

EPOCH: 1989.5

MODELS:
FIELD: BARR/75
TRAPPED PROTONS:
INNER ZN ELEC: A
OUTER ZN ELEC: A
MISSION DURATION:
EVALUATION PHASE:

UN FACTORS: NOT A

Equatorial Region

Northern
Horn

AA

STOP TIME ON TAPE

160.0

170.0

180.0

190.0

200.0

Figure 143

ORBIT: NAVELEX 4
60 DGR/5186-5186 KM

EPOCH: 1989.5

MODELS:

FIELD: BARR/75

TRAPPED PROTONS: AP8

INNER ZN ELEC: AE6

OUTER ZN ELEC: AE17 L0

MISSION DURATION: 60.00 MO

EVALUATION PHASE: SOLAR MAX

UN FACTORS: NOT APPLIED

STOP TIME ON TAPE = 7.716662

NASA-GSFC

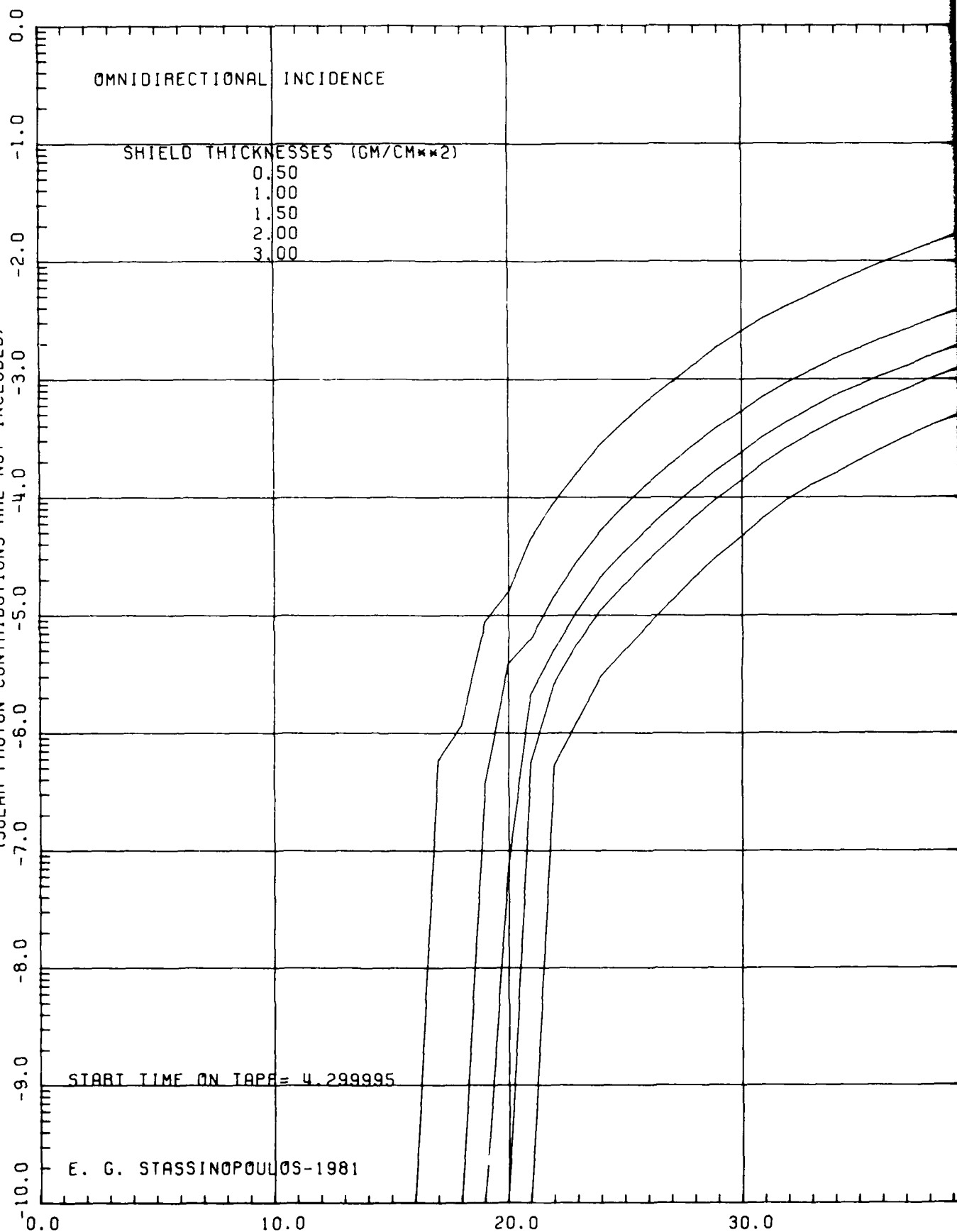
190.0

200.0

210.0

INSTANTANEOUS ALUMINUM PROTON DOSE (RADS)

(SOLAR PROTON CONTRIBUTIONS ARE NOT INCLUDED)



12

Equatorial Region

SAA

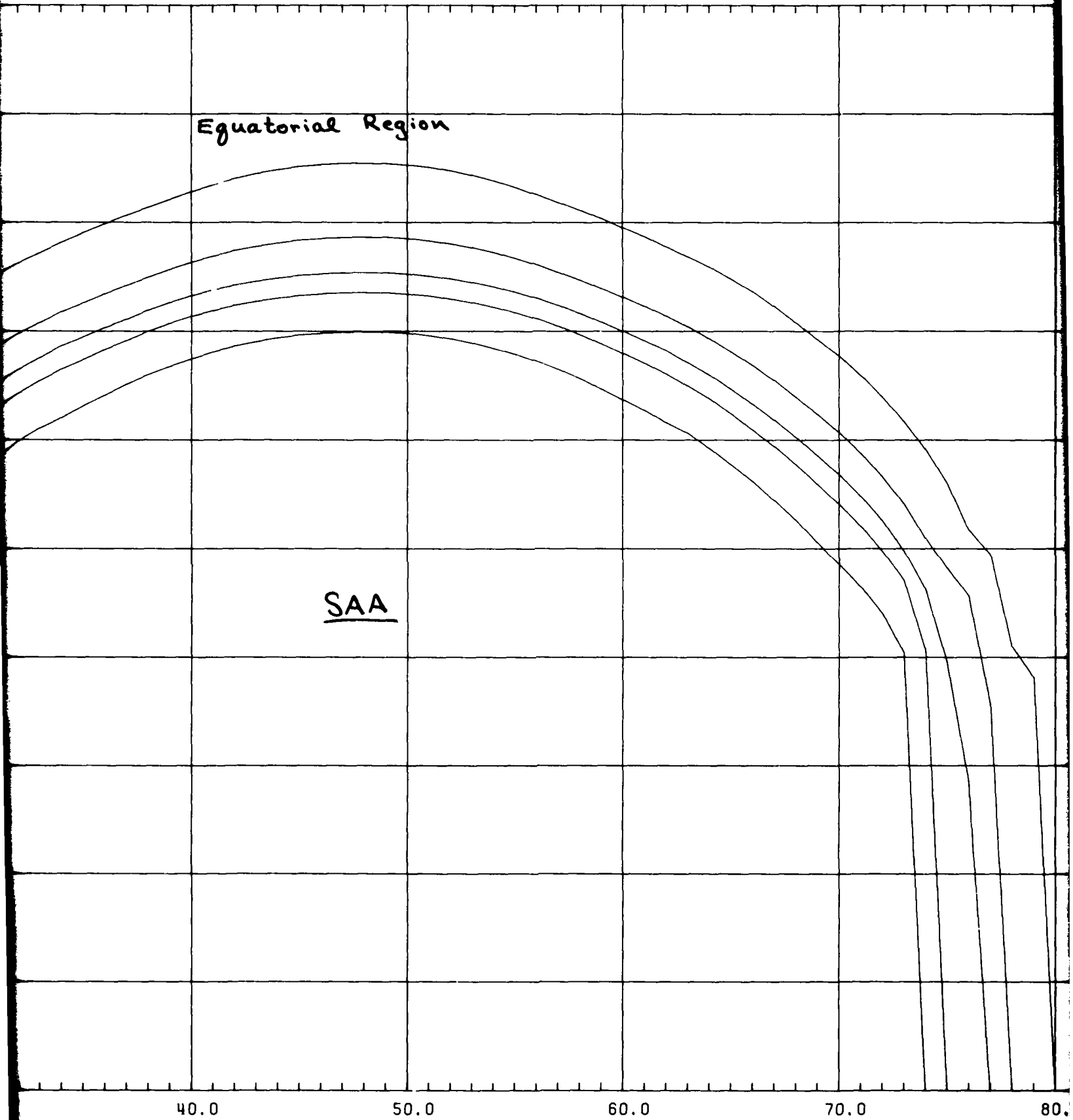
40.0

50.0

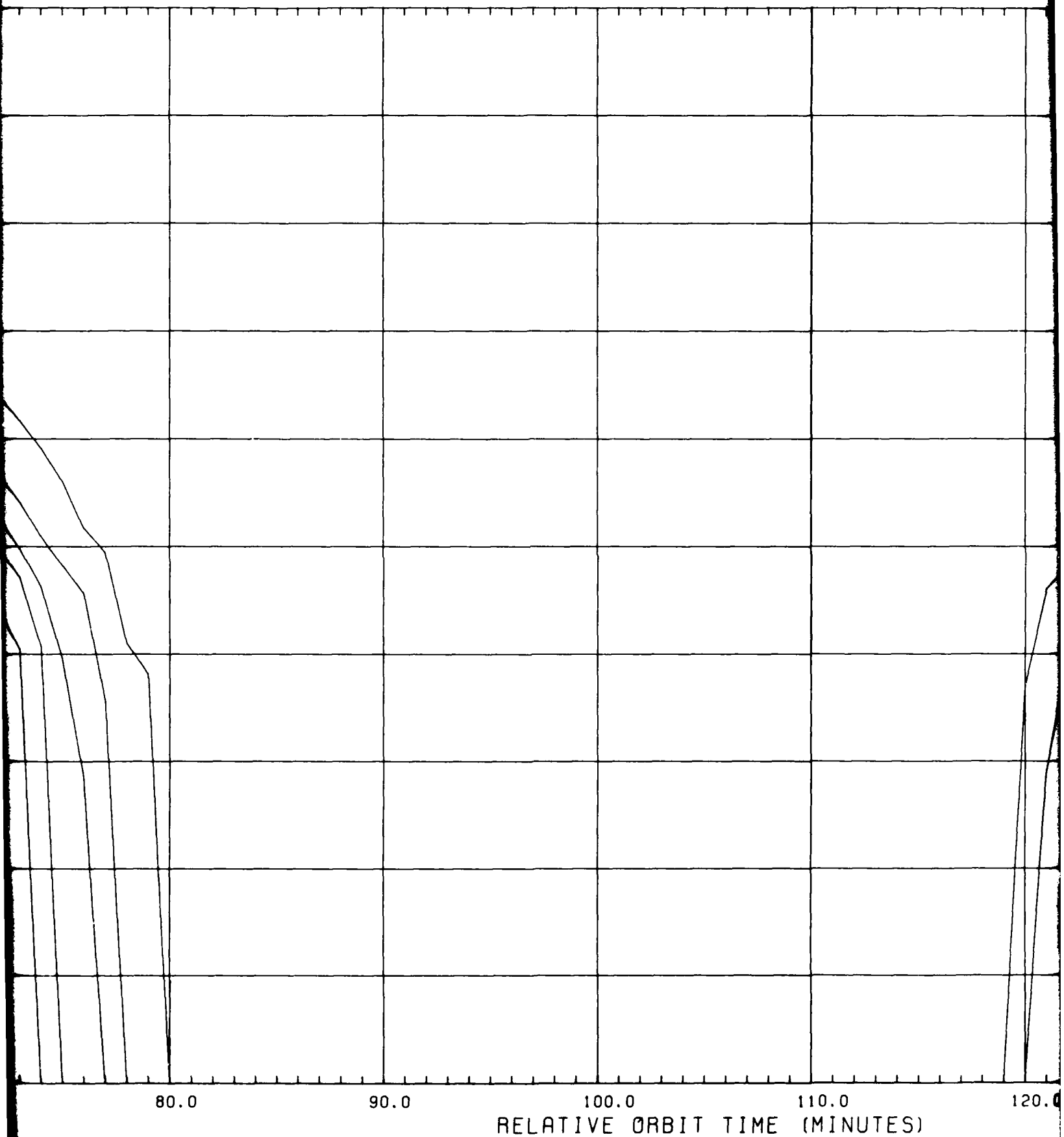
60.0

70.0

80.0



DOSE AT CENTER OF ALUMINUM SPHERES



ERES

4

Equatorial Region

SAA

120.0

130.0

140.0

150.0

160.0

Figure

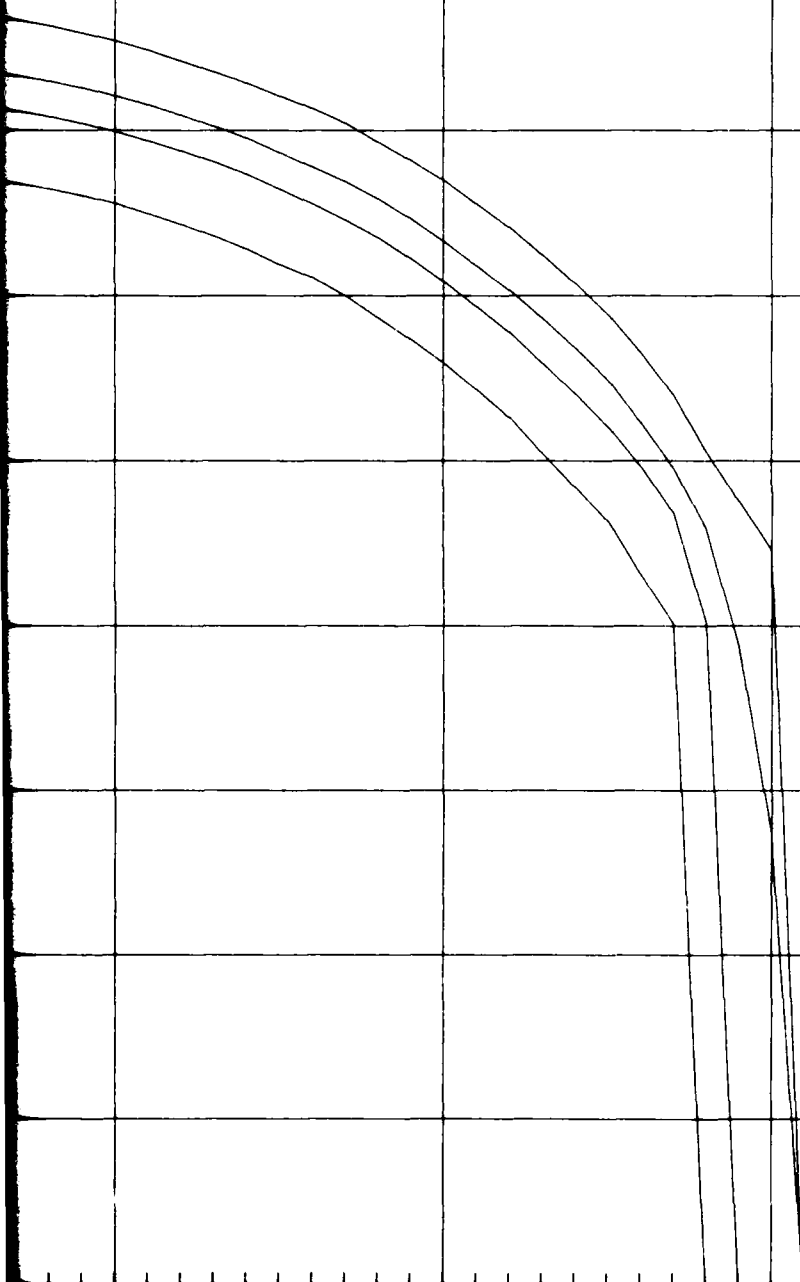
ORBIT: NAVELEX 4
60 DGR/5186-5186 KM

EPOCH: 1989.5

MODELS:
FIELD: BARR/75
TRAPPED PROTONS: AP8
INNER ZN ELEC: AE6
OUTER ZN ELEC: AE17 L8
MISSION DURATION: 60.00 MO
EVALUATION PHASE: SOLAR MAX

UN FACTORS: NOT APPLIED

Region



STOP TIME ON TAPE= 7.716662

NASA-GS

160.0

170.0

180.0

190.0

200.0

Figure 144

ORBIT: NAVELEX 4
60 DGR/5186-5186 KM

EPOCH: 1989.5

MODELS:

FIELD: BARR/75

TRAPPED PROTONS: AP8

INNER ZN ELEC: AE6

OUTER ZN ELEC: AE17 L0

MISSION DURATION: 60.00 M0

EVALUATION PHASE: SOLAR MAX

UN FACTORS: NOT APPLIED

STOP TIME ON TAPE = 7.716662

NASA-GSFC

190.0

200.0

210.0

INSTANTANEOUS ALUMINUM ELECTRON DOSE (RADS)

(PLOTTED DOSE VALUES INCLUDE BREMSSTRAHLUNG CONTRIBUTIONS)

OMNIDIRECTIONAL INCIDENCE

SHIELD THICKNESSES (GM/CM**2)

0.50

1.00

1.50

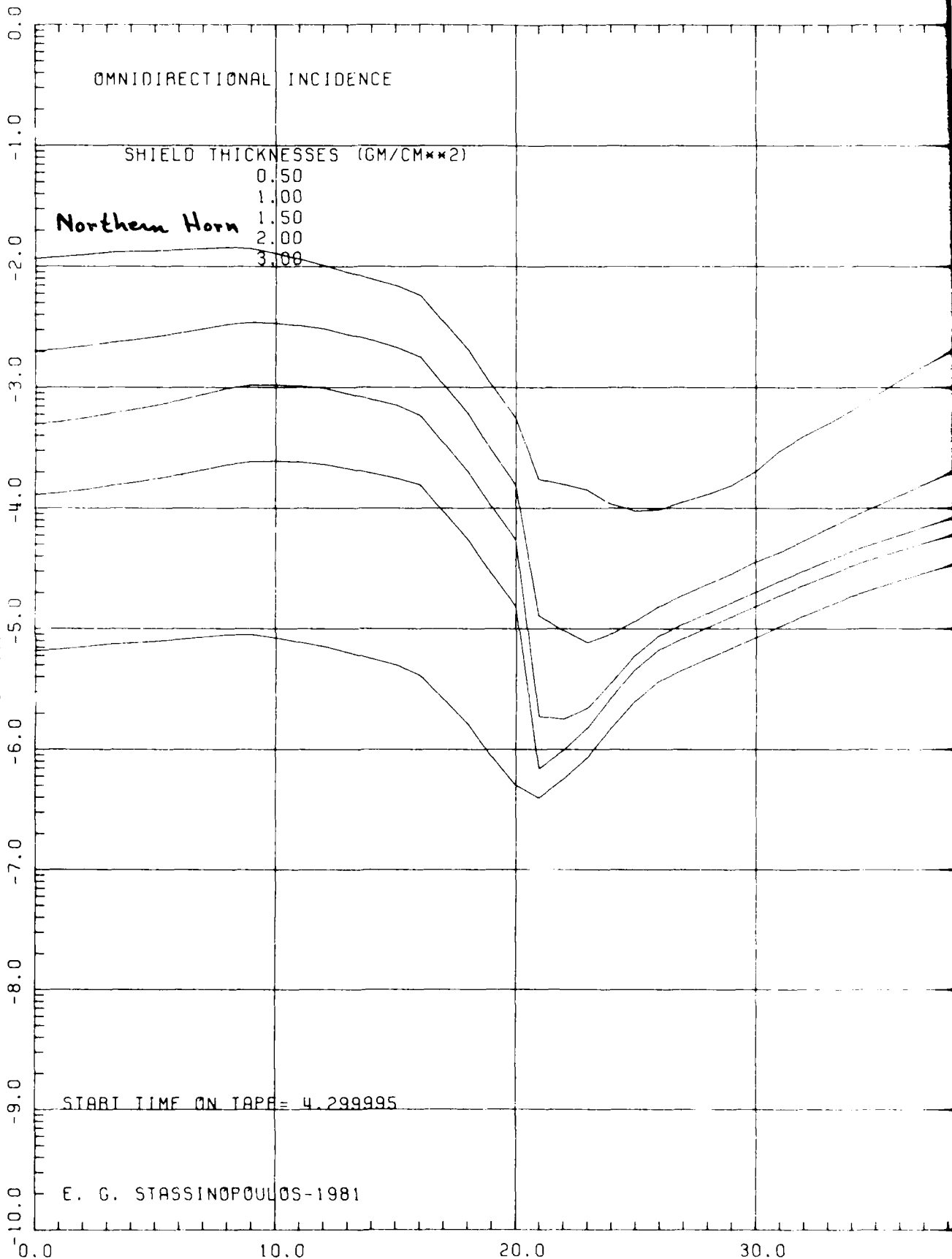
2.00

3.00

Northern Horn

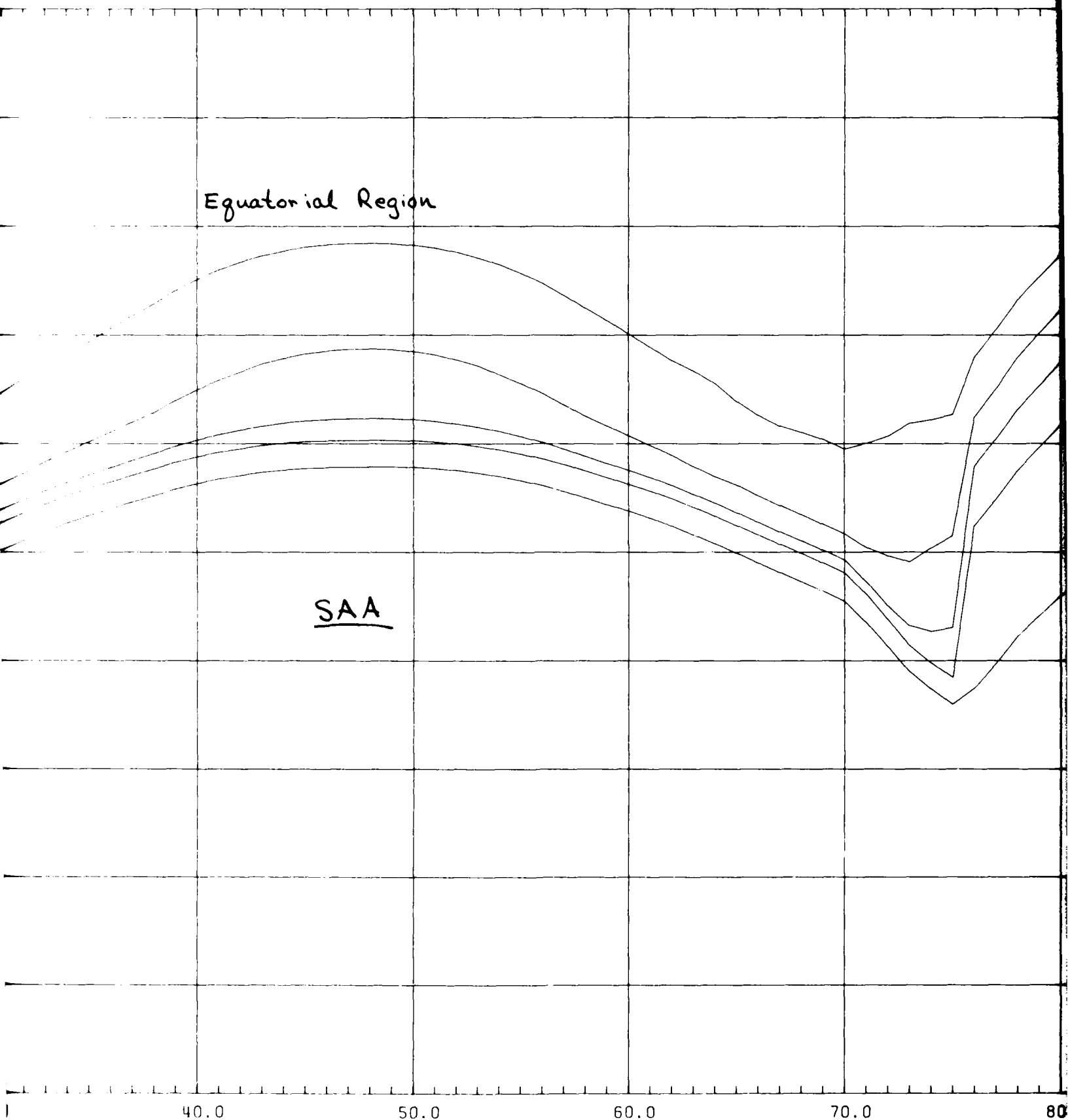
START TIME ON TAPE = 4.299995

E. G. STASSINOPOULOS-1981



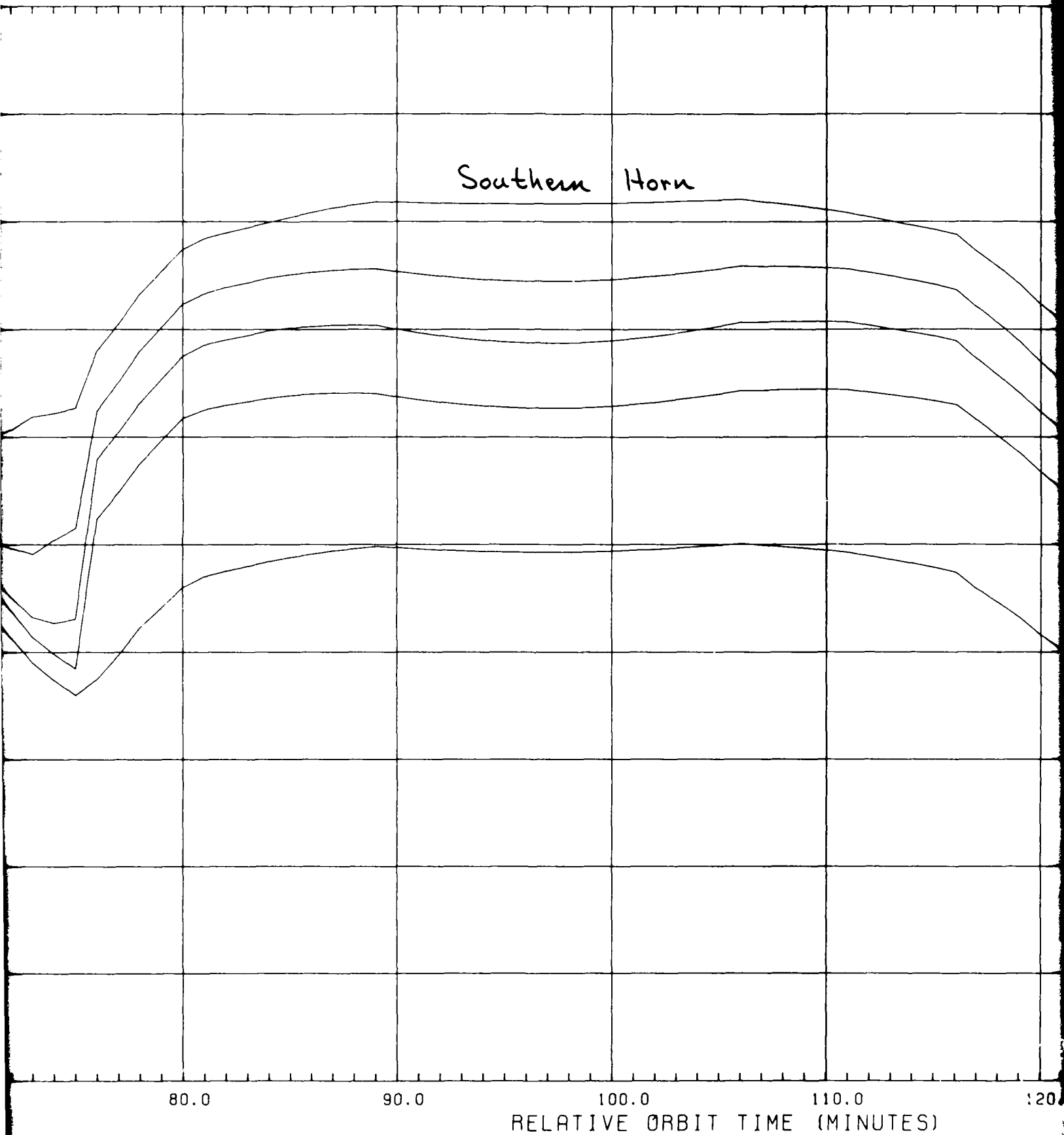
Equatorial Region

SAA



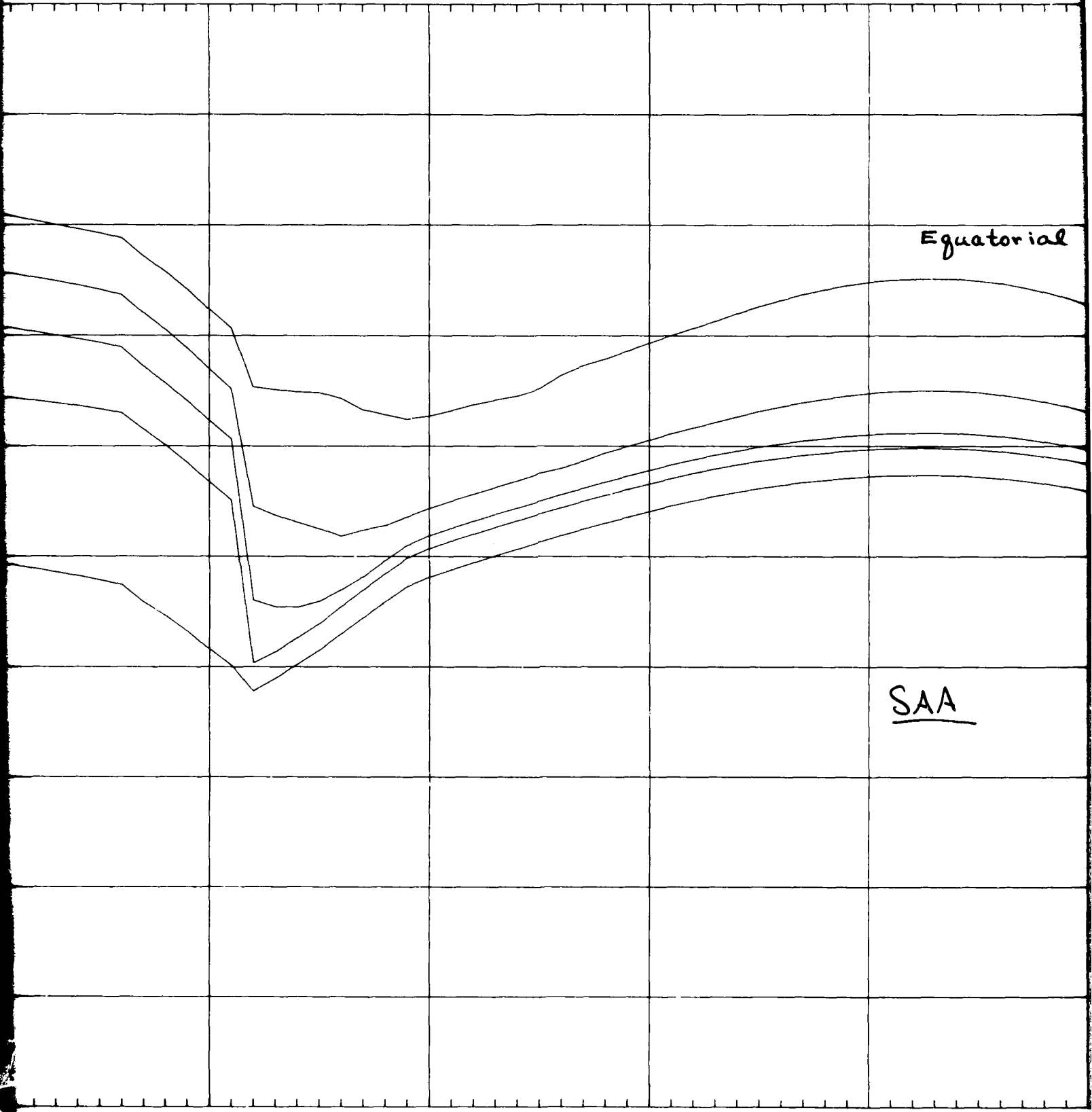
DOSE AT CENTER OF ALUMINUM SPHERES

Southern Horn



4

NUM SPHERES



Equatorial

SAA

MINUTES)

ORBIT: NA
60 DGR/

EPOCH: 19

MODELS:
FIELD: B
TRAPPED
INNER ZN
OUTER ZN
MISSION DU
EVALUATION

UN FACTORS

Northern
Horn

Equatorial Region

SAA

STOP TIME

150.0 160.0 170.0 180.0 190.0

Figure 145

ORBIT: NAVELEX 4
60 DGR/5186-5186 KM

EPOCH: 1989.5

MODELS:

FIELD: BARR/75

TRAPPED PROTONS: AP8

INNER ZN ELEC: AE6

OUTER ZN ELEC: AE17 L0

MISSION DURATION: 60.00 MO

EVALUATION PHASE: SOLAR MAX

UN FACTORS: NOT APPLIED

Northern
Horn

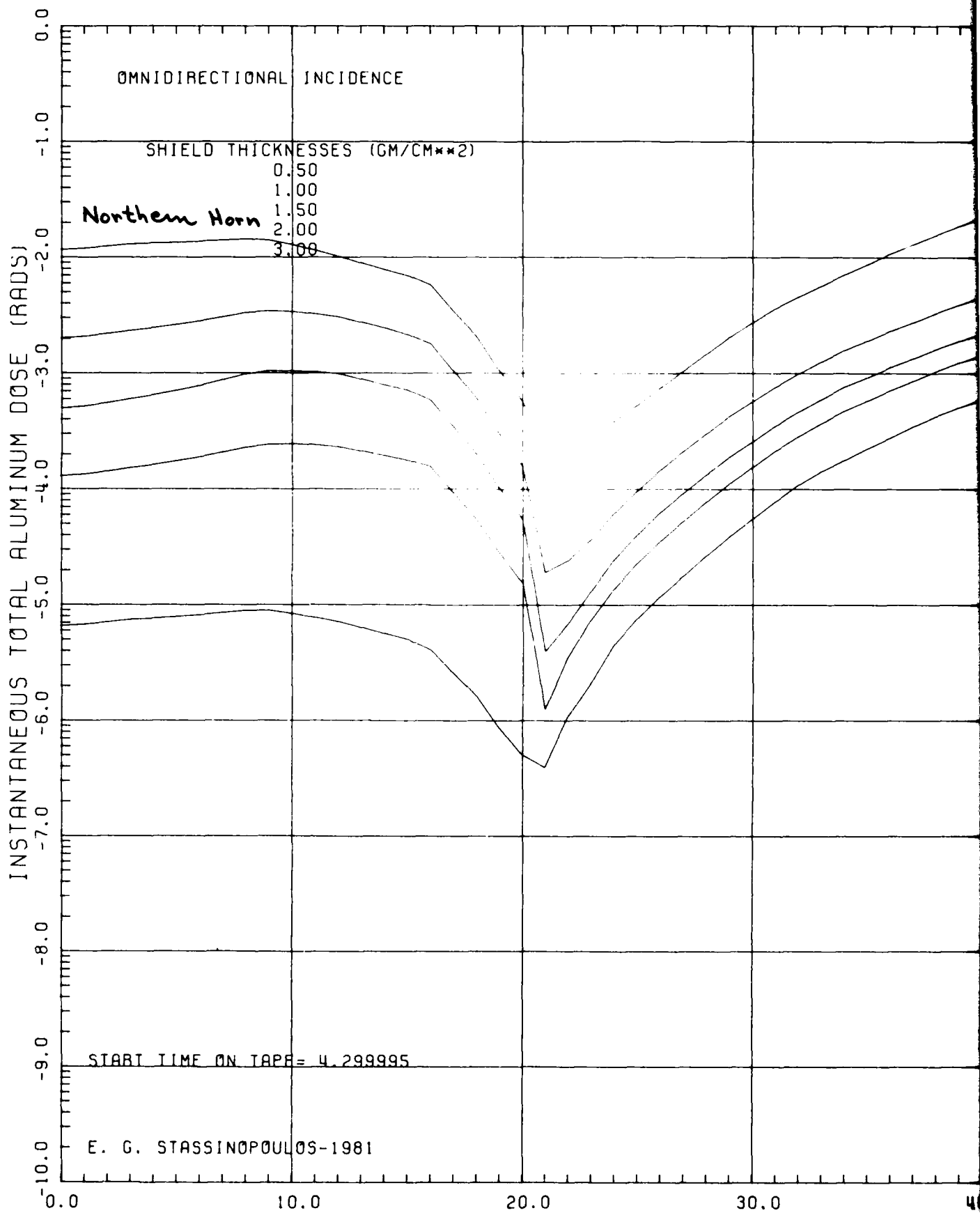
STOP TIME ON TAPE = 7.716662

NASA-GSFC

190.0

200.0

210.0



Equatorial Region

SAA

0

40.0

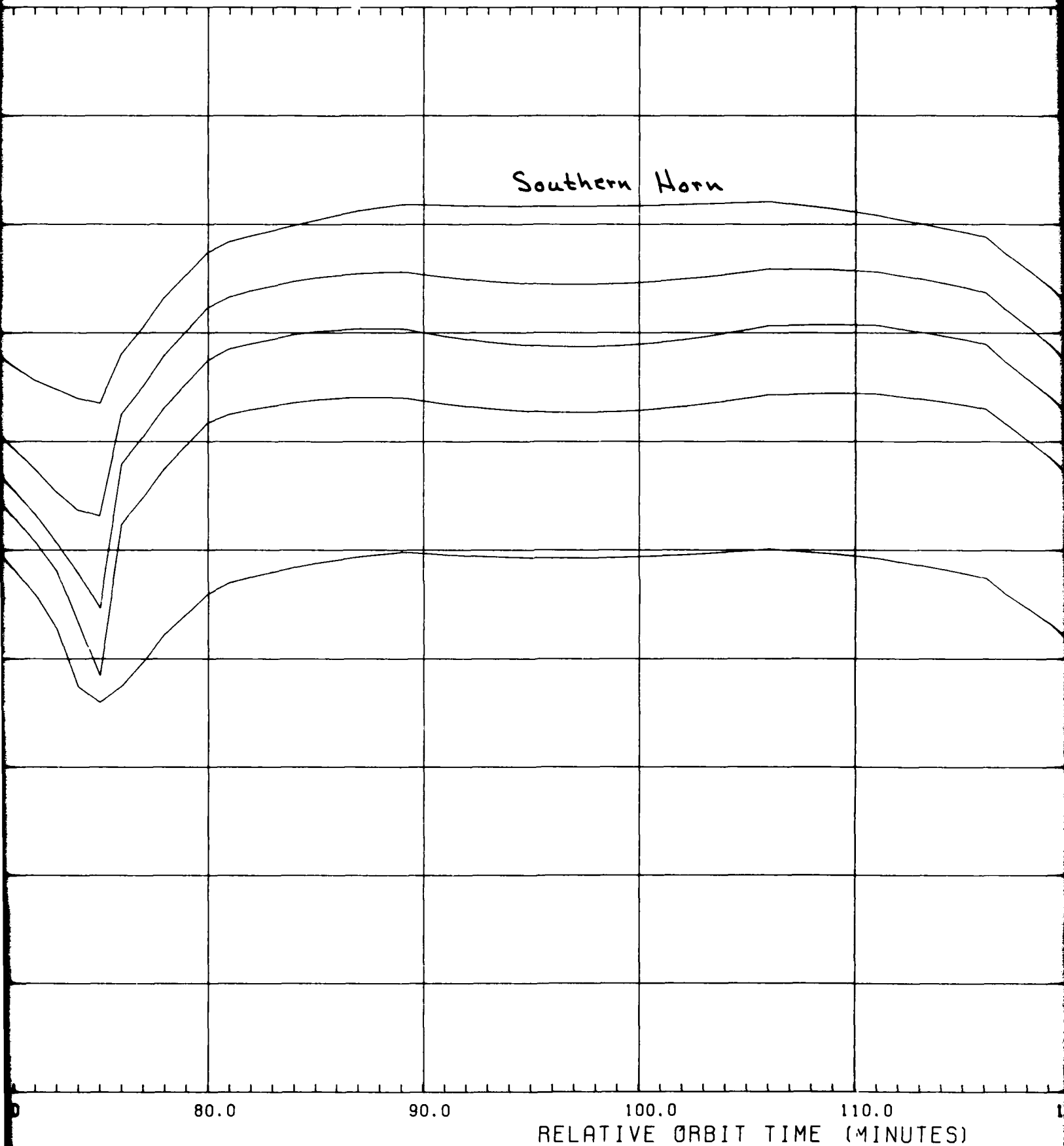
50.0

60.0

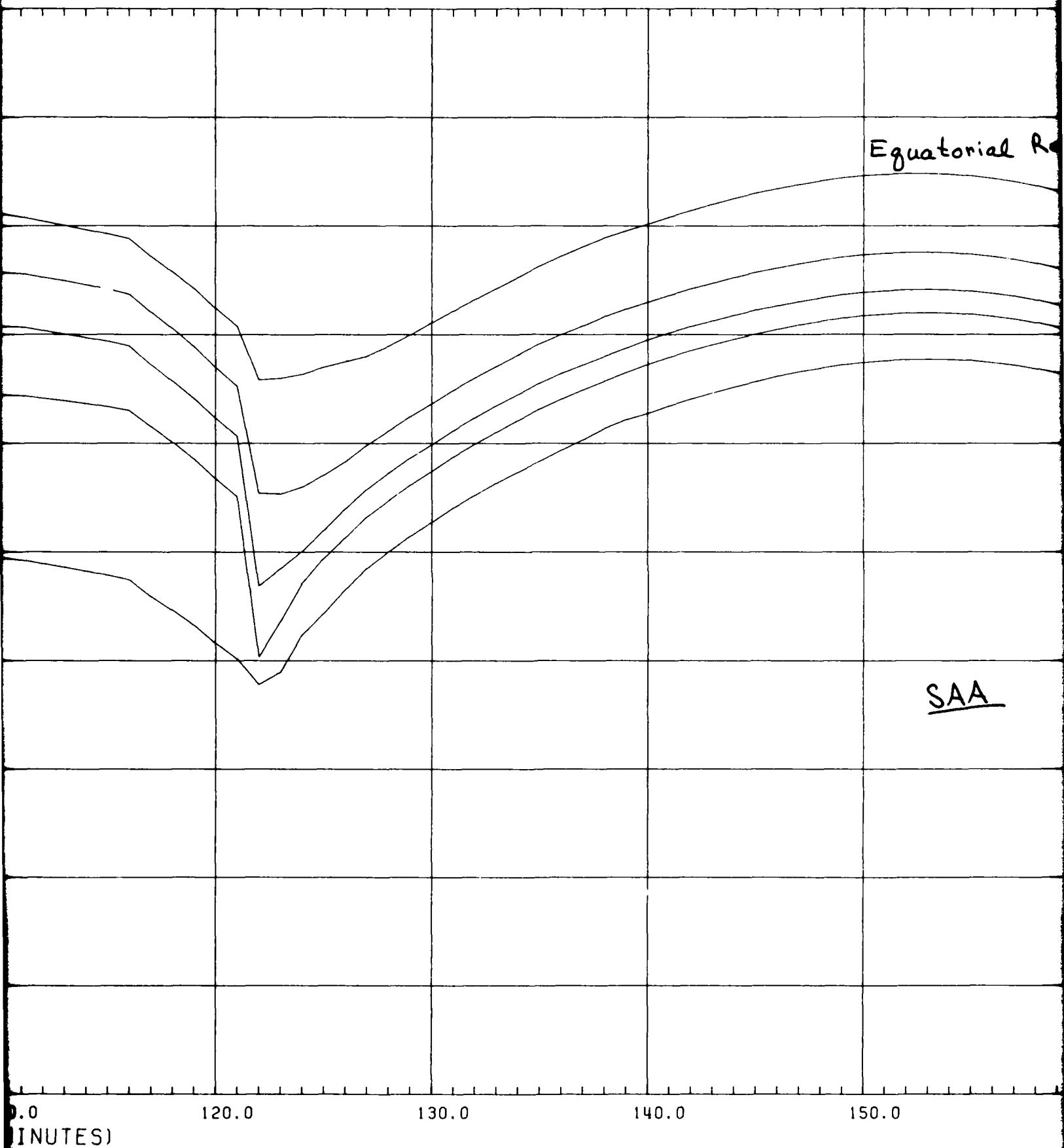
70.0

DOSE AT CENTER OF ALUMINUM SPHERES

Southern Horn



INUM SPHERES



ORBIT: NAVEL
60 DGR/518

EPOCH: 1989

MODELS:
FIELD: BARR
TRAPPED PRO
INNER ZN EL
OUTER ZN EL
MISSION DURA
EVALUATION P

UN FACTORS:

Equatorial Region

Northern
Horn

SAA

STOP TIME ON

0.0 160.0 170.0 180.0 190.0

Figure 146

ORBIT: NAVELEX 4
60 DGR/5186-5186 KM

EPOCH: 1989.5

MODELS:

FIELD: BARR/75

TRAPPED PROTONS: AP8

INNER ZN ELEC: AE6

OUTER ZN ELEC: AE17 L0

MISSION DURATION: 60.00 MO

EVALUATION PHASE: SOLAR MAX

UN FACTORS: NOT APPLIED

Northern
Horn

STOP TIME ON TAPE= 7.716662

NASA-GSFC

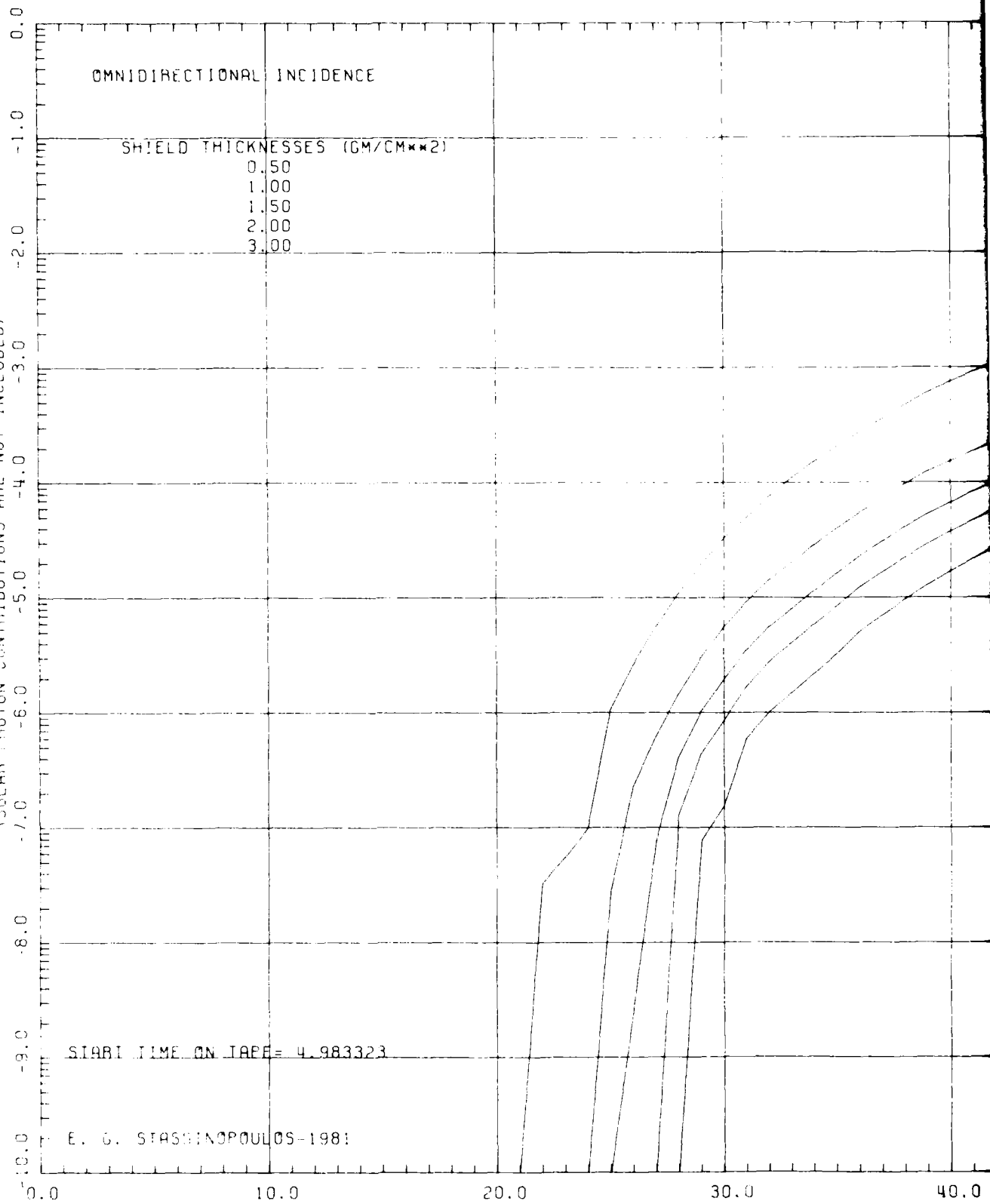
190.0

200.0

210.0

INSTANTANEOUS ALUMINUM PROTON DOSE (RADS)

(SOLAR PROTON CONTRIBUTIONS ARE NOT INCLUDED)



Equatorial Region

SAA

40.0

50.0

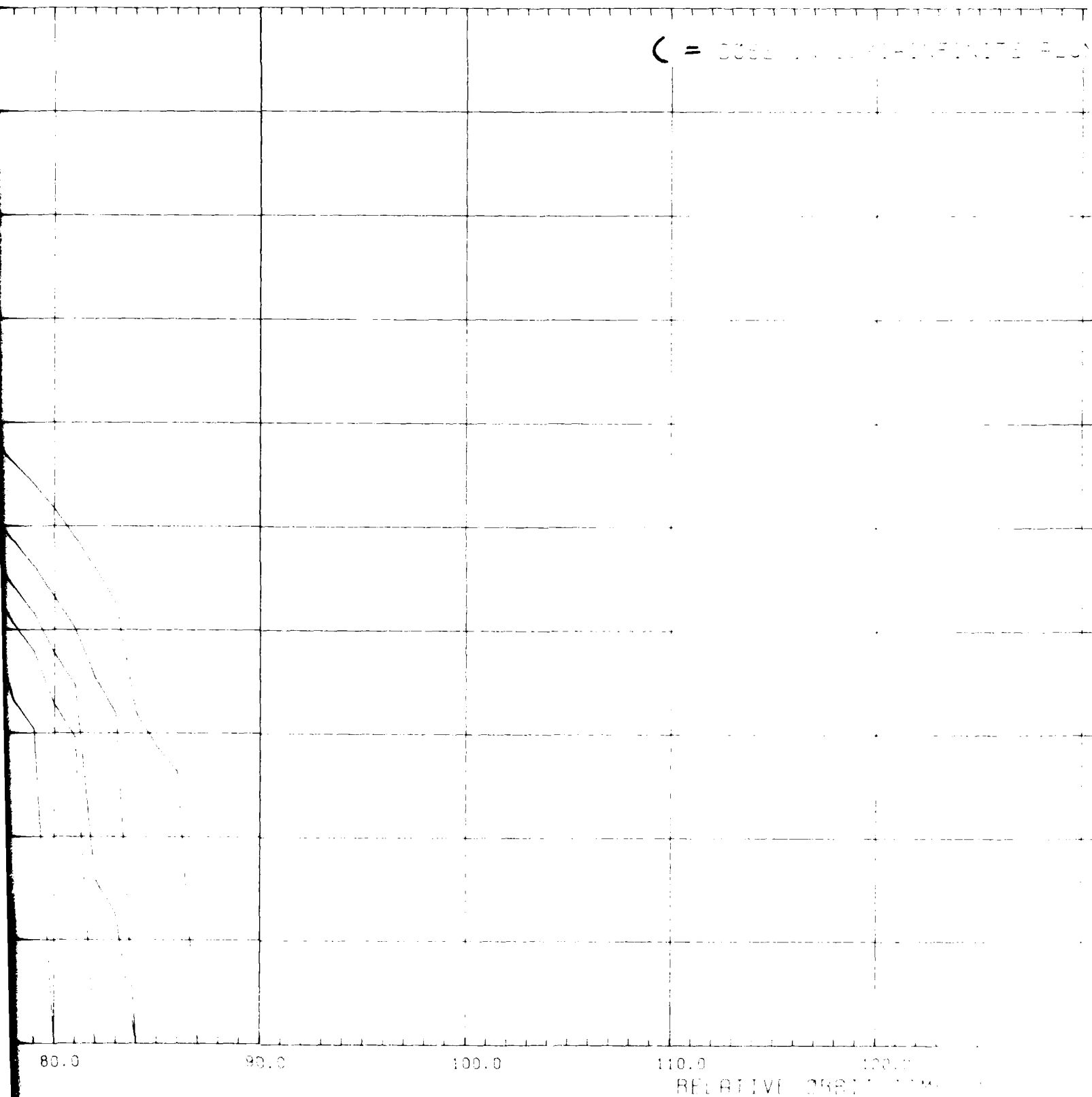
60.0

70.0

80.0

DOSE AT TRANSMISSION SURFACE OF FINITE AL

$C = \text{DOSE AT TRANSMISSION SURFACE OF FINITE AL}$



AD-A141 849

ORBITAL RADIATION STUDY FOR INCLINED CIRCULAR
TRAJECTORIES(U) NATIONAL AERONAUTICS AND SPACE
ADMINISTRATION GREENBELT MD GO... E G STASSINOPOULOS
NOV 81 NASA-GSFC-X-601-81-28

UNCLASSIFIED

F/G 22/3

NL

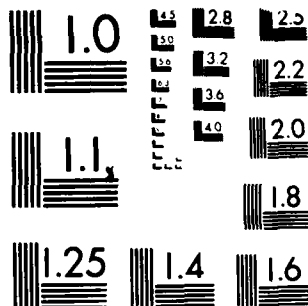
END

DATE

FORMED

7-B4

DTIC



MICROCOPY RESOLUTION TEST CHART
NATIONAL BUREAU OF STANDARDS 1963-A

4

FINITE ALUMINUM SLAB SHIELDS

FINITE ALUMINUM MEDIUM)

Equator

SAA

130.0

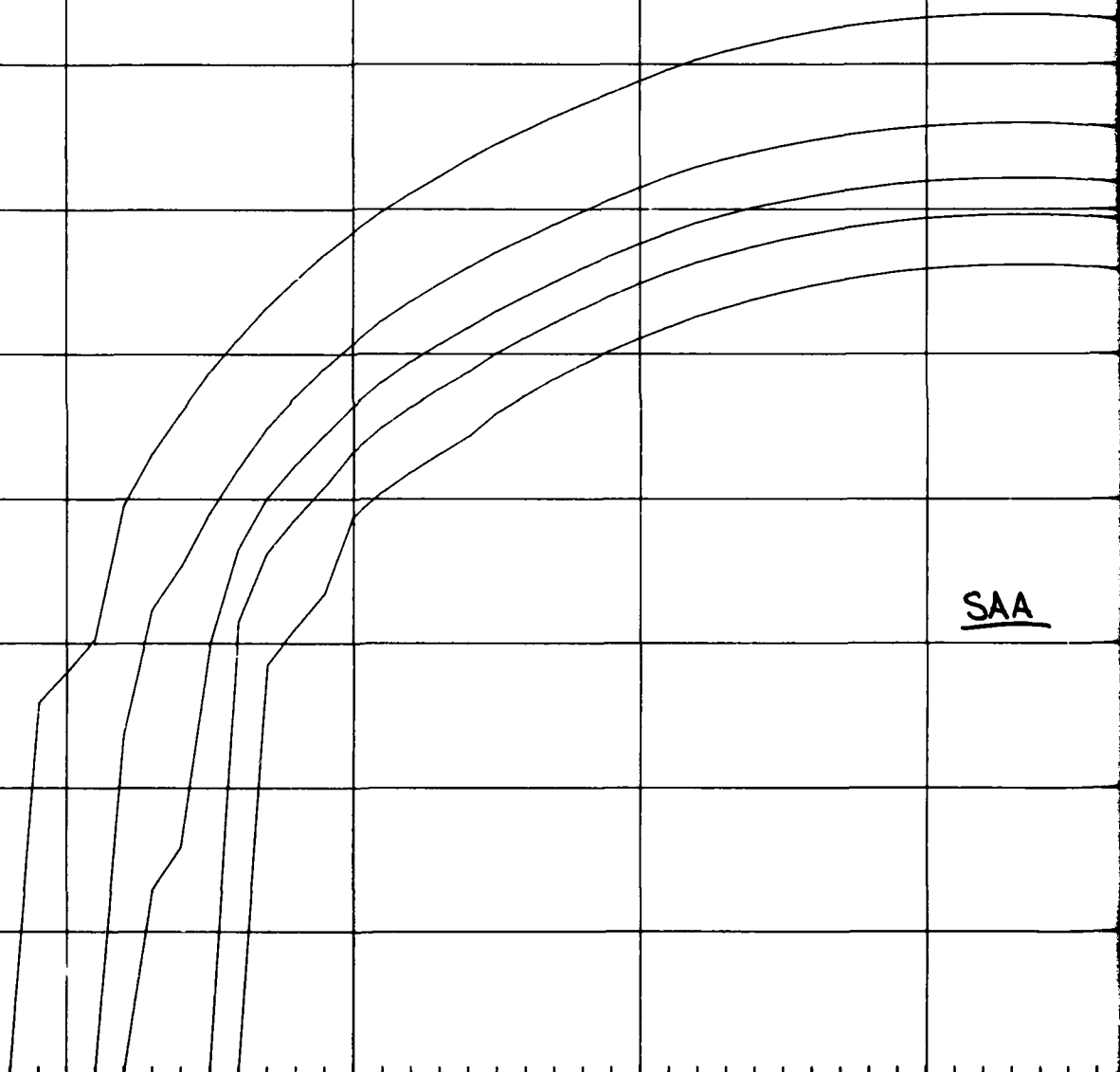
140.0

150.0

160.0

170.0

(MINUTES)



1
5

Equatorial Region

SAA

170.0

180.0

190.0

200.0

210.0

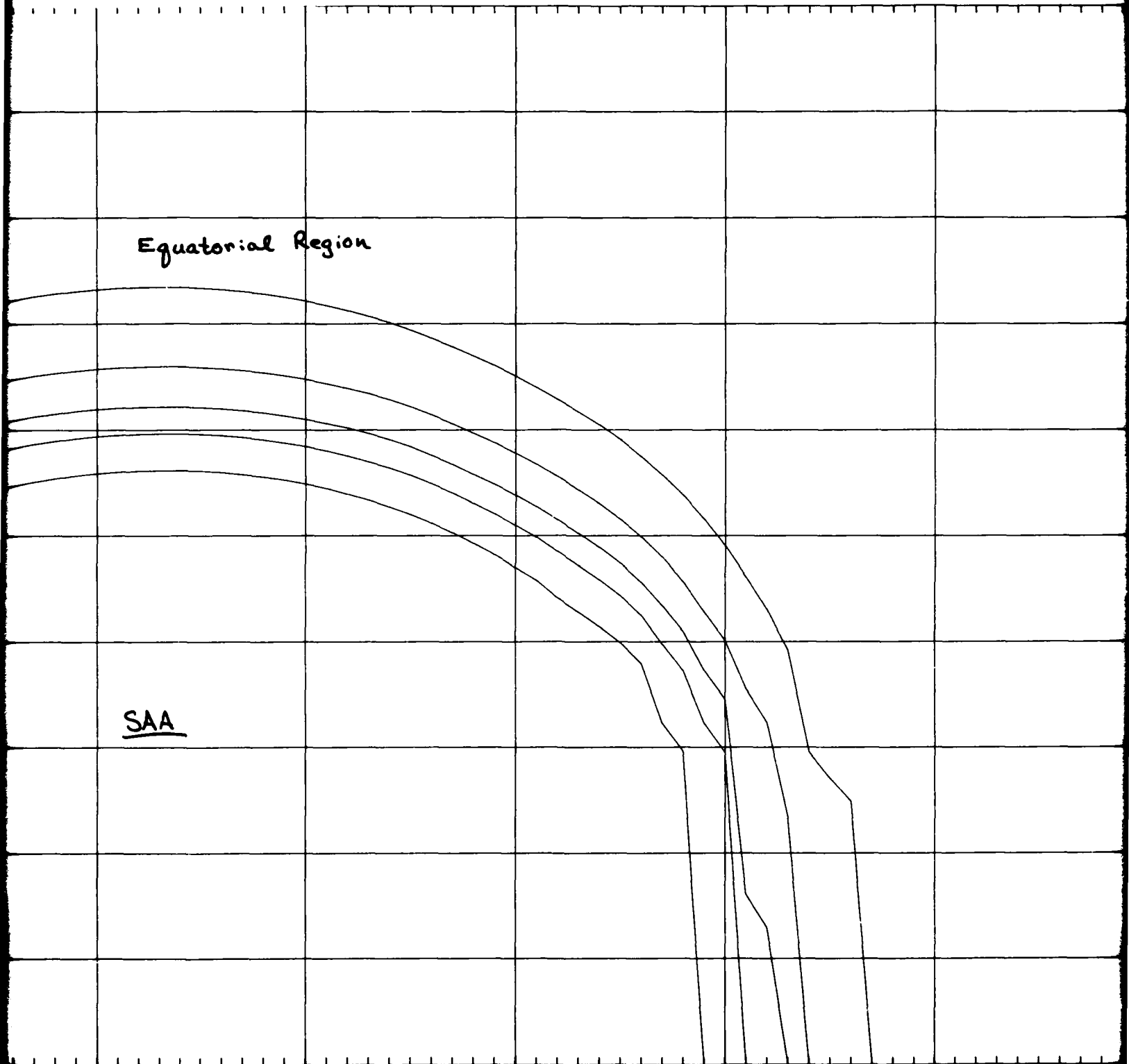


Figure 147

ORBIT: NAVELEX 5
60 DGR/6389-6389 KM

EPOCH: 1989.5

MODELS:
FIELD: BARR/75
TRAPPED PROTONS: AP8
INNER ZN ELEC: AE6
OUTER ZN ELEC: AE17 L0

MISSION DURATION: 60.00 MO
EVALUATION PHASE: SOLAR MAX

UN FACTORS: NOT APPLIED

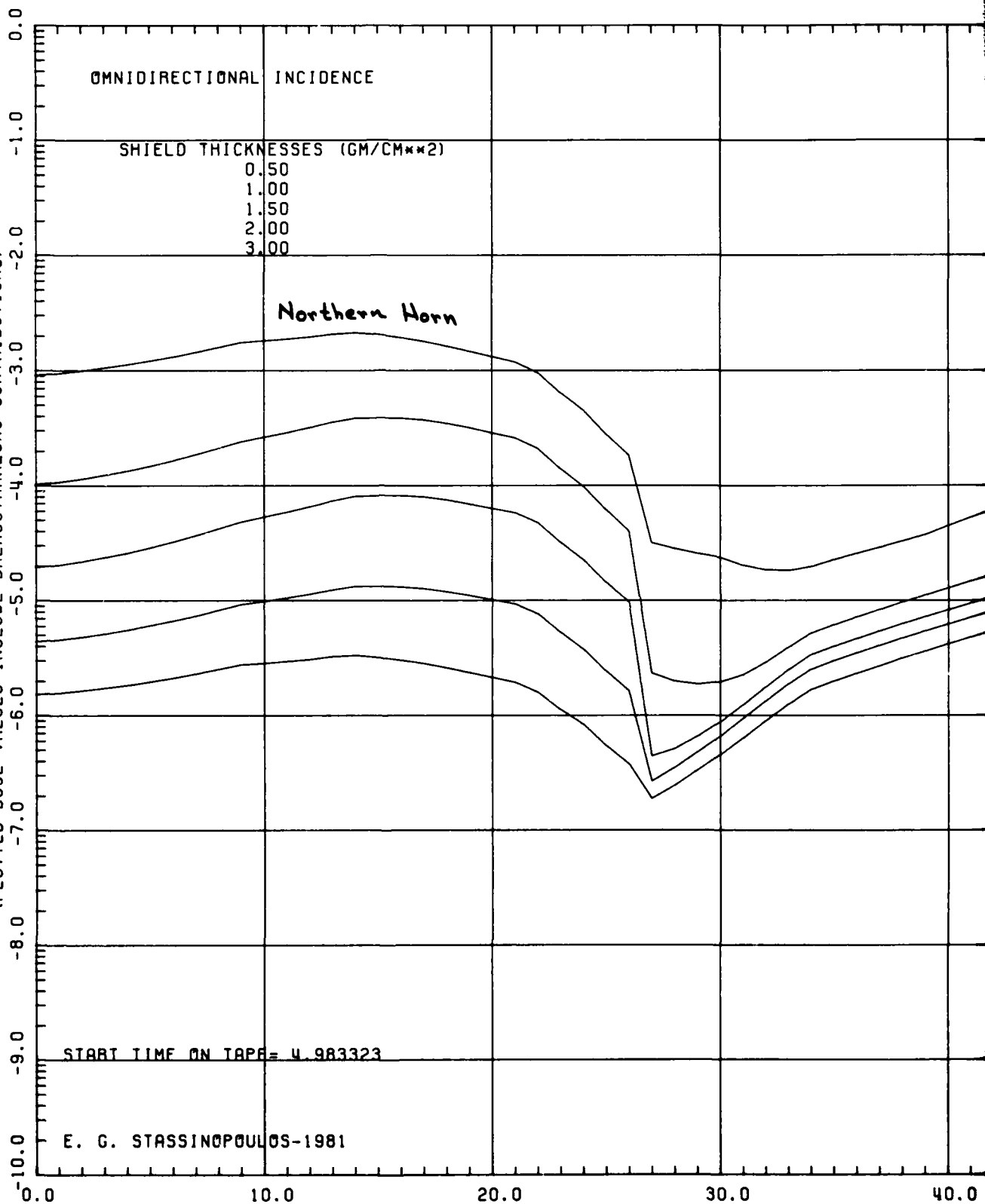
STOP TIME ON TAPE= 8.933318

NASA-GSFC

210.0 220.0 230.0 240.0

INSTANTANEOUS ALUMINUM ELECTRON DOSE (RADS)

(PLOTTED DOSE VALUES INCLUDE BREMSSTRAHLUNG CONTRIBUTIONS)



'2

Equatorial Region

SAA

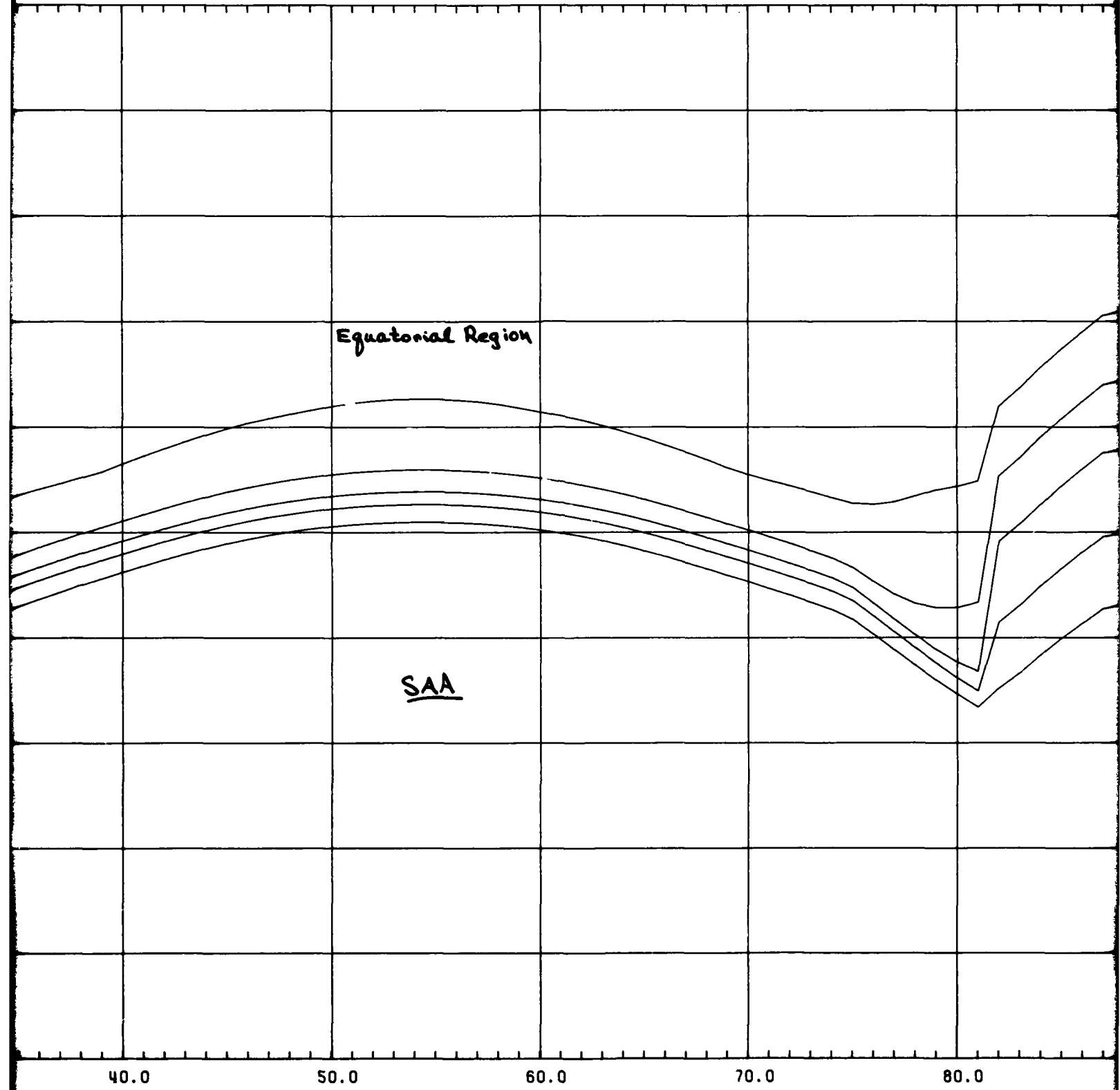
40.0

50.0

60.0

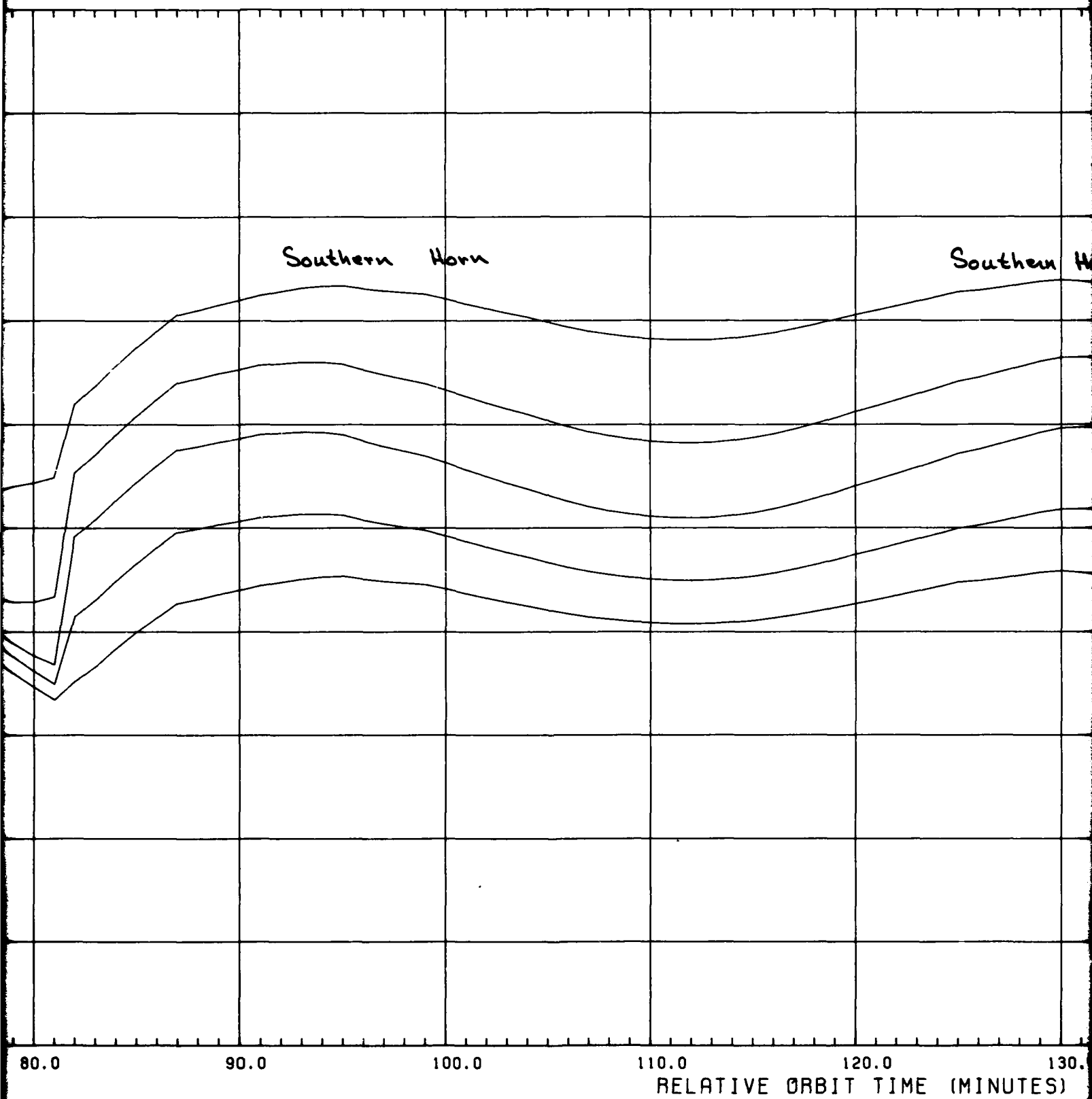
70.0

80.0



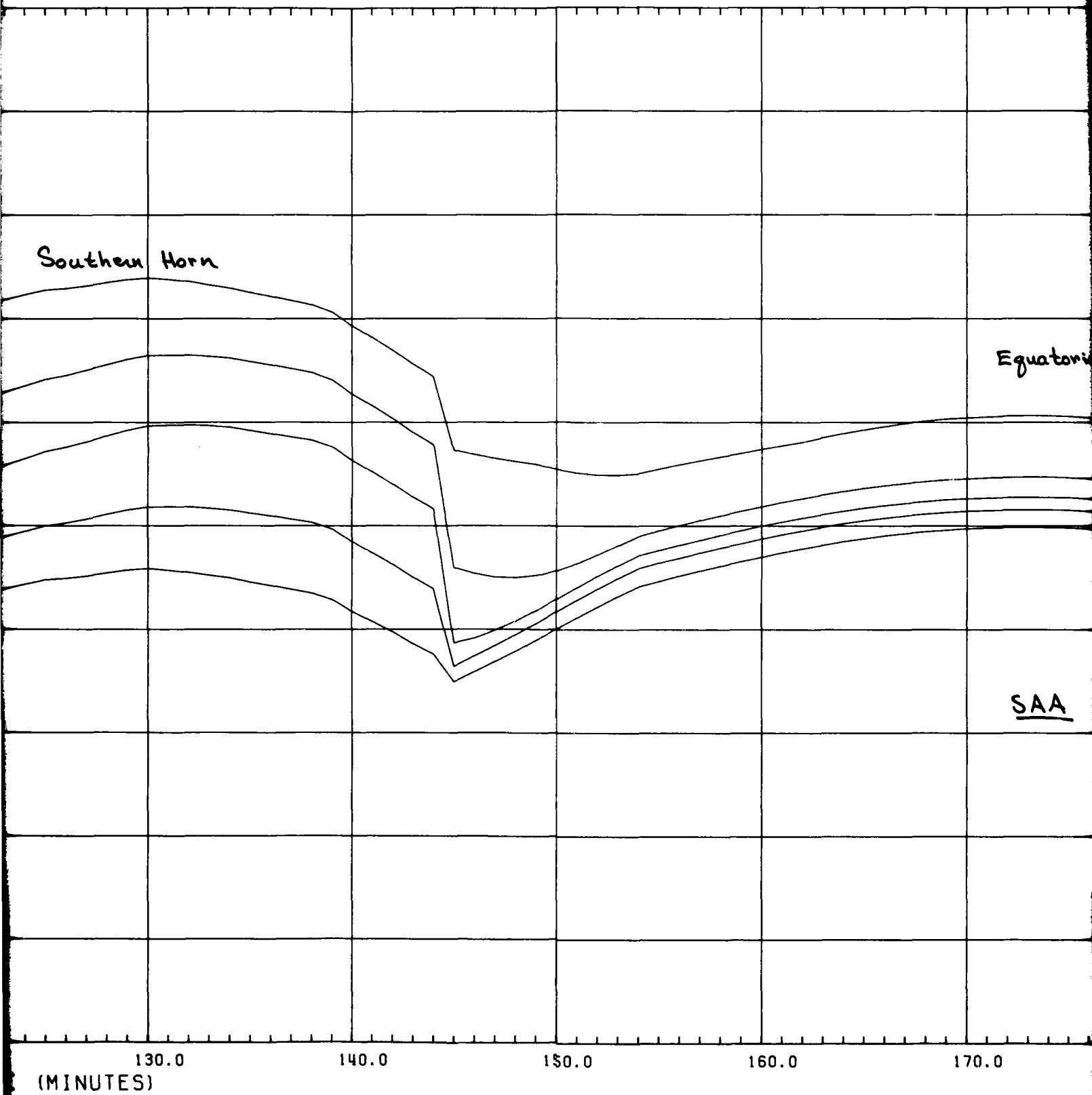
3

DOSE AT TRANSMISSION SURFACE OF FINITE ALU

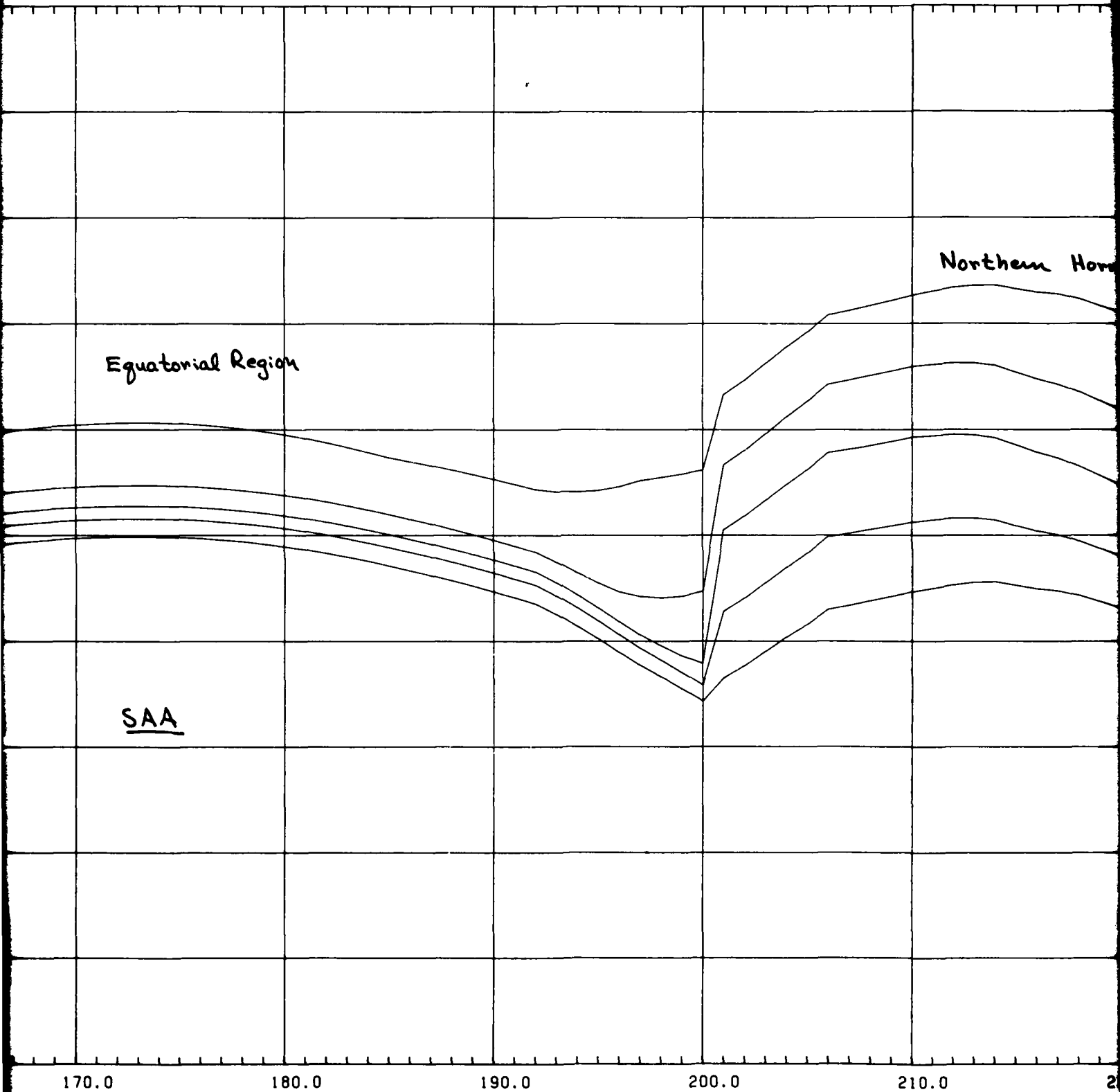


4

FINITE ALUMINUM SLAB SHIELDS



'5



D-3

6

Figure 148

ORBIT: NAVELEX 5
60 DGR/6389-6389 KM

EPOCH: 1989.5

MODELS:

FIELD: BARR/75

TRAPPED PROTONS: AP8

INNER ZN ELEC: AE6

OUTER ZN ELEC: AE17 L0

MISSION DURATION: 60.00 MO

EVALUATION PHASE: SOLAR MAX

UN FACTORS: NOT APPLIED

Northern Horn

STOP TIME ON TAPE= 8.933318

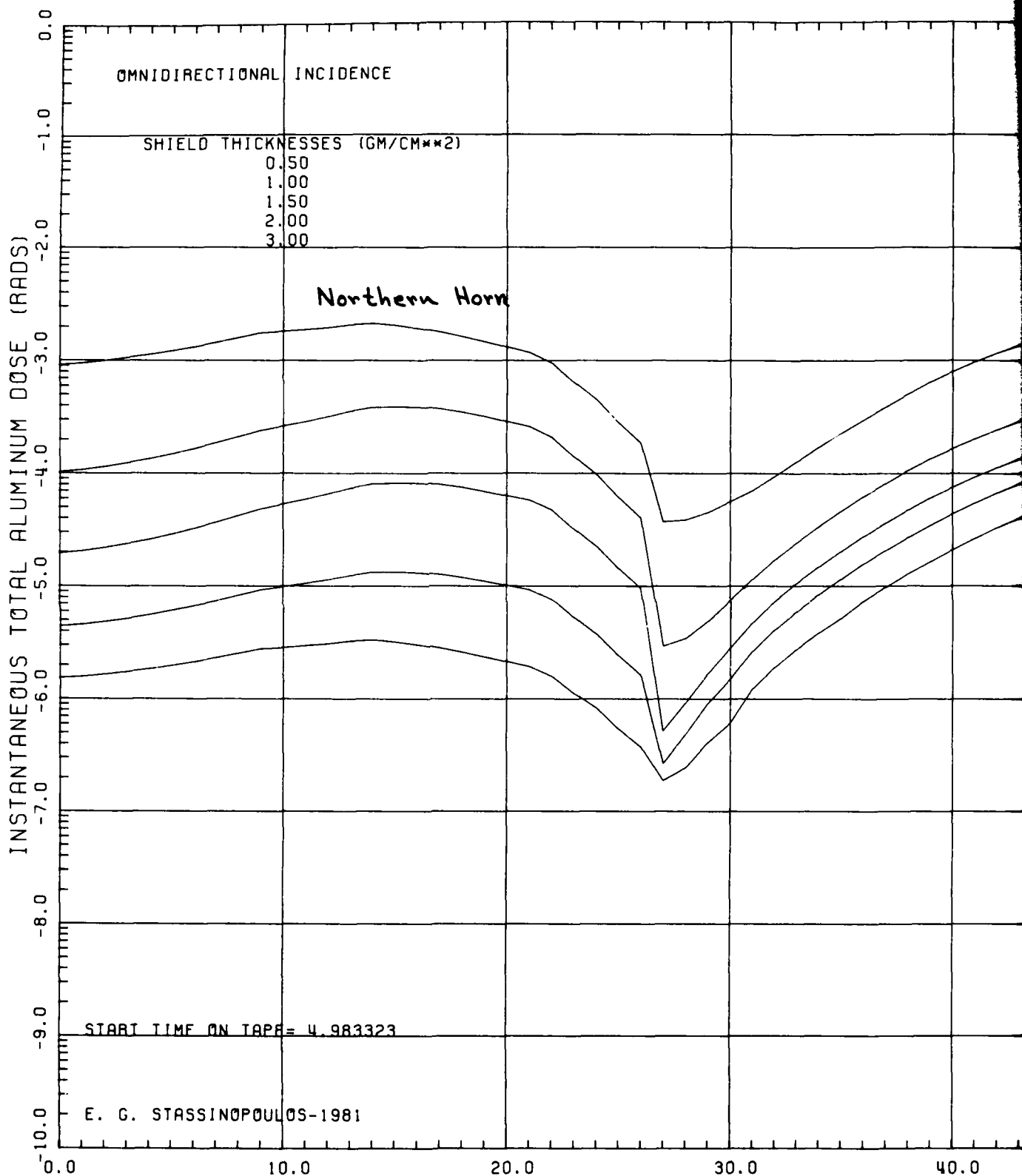
NASA-GSFC

210.0

220.0

230.0

240.0



'2

Equatorial Region

SAA

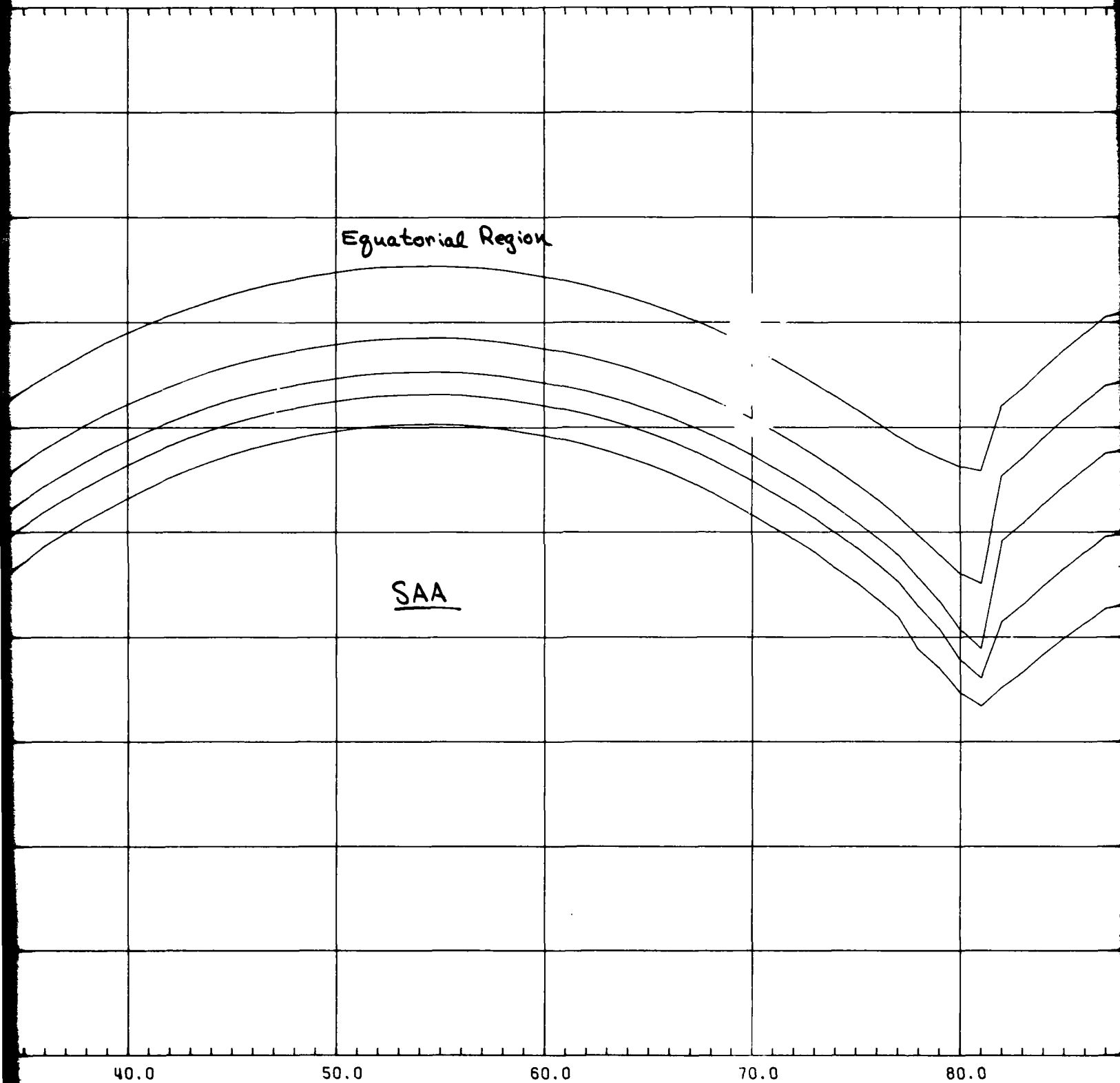
40.0

50.0

60.0

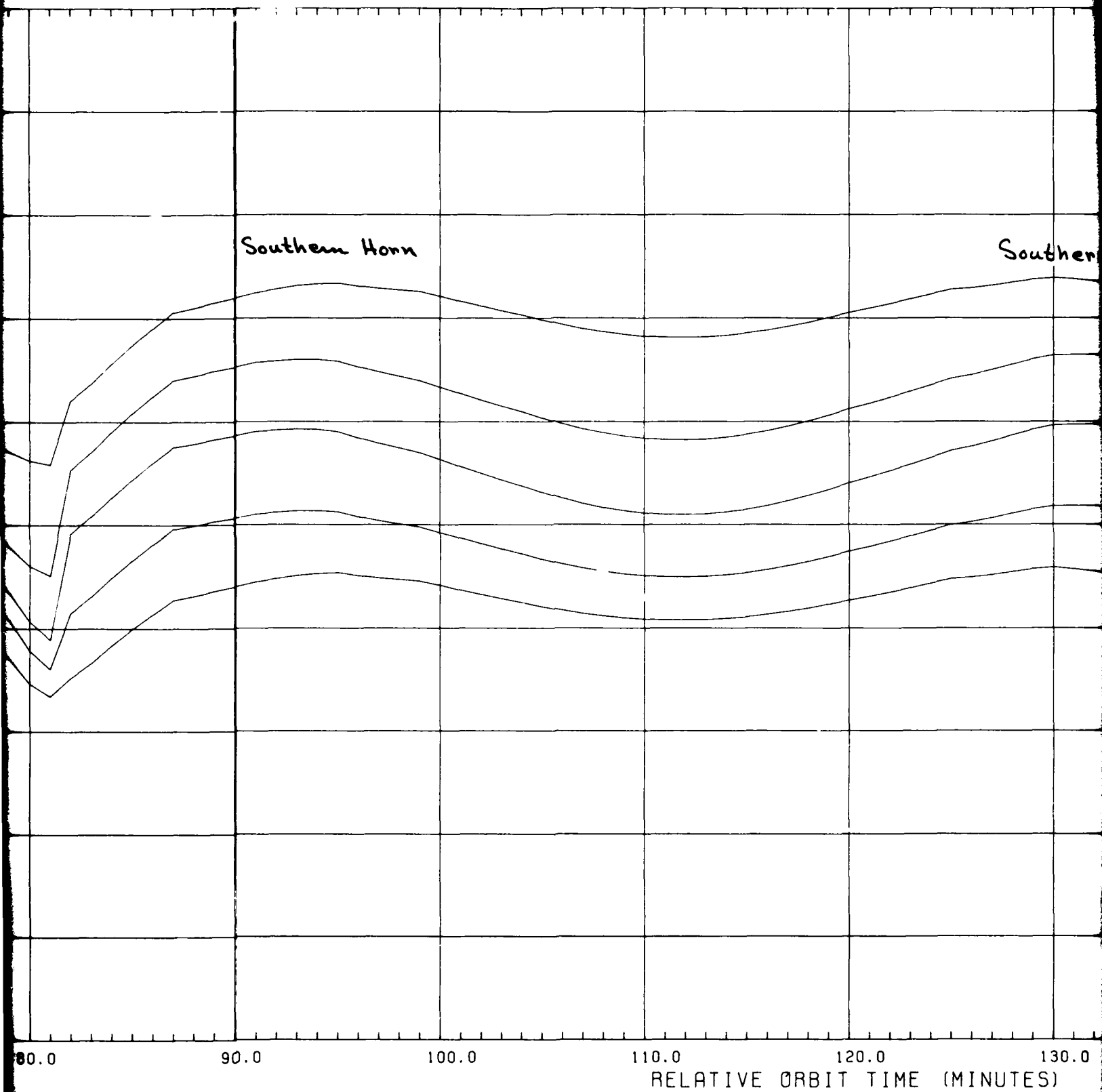
70.0

80.0



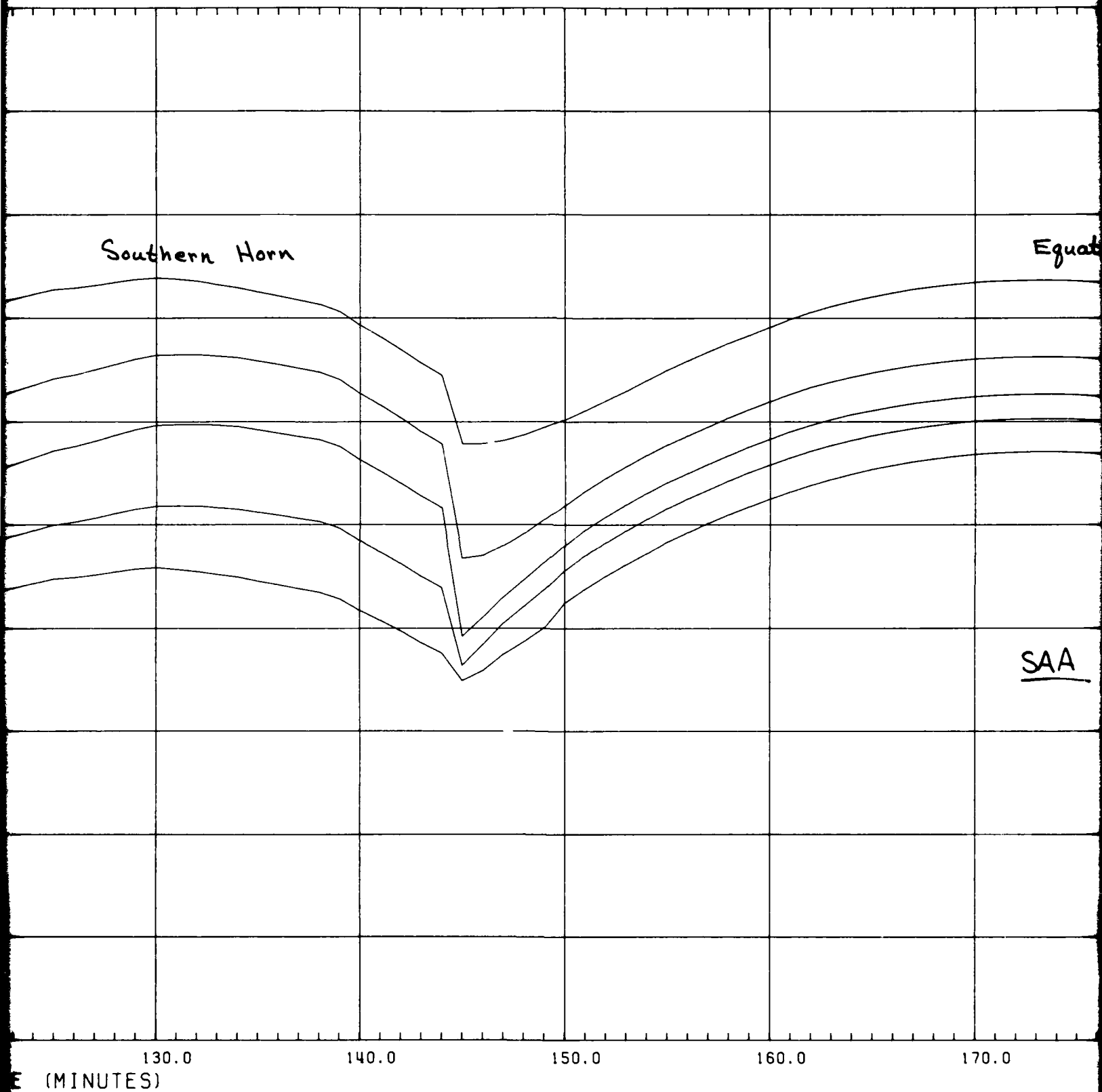
3

DOSE AT TRANSMISSION SURFACE OF FINITE ALUM



'4

F FINITE ALUMINUM SLAB SHIELDS



E (MINUTES)

5

Equatorial Region

Northern Horn

SAA

170.0

180.0

190.0

200.0

210.0

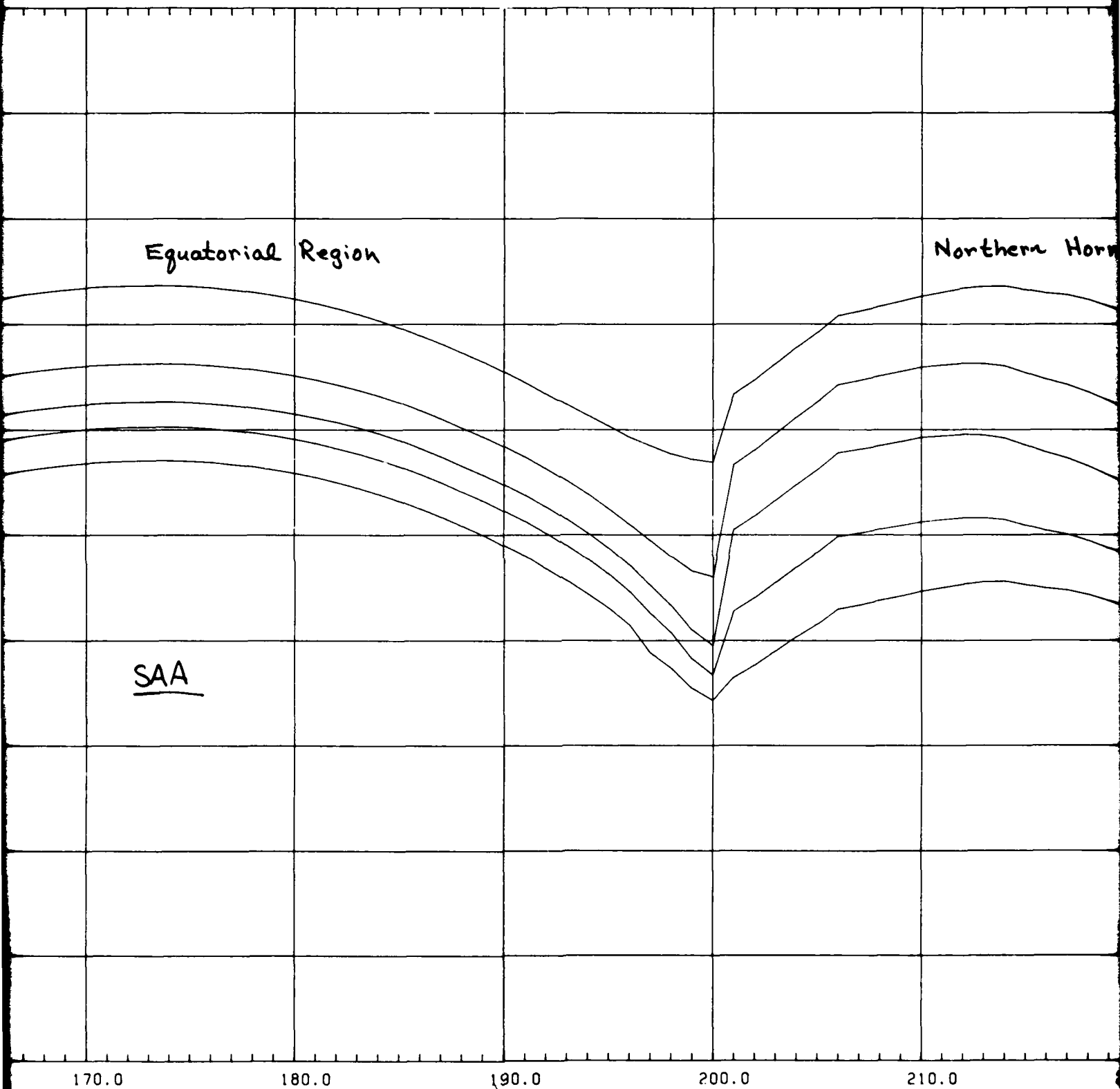


Figure 149

ORBIT: NAVELEX 5
60 DGR/6389-6389 KM

EPOCH: 1989.5

MODELS:
FIELD: BARR/75
TRAPPED PROTONS: AP8
INNER ZN ELEC: AE6
OUTER ZN ELEC: AE17 L0

MISSION DURATION: 60.00 MO
EVALUATION PHASE: SOLAR MAX

UN FACTORS: NOT APPLIED

Northern Horn

STOP TIME ON TAPF= 8.933318

NASA-GSFC

210.0

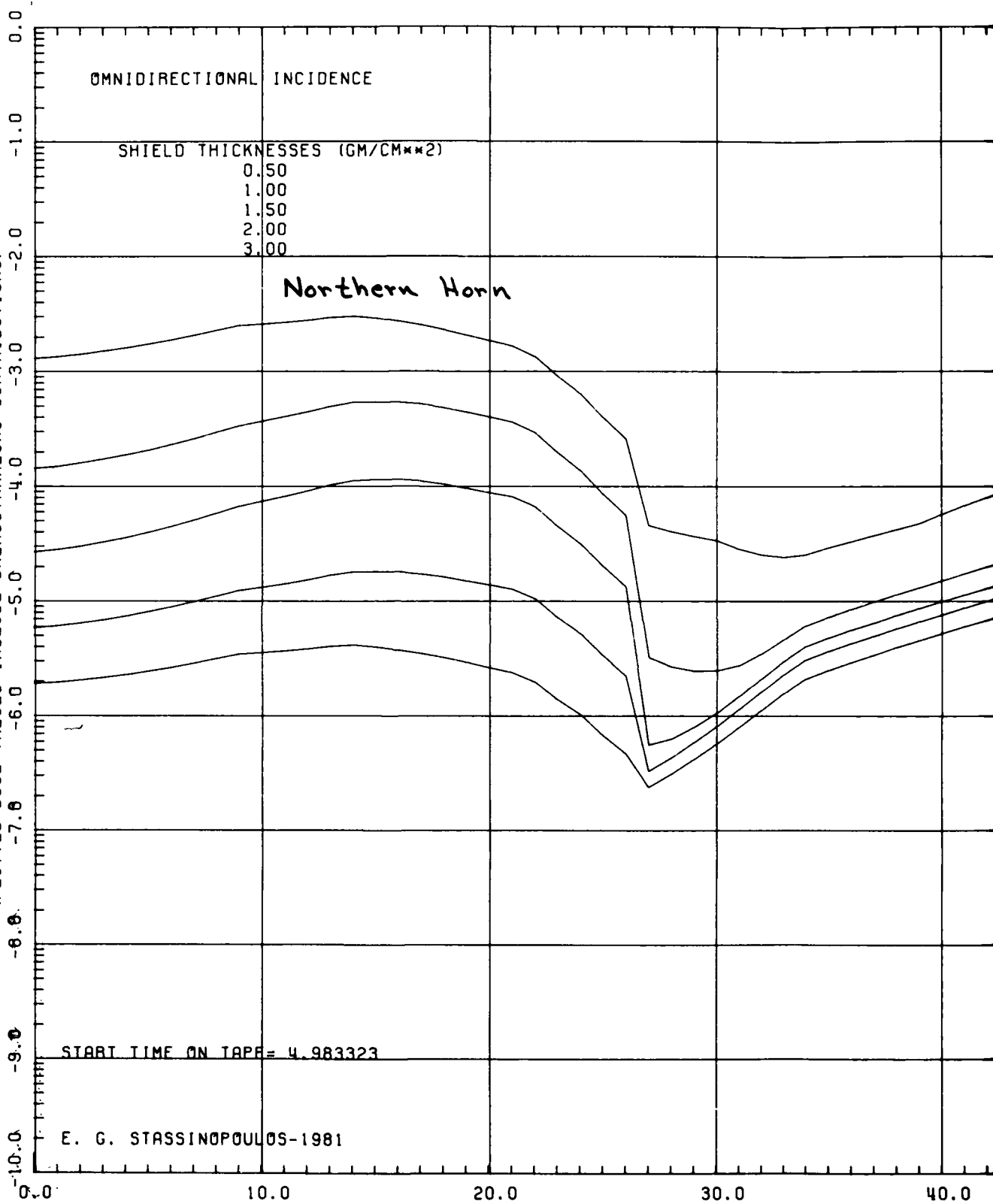
220.0

230.0

240.0

INSTANTANEOUS ALUMINUM ELECTRON DOSE (RADS)

(PLOTTED DOSE VALUES INCLUDE BREMSSTRAHLUNG CONTRIBUTIONS)



2

Equatorial Region

SAA

40.0

50.0

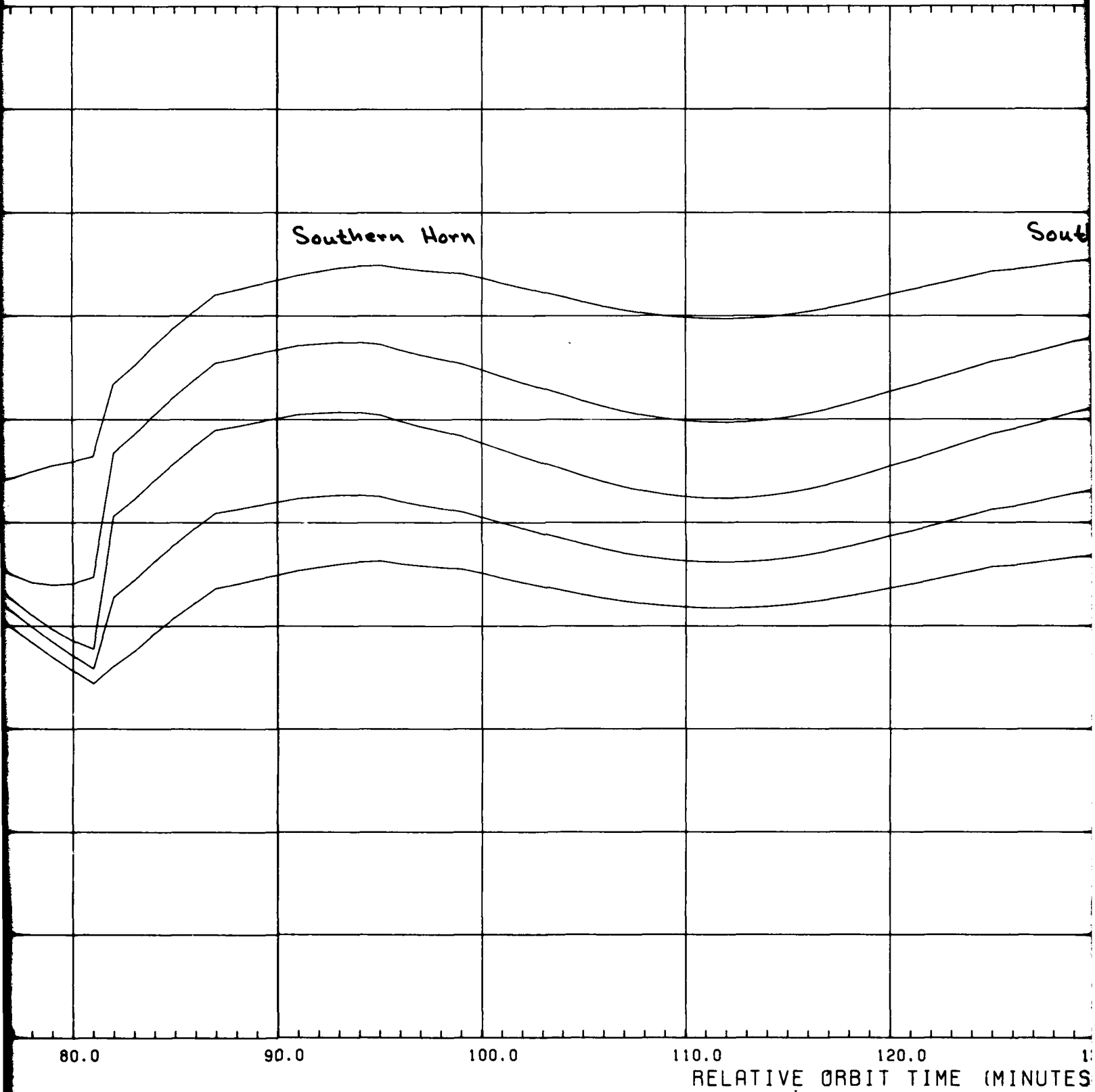
60.0

70.0

80.0

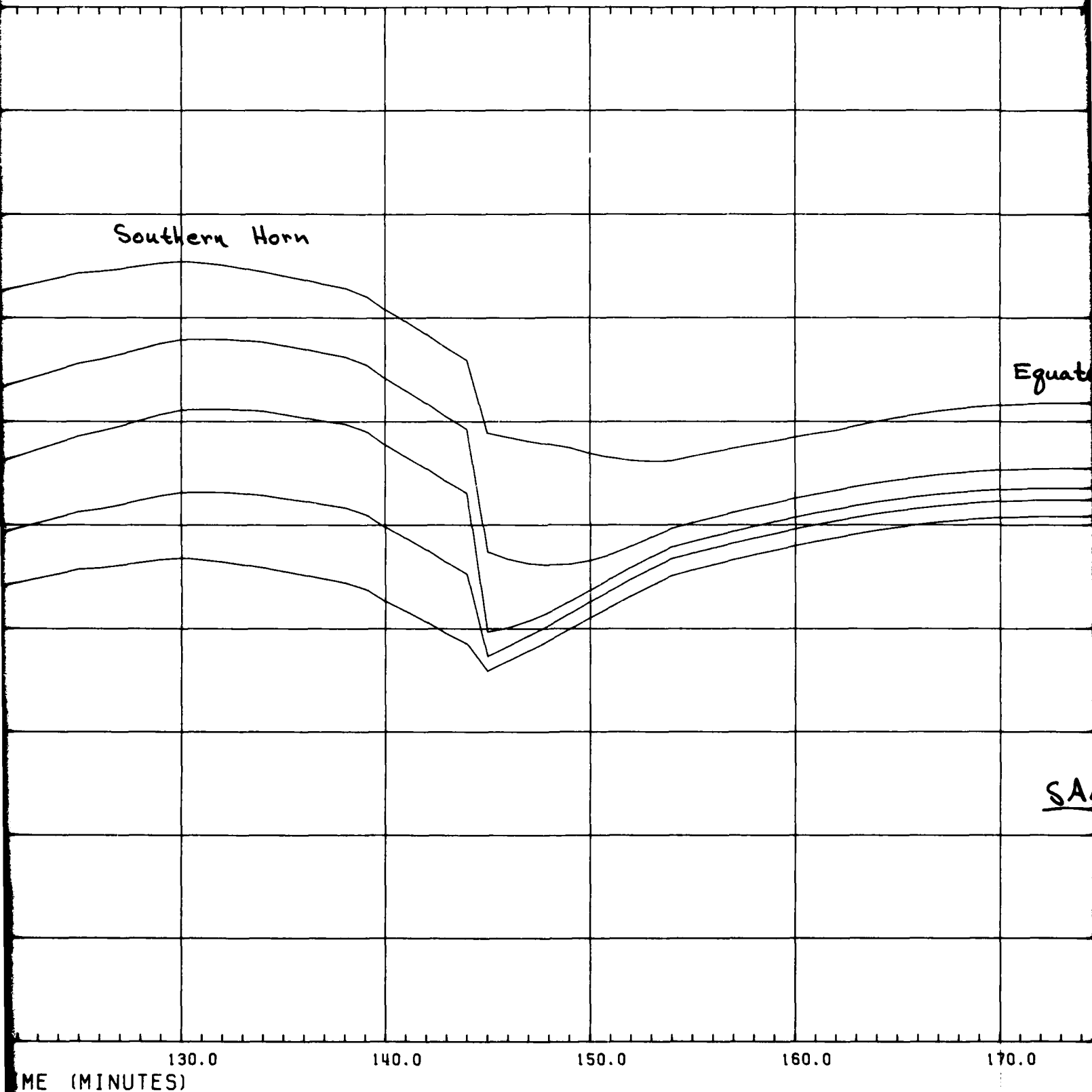
3

DOSE IN SEMI-INFINITE ALUMINUM



4

ITE ALUMINUM MEDIUM



5

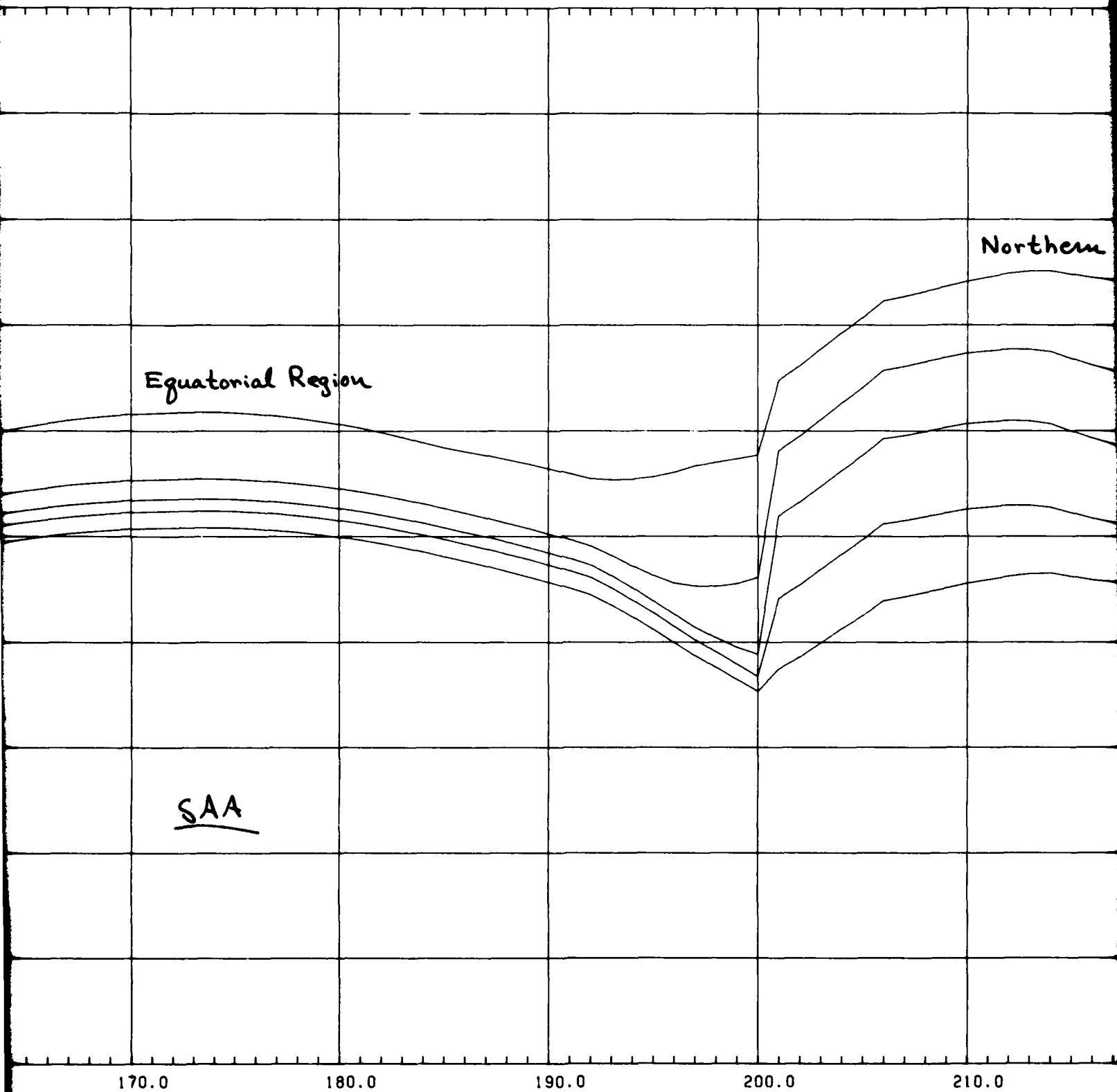


Figure 150

ORBIT: NAVELEX 5
60 DGR/6389-6389 KM

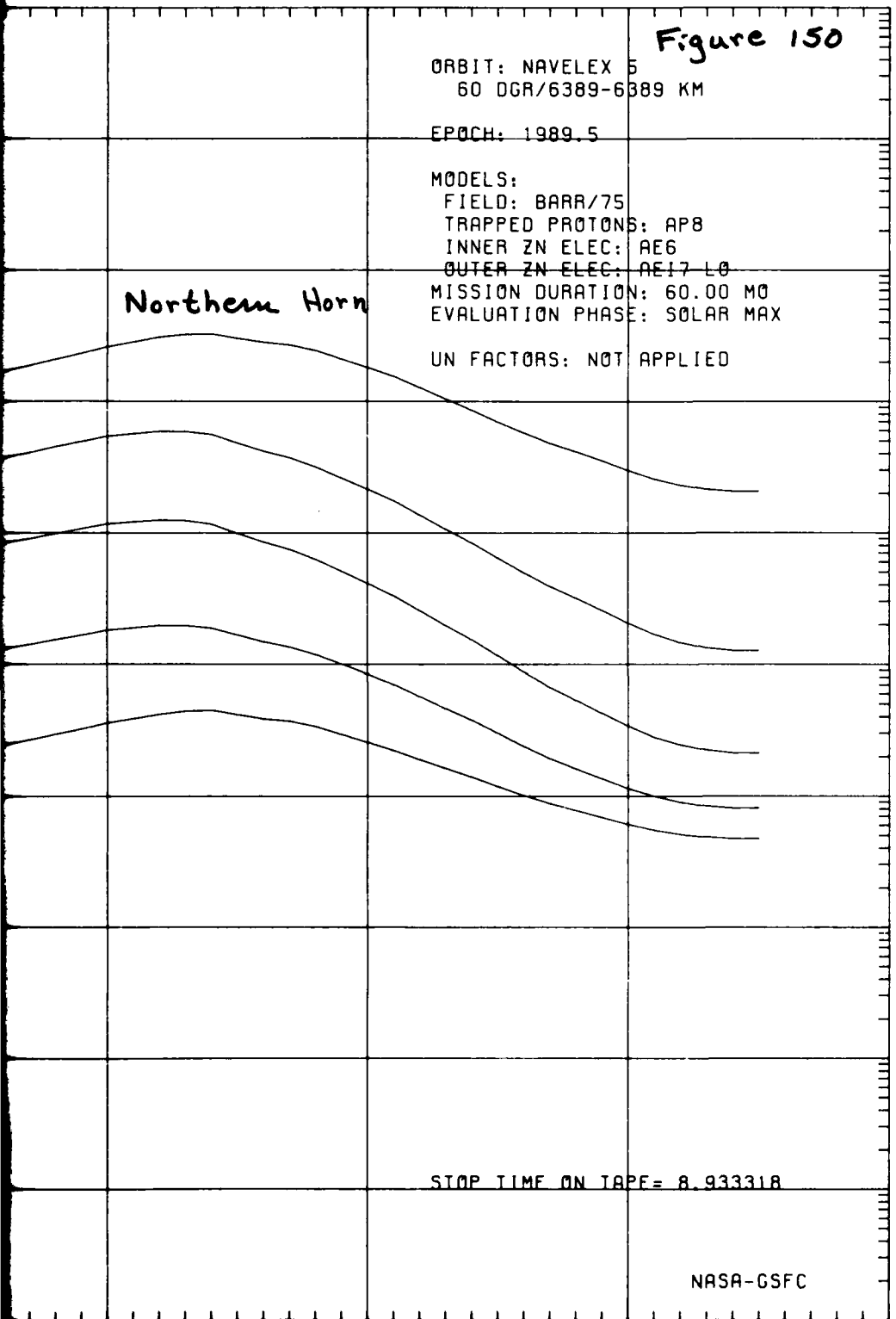
EPOCH: 1989.5

MODELS:
FIELD: BARR/75
TRAPPED PROTONS: AP8
INNER ZN ELEC: AE6
OUTER ZN ELEC: AE17 L0

MISSION DURATION: 60.00 MO
EVALUATION PHASE: SOLAR MAX

UN FACTORS: NOT APPLIED

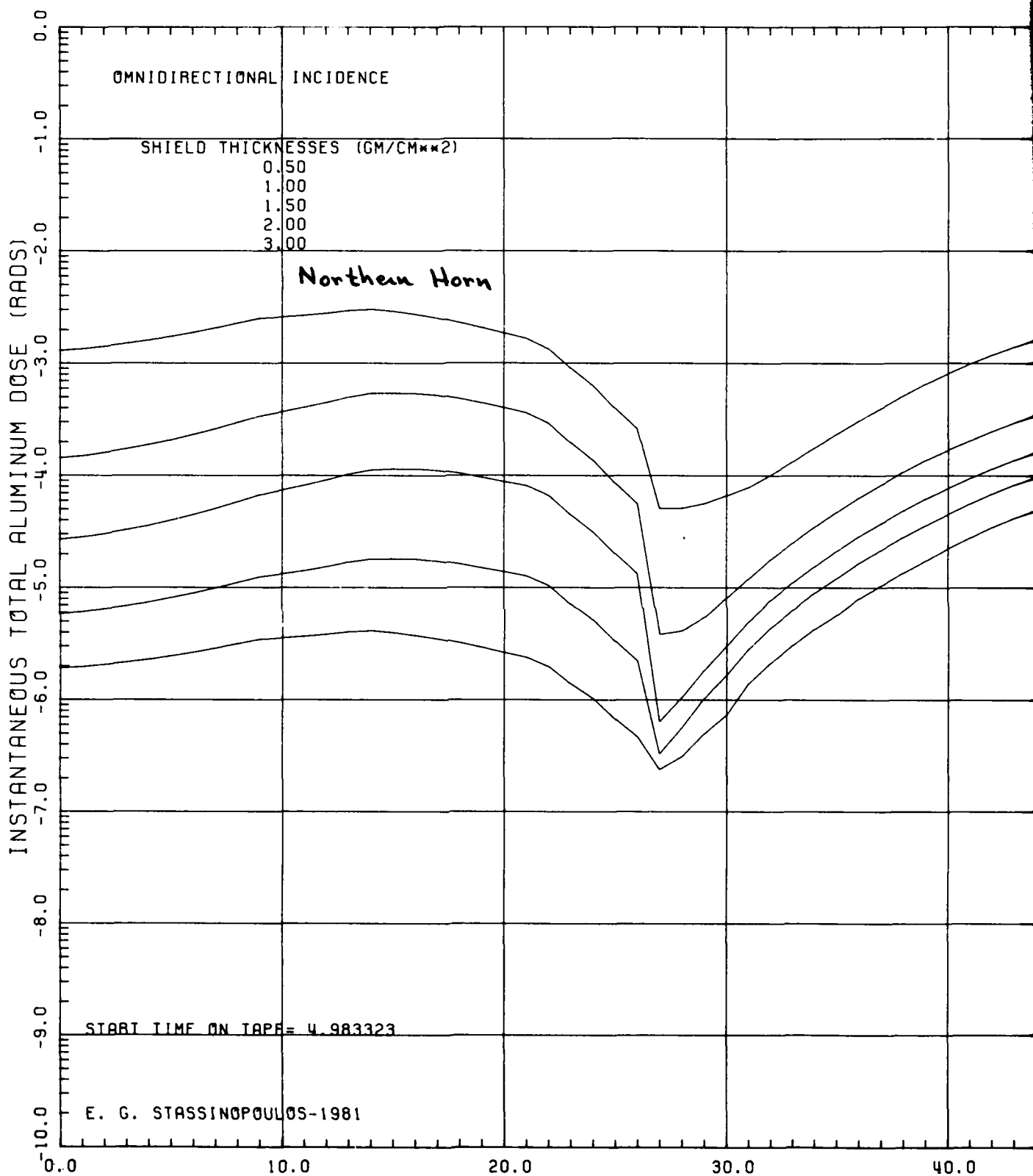
Northern Horn



STOP TIME ON TAPE = 8.933318

NASA-GSFC

210.0 220.0 230.0 240.0



2'

Equatorial Region

SAA

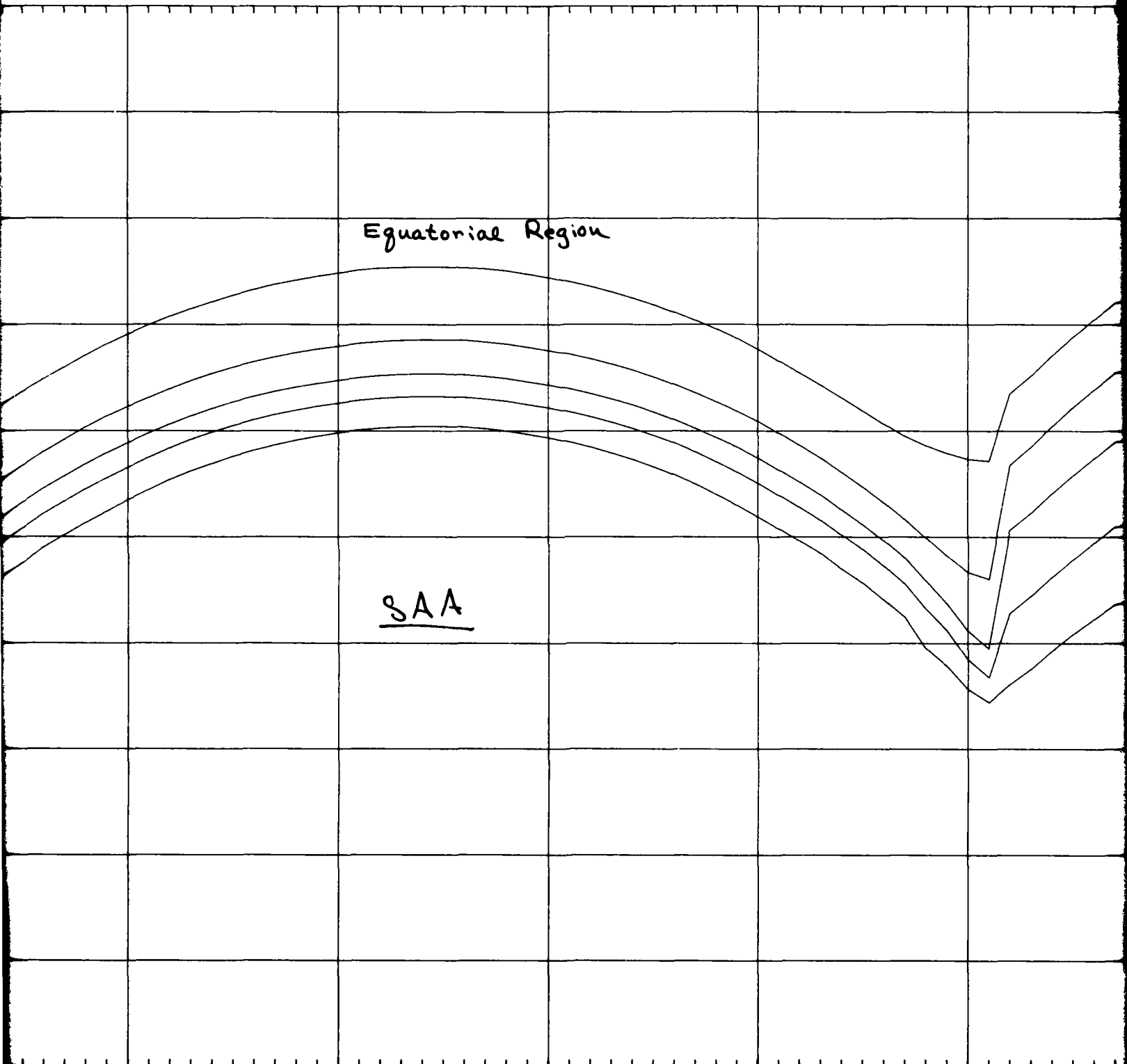
40.0

50.0

60.0

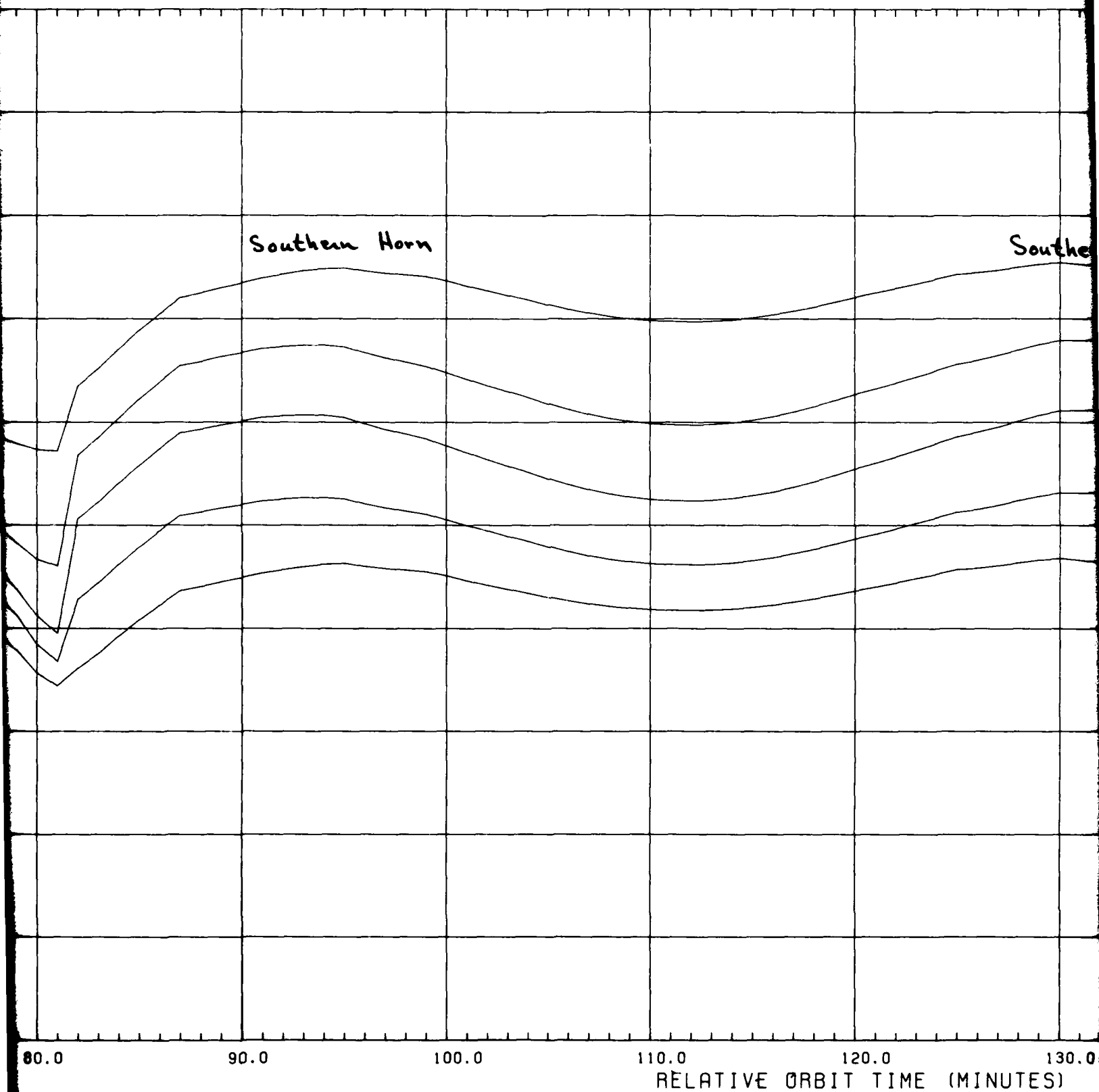
70.0

80.0



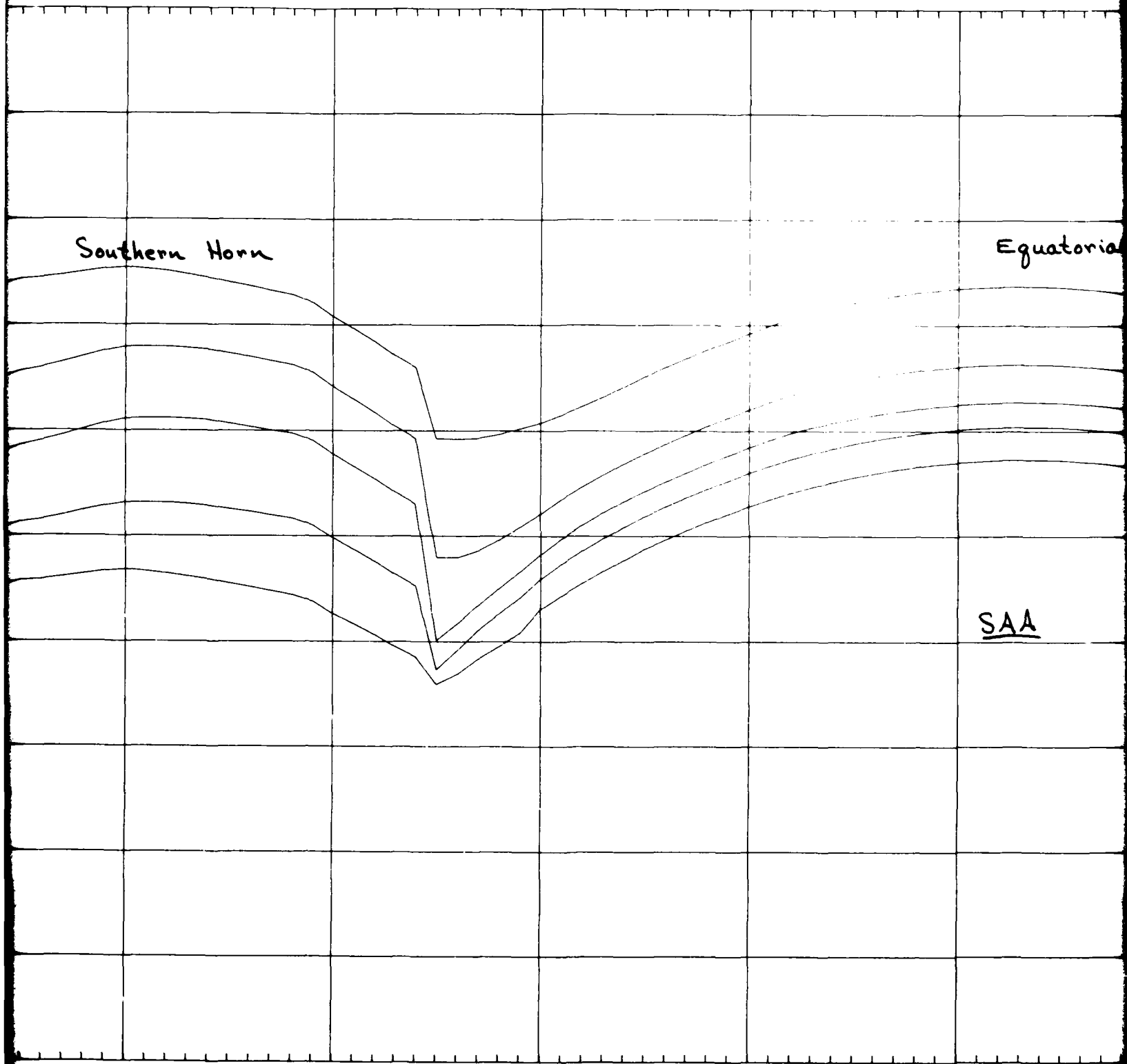
3

DOSE IN SEMI-INFINITE ALUMINUM M



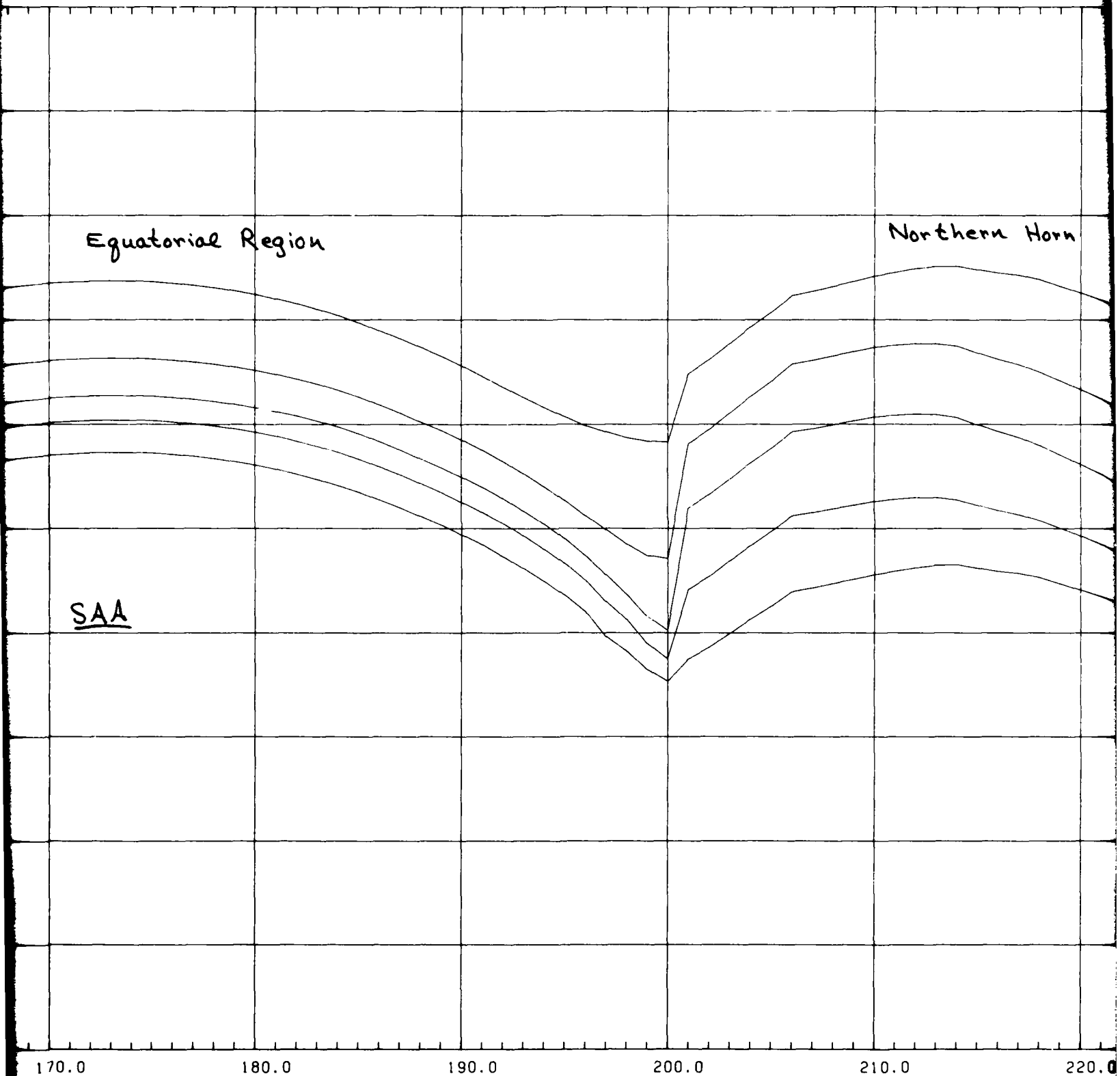
4

ALUMINUM MEDIUM



130.0 140.0 150.0 160.0 170.0
(MINUTES)

51



Equatorial Region

Northern Horn

SAA

170.0 180.0 190.0 200.0 210.0 220.0

Figure 151

ORBIT: NAVELEX 5
60 DGR/6389-6389 KM

EPOCH: 1989.5

MODELS:

FIELD: BARR/75

TRAPPED PROTONS: AP8

INNER ZN ELEC: AE6

OUTER ZN ELEC: AE17 L0

MISSION DURATION: 60.00 MO

EVALUATION PHASE: SOLAR MAX

UN FACTORS: NOT APPLIED

Northern Horn

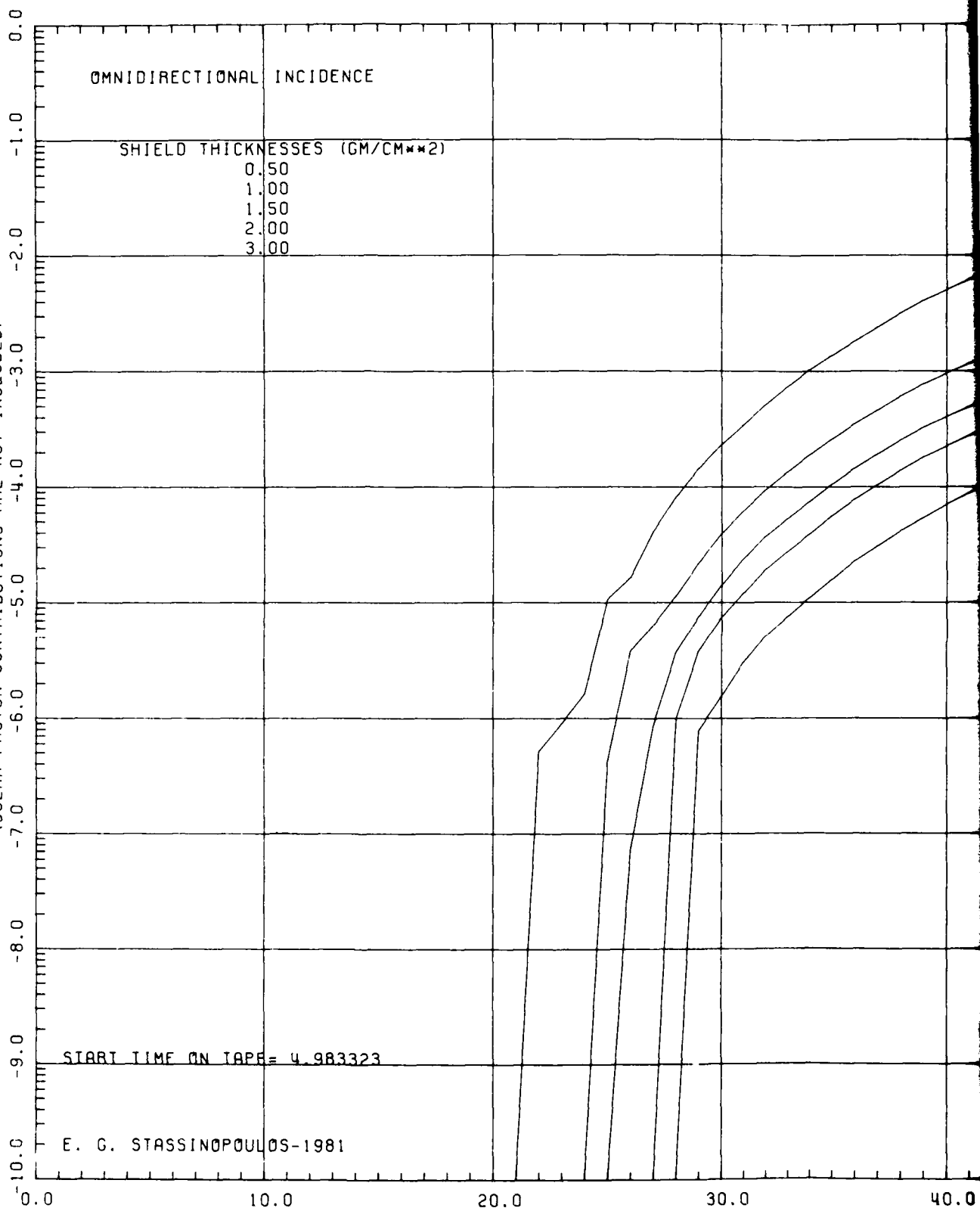
STOP TIME ON TAPE = 8.933318

NASA-GSFC

210.0 220.0 230.0 240.0

INSTANTANEOUS ALUMINUM PROTON DOSE (RADS)

(SOLAR PROTON CONTRIBUTIONS ARE NOT INCLUDED)



1
2

Equatorial Region

SAA

40.0

50.0

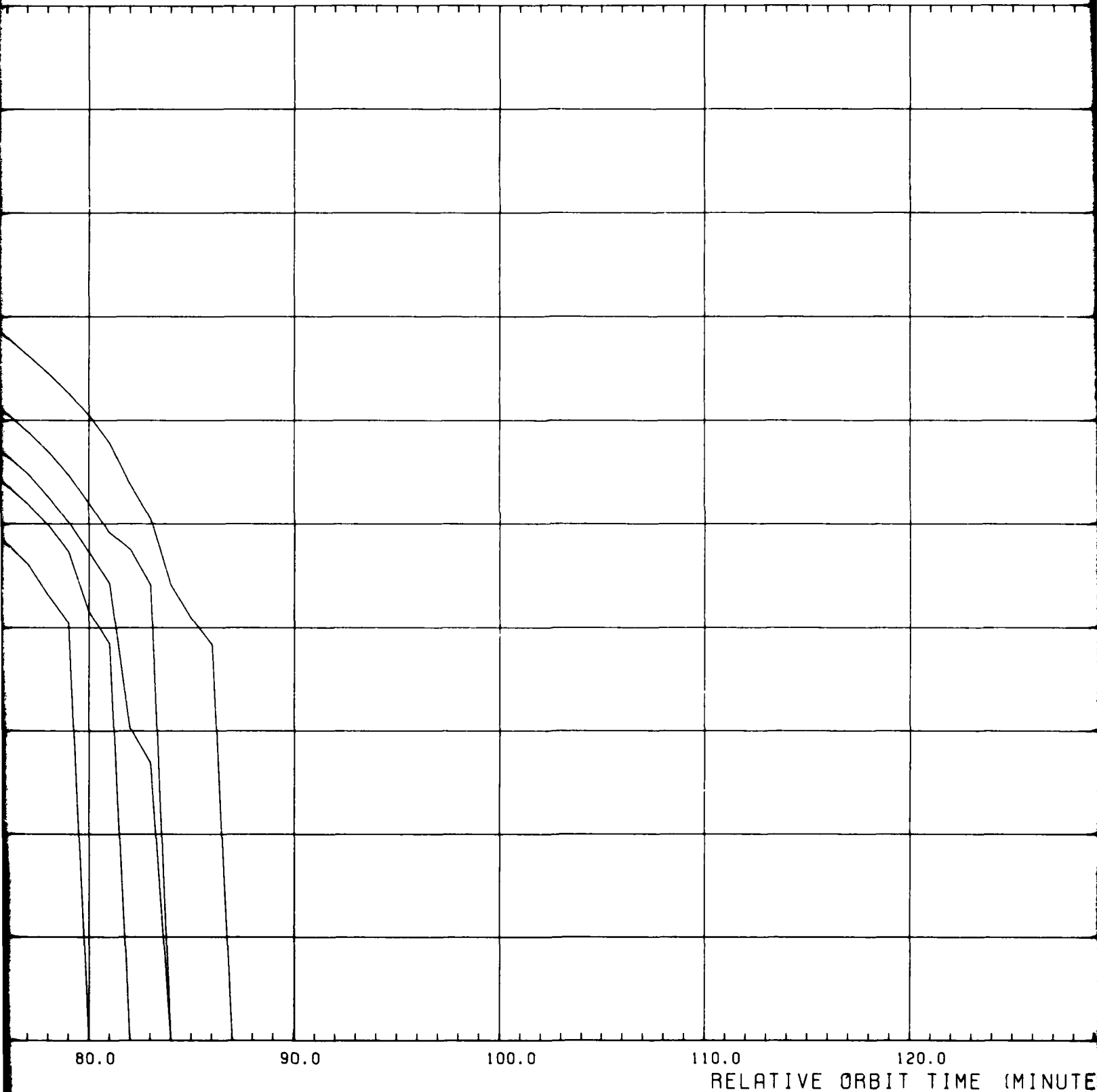
60.0

70.0

80.0

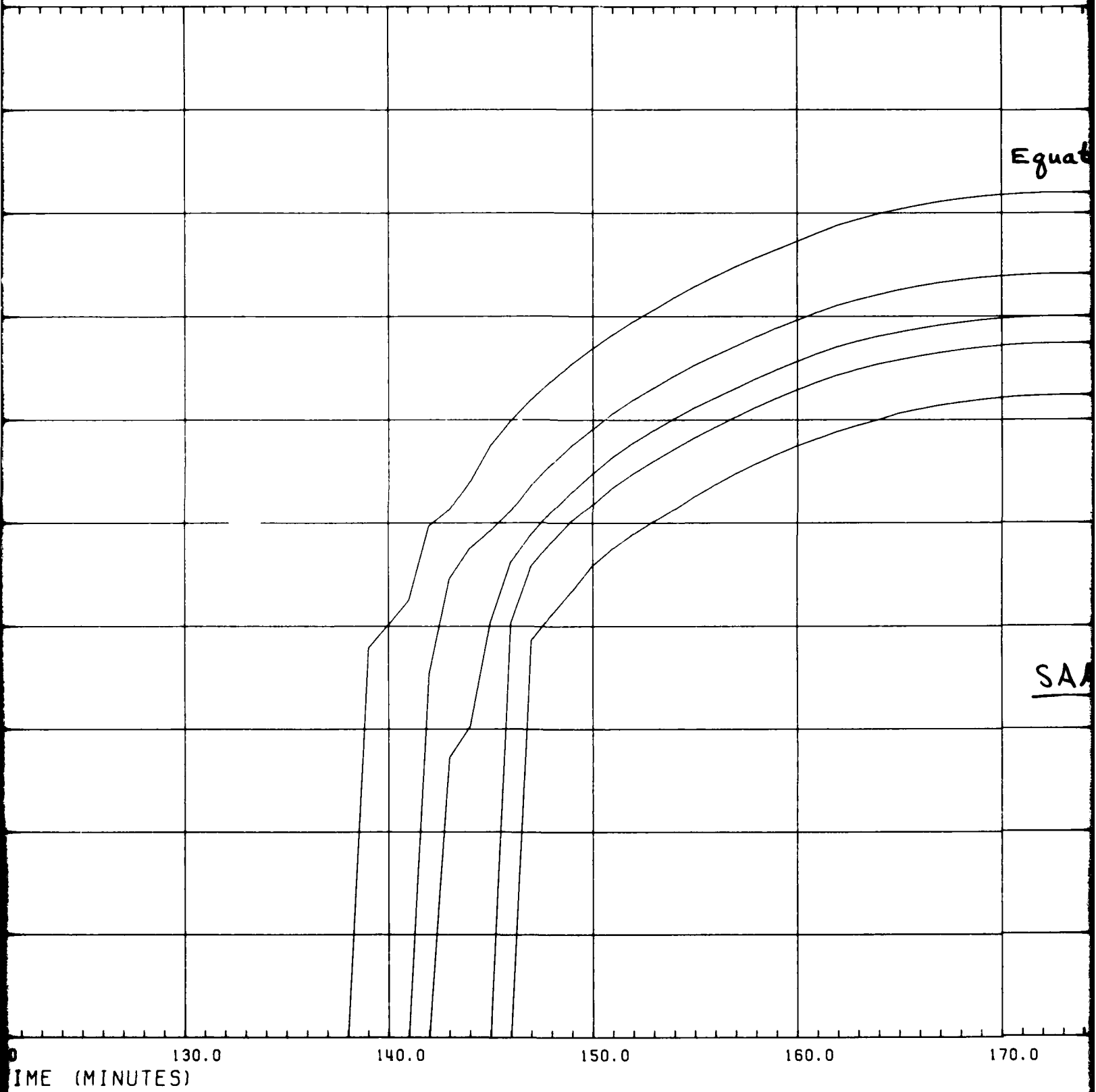
3'

DOSE AT CENTER OF ALUMINUM



4

ALUMINUM SPHERES



5

Equatorial Region

SAA

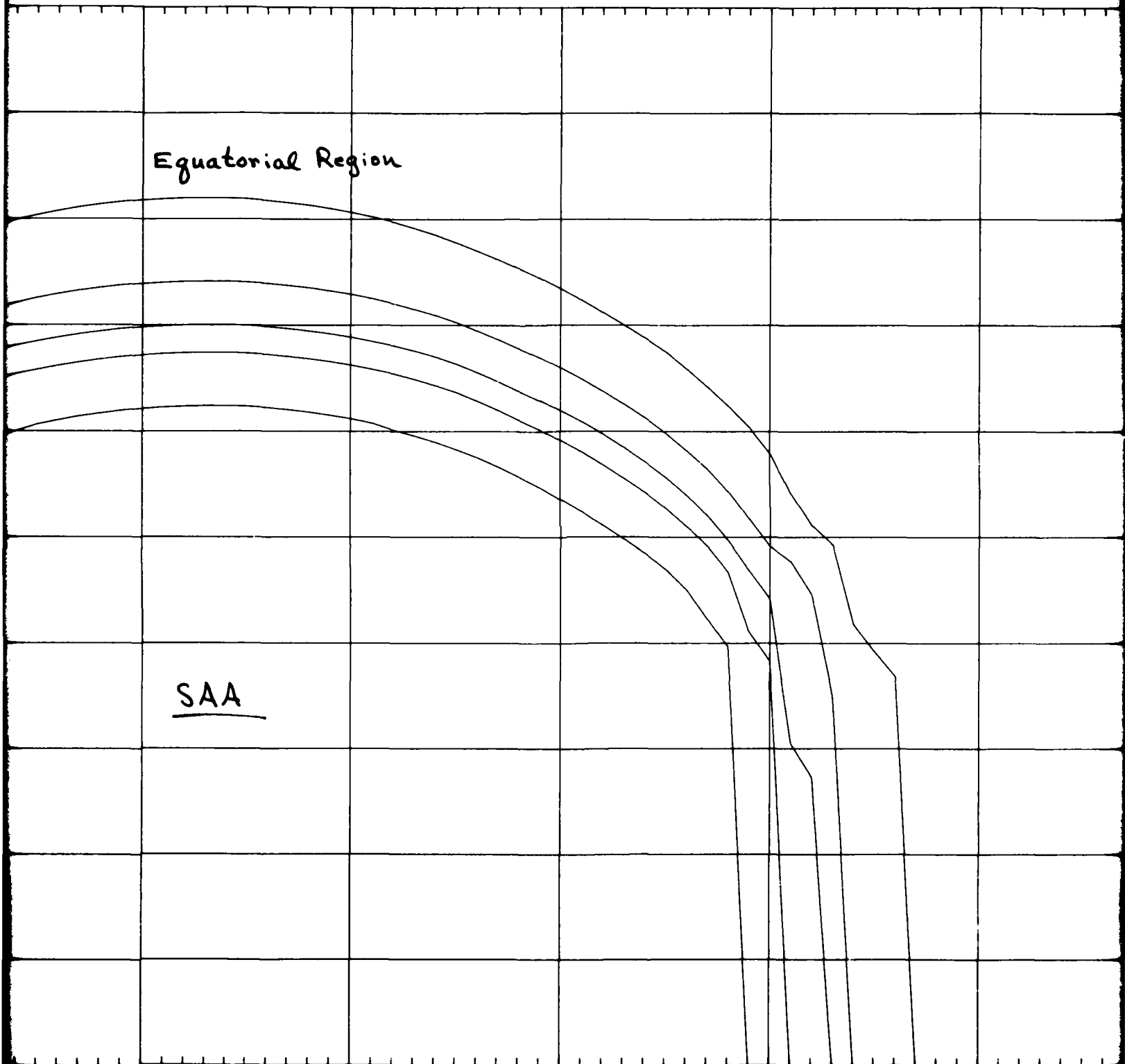
170.0

180.0

190.0

200.0

210.0



6

Figure 15a

ORBIT: NAVELEX 5
60 DGR/6389-6389 KM

EPOCH: 1989.5

MODELS:

FIELD: BARR/75
TRAPPED PROTONS: AP8
INNER ZN ELEC: AEG
OUTER ZN ELEC: AE17-L0

MISSION DURATION: 60.00 MO
EVALUATION PHASE: SOLAR MAX

UN FACTORS: NOT APPLIED

STOP TIME ON TAPE = 8.933318

NASA-GSFC

210.0

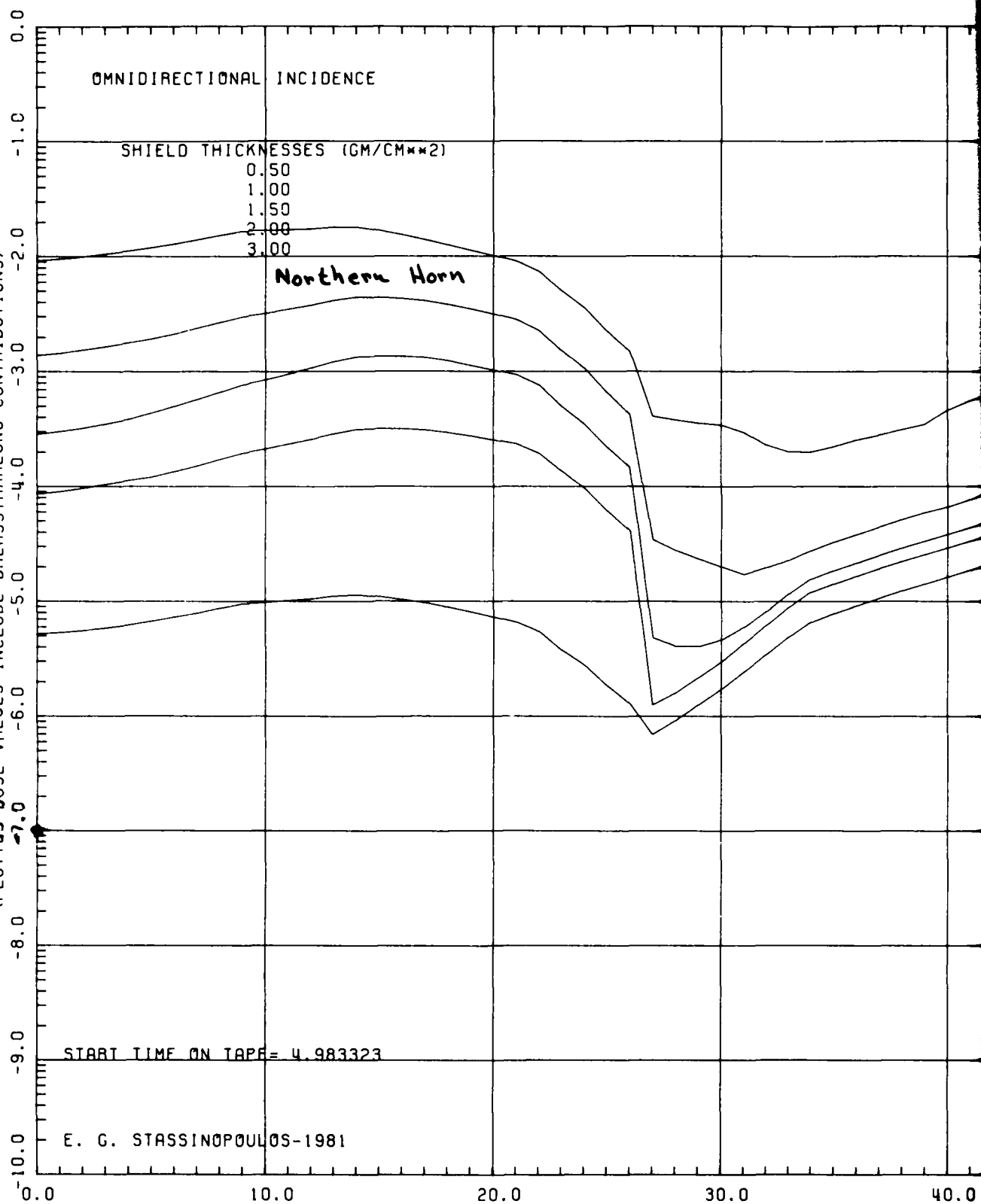
220.0

230.0

240.0

INSTANTANEOUS ALUMINUM ELECTRON DOSE (RADS)

(PLOTTED DOSE VALUES INCLUDE BREMSSTRAHLUNG CONTRIBUTIONS)



2¹

Equatorial Region

SAA

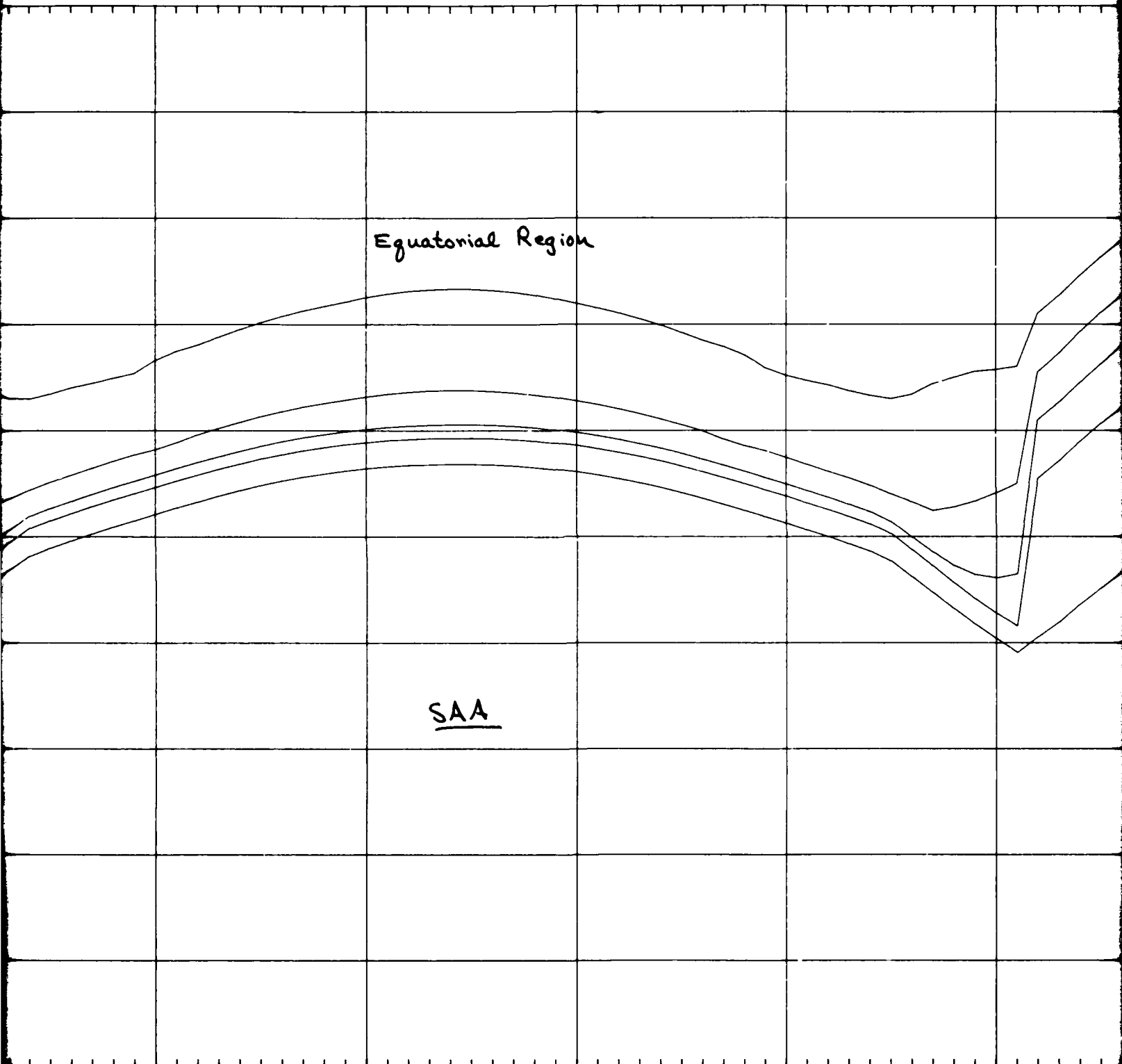
40.0

50.0

60.0

70.0

80.0



3¹

DOSE AT CENTER OF ALUMINUM

Southern Horn

80.0

90.0

100.0

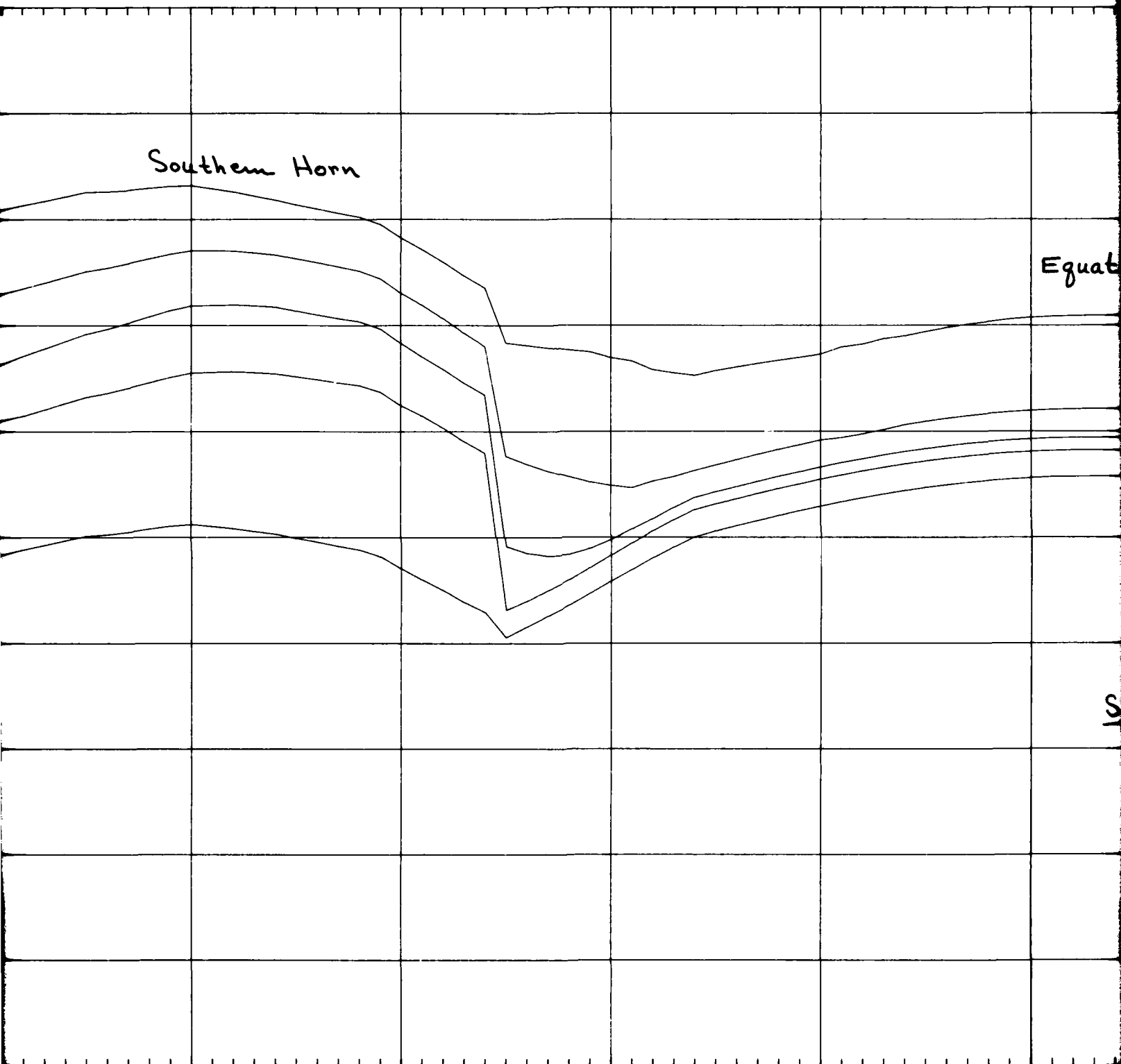
110.0

120.0

RELATIVE ORBIT TIME (MINUT

'4

F ALUMINUM SPHERES



0 130.0 140.0 150.0 160.0 170.0
TIME (MINUTES)

5

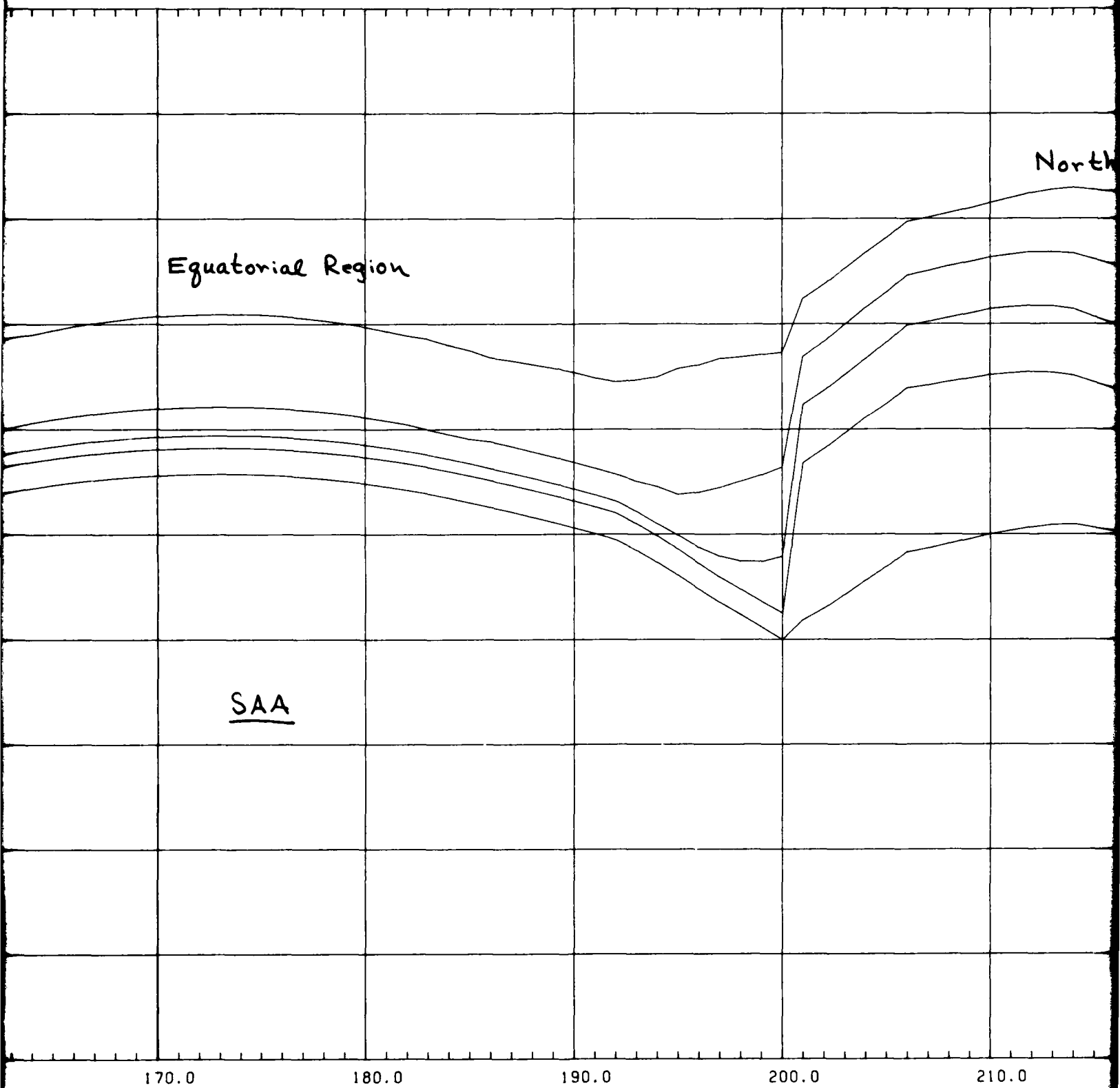


Figure 153

ORBIT: NAVELEX 5
60 DGR/6389-6389 KM

EPOCH: 1989.5

MODELS:
FIELD: BARR/75
TRAPPED PROTONS: AP8
INNER ZN ELEC: AE6
OUTER ZN ELEC: AE17-L0
MISSION DURATION: 60.00 MO
EVALUATION PHASE: SOLAR MAX

UN FACTORS: NOT APPLIED

Northern
Horn

STOP TIME ON TAPF= 8.933318

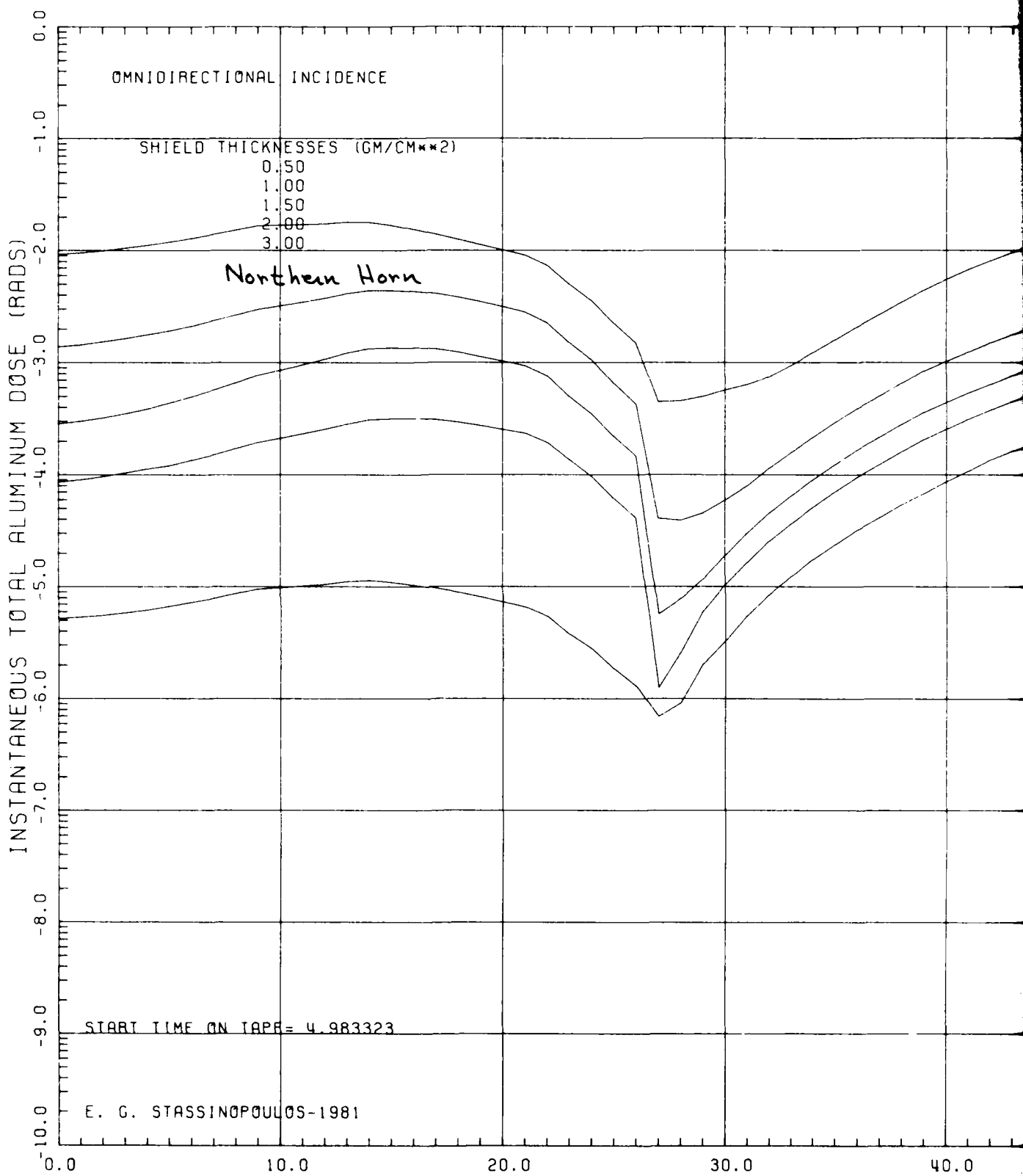
NASA-GSFC

210.0

220.0

230.0

240.0

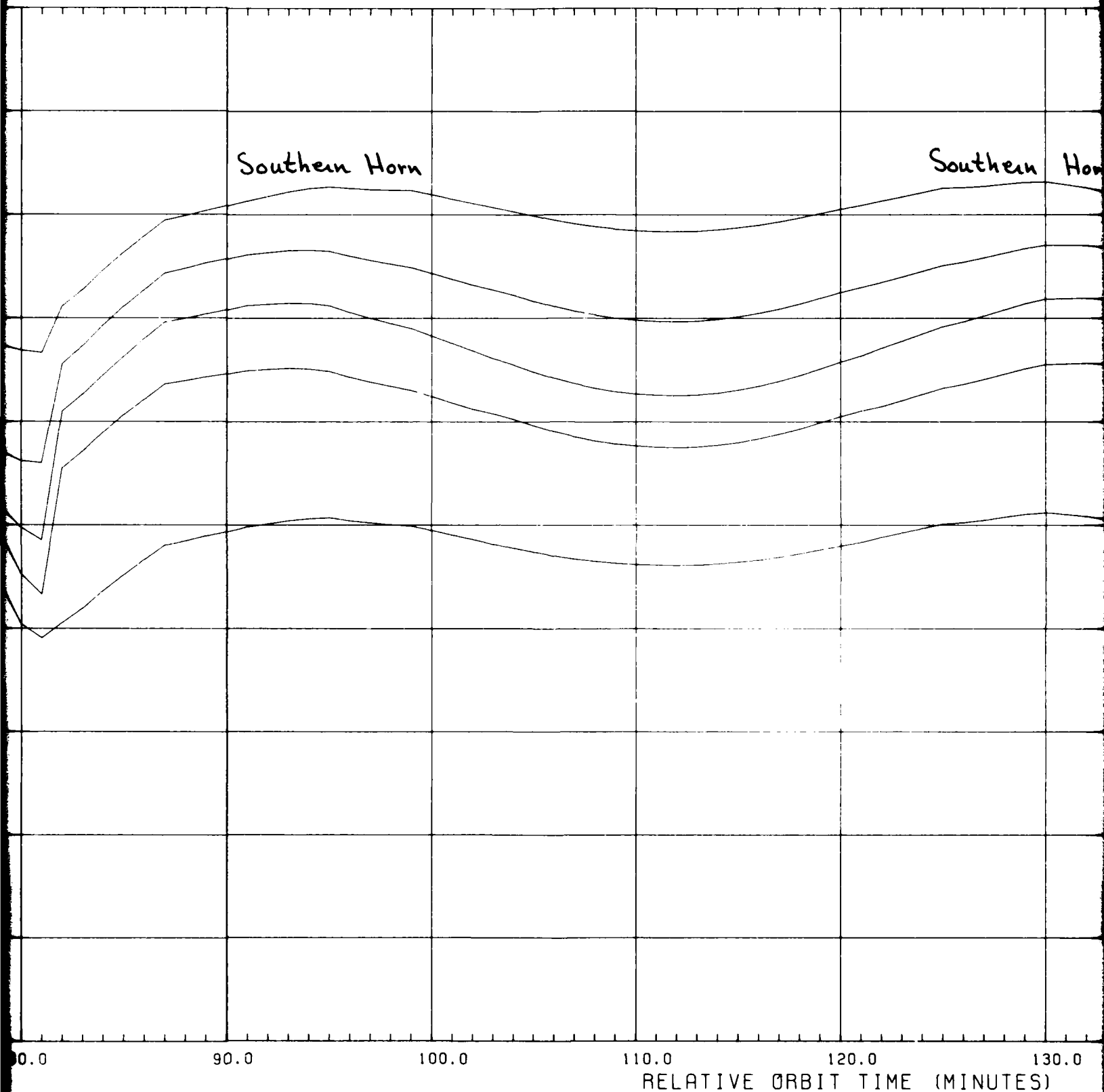


Equatorial Region

SAA

3

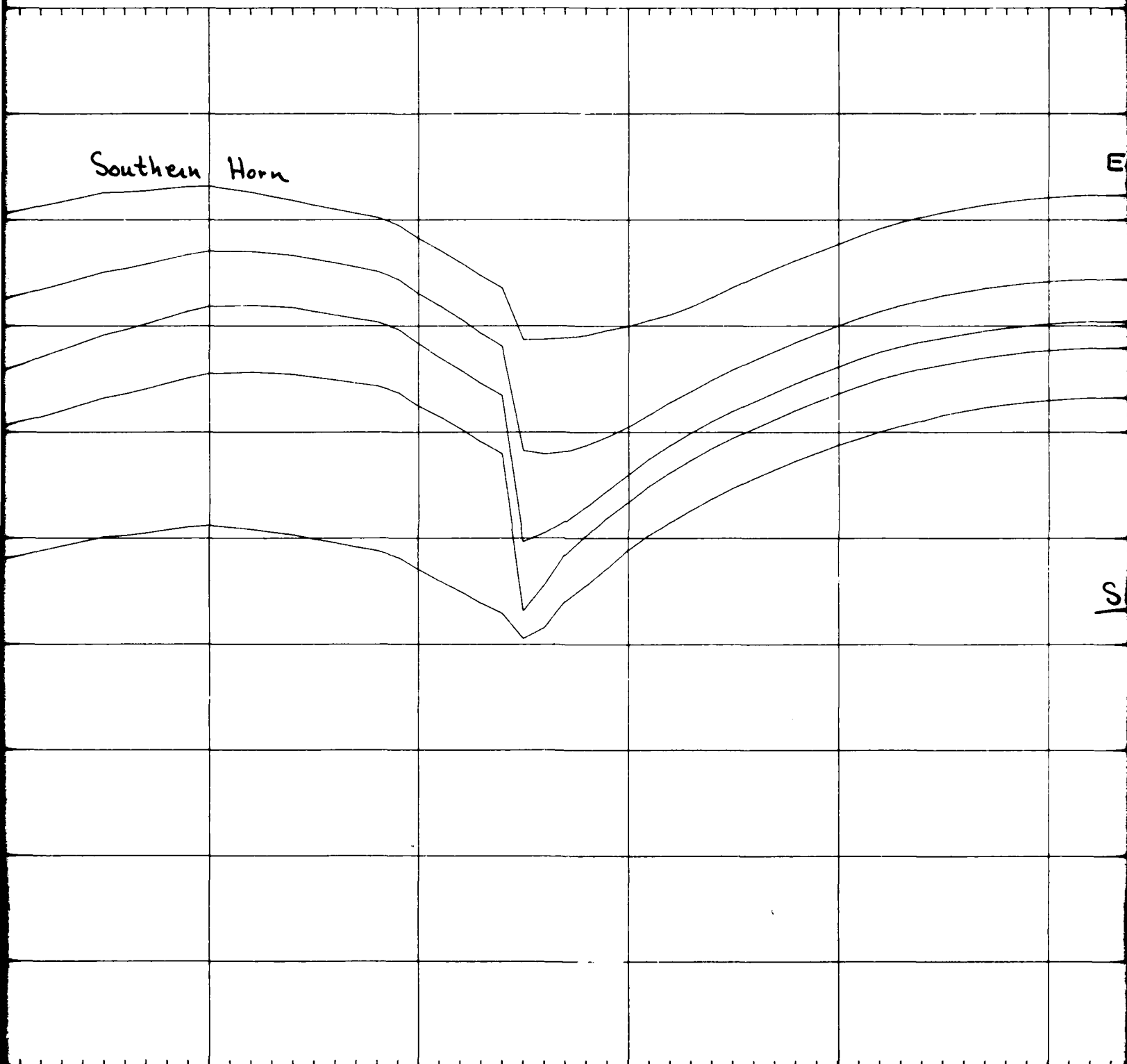
DOSE AT CENTER OF ALUMINUM SPHERE



4'

OF ALUMINUM SPHERES

Southern Horn



0.0 130.0 140.0 150.0 160.0 170.0
TIME (MINUTES)

5

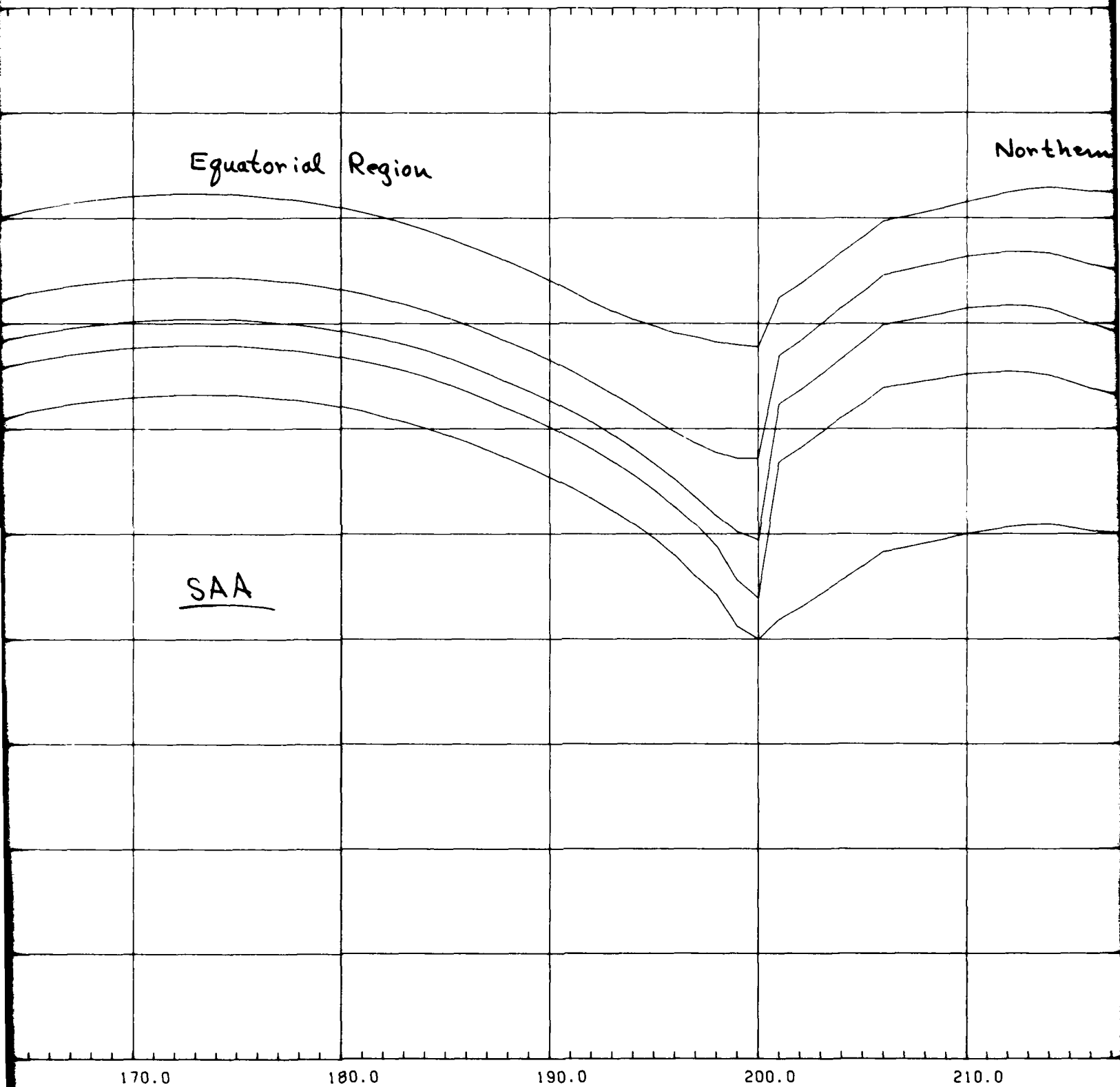


Figure 154

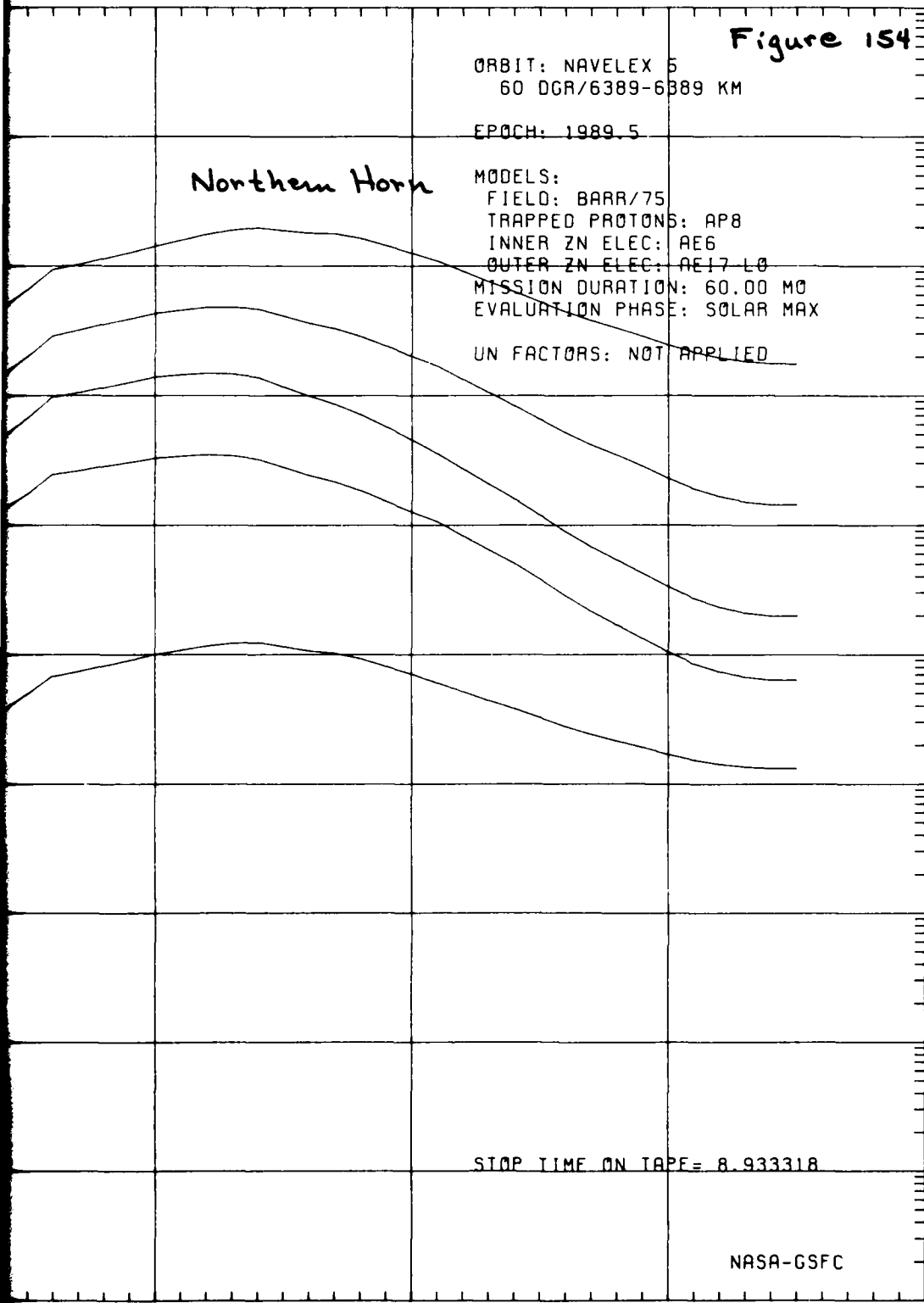
Northern Horn

ORBIT: NAVELEX 5
 60 DGR/6389-6389 KM
 EPOCH: 1989.5
 MODELS:
 FIELD: BARR/75
 TRAPPED PROTONS: AP8
 INNER ZN ELEC: AE6
 OUTER ZN ELEC: AE17 L0
 MISSION DURATION: 60.00 MO
 EVALUATION PHASE: SOLAR MAX
 UN FACTORS: NOT APPLIED

STOP TIME ON TAPE = 8.933318

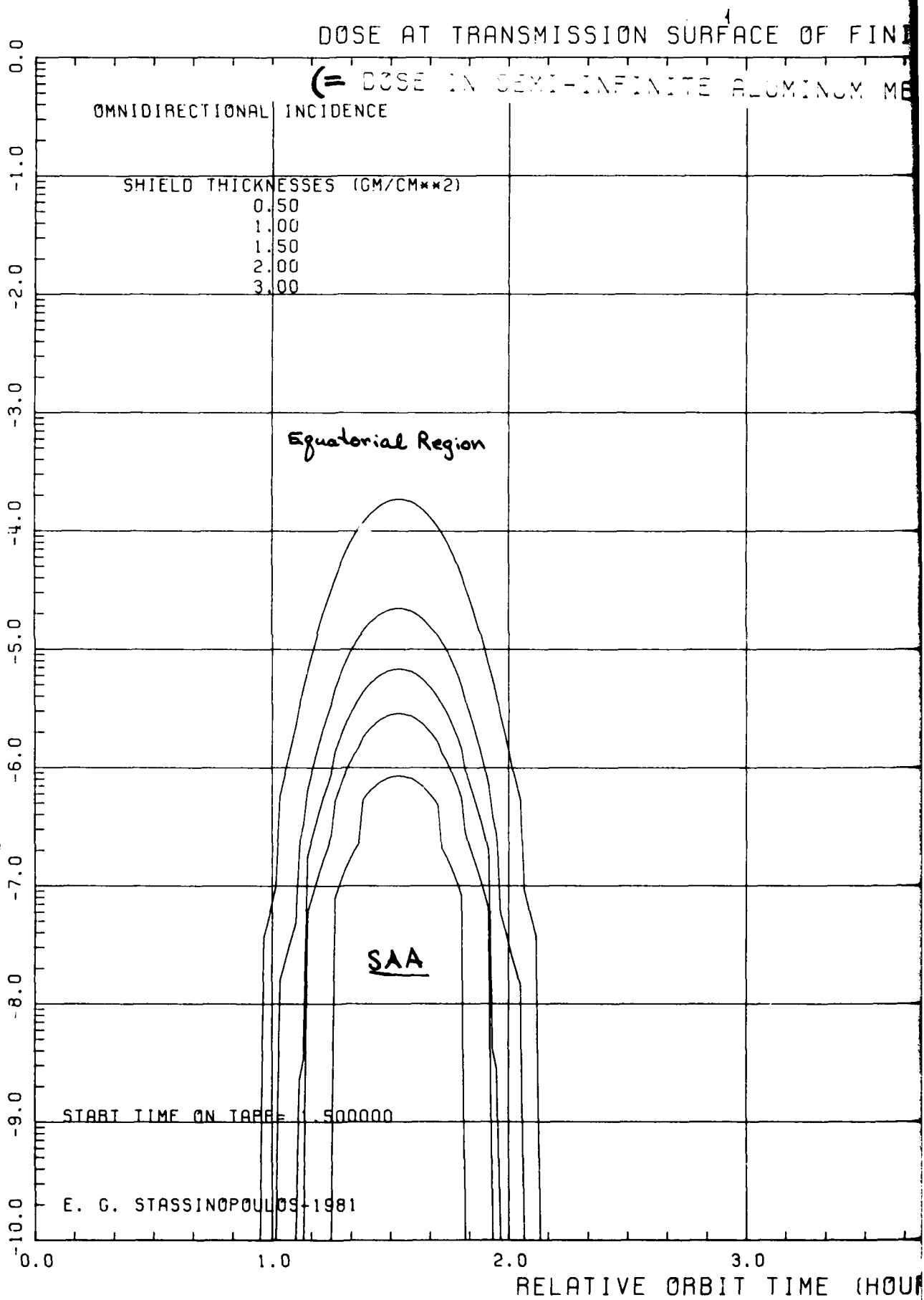
NASA-GSFC

210.0 220.0 230.0 240.0



INSTANTANEOUS ALUMINUM PROTON DOSE (RADS)

(SOLAR PROTON CONTRIBUTIONS ARE NOT INCLUDED)



SURFACE OF FINITE ALUMINUM SLAB SHIELDS

FINITE ALUMINUM MEDIUM)

Figure 155

ORBIT: NAVELEX 6
60 DGR/10371-10371 KM

EPOCH: 1989.5

MODELS:
FIELD: BARR/75
TRAPPED PROTONS: AP8
INNER ZN ELEC: AE6
OUTER ZN ELEC: AE17-L0
MISSION DURATION: 60.00 MO
EVALUATION PHASE: SOLAR MAX

UN FACTORS: NOT APPLIED

Equatorial
Region

SAA

STOP TIME ON TAPE = 7.483318

NASA-GSFC

3.0

4.0

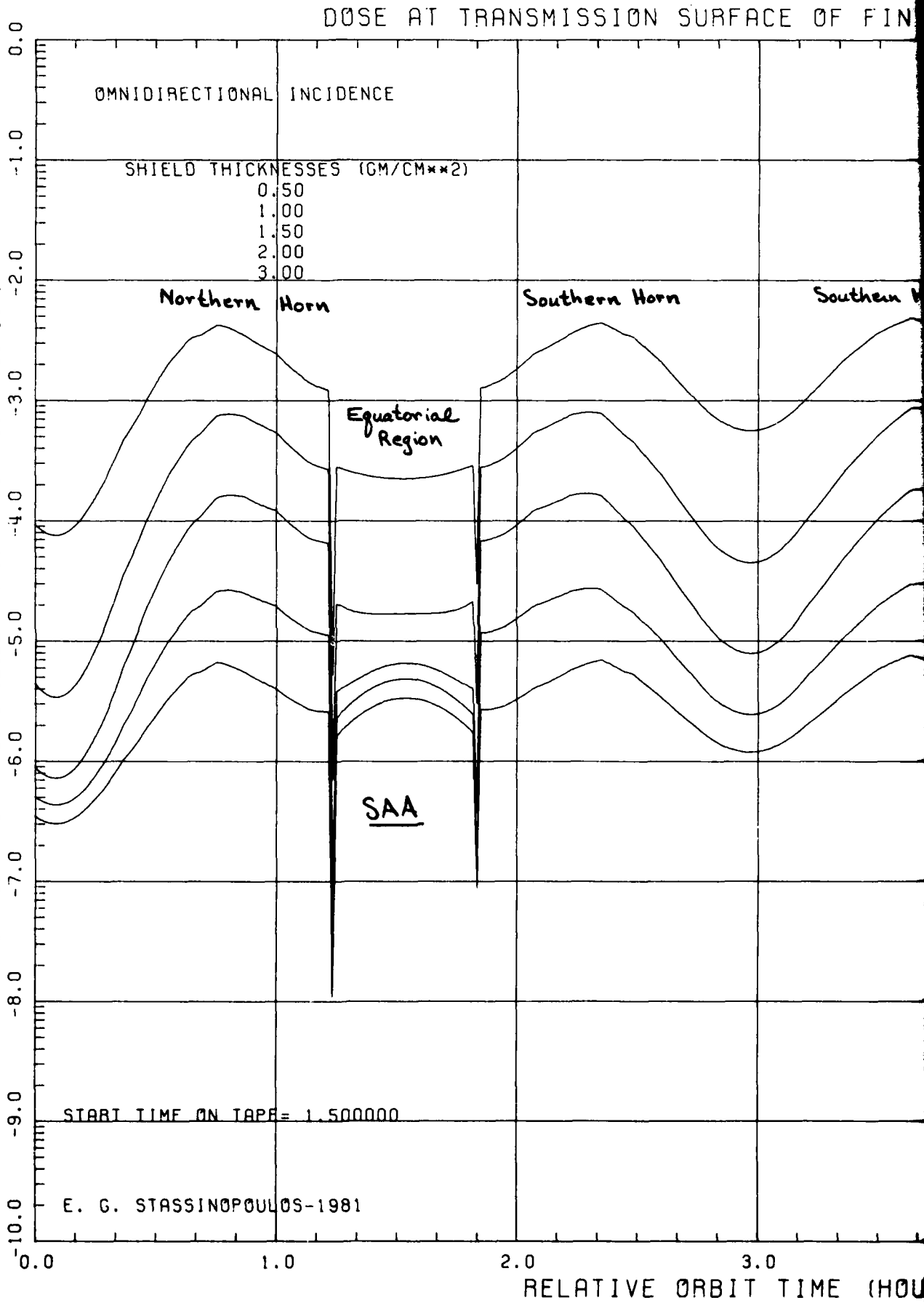
5.0

6.0

ORBIT TIME (HOURS)

INSTANTANEOUS ALUMINUM ELECTRON DOSE (RADS)

(PLOTTED DOSE VALUES INCLUDE BREMSSTRAHLUNG CONTRIBUTIONS)



ON SURFACE OF FINITE ALUMINUM SLAB SHIELDS

Figure 156

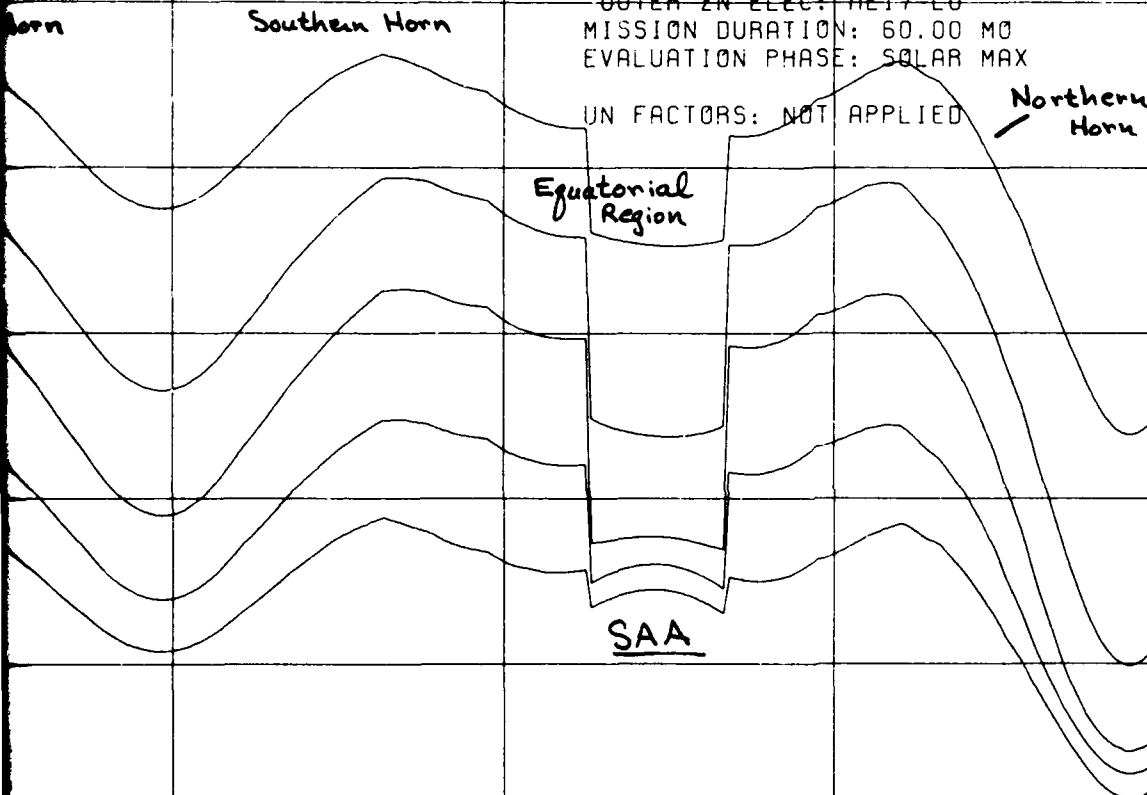
ORBIT: NAVELEX 6
60 DGR/10371-10371 KM

EPOCH: 1989.5

MODELS:
FIELD: BARR/75
TRAPPED PROTONS: AP8
INNER ZN ELEC: AE6
OUTER ZN ELEC: AE17-L0

MISSION DURATION: 60.00 MO
EVALUATION PHASE: SOLAR MAX

UN FACTORS: NOT APPLIED

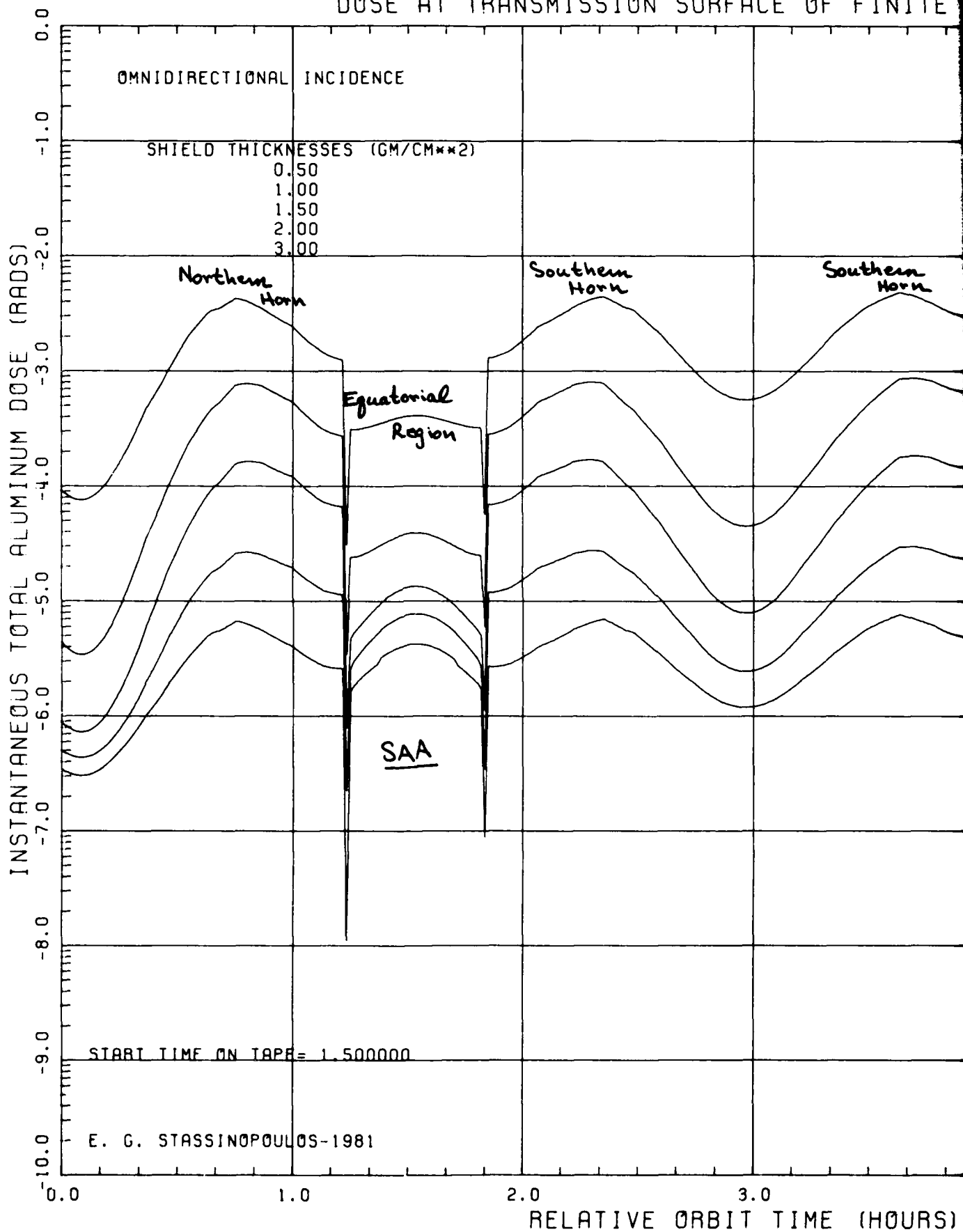


STOP TIME ON TAPE= 7.483318

NASA-GSFC

ORBIT TIME (HOURS)

DOSE AT TRANSMISSION SURFACE OF FINITE



SURFACE OF FINITE ALUMINUM SLAB SHIELDS

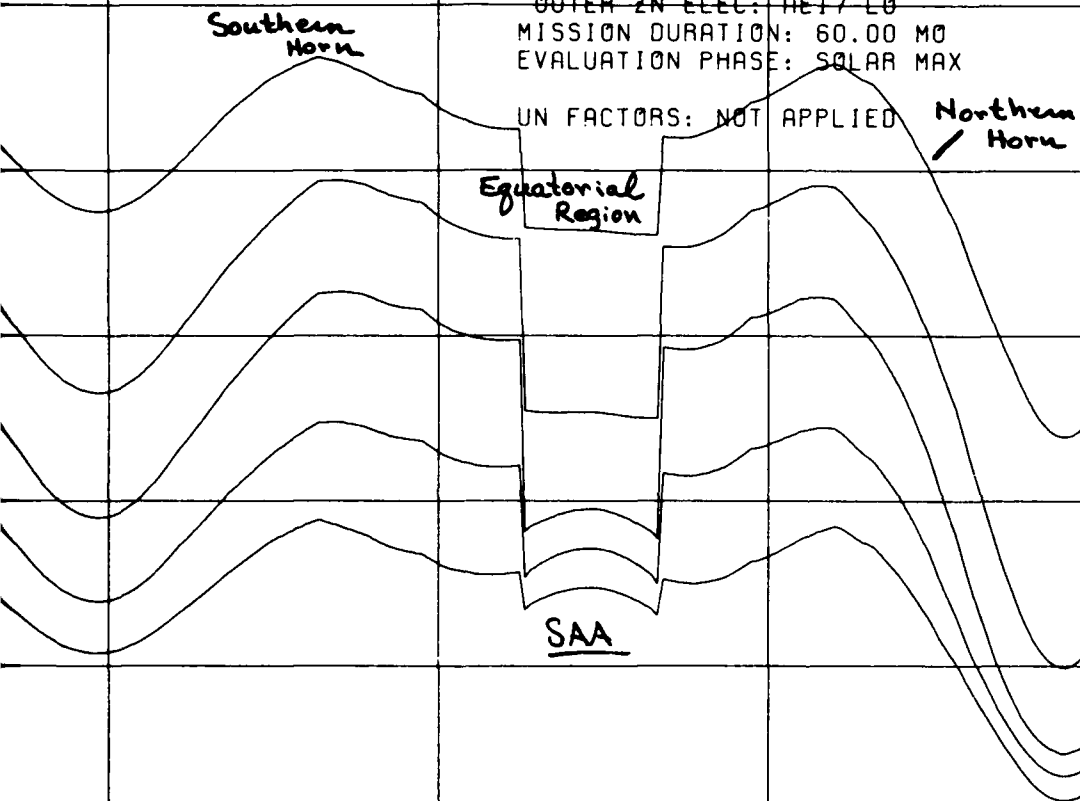
Figure 157

ORBIT: NAVELEX 6
60 DGR/10371-10371 KM

EPOCH: 1989.5

MODELS:
FIELD: BARR/75
TRAPPED PROTONS: AP8
INNER ZN ELEC: AE6
OUTER ZN ELEC: AE17 L0
MISSION DURATION: 60.00 MO
EVALUATION PHASE: SOLAR MAX

UN FACTORS: NOT APPLIED



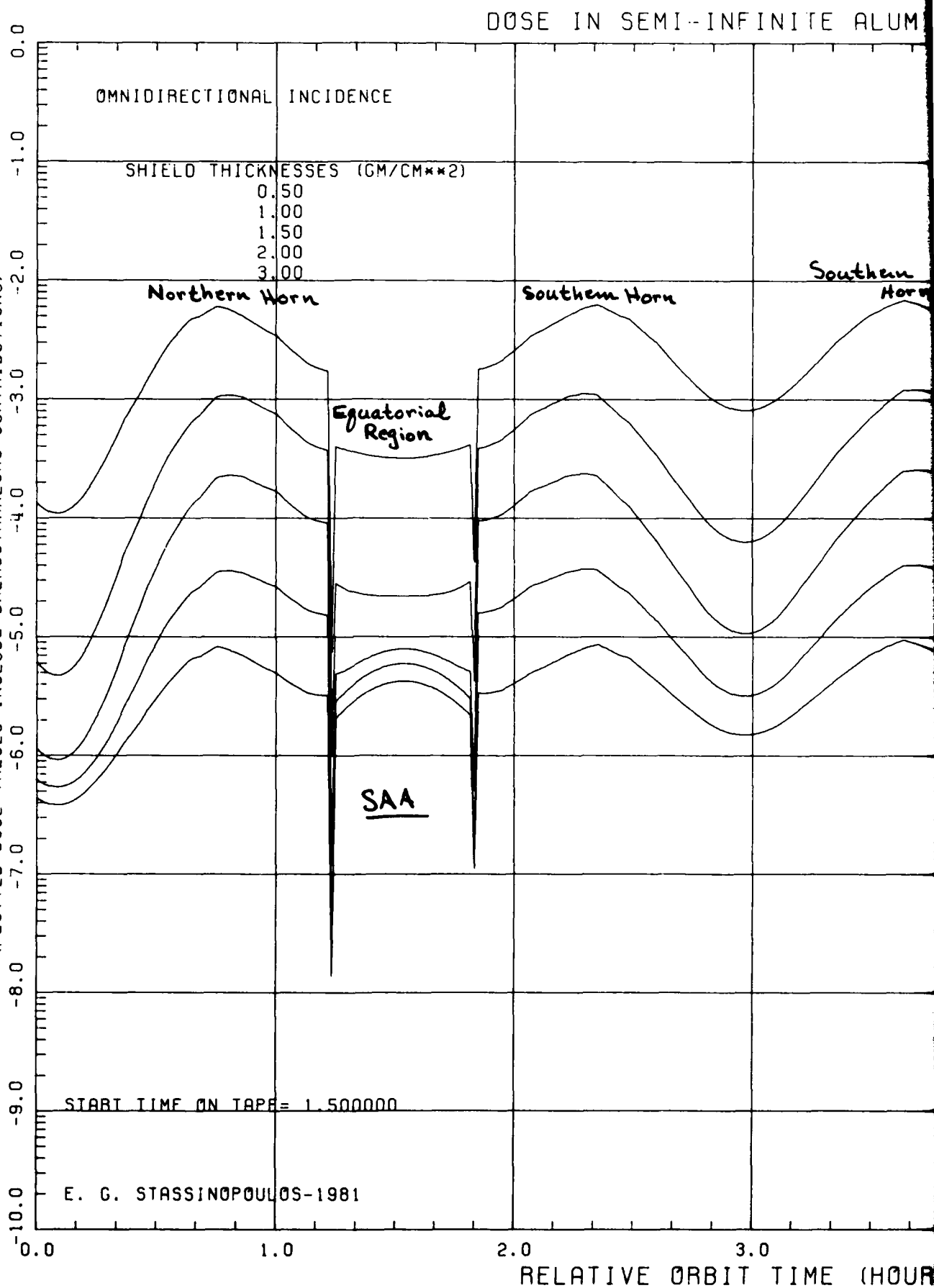
STOP TIME ON TAPE= 7.483318

NASA-GSFC

ORBIT TIME (HOURS)

INSTANTANEOUS ALUMINUM ELECTRON DOSE (RADS)

(PLOTTED DOSE VALUES INCLUDE BREMSSTRAHLUNG CONTRIBUTIONS)



INFINITE ALUMINUM MEDIUM

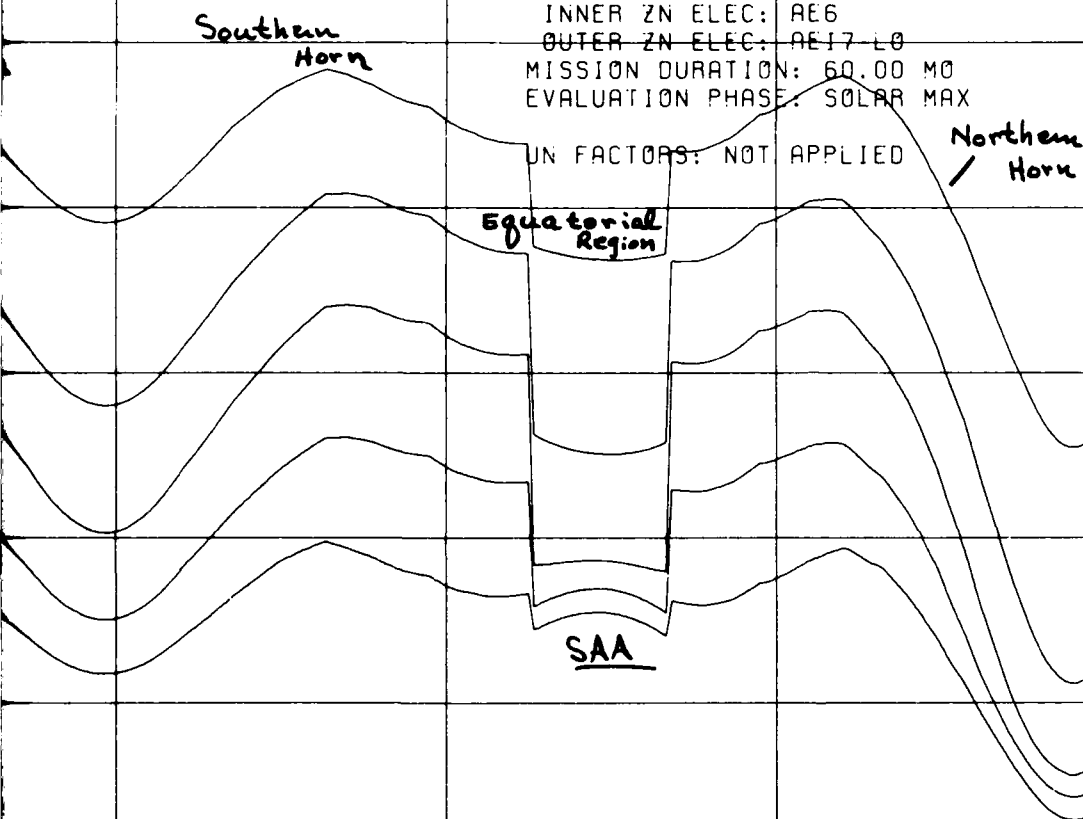
Figure 158

ORBIT: NAVELEX 6
60 DGR/10371-10371 KM

EPOCH: 1989.5

MODELS:
FIELD: BARR/75
TRAPPED PROTONS: AP8
INNER ZN ELEC: AE6
OUTER ZN ELEC: AE17-L0
MISSION DURATION: 60.00 MO
EVALUATION PHASE: SOLAR MAX

UN FACTORS: NOT APPLIED

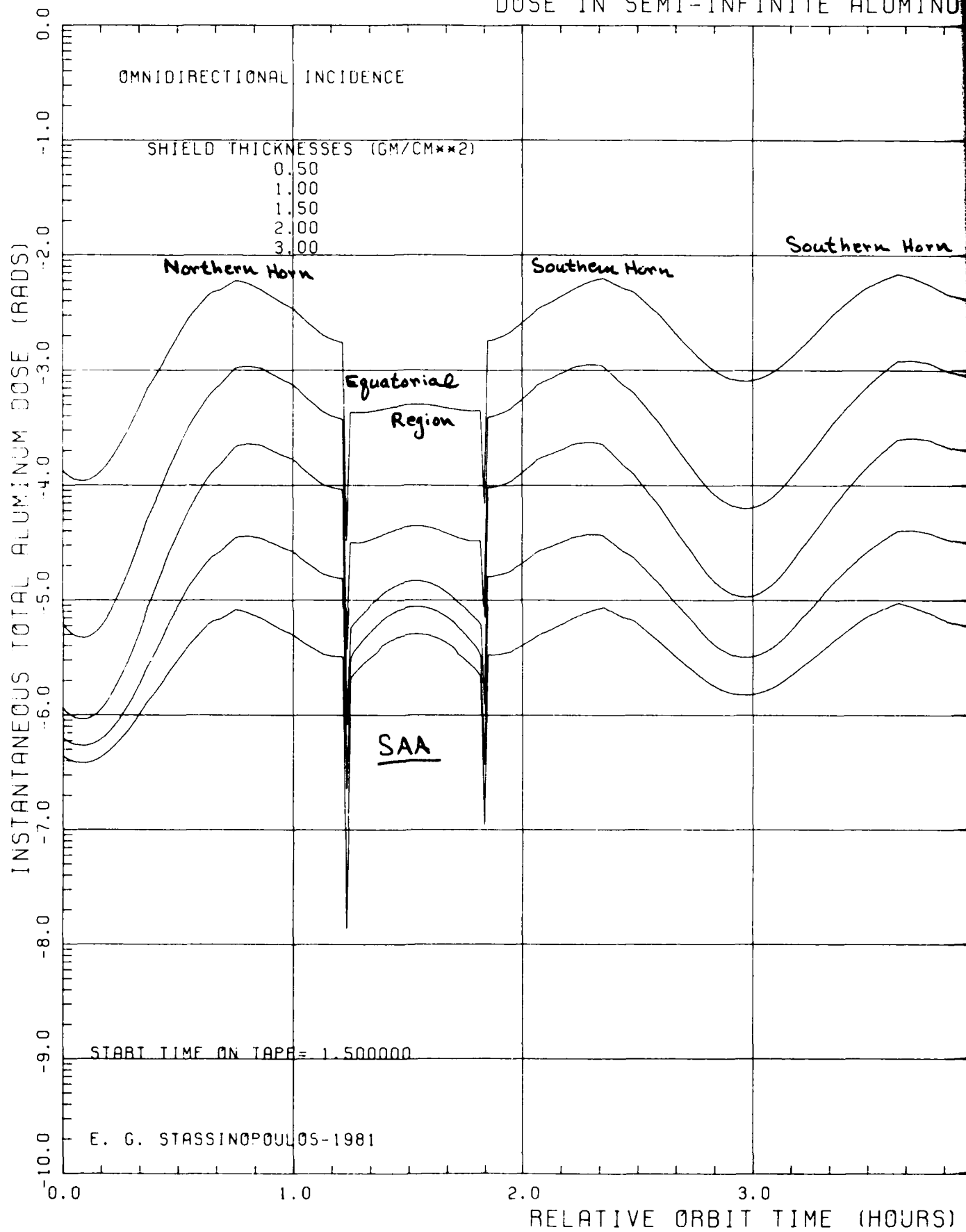


STOP TIME ON TAPE = 7.483318

NASA-GSFC

ORBIT TIME (HOURS)

DOSE IN SEMI-INFINITE ALUMINUM



INFINITE ALUMINUM MEDIUM

Figure 159

ORBIT: NAVELEX 6
60 DGR/10371-10371 KM

EPOCH: 1989.5

MODELS:
FIELD: BARR/75
TRAPPED PROTONS: AP8
INNER ZN ELEC: A65
OUTER ZN ELEC: AE17 L0
MISSION DURATION: 60.00 MO
EVALUATION PHASE: SOLAR MAX

UN FACTORS: NOT APPLIED

Southern Horn

Northern Horn

Equatorial
Region

SAA

STOP TIME ON TAPE= 7.483318

NASA-GSFC

3.0 4.0 5.0 6.0
IT TIME (HOURS)

INSTANTANEOUS ALUMINUM PROTON DOSE (RADS)

(SOLAR PROTON CONTRIBUTIONS ARE NOT INCLUDED)

DOSE AT CENTER OF ALUMINUM

OMNIDIRECTIONAL INCIDENCE

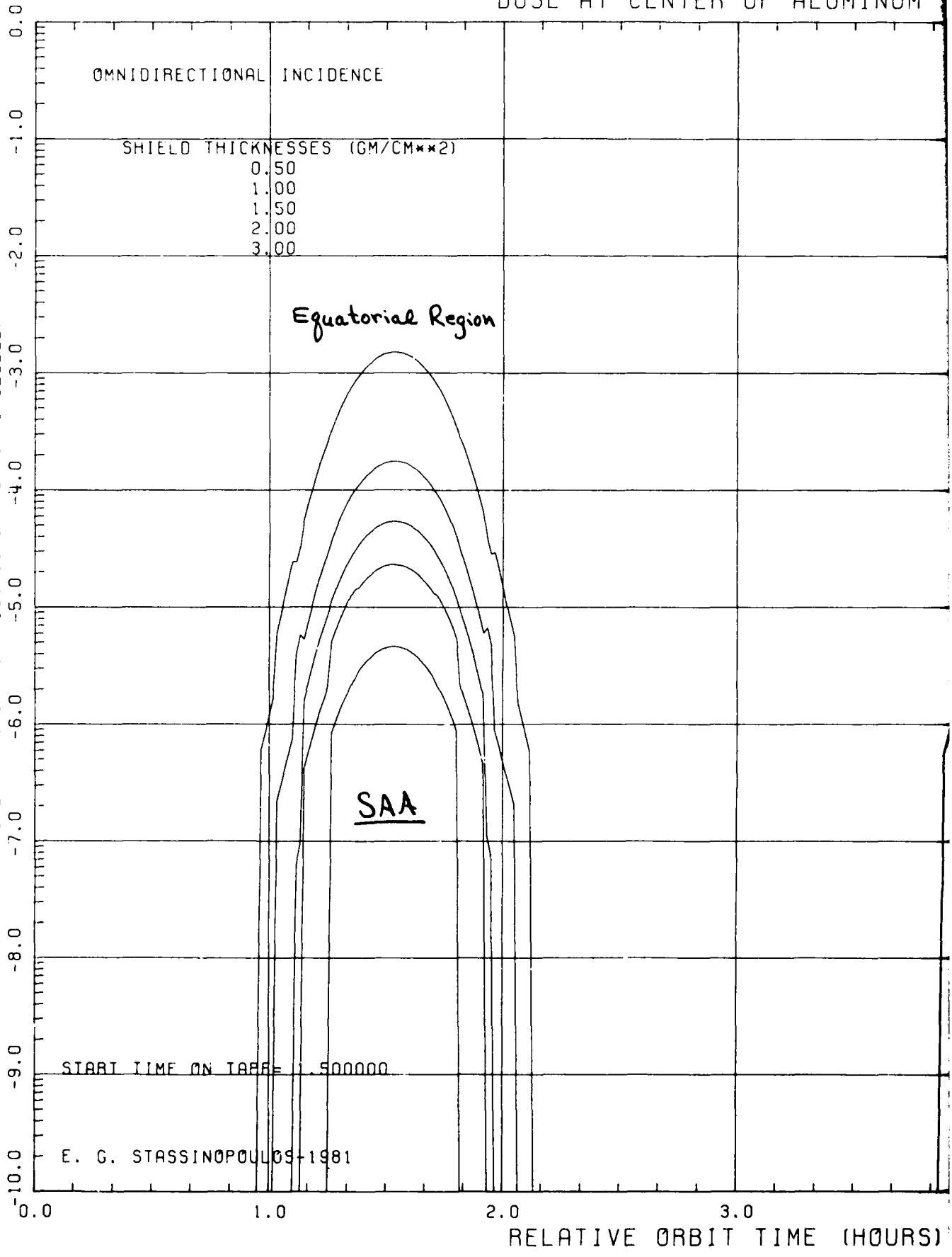
SHIELD THICKNESSES (GM/CM**2)
0.50
1.00
1.50
2.00
3.00

Equatorial Region

SAA

START TIME ON TABLE 1.500000

E. G. STASSINOPOULOS-1981



TER OF ALUMINUM SPHERES

Figure 160

ORBIT: NAVELEX 6
60 DGR/10371-10371 KM

EPOCH: 1989.5

MODELS:
FIELD: BARR/75
TRAPPED PROTONS: AP8
INNER ZN ELEC: AE6
OUTER ZN ELEC: AE17-L0
MISSION DURATION: 60.00 MO
EVALUATION PHASE: SOLAR MAX

UN FACTORS: NOT APPLIED

Equatorial Region

SAA

STOP TIME ON TAPE = 7.483318

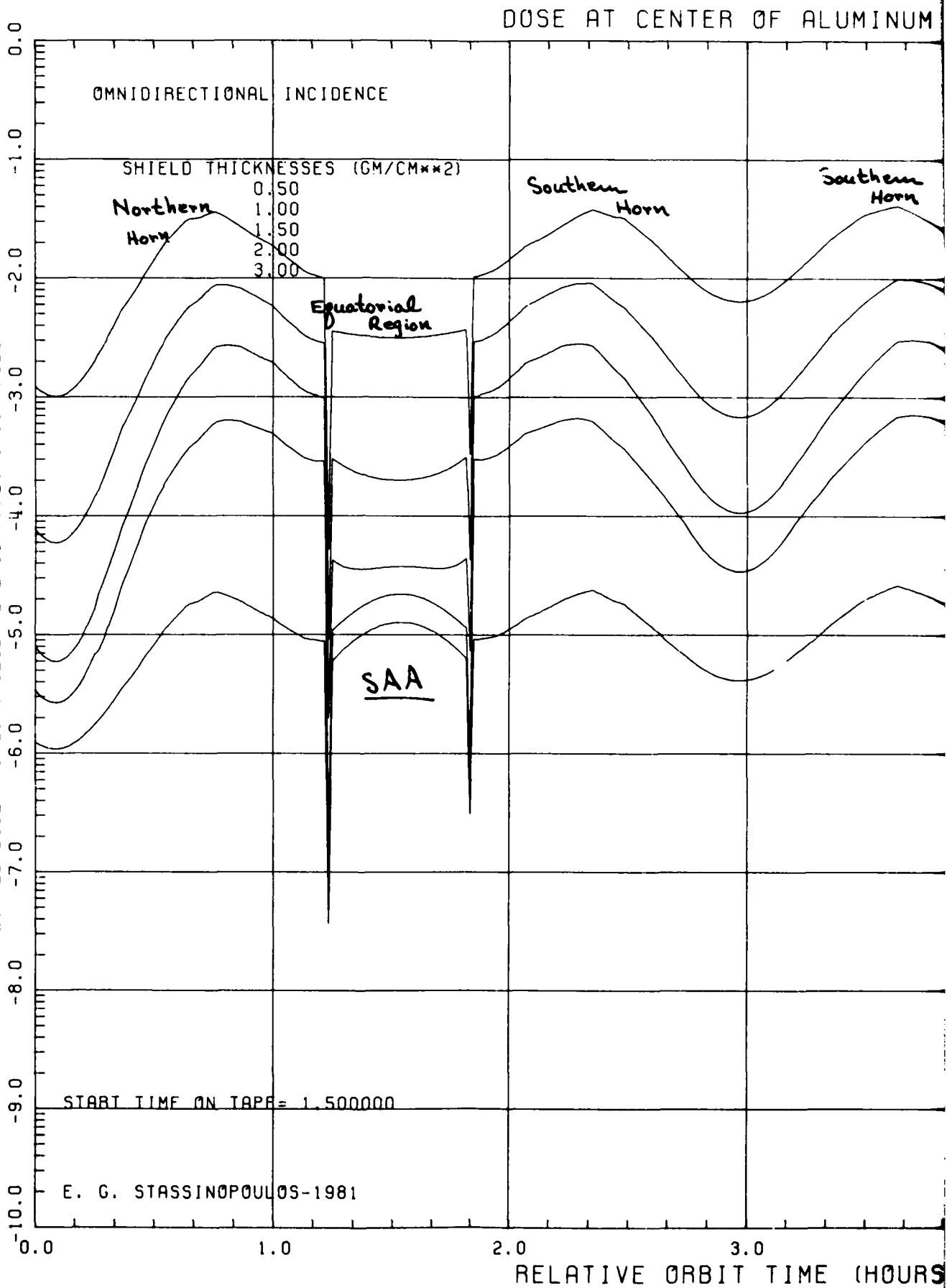
NASA-GSFC

3.0 4.0 5.0 6.0

BIT TIME (HOURS)

INSTANTANEOUS ALUMINUM ELECTRON DOSE (RADS)

(PLOTTED DOSE VALUES INCLUDE BREMSSTRAHLUNG CONTRIBUTIONS)



TER OF ALUMINUM SPHERES

Figure 161

ORBIT: NAVELEX 6
60 DGR/10371-10371 KM

EPOCH: 1989.5

MODELS:
FIELD: BARR/75
TRAPPED PROTONS: AP8
INNER ZN ELEC: AE6
OUTER ZN ELEC: AE17 L0
MISSION DURATION: 60.00 MO
EVALUATION PHASE: SOLAR MAX
UN FACTORS: NOT APPLIED

Southern
Horn

Northern
Horn

Equatorial
Region

SAA

STOP TIME ON TAPE = 7.483318

NASA-GSFC

ORBIT TIME (HOURS)

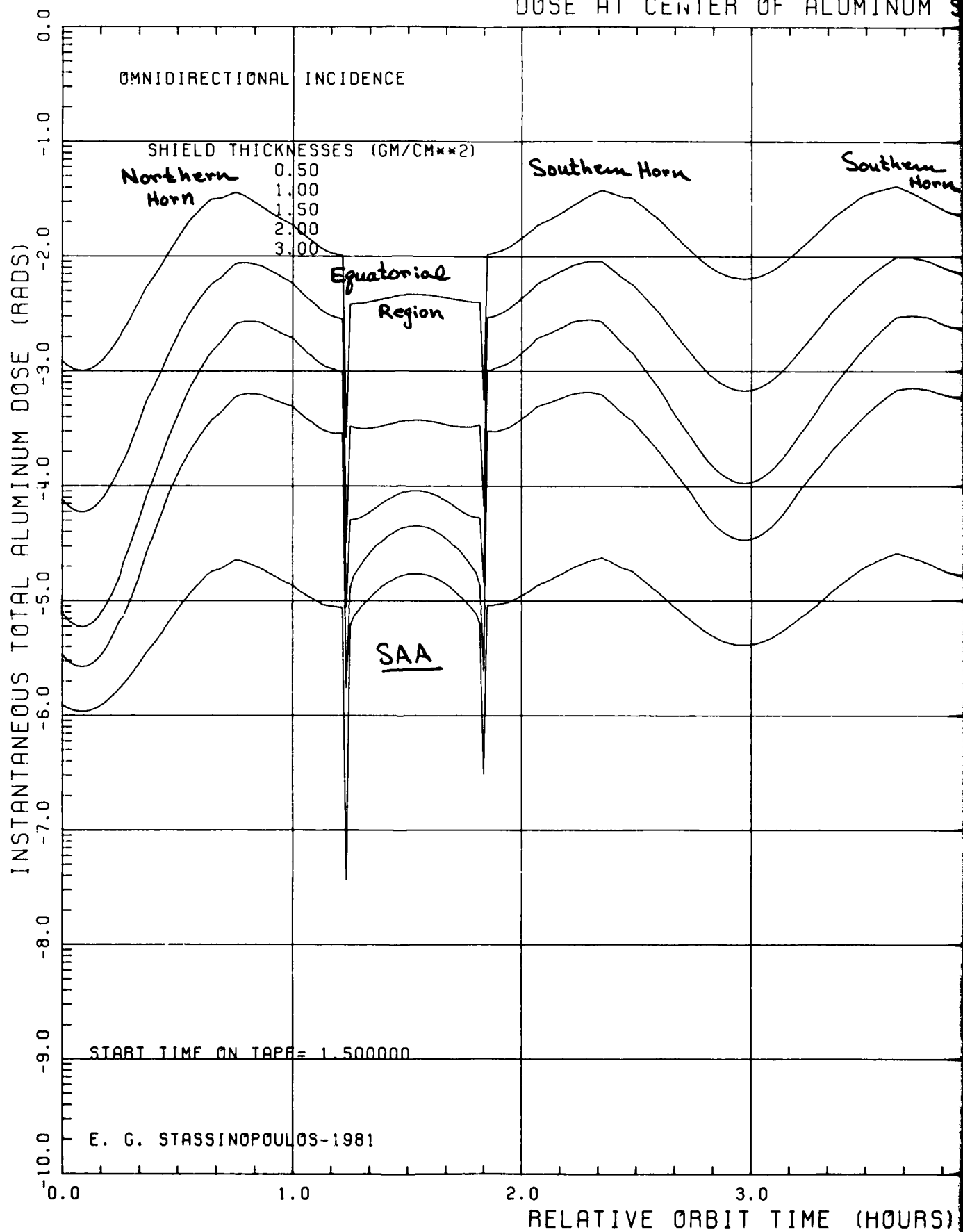
3.0

4.0

5.0

6.0

DOSE AT CENTER OF ALUMINUM S



OF ALUMINUM SPHERES

Figure 162

ORBIT: NAVELEX 6
60 DGR/10371-10371 KM

EPOCH: 1989.5

MODELS:
FIELD: BARR/75
TRAPPED PROTONS: AP8
INNER ZN ELEC: AE6
OUTER ZN ELEC: AE17 L0
MISSION DURATION: 60.00 MO
EVALUATION PHASE: SOLAR MAX

UN FACTORS: NOT APPLIED

*Southern
Horn*
*Northern
Horn*
*Equatorial
Region*

SAA

STOP TIME ON TAPE = 7.483318

NASA-GSFC

3.0 4.0 5.0 6.0
T TIME (HOURS)

**DAT
FILM**

7